
#### Abstract

\title{ of Dissertation: \\ DEVELOPMENT OF A COMPACT HEAT EXCHANGER WITH BIFURCATED BARE TUBES }

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Heat transfer enhancement of air-to-fluid heat exchangers by novel surface or geometry design and optimization is a major research topic. The traditional way of reducing airside thermal resistance is to extend airside heat transfer area by adding fins and the conventional method of reducing fluid side thermal resistance is to use enhanced inner surfaces. These approaches have limitations in further reducing the thermal resistance.

Small diameter (4 and 5 mm ) fin-and-tube heat exchangers, louvered fin minichannel heat exchangers (MCHX), newly studied round bare tube heat exchangers (BTHX) and shape optimized bare tube heat exchangers (sBTHX) with diameter of $0.8 \sim 1.0 \mathrm{~mm}$ were experimentally investigated using air and water to gain the fundamental understanding of heat transfer and the current technology limitations. Correlations of air-side heat transfer coefficient and pressure drop were then developed for BTHX and sBTHX.


To improve current technologies, a novel bifurcated bare tube heat exchanger (referred as bBTHX, hereafter) was proposed in this study. It was numerically investigated and optimized using Parameterized Parallel Computational Fluid Dynamics (PPCFD) and Approximation Assisted Optimization (AAO) techniques. The most unique feature of bBTHX is the addition of bifurcation, which enhances airside heat transfer by creating 3D flow and waterside heat transfer by boundary layer interruption and redevelopment. The airside and waterside pressure drop can also be reduced by proper design and optimization, resulting in smaller fan and pumping power. Compared to MCHX with similar capacity and frontal area, the optimal bBTHX design has $38 \%$ lower total power and $83 \%$ smaller volume and $87 \%$ smaller material volume. Compared to BTHX with similar capacity and frontal area, the optimal design has $28 \%$ lower total power and $11 \%$ smaller volume and $10 \%$ smaller material volume.

The bBTHX design can be widely applied in industry such as automotive radiators, oil coolers, condenser and evaporator. Two applications of this heat exchanger were discussed in detail: car radiator and indoor coil for Hybrid Variable Refrigerant Flow (HVRF) system. The bBTHX car radiator has $30 \%$ lower pumping power, $68 \%$ smaller heat exchanger volume and $67 \%$ less water weight than those of baseline. Moreover, refrigerant charge of HVRF systems with bBTHX is reduced by 40~70\%.

# DEVELOPMENT OF A COMPACT HEAT EXCHANGER WITH BIFURCATED BARE TUBES 

by

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## Dedication

To everyone who believed in me: My family, mentors and friends

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## Nomenclature

| Symbols |  |
| :---: | :---: |
| A | area $\left(\mathrm{m}^{2}\right)$ |
| $A_{f}$ | minimum free flow area ( $\mathrm{m}^{2}$ ) |
| $A_{f}$ | surface area of fin ( $\mathrm{m}^{2}$ ) |
| $A_{0}$ | total air side heat transfer area ( $\mathrm{m}^{2}$ ) |
| $A_{p, i}$ | inside surface area of tubes ( $\mathrm{m}^{2}$ ) |
| $A_{p, o}$ | outside surface area of tubes (m2) |
| $b_{p}^{\prime}$ | slope of a straight line between the outside and inside tube wall temperature ( $\mathrm{J} / \mathrm{kgK}$ ) |
| $b_{r}$ | slope of the air saturation curved at the mean coolant temperature and the inside wall temperature ( $\mathrm{J} / \mathrm{kgK}$ ) |
| $b_{w, m}^{\prime}$ | slope of the air saturation curve at the water film temperature of the fin surface ( J/kgK) |
| $b_{w, p}^{\prime}$ | slope of the air saturation curve at the water film temperature of the tube wall surface ( $\mathrm{J} / \mathrm{kgK}$ ) |
| $C_{0}, C_{1}$ | constant |
| $C_{p}$ | specific heat capacity at constant pressure ( $\mathrm{J} / \mathrm{kgK}$ ) |
| $D_{c}$ | tube outside diameter (include collar) (m) |
| $D_{i}$ | tube inside diameter (m) |
| $E B$ | energy balance (\%) |
| EES | Engineering Equation Solver |
| $f$ | friction factor |
| $f_{i}$ | in-tube friction factors of water |
| $F$ | log mean temperature difference correction factor |
| $G$ | mass flux ( $\mathrm{kg} / \mathrm{m}^{2} \mathrm{~s}$ ) |
| $G_{\text {max }}$ | maximum mass velocity based on minimum flow area ( $\mathrm{kg} \cdot \mathrm{m}^{2}$ ) |
| $h$ | convective heat transfer coefficient ( $\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}$ ) |
| $h_{c, o}$ | sensible capacity transfer coefficient (W/m² ${ }^{\text {a }}$ ) |
| $h_{d, o}$ | mass transfer coefficient ( $\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}$ ) |
| $h_{i}$ | inside heat transfer coefficient ( $\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}$ ) |
| $h_{o, w}$ | total heat transfer coefficient for wet external fin (W/m² ${ }^{2}$ ) |
| HVAC\&R | heating, ventilation, air-conditioning and refrigeration |
| $I_{0}$ | modified Bessel function solution of the first kind, order 0 |
| $I_{1}$ | modified Bessel function solution of the first kind, order 1 |
| $i$ | enthalpy ( $\mathrm{kJ} / \mathrm{kg}$ ) |
| $i_{a}$ | air enthalpy (kJ/kg) |
| $i_{a, \text { in }}$ | inlet air enthalpy ( $\mathrm{kJ} / \mathrm{kg}$ ) |
| $i_{a, m}$ | mean air enthalpy ( $\mathrm{kJ} / \mathrm{kg}$ ) |
| $i_{a, \text { out }}$ | outlet air enthalpy ( $\mathrm{kJ} / \mathrm{kg}$ ) |
| $i_{g}$ | saturated water vapor enthalpy ( $\mathrm{kJ} / \mathrm{kg}$ ) |
| $i_{m}$ | mean enthalpy (kJ/kg) |
| $i_{r, \text { in }}$ | saturated air enthalpy at the inlet water temperature ( $\mathrm{kJ} / \mathrm{kg}$ ) |
| $i_{r, m}$ | mean saturated air enthalpy at the mean water temperature ( $\mathrm{kJ} / \mathrm{kg}$ ) |
| $i_{r, \text { out }}$ | saturated air enthalpy at the outlet water temperature (kJ/kg) |
| $i_{s, f m}$ | saturated air enthalpy at the fin mean temperature ( $\mathrm{kJ} / \mathrm{kg}$ ) |


| $i_{s, f b}$ | saturated air enthalpy at the fin base temperature ( $\mathrm{kJ} / \mathrm{kg}$ ) |
| :---: | :---: |
| $i_{s, p, i, m}$ | mean saturated air enthalpy at the mean inside tube wall temperature ( $\mathrm{kJ} / \mathrm{kg}$ ) |
| $i_{s, p, o, m}$ | mean saturated air enthalpy at the mean outside tube wall temperature ( $\mathrm{kJ} / \mathrm{kg}$ ) |
| $i_{s, w}$ | saturated air enthalpy at the water film temperature ( $\mathrm{kJ} / \mathrm{kg}$ ) |
| $i_{s, w, m}$ | mean saturated air enthalpy at the mean water film temperature of the fin surface ( $\mathrm{kJ} / \mathrm{kg}$ ) |
| $j_{h}$ | Chilton-Coburn $j$-factor of the heat transfer |
| $j_{m}$ | Chilton-Coburn $j$-factor of the mass transfer |
| $K_{0}$ | modified Bessel function solution of the second kind, order 0 |
| $K_{1}$ | modified Bessel function solution of the second kind, order 1 |
| $k$ | thermal conductivity ( $\mathrm{W} / \mathrm{mK}$ ) |
| $k_{f}$ | thermal conductivity of fin (W/mK) |
| $k_{i}$ | thermal conductivity of water ( $\mathrm{W} / \mathrm{mK}$ ) |
| $k_{p}$ | thermal conductivity of tube (W/mK) |
| $k_{w}$ | thermal conductivity of water film (W/mK) |
| $L_{p}$ | tube length (m) |
| Le | Lewis number |
| $\dot{m}$ | mass flow rate ( $\mathrm{g} / \mathrm{s}$ ) |
| NTU | number of transfer unit |
| $N u$ | Nusselt number |
| $O D$ | outer diameter (m) |
| $P$ | pressure (Pa) |
| $\Delta P$ | pressure difference (Pa) |
| $P_{l}$ | longitudinal tube pitch (m) |
| Pr | Prandtl number |
| $P_{t}$ | transverse tube pitch (m) |
| $\dot{Q}$ | heat transfer rate (W) |
| $R e$ | Reynolds number |
| $R e_{D i}$ | Reynolds number based on inside diameter |
| $R_{\text {eq }}$ | equivalent radius for circular fin |
| $R_{\text {ov }}$ | overall thermal resistance (K/W) |
| $r$ | tube inside radius (m) |
| $r_{i}$ | distance from the center of the tube to the fin base (m) |
| $r_{o}$ | distance from the center of the tube to the fin tip (m) |
| Sc | Schmidt number |
| $T$ | temperature (K) |
| $T_{w, m}$ | mean temperature of the water film (K) |
| $T_{p, i, m}$ | mean temperature of the inner tube wall (K) |
| $T_{p, o, m}$ | mean temperature of the outer tube wall (K) |
| $T_{r, m}$ | mean temperature of water (K) |
| $\Delta T$ | temperature difference (K) |
| $\Delta T_{l m}$ | log mean temperature difference for counter-flow (K) |
| $t$ | Fin thickness (m) |
| $U$ | overall heat transfer coefficient ( $\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}$ ) |
| $U_{o, w}$ | wet surface heat transfer coefficient, based on enthalpy difference ( $\mathrm{kg} / \mathrm{m}^{2} \mathrm{~s}$ ) |
| VFR | volume flow rate ( $\mathrm{m}^{3} / \mathrm{s}$ ) |
| $W_{a}$ | humidity ratio of moist air ( $\mathrm{kg} / \mathrm{kg}$ ) |


| $W_{a, m}$ | mean air humidity ratio ( $\mathrm{kg} / \mathrm{kg}$ ) |
| :---: | :---: |
| $W_{s, p, o, m}$ | mean saturated air humidity ratio at the mean outside tube wall temperature (kg/kg) |
| $W_{s, w}$ | saturated air humidity ratio at the water film temperature ( $\mathrm{kg} / \mathrm{kg}$ ) |
| $W_{s, w, m}$ | mean saturated air humidity ratio at the mean water film temperature of the fin surface ( $\mathrm{kg} / \mathrm{kg}$ ) |
| $X_{L}$ | geometric parameter (m) |
| $X_{M}$ | geometric parameter (m) |
| $y_{w}$ | Thickness of condensate water film (m) |
| Greek symbols |  |
| $\rho$ | density ( $\mathrm{kg} / \mathrm{m}^{3}$ ) |
| $\varepsilon$ | fin factor |
| $\sigma$ | contraction ratio |
| $\eta_{0}$ | surface effectiveness |
| $\eta$ | fin efficiency |
| $\delta_{\text {f }}$ | fin thickness |
| $\mu$ | dynamic viscosity ( $\mathrm{Ns} / \mathrm{m}^{2}$ ) |
| Subscripts |  |
| $a$ | air side |
| $i$ | inlet |
| $m$ | mean |
| $n$ | exponent |
| $o$ | outlet |
| $t w$ | tube wall |
| w | wet |
| fr | frontal |
| r | refrigerant |

## Chapter 1: Motivations

### 1.1. Motivations

Air-to-fluid Heat eXchanger (HX) is widely used in industry as automotive radiator, oil cooler, water and glycol cooler, condenser, evaporator and indoor coil for chiller system. The heat exchange efficiency of the heat exchanger has a considerable influence on total energy usage.

As the economy develops and population grows, the demand for energy, material, space and other resources increases dramatically, calling for better energy, material and space usage. Compact heat exchanger that has less material, less envelope volume than traditional heat exchanger is the next generation heat exchanger.

The work presented here is motivated by the need of innovative designs and advanced geometries to improve heat exchanger performance. Heat transfer enhancement has been studied for decades and the traditional way to maximize the overall HX conductance (hA) is to increase surface area by employing extended surfaces, for instance fins, to airside to reduce airside thermal resistance because it accounts for 75~90\% of total thermal resistance. However, fin surfaces have lower heat transfer coefficients compared with primary heat transfer surfaces (tube surface) because of the boundary layer development. Moreover, additional fins inevitably result in more material consumption. Thus, conventional fin-and-tube heat exchanger designs have intrinsic limitation of further improving heat transfer performance giving certain amount of volume and material volume.

### 1.2. Literature Review

1.2.1. Airside heat transfer enhancement

Air-to-fluid heat exchangers in applications are mainly fin-and-tube heat exchangers, including round tube heat exchangers, flat tube heat exchangers and microchannel heat exchangers. A large amount of literature can be found in enhancing heat transfer of these conventional designs. Over 9500 literatures on Heat Transfer Enhancement (HTE) are published per Webb and Kim (2005). Webb and Kim (2005) categorized the heat transfer enhancement techniques into passive methods and active methods, as shown in Table 1. Compound enhancement and combinations of two or more passive and/or active techniques are also found. Passive techniques are the most sought (70.9\%) with emphasis on extended surfaces (Manglik and Bergles, 2004).

The ultimate objective of enhancing heat exchanger performance is to increase the total UA value, which can result in: (a) size reduction usually reflecting cost and weight reduction, (b) increase in heat duty, (c) LMTD reduction leading to better thermodynamic efficiency or (d) pumping power reduction which can be translated into the operating cost reduction. UA can be increased by increasing U, A or both. Since air thermal resistance accounts for 75\%~95\% of the overall HX thermal resistance, efforts to improve air-to-fluid heat exchanger performance mainly focus on air side.

Table 1 Classification of Various HTE Techniques

| Passive Techniques | Active Techniques |
| :---: | :---: |
| Surface coating | Mechanical aids |
| Rough surfaces | Surface vibration |
| Extended surfaces | Fluid vibration |
| Displaced inserts | Electrostatic fields |
| Swirl Flow | Injection |
| Coiled tubes | Suction |
| Surface tension | Jet impingement |
| Additives for liquids |  |
| Additives for gases |  |

Round tube-and-fin heat exchangers (RTHX) with various configurations were proposed and studied in the past decades. RTHX has become a universally used air-to-fluid heat exchanger type. Such heat exchanger consists of mechanically or hydraulically expanded equally spaced parallel tube bundles in staggered or in-lined pattern with continuous various configured fins on the outside of the tube array to improve heat transfer coefficient on airside. Usually liquid heat transfer medium such as water, oil, or refrigerant is forced to flow through the tube bundles while gas heat transfer medium, such as air, flows across the tubes through the fins. Different fins types have been studied comprehensively including plain fins, wavy fins or corrugated fins, louvered fins, offset strip fins, and perforated fins. The shapes of tubes are mostly round or oval. The fundamental characteristics of different fin types are summarized in a recent review paper (Bhuiyan and Islam, 2016) by numerically comparing heat transfer and pressure drop performance of different fin-and-tube heat exchangers under different conditions. Two basic concepts are extensively used for such extended surfaces: (i) special channel shapes, such as wavy channels, which provide mixing due to the boundary layer separation within the channel; (ii) repeated growth and wake destruction of boundary layers, such as offset strip fin, louvered fin and perforated fin (Bhuiyan and Islam, 2016). Another method to enhance heat transfer is vortex generator which not
only interrupts the development of thermal boundary layer but also generates longitudinal vertices and causes flow destabilization and it can be combined with all previous enhanced fin types. Jacobi and Shah (1995) reviewed the vortex generators in detail. Fundamentally heat transfer enhancement comes with pressure drop penalty. These different surfaces discussed above are summarized in Table 2.

Flat tube or mini- and micro- channel heat exchangers (MCHX) are standard heat exchangers for automobile radiators. They are also widely applied as air conditioning evaporators and condensers for residential, industrial and automotive use. Typical flat tube heat exchangers and microchannel heat exchangers are usually designed with multi-louvered fins of which the heat transfer enhancement mechanism was explained previously. The flat tube shape is also beneficial compared with RTHX regarding airside pressure drop. A typical MCHX comprises of a flat tube with multiple small sized ports. The advantage of MCHX over round fin-and-tube heat exchanger and flat tube heat exchanger is on the liquid side. By reducing the tube size, a factor of 10 in the heat dissipation of integrated chips using a microchannel heat exchanger was reported by Tuckerman and Pease (1981). Smaller liquid flow passages lead to higher surface area to volume ratio, higher thermal transport, smaller envelope volume, smaller overall refrigerant charge and higher system efficiency. Specially, the usage of headers simplifies the refrigerant circuitry and reduces the total flow path, resulting in reduced pressure drop.

Table 2 Fin types and HTE mechanism

| Fin type | Picture | HTE mechanism |
| :---: | :---: | :---: |
| Plain fin-and-tube HX (Kays and London, 1984) |  | - Extended heat transfer area |
| Wavy fin-and-tube HX (Kays and London, 1984) |  | - Extended heat transfer area <br> - Lengthened flow path <br> - Improved airflow mixing <br> - Boundary layer interruption and redevelopment at corrugations |
| Corrugated louvered fin-and-tube HX <br> (Bhuiyan and Islam, 2016) |  | - Extended heat transfer area <br> - Boundary layer is interruption and redevelopment at louvers <br> - Improved flow mixing |
| Offset strip fin-and-tube HX <br> (Bhuiyan and Islam, 2016) |  | - Extended heat transfer area <br> - Boundary layer interruption and redevelopment at strips <br> - Improved flow mixing |
| Perforated fin-and-tube HX <br> (Bhuiyan and Islam, 2016) |  | - Boundary layer dissipation in the wake region formed by holes <br> - Improved flow mixing |
| Vortex generator (Jacobi and Shah, 1995) | . | - Boundary layer interruption and redevelopment <br> - Generate longitudinal vortices <br> - Cause flow destabilization |

Recently, small diameter ( $<5 \mathrm{~mm}$ ) finless heat exchangers have been proposed and investigated. Bacellar (2014) numerically investigated bare tube heat exchangers and plain fin-and-tube heat exchangers with diameter of $2 \sim 5 \mathrm{~mm}$ and developed correlations based on CFD simulation results. Paitoonsurikarn et al. (2000) numerically found that bare tube heat exchanger can achieve a much larger air-side heat transfer coefficient ( $300 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$ ) with the air velocity range of $1 \sim 6 \mathrm{~m} / \mathrm{s}$ by reducing tube diameter to $0.3 \sim 0.5 \mathrm{~mm}$. Thus, finless designs using bare tubes with hydraulic diameter less than 1 mm can exceed the air-side heat transfer performance of conventional heat exchangers.
1.2.2. Heat transfer and friction characteristics of water flow in macro and micro-tubes ( $<4 \mathrm{~mm}$ )

In current study, water is used as the in-tube side fluid and single-phase heat transfer and friction characteristics are analyzed. As it is the first time to investigate the performance of the novel geometry, the single-phase characteristics are needed as 'building block' (Brognaux et al., 1997) for the development of the needed two-phase heat transfer mechanisms and correlations. So here the heat transfer enhancement approaches for in-tube side of bare tube are briefly summarized and the heat transfer and friction characteristics of water flow in macro and micro-tubes are analyzed. .

Although the airside heat transfer enhancement is significant, tube side heat transfer performance enhancement is also a major topic. Similar to the airside, there are also passive and active heat transfer enhancement methods. Passive heat transfer enhancement methods include enhanced inner surfaces (micro-fin tubes, inner grooved tubes) and inserts (coils, twisted tape). Enhanced inner surfaces are commonly used in air-cooled compact heat exchanger designs because they can produce enhanced tube side heat transfer coefficient with a small pressure drop penalty. Enhanced surfaces promote turbulence and reduce the thickness of the boundary layer, leading to higher local heat transfer coefficient.

Brognaux et al. (1997) investigated heat transfer and friction characteristics for singlephase flow in single-grooved and cross grooved micro-fin tubes with outer diameter of 15.987 mm . It was found that micro-fin tubes had enhanced heat transfer coefficient as high as 1.8 times that of smooth tubes. There is also research on two-phase flow in micro-fin tube with even smaller diameter (4~7 mm) (Hu et al., 2008, Mancin et al., 2016).

However, in current study, the diameter range is below 4 mm . As far as the author knows, there is no available microgroove or micro-fin tubes within this diameter range, especially for diameter that is less than 1 mm . Thus, one main task of current research is to find a way to enhance heat transfer in macro and micro tubes (Kandlikar, 2002).

Conventional forced convection heat transfer and friction correlations which were derived from tubes with diameter much larger than micro-tubes were examined by many researchers. Here is a summary:

## Friction factors

Yang et al. (2003) measured the friction factors of water flow in tubes with diameter ranging from 0.5 to 4 mm . They found that there is no significant discrepancy for water flow in small diameter tubes compared with large diameter tubes. Yang and Lin (2007) also found that for tubes diameter ranging from 100 to $1100 \mu \mathrm{~m}$, the test results agree very well with the conventional Poiseuille $\left(f=16 / R e_{d}\right)$ and Blasius $\left(f=0.079 R e_{d}^{-0.25}\right)$ equations in laminar and turbulent regime, respectively. Comparable results were reported by Lelea et al. (2004) for diameter range of $0.1 \sim 0.5 \mathrm{~mm}$, Li et al. (2003) for dimeter of 79.9 to $205.3 \mu \mathrm{~m}$ and Celata et al. (2002) for diameter of $130 \mu \mathrm{~m}$. So, the conventional theories are applicable for flow in the size range of current study $(0.5 \sim 4 \mathrm{~mm})$. This is the theoretical basis of baseline bare tube heat exchanger CFD simulation results verification for pressure drop.

## Fully developed heat transfer

Yu et al. (3) found the Nusselt numbers for water cooling in turbulent regime were considerably larger than those would be predicted for larger tubes, suggesting the Reynolds analogy does not apply for micro tube flow. Yen et al, (9) measured heat transfer performance of
laminar refrigerant R -123 in 0.3 mm diameter tube and found the results were in reasonable agreement with the analytical laminar constant heat flux value ( $\mathrm{Nu}=4.36$ ). However, in these experiments, the temperature was measured by direct attaching K-type thermocouple on the tube wall. Yang and Lin (2007) pointed out that the measurement accuracy of micro-tube wall temperature is significant because the diameter of the sensors for measuring micro-tube surface temperature is comparable to the size of the micro-tube itself. This may result in inaccurate temperature measurement because of sensor wire thermal shunt. A non-contacted liquid crystal thermography (LCT) method was proposed to measure the surface temperature of micro-tubes to avoid the thermal shunt and thermocouple contact problems. Samples with diameter of 123 to 962 $\mu \mathrm{m}$ were tested. They found the conventional heat transfer correlations for laminar and turbulent flow can be well applied for predicting the fully developed heat transfer performance in microtubes. The transition occurs at Reynolds number from 2300 to 3000 , which is the same range as that for conventional tubes. Correlations examined for turbulent flow include Dittus and Boelter (1930) correlation, Petukhov and Popov (1963) correlation and Gnielinski (1976) correlation. Thus, we can conclude there is no significant size effect for water flow in tubes within the diameter range in current study $(0.5 \sim 4 \mathrm{~mm})$. This serves as the theoretical basis of baseline bare tube heat exchanger CFD simulation results verification for heat transfer.

### 1.2.3. Nature inspired heat exchangers

Nature has inspired many scientists and engineers to solve problems through observation and mimicry. One such example is heat transfer enhancement. The enormous natural heat and mass transfer phenomena have led engineers to seek solutions to heat transfer enhancement problems from nature. Fractal geometries are found in respiratory and vascular systems of plants and animals, such as blood vessels, human lungs, leaves, coastlines, etc. Inspired by this, fractal heat exchangers
have been developed and are found to have intrinsic advantage of minimized flow resistance and strong heat transfer capability.

Significant amount of research has been conducted on the theory of fractals (Murray, 1926; Sherman, 1981; Mandelbrot, 1982; West, 1997; Bejan et al., 2008; Bejan and Lorente, 2006, 2007, 2011; Bejan, 1997, 2002, 2003; Xu and Yu , 2006), and the main findings are summarized in Table
3.

Table 3 Fractal theory development

| Researchers | Major findings |
| :--- | :--- |
| Murray, 1926 | Developed Murray's law: The cube of the radius of a parent branch equals the sum of the cubes of <br> the radii of daughter branches. |
| Sherman, 1981 | Found when Murray's law was obeyed a functional relationship exists between channel diameters <br> and various flow characteristics such as wall shear stress, velocity profile, and pressure gradient. |
| Mandelbrot, 1982 | Described fractal structure from nature: coastlines, leaves and clouds. |
| West, 1997 | Developed scaling laws for a bulk fluid transport problem to minimize the flow work. |
| Bejan et al., 2008; Bejan and Lorente, <br> $2006,2007,2011 ;$ Bejan, 1997, 2002,, <br> 2003 | Developed Constructal Theory: For a finite-size flow system to persist in time (to survive) its <br> configuration must evolve in such a way that it provides an easier access to the currents that flow <br> through it. |
| Xu and Yu, 2006 | Analyzed the transport properties including electrical conductivity, heat conduction, convective <br> heat transfer, laminar flow, and turbulent flow in the networks and derived the scaling exponents <br> of the transport properties in the networks. |

For application, the fractal channels (FC) are mainly used for electronic cooling. Thus, most research focuses on comparing its performance with traditional serpentine channel (SC) and parallel channel (PC). The major findings are summarized in Table 4. Even though there are various application of fractal channels, including heat sink (as shown in Table 2), fuel cell (Senn and Poulikakos, 2004), microreactor (Chen et al., 2011; Chen et al., 2015) distributor (Tondeur and Luo, 2004; Luo et al. 2007; Guo et al., 2014), collector (Guo et al., 2014), tube and shell heat exchanger (Guo et al., 2014), spindle (Xia et al., 2015), Si/Ge nanocomposite (Chen et al., 2015), etc. However, most of the research focus on heat sink for electronic cooling due to the inherent advantage of temperature uniformity of fractal structure. For fluid types, research cover liquid-toliquid (Tondeur and Luo, 2004; Luo et al., 2007; Guo et al., 2014), solid-to-liquid (as shown in Table 2), solid-to-two phase (Daniels et al. 2011; Daniels et al., 2007; Zhang et al., 2011), solid-to-gas (Chen et al., 2014; Chen et al., 2011; Chen et al., 2015), and solid-to-solid (Chen et al.,

## 2015) heat exchangers; however, no research has been done to liquid-to-gas heat exchangers,

## which is a research gap and should be investigated.

Table 4 Summary of major findings in research on fractal heat sinks

| Ref. | Major findings |
| :---: | :---: |
| $\begin{aligned} & \hline \text { Pence, } \\ & 2002 \end{aligned}$ | Compared with parallel channels (PC) with equal surface area, fractal channels (FC) has: <br> $1.60 \%$ lower pressure drop for the same total flow rate and $30^{\circ} \mathrm{C}$ lower wall temperature under identical pumping power condition. <br> 2. $50 \%$ lower density with similar maximum wall temperature and pressure drop. |
| Chen and Cheng, $2002$ | Compared with PC with equal surface area, FC has: <br> 1. Higher total heat transfer rate; <br> 2. Lower total pressure drop; <br> 3. Larger fractal dimension or a larger total number of branch levels will result in a stronger heat transfer capability with a smaller pumping power. |
| Senn and Poulikakos, 2004 | 1. Compared with SC with same heat transfer area and same rectangular area, FC has larger heat transfer capability and 50\% lower pressure drop; <br> 2. Pressure drop from bifurcation is substantial and not negligible; <br> 3. Lower pressure results from the not fully developed flow in higher branch level; <br> 4. Secondary flow motions initiates at bifurcations; <br> 5. Transverse vortices create recirculation at bifurcations that result in hot spots at the inner corners of bifurcations; <br> 6. Longitudinal vortices result in enhanced thermal mixing and a decrease in the required flow rate for heat transfer; <br> 7. Laminar mixing by secondary flow motions improves local Nusselt number. |
| Alharbi et al., 2003 | Compared with Pence's 1-D model, the 3D model: <br> 1. Predicts a $20 \%$ lower total pressure drop for fractal channels but similar for straight one, this is due to pressure recovery at bifurcations that results from an increase in flow area; <br> 2. Predicts pressure drop $17 \%$ higher for SC when using temperature dependent properties, but similar for FC ; <br> 3. Has the reinitiating assumption, which seems to provide plausible trends in pressure distribution. |
| Alharbi et al., 2004 | 1. FC has $75 \%$ lower temperature variation and a $10 \%$ pressure-drop penalty compared with the PC ; 2. The assumption of constant properties is not suitable for high heat flux condition. |
| Enfield et al., 2004 | 1. Developed a 2D model for predicting concentration profiles and degree of mixing (DoM); <br> 2. Developed a non-dimension number and a design guideline to determine the optimal number of branch levels to minimize pressure drop and maximize DoM for a fixed initial parent channel width, total path length, and channel depth. |
| Wang et al., 2006 | Compare FC with PC and SC, FC has: <br> 1. The best temperature uniformity; <br> 2. Lower pressure drop than SC but higher pressure drop than PC; <br> 3. Reduced risk of accidental blockage of channel segments; <br> 4. Reduced potential of thermal damage due to the reduced risk of blockage; <br> 5. Increased number of parent channels and branch levels resulted in increased temperature uniformity. |
| Wang et al., 2007 | 1. Pressure drop increases as bifurcation angle increases with a decreasing increasing rate and $30^{\circ}$ is the optimal angle; <br> 2. Channels with bifurcation angle of $180^{\circ}+180^{\circ}$ has a lower pressure drop compared with PC due to pressure recovery at bifurcation; <br> 3. Increasing angle also increases the risk of appearance of hotspot near the bifurcation; <br> 4. More uniform distribution of the outlet mass flow can be achieved with increased bifurcation angles, but the gradient is reduced with increasing angles. |
| Hong et al., 2007 | 1. A modified structure was proposed to address the hotspot issue (by adding serpentine channel structure at the end of highest branches); <br> 2. Hotspot appears at the highest branch (4th) due to assumption of conjugate heat transfer; <br> 3. Effect of bifurcation on pressure drop becomes more obvious for higher flow rate, resulting in a non-linear relationship between pressure drop and mass flow rate, unlike the linear one for PC; <br> 4. The modified FC is much better than that of PC with respect to pressure drop, thermal resistance and temperature uniformity; and this advantage is much more obvious when the flow rate or the pressure drop is low, which is favored because high pressure drop is not recommended in practice for the design of microsystems. |
| Chen et al., 2010 | 1. FC has considerable advantages over SC in both heat transfer and pressure drop; <br> 2. FC has inherent advantage of uniform temperature on the heating surface than SC. <br> 3. The local pressure loss due to confluence flow is found to be larger than that due to difluence flow. |
| Wang et al., 2010 | 1. Leaf-like flow networks has lower pressure drop and higher heat transfer coefficient than symmetric tree-like ones. |
| Yu et al., 2012 | 1. FC has a much higher heat transfer coefficient at the cost of a much higher pump power compared with PC with the same heat transfer area. <br> 2. AR (aspect ratio=height/width) of microchannel plays a very important role when considering pressure loss, heat transfer coefficient, and COP; <br> 3. FC with lowest AR has highest COP , but the one with highest AR has the highest ratio of COP over COP of PC . |
| Zhang et al., 2013 | 1. Small aspect ratio is preferred for a smaller pressure drop and a larger heat transfer rate; <br> 2. A high branch level produced a high pressure drop and a large heat transfer rate; <br> 2. The bends with fillets for the fractal-like microchannel reduce the local minor pressure losses, compared with that with the $90^{\circ}$ bends, resulting in a lower overall pressure drop. |


| Zhang et al., 2015 | 1. Both the flow rate and the AR have large influences on the evolution of the vortices, which promote the fluid mixing and <br> enhance the efficiency of heat transfer; <br> 2. FC with a smaller AR of 0.333 was verified to have lower pressure drop and better heat transfer performance within all the <br> other microchannel networks under investigation in the study; |
| :--- | :--- |
|  | 3. Observed transverse and longitudinal vorticities, secondary flow and recirculation flow motions; <br> 4. Confluence flow has a larger pressure drop than diffluent flow, but not much difference. |

1.2.4. Numerical study of compact heat exchangers

The design and sizing of heat exchangers involve many complex procedures and calculations. The convective heat transfer coefficients between fluids and walls are key variables in the design procedure. Heat transfer coefficients are functions of flow velocity, fluid properties, internal tube geometry and dimension, external tube geometry and dimension. For simple geometries, heat transfer coefficients are available in literature for single-phase flow at laminar, turbulent or transitional conditions (Sunden, 2007). However, for complex heat exchanger geometries or novel heat exchangers, analytical solutions and correlations in literature are no longer applicable. As a numerical solution methodology of governing equations for mass conservation, momentum, heat transfer and other transport process, Computational fluid dynamics (CFD) or computational heat transfer (CHT) or numerical heat transfer (NHT) has become a powerful tool for HX design nowadays (Sunden, 2007). In Sunden's review, the CFD can be applied to HX simulation in two ways: (i) entire heat exchanger or (ii) unitary cell. In the first way, the entire heat exchanger or the heat transferring surface is modeled by using large scale or coarse computational meshes or by applying a local averaging or porous medium approach. However, when applying this method, several assumptions including physical properties and characteristics should be known beforehand which is not practical for prediction. The advantage is the computational cost is relatively low. Another way is to identify modules or group of modules that repeat themselves in a periodic or cyclic manner in the main flow direction. This method enables accurate calculations by including more details in the model. Patankar et al. (1980) first introduced
stream wise periodic flow and heat transfer which becomes a common method for HX research, especially for prediction of novel surfaces and complex geometry. The CFD model should be validated using experimental data to demonstrate the validity of the computation. Abdelaziz et al. (2010) reported uncertainty of $\pm 10 \%$ in airside capacity against experimental data for a novel HX. Xiaoping et al. (2010) also reported $\pm 10 \%$ uncertainty for airside heat transfer coefficient predication against experimental data for a louvered fin microchannel heat exchanger. In Chen et al. (2016)'s study of a printed circuit heat exchanger, the largest deviation for prediction of Nusselt number is found to be $6.5 \%$ and that for capacity is $6.14 \%$. CFD has also been applied to investigate the airside flow maldistribution (Yaïci et al., 2016).

### 1.2.5. Heat exchanger design optimization

Design optimization involving CFD calculation is a major research topic, especially in novel heat exchanger design. To do optimization, the first step is to automate CFD calculation to make it easier to explore the design space. The reasons are, first although commercial software allows one to build model and mesh, it is hard to change the topology and second, the calculation time can be reduced by taking advantage of parallel calculation.

Hilbert et al. (2006) and Abdelaziz et al. $(2009,2010)$ both introduced such automated method and Abdelaziz named it Parallel Parameterized CFD (PPCFD). This PPCFD method relies on the commercial software Gambit (2017) for geometry and mesh generation and FLUENT ${ }^{\circledR}$ (2017) for solving the flow and energy equations. It has four steps:

- Generate profile for design variables and parameters
- Generate journal files for both Gambit and FLUENT ${ }^{\circledR}$
- Execute Gambit and FLUENT ${ }^{\circledR}$. using the journal files to generate geometry, generate mesh and run simulation in FLUENT ${ }^{\circledR}$
- Post process output data to obtain thermal-hydraulic performance

A batch file generated alongside executes these four steps. Depending upon the number of processors available, different cases can be run in parallel. It is reported that this automation technique saves more than $90 \%$ engineering time compared to conventional CFD modeling (Abdelaziz et al., 2010).

Even though the PPCFD technique can help reduce the computational time by automating the simulation procedure and using parallel computation, it may still result in huge computational cost when implementing this method into heat exchanger optimization problem. Therefore approximation-assisted optimization (AAO) has been applied for optimization using numerical simulations. Huang et al. (2015) summarized the optimization methods into five categories as shown in Table 5.

Table 5 Summary of HX optimization methods (adapted from Huang et al. 2015)

| Methodology | Expertise | Relative Computational Cost |
| :---: | :---: | :---: |
| Exhaustive Search | Low | $10,000,000$ |
| Random Search | Low | $1,000,000$ |
| Parametric Analysis | Low | 100,000 |
| Gradient-Based Methods | Medium | 10,000 |
| Heuristic Methods (e.g. GA's, MOGA, etc.) | Medium | 10,000 |
| Approximation Assisted Optimization <br> (offline) | High | $100-1000$ |
| Approximation Assisted Optimization <br> (online) | High | 100 |

One key procedure in approximation assisted optimization (AAO) is metamodeling. Metamodels are statistical approximations that are used to replace the actual models with acceptable estimation errors to significantly reduce the computation amount. Metamodeling
involves three steps (Simpson et al., 2001): (a) choose an experimental design for generating data, (b) choose a model to represent the data, and then (c) fit the model to the observed data. There are several options for each of these steps, as shown in Figure 1. First, AAO requires an initial set of sample points in the design space. It is recommended to use Design of Experiments (DoE). Classical designs tend to allocate points on or near the design space boundaries and leave a few points in the center, including full and fractional factorial designs, central composite designs and Box-Behnken designs. However, a good DoE should fill the entire design space instead of focusing only on the boundaries or at the center (Sacks et al., 1989, Jin et al., 2001), thus space filling designs should be applied, including Latin Hypercubes (LHC) (McKay, 1979), mean squared error (Jin et al., 2002), integrated mean squared error (Sacks et al., 1989), maximin distance approach (Johnson et al., 1990), orthogonal arrays (Taguchi, 1987; Owen, 1992), Hammersley sequences (Kalagnanam and Diwekar, 1997). Various metamodeling approaches in engineering optimization have been reported, including response surface techniques (Otto et al., 1996; Sobieski et al., 1998), Kriging (Jones et al., 1998), artificial neural networks (ANNs) (Fonseca et al., 2003), and inductive learning (Langley and Simon, 1995). Simpson et al. (2001) reviewed several of these techniques in detail.

Among them, Kriging meta-model technique is most widely used for heat exchangers because of its flexibility and suitability (Jones et al., 1998). The approximation can be done in both offline and online manners. In offline mode, the metamodeling procedure is carried out before any optimization is conducted, while in online approximation, an existing metamodel is updated during optimization with new points sampled in intermediate stages based on the progress of the optimizer. Recent air-refrigerant heat exchanger optimization research using AAO techniques are summarized in Table 6.


Figure 1 Techniques for metamodeling
(Simpson et al., 2001)

Table 6 Summary of air-refrigerant heat exchanger optimization using AAO techniques

| Authors | Experimental design | Meta- <br> model | Objectives |
| :---: | :---: | :---: | :---: |
| Aute (2008) | Space-Filling Cross- <br> Validation Tradeoff <br> (SFCVT) | Kriging | Multi-objective: <br> Min: ADP, V |
| Abdelaziz <br> $(2009)$ | MED | Kriging | Max: heat transfer capacity per <br> frontal area, per heat exchanger <br> volume, and per material. |
| Khaled et al. <br> $(2010)$ | Space-Filling Cross- <br> Validation Tradeoff <br> (SFCVT) | Kriging | Single-objective: <br> Min: ADP <br> Max: AHTC |
| Baceller <br> $(2016)$ | Latin Hypercube <br> Sampling (LHS) | Kriging | Multi-objective: <br> Min: ADP, V |

Based on the literature review, the main gaps are as follows:

- Air-side heat transfer coefficient of air-to-fluid heat exchanger still has the potential to be improved, and so is the fluid side. Which means the air-to-fluid heat exchanger still has the potential to have less pressure drop, less volume and less material volume when delivering the same capacity.
- Air flow is limited to be two dimensional on primary heat transfer surface (tube surface) for bare tube heat exchangers. There is no bare tube heat exchanger design that utilizes 3D flow on primary heat transfer surface.
- Nature inspired heat exchange devices haven't been systematically reviewed and there lacks design guideline for nature inspired thermal device design.
- Though numerical studied, small diameter fin-and-tube heat exchangers and bare tube heat exchangers have not been widely experimentally investigated.


### 1.3. Objectives

There are three main objectives of current research: (1) experimentally investigate the performance of state-of-the-art heat exchangers to gain the heat exchanger design fundamentals, including traditional louvered finned mini-channel heat exchangers, small diameter fin-and-tube heat exchanger ( $4 \sim 5 \mathrm{~mm}$ ), newly studied round and shape optimized bare tube heat exchanger ( $0.8 \sim 1 \mathrm{~mm}$ ); (2) invent, simulate and optimize a novel heat exchanger that has improved thermal and hydraulic performance as compared to baselines on both airside and fluid side and validate the novel heat exchanger performance against experimental data, (3) analyze the applicability of the
novel heat exchanger; and discuss its applicability as automotive radiator and indoor coil for hybrid variable flow system in detail.

Investigation of the state-of-the-art heat exchangers performance encompasses: (a) design and construct an ASHRAE standardized air-to-water/Ref. test facility for heat exchanger capacity ranging from 1 to 10 kW with uncertainty of less than $\pm 5 \%$; (b) experimentally investigate the single-phase transfer using water and air for mini-channel heat exchangers, small diameter fin-and-tube heat exchanger $(4 \sim 5 \mathrm{~mm})$ and bare tube heat exchanger ( $0.8 \sim 1 \mathrm{~mm}$ ); (c) experimentally investigate the two-phase heat transfer using refrigerant (R410A) and air for fin-and-tube heat exchangers; (d) develop air-side heat transfer coefficient and pressure drop correlations for bare tube heat exchangers ( $0.8 \sim 1 \mathrm{~mm}$ ) which have been rarely investigated in literature; (e) analyze the thermal and hydraulic mechanisms of distinct fins and tube shapes to gain the fundamental understanding of heat transfer and address potentials and limitations of compact air-to-fluid heat exchanger design.

Novel air-to-fluid heat exchanger design optimization includes: (a) review nature inspired heat exchange devices comprehensively and develop a design guideline for nature-inspired heat exchangers; (b) invent a heat exchanger with newly defined geometry; (b) conduct numerical simulation and parametric study on both airside and waterside using Computational Fluid Dynamics (CFD) software; (c) conduct experimental tests to validate CFD simulation; (d) explore thermal and hydraulic characteristics of the novel air-to-refrigerant heat exchanger on both airside and waterside; (d) develop Parallel Parameterized CFD (PPCFD) technique in ANSYS ${ }^{\circledR}$ Workbench ${ }^{\mathrm{TM}}$ and couple it with Approximation Assisted Optimization (AAO) technique; (e) optimize current heat exchanger so that it has $30 \%$ lower total pumping power, $30 \%$ less volume and $30 \%$ less material volume than those of baseline louvered fin mini-channel heat exchanger
with similar capacity, same or smaller frontal area, similar aspect ratio and same working condition; (f) optimize current heat exchanger so that it has $10 \%$ lower total pumping power, $10 \%$ less volume and $10 \%$ less material volume than those of baseline bare tube heat exchanger with similar capacity, same or smaller frontal area, similar aspect ratio and same working condition.

Applicability analysis of the novel heat exchanger includes: (a) design an air-to-water automotive radiator that has $30 \%$ lower total pumping power, $30 \%$ less volume and $30 \%$ less material volume than those of baseline which is the widely used louvered fin and flat tube heat exchanger that has been tested; (b) model a Hybrid Variable Refrigerant Flow (HVRF) system numerically with current heat exchanger design as indoor coil that has $30 \%$ less charge than those of baseline Refrigerant Flow (VRF) system; and, investigate the performance of HVRF system with different refrigerants, including R410A, R290 and R600a.

### 1.4. Dissertation organization

This dissertation is organized so that the research motivation, literature review and research objectives are presented in Chapter 1. Chapter 2 summarizes the research approaches including experimental study, CFD simulation, approximation techniques and multi-scale optimization. Chapter 3 discusses the test results of all baseline heat exchangers that were tested using water and air under both dry and dehumidifying conditions. The novel bifurcated bare tube heat exchanger (bBTHX) is presented in Chapter 4, 5 and 6. In Chapter 4, the bBTHX design concept is shown and parametric study results are discussed for both airside and waterside. The thermal and hydraulic mechanisms are discussed in detail to demonstrate the advantages. In Chapter 5, the metamodel and bBTHX solver are explained and the optimization results are discussed. Two optimization cases are studied. In Chapter 6, two applications of current design are described.

Chapter 7 draws the main conclusions, summarizes the contributions and related publications and provides recommendations for future work. Test facility pictures, HX blockage test results, HX experimental data, Grid Convergence Index (GCI) calculation data, meta-model data and optimization results are listed in the appendices.

## Chapter 2: Research Approach

This chapter gives a detailed description of research approaches of experimental test, Computational Fluid Dynamics (CFD) simulation and optimization.

### 2.1. Experimental Study

### 2.1.1. Test facility

The experimental setup (shown in Figure 2) consists of a closed-loop wind tunnel, a water system, a refrigerant pumped system and a data acquisition system, which is capable of testing heat exchangers from 100 W to 10 kW capacity under a wide range of operating conditions.

The closed-loop wind tunnel was designed and constructed based on the ASHRAE standard 41.2 (ASHRAE, 1987). It included three parts: (1) a test section with two different duct sizes to test heat exchangers with cross section area as large as $0.66 \times 0.66 \mathrm{~m}$, (2) a flow measurement chamber with nozzle grid to measure air-side mass flow rate and (3) an air return duct with three air handling coils with hot water, cold water and glycol water and a humidifier to control the inlet air temperature and humidity. A variable speed fan was installed to allow different air velocities through the wind tunnel.

Two ducts were built independently as test sections with the cross-section area of $0.33 \times$ 0.33 m and $0.66 \times 0.66 \mathrm{~m}$, respectively. Both the test sections have the same length of 2.5 m . There is no difference in construction materials and instrumentations between the small and large ducts except for the cross-section area. During a heat exchanger test, one test section is selected based on heat exchanger size, and the other is blocked meanwhile. Both test sections were built using 9.53 mm polypropylene plates with a thermal conductivity of $0.5 \mathrm{~W} / \mathrm{mK}$ and insulated from the
inside using 50.8 mm thick polystyrene foams with a thermal conductivity of $0.12 \mathrm{~W} / \mathrm{mK}$, to prevent heat loss from or into the duct.

The heat exchanger prototype was placed inside the smaller test section. The gaps between the heat exchanger and the section were completely blocked with plates to avoid bypass flow, with leading edge in the air flow direction. Two sampling trees which were branch structure made of thin copper tubes with uniform sampling holes distributed over cross section area were installed at the upstream and downstream of heat exchanger. Sample air was drawn from the sampling tree by a blower mounted outside the duct and supplied back into the duct to ensure same amount of air flow. Air flow rate inside the sampling tree was only $1 \mathrm{~L} / \mathrm{min}$ through a moisture resistant tube with an inner diameter of 3.2 mm thus the power consumption of blower is too small to cause any significant temperature increase of the air. Along the sampling air tube, there was a 4 -wire Class 1/10DIN RTD sensor and a chilled mirror hygrometer chamber to measure the temperature and dew point temperature of sample air. Two sets of mixers and settling means were installed before and after sampling devices to ensure the uniformity of air stream.


Figure 2 Heat exchanger test facility - wind tunnel and water loop

Settling means are two layers of metal mesh with $51 \%$ open area each which also functions as supporting structure of sampling trees and thermal couple grids with six and nine T-type thermocouples for small and big test sections, respectively. Readings of thermocouples were used as a supplement of RTD for temperature measurement and as a mean of air uniformity check. RTD readings was used in the data reduction due to its high accuracy. Both barometric and differential
pressure transducers were installed with ASHRAE standard pressure tap to measure absolute pressure before heat exchanger and pressure drop across heat exchanger.

Air flow measurement chamber is in the downstream of heat exchanger testing section. A nozzle grid of seven ASHRAE-standard nozzles with diameter of $0.051,0.076,0.076,0.127$, $0.127,0.178$ and 0.178 m were installed in the middle of this chamber to cover the volume flow rate range of 0.03 to $1.65 \mathrm{~m}^{3} / \mathrm{s}$. Nozzles are plugged when not in use. Two sets of settling means were installed before and after the nozzle grid to straighten the flow. A sampling tree integrated with a four-wire-class 1/10DIN RTD sensor was installed in the upstream of nozzle grid to measure the air temperature of nozzle. Both barometric and differential pressure transducers were connected to the standard pressure tap ring to measure the absolute pressure before nozzle and pressure drop across nozzle. Nozzles were calibrated under the procedure described in ASHRAE standard 41.2 (AHRAE, 1987). This procedure is based on energy conservation, which means heat output from electric heater should be equal to heat input into the air inside the duct. Electric heaters with capacities of $0.5,4,7$ and 9 kW were used to ensure the air temperature rise to be no less than $10^{\circ} \mathrm{C}$ to reduce uncertainty. Watt meter with a high accuracy of $\pm 0.2 \%$ of reading was used to measure the heater's actual power $\left(\dot{Q}_{\text {wattmeter }}\right)$. This should equal the heating capacity of the air which was calculated using air flow rate and air temperature difference, as shown in Equation (1). Air flow rate was measured by nozzle and was calculated by using equations from ASHRAE standard 41.2 (ASHRAE, 1987) without correction factor.

$$
\begin{equation*}
\dot{Q}_{\text {nozzle }}=\rho V F R_{\text {nozzle }} C_{p, a} \Delta T \tag{1}
\end{equation*}
$$

Comparison between the actual heater output (with an uncertainty of $\leq 3 \%$ ) and heating capacity measured by nozzle (with an uncertainty of $\leq 1 \%$ ) are shown in Figure 3. The differences
for all flow rates are within $\pm 4.8 \%$, indicating that no correction factor is needed for this nozzle grid.

Air return duct is a metal duct with outside insulation to reduce heat loss. Three coils were installed to control the air inlet temperature. A tankless gas water heater with the highest outlet temperature of $185^{\circ} \mathrm{F}$ heated the hot water loop. A chiller with the lowest outlet temperature of $5^{\circ} \mathrm{C}$ and $-11^{\circ} \mathrm{C}$ cools the cold-water loop and glycol water loop respectively. Three mixing valves are located on each water line to regulate the temperature of each coil thus to control the temperature of inlet air of test heat exchanger. Cold water and glycol water lines are also used to dehumidify the air while an electric steam humidifier is used to increase the humidity.

The water loop of the heat exchanger test facility consists of (1) a variable speed gear pump to provide different steady water flow rates, (2) two 4-wire Class 1/10 DIN RTD sensors and two absolute pressure transducers installed at the inlet and outlet of heat exchangers, (3) a differential pressure transducer to measure the pressure drop across the heat exchanger, (4) a Coriolis mass flow meter to measure mass flow rate and density of water, (5) a water buffer tank to eliminate small fluctuations of water temperature and (6) a plate heat exchanger that exchangers heat from a tankless gas water heater to water inside the water tank.


Figure 3 Comparison of air flow rate measured by wattmeter and nozzles

## Refrigerant Pumped Loop



Figure 4 Heat exchanger test facility - refrigerant pumped loop

Besides water loop, a refrigerant pumped loop was built as well to conduct two-phase testing. The refrigerant pumped loop consists of (1) a variable speed gear pump to provide different refrigerant flow rates, (2) four 4-wire Class $1 / 10$ DIN RTD sensors and two absolute pressure transducers installed at the inlet and outlet of heat exchangers, (3) a differential pressure transducer
to measure the pressure drop across the heat exchanger, (4) a Coriolis mass flow meter to measure mass flow rate and density of refrigerant, (5) one 5 kW electric heater and one 10 kW electric heater to heat refrigerant and watt meters to measure power consumption in order to back calculate inlet quality, (6) a receiver, (7) a plate heat exchanger to balance heat transfer, and (8) a glycol water/chilled water tank connected to plate heat exchanger to control the temperature. This refrigerant loop can handle refrigerant like R410A, R134a etc.

All data were recorded after steady state was reached for each test. All data signals of instrumentation were collected by a data acquisition system (DAQ) and transmitted to a computer.
2.1.2. Instrumentation and data acquisition system

Table 7 summarizes the instruments installed in the test facility and the corresponding uncertainties. National Instrument's compact field point models were installed as the data acquisition (DAQ) systems. All the measured signals were collected using LabVIEW program. In addition to receiving signals, the program also provided signals output to system so that a proportional-integral-derivative (PID) control was used to control temperatures, mass flow rates and pressures.

Table 7 Measurement instrument

| Instrument | Type | Manufacturer | Model | Range | Systematic <br> Uncertainty |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Mass Flow <br> Meter | Coriolis | Micro Motion | 2700 | $0 \sim 500 \mathrm{~g} / \mathrm{s}$ | $0.1 \%$ |
| RTD | Resistance | Omega | PR-25AP | $-200 \sim 800^{\circ} \mathrm{C}$ | $0.03 \sim 0.07{ }^{\circ} \mathrm{C}$ |
| Thermocouple | T | Omega | Tt-T-24-100 | $-250^{\circ} \sim 350{ }^{\circ} \mathrm{C}$ | $0.5^{\circ} \mathrm{C}$ |
| Dew point <br> sensor | Chilled <br> mirror <br> hygrometer | EdgeTech | DewTrak II Chilled <br> Mirror Transmitter | $-40 \sim 60{ }^{\circ} \mathrm{C}$ | $\pm 0.2^{\circ} \mathrm{C}$ <br> dew/frost point |
| Barometric | Strain | Setra | $2781600 \mathrm{MA1B2BT1}$ | $60 \sim 110 \mathrm{kPa}$ | $\pm 100 \sim \pm 200 \mathrm{~Pa}$ |
| Differential | Strain | Setra | 2641001 WD 11 T 1 F | 0 to $1 " \mathrm{~W} . \mathrm{C}$. | $(+/-0.25 \% \mathrm{FS}) \pm$ <br> 0.62 Pa |
|  |  | 2641005 WD 11 T 1 F | 0 to $0.5 \% \mathrm{~W} . \mathrm{C}$. | $(+/-0.25 \% \mathrm{FS}) \pm$ <br> 3 Pa |  |
| Watt meter | Suspended <br> coil torsion | Ohio <br> Semitronics Int | GH-020D | $0 \sim 4 \mathrm{~kW}$ | $\pm 0.2 \%$ of <br> reading |

### 2.1.3. Energy balance

For each test, the capacity was measured from both air side and water/refrigerant side. Per ASHARE standard 33, the difference between these two capacities, which is defined as energy balance, should be less than $\pm 5 \%$. The definition of energy balance is shown in Equation (2).

$$
\begin{equation*}
E B=\frac{\dot{Q}_{a}-\dot{Q}_{w / r e f}}{\dot{Q}_{a}+\dot{Q}_{w / r e f}} \times 2 \times 100 \% \tag{2}
\end{equation*}
$$

### 2.1.4. Uncertainty analysis

Total uncertainty is the summation of systematic uncertainty and random uncertainty. Systematic uncertainty is caused by measurement and instrumentation which is defined as the difference between the true value and the value that instrument can measure. Systematic uncertainty sources include imperfect calibration of instruments, changes in the environment and imperfect methods of observation. Random uncertainty is caused by predictable fluctuation in reading which is usually represented by standard deviation. Random uncertainty sources include lack of sensitivity, random noises and statistical processes.

Data could be classified into directly measured data (e.g. temperature $T$ ) and calculated data (e.g. heating capacity $\dot{Q}$ ). For directly measured data, uncertainty sources include instrument systematic error and random error which is represented as the standard deviation of data series. Total uncertainty of calculated parameter is determined by uncertainty propagation with total uncertainties of directly measured parameters. Pythagorean summation is used for uncertainty propagation. For a calculated value $f$, which is calculated using measured value $x_{1}, x_{2}, \cdots, x_{n}$ as shown in Equation (3), the total uncertainty of $f$, is then calculated as in Equation (4). Equation (5) and (6) is an example of uncertainty propagation for enthalpy. In current study, the uncertainty propagation is conducted using software Engineering Equation Solver (EES). Experimental uncertainties of the key parameters are summarized in Table 8. Uncertainty was evaluated for each test in current study.

$$
\begin{align*}
& f=f\left(x_{1}, x_{2}, \cdots, x_{n}\right)  \tag{3}\\
& \omega_{f}=\sqrt{\left(\frac{\partial f}{\partial x_{1}} \omega_{x_{1}}\right)^{2}+\left(\frac{\partial f}{\partial x_{2}} \omega_{x_{2}}\right)^{2}+\cdots+\left(\frac{\partial f}{\partial x_{n}} \omega_{x_{n}}\right)^{2}}  \tag{4}\\
& h=f(P, T)  \tag{5}\\
& \omega_{h}=\sqrt{\left(\frac{\partial f}{\partial P} \omega_{P}\right)^{2}+\left(\frac{\partial f}{\partial T} \omega_{T}\right)^{2}} \tag{6}
\end{align*}
$$

Table 8 Total uncertainty in key parameters

|  | Air | Water |
| :---: | :---: | :---: |
| Temperature | $\pm 0.18 \mathrm{~K}$ | $\pm 0.14 \mathrm{~K}$ |
| Flow rate | $\pm 0.8 \sim 1.7 \%$ | $\pm 0.2 \sim 0.3 \%$ |
| Pressure drop | $\pm 1.5 \sim 2.1 \%$ | $\pm 1.4 \sim 2.5 \%$ |
| Capacity | $\pm 1.2 \sim 2.1 \%$ | $\pm 1.6 \sim 3.1 \%$ |

2.1.5. Data reduction

For heat exchanger, both dry and wet condition tests were conducted, thus here the data reduction method are explained separately.

## Dry condition

The heat exchanger capacities under dry conditions were calculated for air-side and waterside using Equation (7) and (8), respectively,

$$
\begin{align*}
& \dot{Q}_{a}=C_{p, a} \dot{m}_{a}\left(T_{a, o}-T_{a, i}\right)  \tag{7}\\
& \dot{Q}_{r}=C_{p, r} \dot{m}_{r}\left(\mathrm{~T}_{\mathrm{r}, \mathrm{i}}-\mathrm{T}_{\mathrm{r}, \mathrm{o}}\right) \tag{8}
\end{align*}
$$

For further data reduction, the arithmetic average of air-side capacity $\dot{Q}_{a}$ and water-side capacity $\dot{Q}_{r}$ are used as the heat exchanger capacity, $\dot{Q}$, represented by Equation (9).

$$
\begin{equation*}
\dot{Q}=\frac{\dot{Q}_{a}+\dot{Q}_{r}}{2} \tag{9}
\end{equation*}
$$

To calculate the air heat transfer coefficient, Wilson plot method was used. The Wilson plot method was first proposed by Wilson (Wilson, 1915), and is a widely-used method to determine the convective HTC using experimental data (Fernández-Seara, 2007). In this case, Wilson plot method was used to find the air-side HTCs for different flow rates. The overall thermal resistance could be expressed as the summation of water-side convective thermal resistance, $R_{w}$, tube wall thermal resistance $R_{t w}$ and air-side convective thermal resistance $R_{a}$. Thermal resistance due to fluid fouling was neglected. Thus, the overall thermal resistance could be written as Equation (10).

$$
\begin{equation*}
R_{o v}=R_{a}+R_{t w}+R_{r} \tag{10}
\end{equation*}
$$

Based on experimental data, overall thermal resistance ( $R_{o v}$ ) could be evaluated using either $\varepsilon$-NTU method or LMTD method. LMTD method was chosen for this study because the inlet and outlet temperatures of both fluids are known. For a cross-flow heat exchanger, we could evaluate the heating capacity as Equation (11).

$$
\begin{equation*}
\dot{Q}=U A F \Delta T_{l m} \tag{11}
\end{equation*}
$$

Where $\Delta T_{l m}$ is the log mean temperature difference for counter-flow configuration, and F is the $\log$ mean temperature difference correction factor for the cross-flow heat exchanger, then F is determined by temperature effectiveness, heat capacity rate ratio. Thus, overall thermal resistance could be expressed as Equation (12).

$$
\begin{equation*}
R_{o v}=\frac{1}{U A}=\frac{F \Delta T_{l m}}{\dot{Q}} \tag{12}
\end{equation*}
$$

We kept the air-side HTC, the air flow rate constant while water flow rate was varied. Wilson (1915) theorized that if the mass flow of the water was modified, then the change in the overall thermal resistance would mainly be due to the variation of the air-side HTC (while the remaining thermal resistances remained nearly constant). Since air inlet temperature and water inlet temperature were constant, we could assume the thermal resistance of air-side and tube wall to be a constant number. Air-side convective thermal resistance $R_{a}$ and tube wall thermal resistance $R_{t w}$ could be expressed as Equation (13).

$$
\begin{equation*}
R_{t w}+R_{a}=C_{1} \tag{13}
\end{equation*}
$$

The convective HTC of water was proportional to a power of the velocity as written as Equation (14).

$$
\begin{equation*}
h_{r}=C_{2} v_{r}^{n} \tag{14}
\end{equation*}
$$

Here the coefficient $C_{2}$ and exponent of water velocity $n$ are unknowns. By combining Equation (12), (13) and (14), the regression form is derived as Equation (15).

$$
\begin{equation*}
R_{o v}=\frac{1}{C_{2} A_{w}} * \frac{1}{v_{w}^{n}}+C_{1} \tag{15}
\end{equation*}
$$

Thus, a linear regression was applied to obtain the values of $C_{1}, C_{2}$ and best curve fitting was used to find $n$. Then the air-side thermal resistance $R_{a}$ could be calculated using Equation (13).

For heat exchanger with fins, the air-side thermal resistance can be represented as Equation (16). Thus, to find air-side heat transfer coefficient, iteration must be conducted to determine fin efficiency using Equation (17) to (22).

$$
\begin{equation*}
R_{a}=\frac{1}{\eta_{0} h A_{0}} \tag{16}
\end{equation*}
$$

The $\eta_{0}$ here is the surface effectiveness, and is related to the fin surface area, total surface area and fin efficiency by Schmidt (1949) equation.

$$
\begin{equation*}
\eta_{0}=1-\frac{A_{f}}{A_{0}}(1-\eta) \tag{17}
\end{equation*}
$$

where
$\eta=\frac{\tanh (m r \phi)}{m r \phi}$
$m=\sqrt{\frac{2 h}{k_{\mathrm{f}} \delta_{\mathrm{f}}}}$
$\phi=\left(\frac{R_{e q}}{r}-1\right)\left[1+0.35 \ln \left(R_{e q} / \mathrm{r}\right)\right]$
$R_{e q}=1.27 \frac{X_{M}}{r}\left(\frac{X_{L}}{X_{M}}-0.3\right)^{1 / 2}$ for staggered tube layout

$$
\begin{equation*}
R_{e q}=1.28 \frac{X_{M}}{r}\left(\frac{X_{L}}{X_{M}}-0.2\right)^{1 / 2} \text { for single row coil and inline layout } \tag{22}
\end{equation*}
$$

For finless heat exchangers, $\eta_{0}=1$, thus $h$ can be directly calculated using Equation (16).

The Chilton-Colburn j factor and f factor is expressed as follows:

$$
\begin{align*}
& j_{a}=\frac{h_{a}}{G_{\max } C_{p, a}} \operatorname{Pr}^{2 / 3}  \tag{23}\\
& f_{a}=\frac{A_{c}}{A} \frac{\rho_{m}}{\rho_{a, i}}\left[\frac{2 \Delta P \rho_{a, i}}{G_{\max }^{2}}-\left(1+\sigma^{2}\right)\left(\frac{\rho_{a, i}}{\rho_{a, o}}-1\right)\right] \tag{24}
\end{align*}
$$

## Wet condition

Data reduction for wet condition is more complicated. Basically, the present reduction method is based on the Threlkeld (1970) method. The main steps are to calculate overall wet heat transfer coefficient $h_{o, w}$ and sensible capacity transfer coefficient $h_{c, o}$ first and then calculate mass transfer coefficient $h_{d, o}$. Details are as below.

The total heat transfer rate for air side is shown in Equation (25). The water side heat transfer rate is the same as shown in Equation (8).

$$
\begin{equation*}
\dot{Q}_{a, w}=\dot{m}_{a}\left(i_{a, o}-i_{a, i}\right) \tag{25}
\end{equation*}
$$

The overall heat transfer coefficient, based on the enthalpy potential is given as follows:

$$
\begin{equation*}
\dot{Q}_{w}=U_{o, w} A_{o} \Delta i_{m} F \tag{26}
\end{equation*}
$$

where $\Delta i_{m}$ is the mean enthalpy difference for counter flow coil,

$$
\begin{equation*}
\Delta i_{m}=i_{a, m}-i_{r, m} \tag{27}
\end{equation*}
$$

The mean enthalpy difference for the counter flow configuration is listed in Equation (28) and (29) according to Bump (1963) and Myers (1967).

$$
\begin{gather*}
i_{a, m}=i_{a, \text { in }}+\frac{i_{a, \text { in }}-i_{a, \text { out }}}{\ln \left(\frac{i_{a, \text { in }}-i_{r, \text { out }}}{i_{a, \text { out }}-i_{r, \text { in }}}\right)}-\frac{\left(i_{a, \text { in }}-i_{a, \text { out }}\right)\left(i_{a, \text { in }}-i_{r, \text { out }}\right)}{\left(i_{a, \text { in }}-i_{r, \text { out }}\right)-\left(i_{a, \text { out }}-i_{r, \text { in }}\right)}  \tag{28}\\
i_{r, m}=i_{r, \text { out }}+\frac{i_{r, \text { out }}-i_{r, \text { in }}}{\ln \left(\frac{i_{a, \text { in }}-i_{r, \text { out }}}{i_{a, \text { out }}-i_{r, \text { in }}}\right)}-\frac{\left(i_{r, \text { out }}-i_{r, \text { in }}\right)\left(i_{a, \text { in }}-i_{r, \text { out }}\right)}{\left(i_{a, \text { in }}-i_{r, \text { out }}\right)-\left(i_{a, \text { out }}-i_{r, \text { in }}\right)} \tag{29}
\end{gather*}
$$

The overall heat transfer coefficient is related to the individual heat transfer resistance (Myers, 1967) as follows:

$$
\begin{equation*}
\frac{1}{U_{o, w}}=\frac{b_{r}^{\prime} A_{0}}{h_{i} A_{p, i}}+\frac{b_{r}^{\prime} A_{0} \ln \left(\frac{D_{c}}{D_{i}}\right)}{2 \pi k_{p} L_{p}}+\frac{1}{h_{o, w}\left(\frac{A_{p, o}}{b_{w, p}^{\prime} A_{0}}+\frac{A_{f} \eta_{f, w}}{b_{w, m}^{\prime} A_{0}}\right)} \tag{30}
\end{equation*}
$$

where

$$
\begin{equation*}
h_{o, w}=\frac{1}{\frac{C_{p, a}}{b_{w, m}^{\prime} h_{c, o}}+\frac{y_{w}}{k_{w}}} \tag{31}
\end{equation*}
$$

$y_{w}$ in Equation (31) is the thickness of the water film. A constant of 0.005 inch was proposed by Myers (1967). In practice, $\left(y_{w} / k_{w}\right)$ accounts for only $0.5-5 \%$ compared to $\left(C_{p, a} / b_{w, m}^{\prime} h_{c, o}\right)$, and has often been neglected by previous investigators. As a result, this term is not included in the final analysis.

$$
\begin{array}{lr}
b_{r}^{\prime}=\frac{i_{s, p, i, m}-i_{r, m}}{T_{p, i, m}-T_{r, m}} & \begin{array}{c}
\text { Slope of the air saturation curved at the mean } \\
\text { coolant temperature and the inside wall temperature }
\end{array} \\
b_{p}^{\prime}=\frac{i_{s, p, o, m}-i_{s, p, i, m}}{T_{p, o, m}-T_{p, i, m}} & \text { Slope of a straight line between the outside and }  \tag{33}\\
\text { inside tube wall temperature }
\end{array}
$$

$$
b_{w, p}^{\prime}=\frac{\Delta i_{s, w, p}}{\Delta T_{w, p}} \quad \begin{align*}
& \text { Slope of the air saturation curve at the water film }  \tag{34}\\
& \text { temperature of the tube wall surface }
\end{align*}
$$

$b_{w, m}^{\prime}=\frac{\Delta i_{s, w, m}}{\Delta T_{w, m}}$
Slope of the air saturation curve at the water film temperature of the fin surface

Tube-side heat transfer coefficient, $h_{i}$ is evaluated from the Gnielinski correlation (Gnielinski, 1976).

$$
\begin{equation*}
h_{i}=\frac{\left(f_{i} / 8\right)\left(R e_{D i}-1000\right) \operatorname{Pr}}{1+12.7 \sqrt{\frac{f_{i}}{8}}\left(\operatorname{Pr}^{\frac{2}{3}}-1\right)} \cdot \frac{k_{i}}{D_{i}}, 2300<R e_{D i}<10^{4} \tag{36}
\end{equation*}
$$

Where $f_{i}$ is calculated using the correlation developed by Petukhov (1970), as shown below,

$$
\begin{equation*}
f_{i}=\frac{1}{\left(0.79 \ln R e_{D i}-1.64\right)^{2}}, 3000<R e_{D i}<5 \times 10^{6} \tag{37}
\end{equation*}
$$

Note that the Reynolds number used in Equations (36) and (37) is based on the inside diameter of the tube.

To calculate the overall heat transfer coefficient, similarly, the wet fin efficiency must be evaluated through iteration. The wet fin efficiency, defined by Threlkeld (1970), is shown in Equation (38).

$$
\begin{equation*}
\eta_{f, w e t}=\frac{i-i_{s, f m}}{i-i_{s, f b}} \tag{38}
\end{equation*}
$$

where $i_{s, f m}$ is the saturated air enthalpy at the man temperature of fin and $i_{s, f b}$ is the saturated air enthalpy at the fin base temperature. The fin efficiency under wet condition is calculated using Equations (39) and (40).

$$
\begin{align*}
& \eta_{f, \text { wet }}=\frac{2 r_{i}}{M_{r}\left(r_{o}^{2}-r_{i}^{2}\right)}\left[\frac{K_{1}\left(M_{T} r_{i}\right) I_{1}\left(M_{T} r_{o}\right)-K_{1}\left(M_{T} r_{o}\right) I_{1}\left(M_{T} r_{i}\right)}{K_{1}\left(M_{T} r_{o}\right) I_{o}\left(M_{T} r_{i}\right)-K_{o}\left(M_{T} r_{i}\right) I_{1}\left(M_{T} r_{o}\right)}\right]  \tag{39}\\
& M_{T}=\sqrt{\frac{2 h_{o, w}}{k_{f} t}} \tag{40}
\end{align*}
$$

The air side heat transfer coefficient under wet conditions was calculated using the following procedure (Wang et al., 1997):

1. Calculate total heat transfer coefficient $U_{o, w}$ using Equations (26) to (29)
2. Calculate fluid side heat transfer coefficient $h_{i}$ using Equation (36)
3. Assume an arbitrary value for the condensate film temperature $T_{w, m}$, and calculate

$$
b_{w, m}^{\prime}
$$

4. Iteratively calculate air side overall heat transfer coefficient under wet condition $h_{o, w}$ using Equation (30). Fin efficiency is evaluated using Equations (39) and (40)
5. Calculate $i_{s, w, m}$ by using the following Equation,

$$
\begin{equation*}
i_{s, w, m}=i_{a, m}-\frac{C_{p, a} h_{o, w} \eta_{f, w e t}}{b_{w, m}^{\prime} h_{c, o}} \times\left(1-U_{o, w} A_{0}\left[\frac{b_{r}^{\prime}}{h_{i} A_{p, i}}+\frac{b_{p}^{\prime} \ln \left(\frac{D_{c}}{D_{i}}\right)}{2 \pi k_{p} L_{p}}\right]\right)\left(i_{a, m}-i_{r, m}\right) \tag{41}
\end{equation*}
$$

6. Calculate $T_{w, m}$ from $i_{s, w, m}$. If $T_{w, m}$ derived here is not equal that is assumed in step (3), then repeat step (3) $\sim$ (6) until $T_{w, m}$ is constant

Obtain overall heat transfer coefficient and calculate sensible capacity transfer coefficient using Equation (31).

The next step is to calculate mass transfer coefficient $h_{d, o}$. Simultaneously heat and mass transfer process can be described by the process line equation from Threlkeld (1970):

$$
\begin{equation*}
\frac{d i_{a}}{d W_{a}}=L e \frac{\left(i_{a}-i_{s, w}\right)}{\left(W_{a}-W_{s, w}\right)}+\left(i_{g}-2501 L e\right) \tag{42}
\end{equation*}
$$

where $R$ represent the ratio of sensible capacity transfer characteristics to the mass transfer performance,

$$
\begin{equation*}
L e=\frac{h_{c, o}}{h_{d, o} C_{p, a}} \tag{43}
\end{equation*}
$$

However, for fin-and-tube heat exchanger, the saturated air enthalpy $\left(i_{s, w}\right)$ at the mean temperature at the fin surface is different from that at the fin base, thus Equation (42) does not correctly describe the dehumidification process. A modification of the process line on the psychrometric chart corresponding to the fin-and-tube heat exchanger is described as follows:

From the energy balance of the dehumidification, one can get Equation (44).

$$
\begin{equation*}
\dot{m}_{a} d i_{a}=\frac{h_{c, o}}{C_{p, a}} d A_{p, o}\left(i_{a, m}-i_{s, p, o, m}\right)+\frac{h_{c, o}}{C_{p, a}} d A_{f}\left(i_{a, m}-i_{s, w, m}\right) \tag{44}
\end{equation*}
$$

From conservation of water condensate, one can get Equation (45).

$$
\begin{equation*}
\dot{m}_{a} d W_{a}=h_{c, o} d A_{p, o}\left(W_{a, m}-W_{s, p, o, m}\right)+h_{d, o} d A_{f}\left(W_{a, m}-W_{s, w, m}\right) \tag{45}
\end{equation*}
$$

Dividing Equation (44) by Equation (45) yields

$$
\begin{equation*}
\frac{d i_{a}}{d W_{a}}=\frac{L e \cdot\left(i_{a, m}-i_{s, p, o, m}\right)+L e \cdot(\varepsilon-1) \cdot\left(i_{a, m}-i_{s, w, m}\right)}{\left(W_{a, m}-W_{s, p, o, m}\right)+(\varepsilon-1) \cdot\left(W_{a, m}-W_{s, w, m}\right)} \tag{46}
\end{equation*}
$$

where

$$
\begin{equation*}
\varepsilon=\frac{A_{o}}{A_{p, o}} \tag{47}
\end{equation*}
$$

The mass transfer coefficient and $h_{d, o}$ is calculated following the procedures:

1. Calculate $W_{s, p, o, m}$ and $W_{s, w, m}$ from $i_{s, p, o, m}$ and $i_{s, w, m}$
2. Assume an arbitrary value for $L e$
3. Calculate the humidity ratio of the outlet air from experiment
4. Calculate the humidity ratio of the outlet air by Equation (45). If the humidity ratio calculated from step (4) equals that from step (3), then stop, else repeat step (2) to (4)
5. Determine $h_{d, o}$ from Equation (43)

Air-side heat and mass transfer coefficient and friction factor of fin-and-tube heat exchanger under wet condition are evaluated using Chilton-Colburn $j$, Chilton-Colburn $j_{m}$ and $f$ factor as follows:

$$
\begin{align*}
& j=\frac{h_{a}}{G_{\max } C_{p, a}} \operatorname{Pr}^{2 / 3}  \tag{48}\\
& j_{m}=\frac{h_{m}}{G_{\max }} S c^{2 / 3}  \tag{49}\\
& f=\frac{\Delta P_{a} \rho_{a} D_{h}}{2 G_{\max }^{2} L_{s}} \operatorname{Pr}^{2 / 3} \tag{50}
\end{align*}
$$

### 2.2. CFD Simulation

2.2.1. Physics and governing equations

CFD simulation for all heat exchangers are conducted using ANSYS ${ }^{\circledR}$ Workbench ${ }^{\text {TM }}$ 18.0. First, Solidworks ${ }^{\circledR}$ generates geometry file and exported it into Workbench ${ }^{\text {TM }}$. Then, Meshing ${ }^{\text {TM }}$, embedded meshing software in Workbench ${ }^{\text {TM }}$ does the meshing; and finally, FLUENT ${ }^{\circledR}$ runs the simulation.

The governing equations of continuity, momentum (Navier-Stokes) and energy are listed as Equation (51) to (53).

$$
\begin{align*}
& \nabla \cdot(\rho \vec{u})=0  \tag{51}\\
& \nabla \cdot(\rho \vec{u} \vec{u})=-\nabla p \nabla \cdot\left[\mu\left(\nabla \vec{u}+\nabla \vec{u}^{T}-\frac{2}{3} \nabla \cdot \vec{u}\right)\right]  \tag{52}\\
& \nabla \cdot\left[\rho \vec{u}\left(h+\frac{u^{2}}{2}\right)\right]=\nabla \cdot(k \nabla T) \tag{53}
\end{align*}
$$

For air side simulation, the assumptions are:

- Three-dimensional, single phase, steady state flow;
- Negligible gravitational effects;
- Ideal gas model for density calculation;
- Temperature dependent properties estimated with polynomial curve fitting other than density;
- Pressure work and kinetic energy are negligible.

In current study, eddies are expected at boundary layer detachment despite low Reynolds numbers. Since the transition between laminar and turbulent regimes is unknown and must be solved by the simulation, the two-equation k- $\varepsilon$ realizable (RKE) model (Shih and Zhu, 1994) was used with enhanced wall functions enabled in every simulation. Another advantage of using turbulence model is that they can better solve a larger range of problems which is desired when simulating many samples using same CFD settings. Compared with standard k- $\varepsilon$ model, RKE has improved performance for problems involving boundary layers under strong adverse pressure gradients or separation, rotation, recirculation and strong streamline curvature thus is suitable for problems in current study. And it has a higher rate of convergence when using RKE compared to
other models for many CFD simulations (Baceller, 2016). For corresponding bare tube simulation, same assumptions were applied.

Thermal properties for air, such as specific heat, thermal conductivity and viscosity are estimated with polynomial curve fitting as function of temperature (Baceller, 2016) and density is based on ideal-gas model.

For water side simulation, the assumptions are:

- Three-dimensional, single phase, steady state flow;
- Negligible gravitational effects;
- Temperature dependent properties estimated with polynomial curve fitting;
- Pressure work and kinetic energy are negligible.

The reason of choosing $\mathrm{k}-\varepsilon$ realizable (RKE) model with enhanced wall functions for waterside simulation is similar to that of airside.

Turbulent boundary conditions were set to default: 5\% turbulence intensity and viscosity ratio of 10 . The pressure-velocity coupling scheme used is the Coupled solver available in FLUENT ${ }^{\circledR}$. A second order upwind space discretization is set to ensure better accuracy. Convergence criteria is defined as $1.0 \mathrm{e}-5$ for continuity and velocities, $1.0 \mathrm{e}-6$ for energy, and $1.0 \mathrm{e}-$ 3 for turbulent kinetic energy $(\mathrm{k})$ and eddy viscosity ( $\varepsilon$ ). If the simulation does not meet the criteria, however it stabilizes into a solution, we assume that if the standard deviation of the last 100 iterations is less than $0.5 \%$ of the average, then it is converged.
2.2.2. Airside computational domain

## BTHX computational domain:

The calculation domain for BTHX is shown in Figure 5 as well as the air flow direction. It is a two-dimensional cross section of two row staggered tubes of the heat exchanger. Boundary conditions are defined as:

- Constant and homogeneous velocity distribution at inlet
- Constant pressure at outlet (0.0 Pa gauge)
- Periodic flow at top and bottom of computational domain
- Tubes as walls, tube wall temperature is fixed at 350 K
- The air inlet temperature is fixed at 300 K

A triangular mesh element was set for the models and a refined boundary layer mesh at tube walls was modeled to capture the momentum and thermal boundary layer development with higher accuracy.


Figure 5 BTHX computational domain

## bBTHX computational domain:

For airside simulation, there are different computational domain due to different simulation purpose. The main difference is tube bank number. For parametric study, only two tube banks are simulated while for metamodeling, tube bank numbers are from 1 to 15 . The computational domain of bBTHX for parametric study is shown in Figure 6. It is a three-dimensional cross section of two rows of staggered tubes of the heat exchanger. End effects are neglected. Boundary conditions are defined as:

- Constant and homogeneous velocity distribution at inlet
- Constant pressure at outlet (0.0 Pa gauge)
- Periodic boundaries at left-right plane (xz-plane) and top-bottom plane (xy-plane) of computational domain
- Tubes as walls, tube wall temperature is fixed at 350 K
- The air inlet temperature is fixed at 300 K

Although hexahedron mesh is a robust meshing method with excellent stability and computational cost, it is time consuming in terms of generation. Time saved during computation sometimes cannot compensate the time consumed during generation, which is especially true for complicated geometries. Along with the increase of computing power, tetrahedral elements, which can be generated automatically even for complex shapes, have been paid more attention to. Moreover, it was observed that results obtained with quadratic tetrahedral elements and hexahedral elements were equivalent in terms of both accuracy and CPU time (Cifuentes and Kalbag, 1992; Wang et al., 2004), and the stability was found to be excellent even with irregular geometry such as human organs (Bourdin and Trosseille, 2007). Thus, for current design, quadratic tetrahedral mesh element was used. Thus, the choice of an unstructured mesh was justified by the greater easiness in controlling and adapting the mesh quality in an automatic way during optimization. And this has been proven to be accurate enough in tube bundle simulations (Ranut et al., 2014).


Figure 6 bBTHX computational domain

For metamodeling, tube bank ranges are from one to 15 , the computational domain of 15 banks are shown in Figure 7.


Figure 7 bBTHX computational domain - 15 banks
2.2.3. Waterside computational domain

## BTHX computational domain:

The waterside computational domain is shown as Figure 8. A triangular mesh element is set for the models and a refined boundary layer mesh at tube walls is modeled to capture the momentum and thermal boundary layer development with higher accuracy. Here are the settings:

- Constant and homogeneous velocity distribution at inlet
- Constant pressure at outlet (0.0 Pa gauge)
- Tube wall temperature is fixed at 350 K
- The water inlet temperature is fixed at 360 K

Total tube length is 30 times diameter. The first half (15D) is there to ensure the second half is fully developed flow, as shown in Figure 9. Results of the second half are used to calculate heat transfer coefficient and pressure drop.


Figure 8 Waterside BTHX computational domain



## bBTHX computational domain:

The waterside computational domain is the same for parametric study and metamodeling. To give an accurate estimation of the tube side heat transfer and pressure drop, the entry length and the fully developed region need to be modeled separately. However, the entry length varies due to different geometries, thus the first task is to determine the entry length of each geometry. Since what really matters is the determination of computational domain for developing flow and fully developed flow instead of finding the entry length equation, we can only focus on heat exchanger with largest Reynolds number. Another parameter needs to be minimal is $\left(l_{1}+l_{2}\right)$ (see chapter 4 for definition of $l_{1}$ and $l_{2}$ ) because in current design, the boundary layer reinitiates at each bifurcation, thus the flow pattern becomes stable after several bifurcations.

In the design domain, the one with largest Reynolds number and smallest $\left(l_{1}+l_{2}\right)$ is the one with diameter equals 4 mm , bifurcation angle equals $60^{\circ}$, water velocity equals $1 \mathrm{~m} / \mathrm{s}$ and length ratio equals 0.5 .


Figure 10 Velocity contour at mid-plane, model A (ID $=4 \mathrm{~mm}, \theta=60^{\circ}, V_{w}=1 \mathrm{~m} / \mathrm{s}, \mathrm{LR}=0.5$ )


Figure 11 Velocity contour at mid-plane, model B
(ID $=\mathbf{4} \mathrm{mm}, \boldsymbol{\theta}=60^{\circ}, V_{w}=1 \mathrm{~m} / \mathrm{s}, \mathrm{LR}=0.5$ )


Figure 12 Velocity contour at mid-plane, 4-segment model (ID=4 mm, $\boldsymbol{\theta}=60^{\circ}, V_{w}=\mathbf{1 m} / \mathrm{s}, \mathrm{LR}=0.5$ )

Two models were built: Model A is with uniform velocity inlet and model B is with fully developed flow inlet. In reality, the tubes are connected with header, thus the inlet condition can
be assumed to be uniform water velocity. Figure 10 shows the velocity contour of the middle plane of model A. This computational domain is divided into 24 segments. To simulate the fully developed region, a straight tube is added to the inlet to ensure the inlet flow is fully developed when it comes to the bifurcation. The length of straight tube is 15 times the diameter. The velocity contour of model B is shown in Figure 11. Comparing the two enlarged figures, it is obvious that the major difference is in segment-01. The addition of bifurcation makes the velocity field almost identical after segment-02.


Figure 13 Normalized WHTC (a) and WDP (b) comparison
Waterside heat transfer coefficient (WHTC) and waterside pressure drop (WDP) are plotted in Figure 13 (a) and (b). From the two graphs, it should be seen that both WHTC and WDP of model A become stable around a certain value starting with segment 2 . The maximum deviation is around $\pm 2 \%$. For model B, WDP and WHTC becomes stable starting with segment 1 . It should be noted that all values are normalized based on the averaged values of model A starting with segment 2.

In current study, to simplify the calculation and save computational time, a 4-segment model is used. The velocity contour is shown in Figure 12. It looks almost identical to the first 4 segments in Figure 10. In Figure 13, results of 4-segment model are very close to the first 4
segments of model A. Thus 4-segment model's results can be used for developing flow and fully developed flow simultaneously. Results of segment $1 \sim 2$ is used for developing flow, and averaged results of segment $3 \sim 4$ is used for fully developed flow.

The selected point is the worst scenario, here is the verification using a random point. As shown in Figure 14, the results of 4-segment model match that with model A very well.


Figure 14 WHTC (a) and WDP (b) comparison
(ID $=1 \mathrm{~mm}, \boldsymbol{\theta}=10^{\circ}, \mathrm{V}_{\mathrm{w}}=0.1 \mathrm{~m} / \mathrm{s}, \mathrm{LR}=0.5$ )

Thus, the waterside computational domain is shown as Figure 15. A tetrahedral mesh element is set for the models and a refined boundary layer mesh at tube walls is modeled to capture the momentum and thermal boundary layer development with higher accuracy. Here are the settings:

- Constant and homogeneous velocity distribution at inlet
- Constant pressure at outlet (0.0 Pa gauge)
- Periodic boundaries at left-right plane (xz-plane) and top-bottom plane (xy-plane) of computational domain
- Tubes as walls, tube wall temperature is fixed at 350 K
- The water inlet temperature is fixed at 360 K


Figure 15 Waterside computational domain
2.2.4. CFD data reduction

UA-LMTD method was applied to calculate airside and waterside convention heat transfer coefficient, as shown in Equation (54) and (55).

$$
\begin{align*}
& \dot{Q}=\dot{m} \cdot c_{p}\left|T_{o}-T_{i}\right|=h \cdot A_{o} \cdot \Delta T_{m l}  \tag{54}\\
& h=\frac{\dot{m} \cdot c_{p}}{A_{o}} \cdot \frac{\left|T_{o}-T_{i}\right|}{\Delta T_{m l}}=\frac{\dot{m} \cdot c_{p}}{A_{o}} \cdot \frac{\left|T_{o}-T_{i}\right|}{\frac{\left[\left(T_{w}-T_{i}\right)-\left(T_{w}-T_{o}\right)\right]}{\ln \left[\left(T_{w}-T_{i}\right) /\left(T_{w}-T_{o}\right)\right]}} \tag{55}
\end{align*}
$$

For pressure drop, the dynamic pressure difference between inlet and outlet is negligible compared to static pressure difference. Thus, static pressure difference between inlet and outlet is regarded as the total pressure difference.

### 2.2.5. CFD Grid Uncertainty Analysis

The Grid Convergence Index (GCI) method (Roache, 1993; Roy and Oberkampf, 2011; ASME, 2009), which is based on Richardson extrapolation method (Richardson, 1910), is an acceptable and a recommended method to evaluate the grid convergence.

Here is the recommended procedure for GCI calculation (Procedure for Estimation and Reporting of Uncertainty Due to Discretization in CFD Applications, 2008):

Step 1: Define a representative cell, mesh, or grid size $h$ as follows for two-dimensional and three dimensional computational domains.

$$
\begin{equation*}
h_{2 D}=\left[\frac{\sum \Delta A_{i}}{N}\right]^{1 / 2} \tag{56}
\end{equation*}
$$

$$
\begin{equation*}
h_{3 D}=\left[\frac{\sum \Delta V_{i}}{N}\right]^{1 / 3} \tag{57}
\end{equation*}
$$

where $\Delta V_{i}$ is the volume and $\Delta A_{i}$ is the area of the $i$ th cell, and $N$ is the total number of cells used for computation.

Step 2: Select three significantly different sets of grids where the element size ratio between subsequent grid resolutions is no less than 1.3. This value of 1.3 is based on experience and not on formal derivation. The refinement should be done systematically and the use of geometrically similar cells is preferable.

Step 3: Let $h_{1}<h_{2}<h_{3}$ and $r_{21}=h_{2} / h_{1}, r_{32}=h_{3} / h_{2}$, calculate the observed order of accuracy $p^{*}$ using Equation (58) through (61). Note that $q(p)=0$ for $r=$ const. and $\varphi$ is a selected variable which is critical to the conclusions being reported. $\varphi_{k}$ is the solution on the $k_{t h}$ grid.

$$
\begin{align*}
& \hat{p}=\left(\frac{1}{\ln r_{21}}\right)\left(\ln \left|\frac{\varepsilon_{32}}{\varepsilon_{21}}\right|+q(\hat{p})\right)  \tag{58}\\
& q(\hat{p})=\ln \left(\frac{r_{2 \hat{p}}^{\hat{p}}-s}{r_{32}^{\hat{\hat{p}}}-s}\right)  \tag{59}\\
& s=\operatorname{sgn}\left(\varepsilon_{21} / \varepsilon_{32}\right)  \tag{60}\\
& \varepsilon_{21}=\varphi_{2}-\varphi_{1} \tag{61}
\end{align*}
$$

Step 4: Calculate the extrapolated values as in Equation (62).

$$
\begin{equation*}
\varphi_{e x t}^{21}=\frac{r_{21}^{p} \varphi_{1}-\varphi_{2}}{\left|r_{21}^{p}-1\right|} \tag{62}
\end{equation*}
$$

Step 5: Calculate and report the error estimates using Equation (63) and the grid convergence index (GCI) in Equation (64).

$$
\begin{align*}
& e_{e x t}^{21}=\left|\frac{\varphi_{e x t}^{21}-\varphi_{1}}{\varphi_{e x t}^{21}}\right|  \tag{63}\\
& G C I_{\text {fine }}^{21}=\frac{F_{s} \cdot e_{a}^{21}}{r_{21}^{p}-1} \tag{64}
\end{align*}
$$

where $F_{s}$ is factor of safety. If solutions on three grids are available, then the following rules should be followed. It should be noticed that in Equation (62) and (64), order of accuracy $p$ is different from the observed order of accuracy $\hat{p}$. When the observed order of accuracy $\hat{p}$ agrees with the formal order $p_{f}$ within $10 \%$, then the formal order of accuracy along with a safety factor of 1.25 is used in the GCI calculation. When the observed order of accuracy does not agree within $10 \%$, then a factor of safety of 3.0 is used. And the order of accuracy is limited between 0.5 and the formal order. Setting the upper bound is because that allowing the order of accuracy to be much larger than the formal order causes the uncertainty estimates to be unreasonably small since the GCI goes to zero as $p$ goes to infinity. Setting the lower bound is because that allowing the order of accuracy to go to zero causes the uncertainty estimate to approach infinity. These are summarized in Table 9 (Roy and Oberkampf, 2011). But if solutions on only two grids are available, $F_{S}=3.0$ (Roache, 1993).

## Table 9 Implementation of GCI

| $\left\|\frac{\hat{p}-p_{f}}{p_{f}}\right\|$ | $F_{s}$ | $p$ |
| :---: | :---: | :--- |
| $\leq 0.1$ | 1.25 | $p_{f}$ |
| $>0.1$ | 3.0 | $\min \left(\max (0.5, \hat{p}), p_{f}\right)$ |

The uncertainties of boundary points of the design space are fundamentally larger than that of other points because the combinations of lower and upper bounds yield the most skewed computational domains, thus having a higher potential for poorer mesh elements in terms of size and aspect ratios. Thus, the GCI method is employed for the $2^{\mathrm{n}}$ samples represented by all variable combinations of 0 's and 1's for an n-dimension design space plus one central sample. (Bacellar, 2016)

Airside GCI results: The numerical uncertainties of heat transfer coefficient and pressure drop are $2.2 \%$ and $4.0 \%$ for finer mesh, and $3.0 \%$ and $4.5 \%$ for coarse mesh, respectively, as shown in Figure 16. Mesh 2 (intermediate mesh) was selected as the final mesh.

Waterside GCI results: The numerical uncertainties of heat transfer coefficient and pressure drop are $4.0 \%$ and $3.6 \%$ for finer mesh, and $4.9 \%$ and $5.6 \%$ for coarse mesh, respectively, as shown in Figure 17. Mesh 2 (intermediate mesh) was selected as the final mesh.


Figure 16 Airside GCI results


Figure 17 Waterside GCI results
2.2.6. New CFD automation approach

To do Approximation Assisted Optimization method, CFD automation is a necessity. In this dissertation, the CFD simulation was carried out using a parallel parameterized computational fluid dynamics (PPCFD) simulation, which was proposed by Abdelaziz (2009) and realized in ANSYS ${ }^{\circledR}$ Workbench ${ }^{\text {TM }} 18.0$ using two novel approaches proposed by the author. PPCFD automatically generates mesh and CFD journal files, runs the files and performs post processing to summarize and analyze the results. In Abdelaziz (2009)'s approach, Gambit was used for geometry and meshing and ANSYS ${ }^{\circledR}$ FLUENT $^{\circledR}$ for simulations. Gambit, which was the only platform that allowed journaling and scripting therefore suitable for the automation method, is no longer available. Two novel approaches are proposed here to realize the PPCFD procedure in ANSYS ${ }^{\circledR}$ Workbench ${ }^{\text {TM }}$.

The detailed steps of approach A are shown in Figure 18 (a) and as follows:

1) Generate design of experiment (DOE) table for parameters;
2) Use external code to generate a series of batch file, journal files for Workbench ${ }^{\mathrm{TM}}$, DesignModeler ${ }^{\text {TM }}$, Meshing ${ }^{\text {TM }}$ and FLUENT ${ }^{\circledR}$;
3) Execute journal files for Workbench ${ }^{\mathrm{TM}}$, DesignModeler ${ }^{\mathrm{TM}}$, Meshing ${ }^{\mathrm{TM}}$ and FLUENT $^{\circledR}$ using the journal files to generate geometry, generate mesh and run simulation in Workbench ${ }^{\text {TM }}$;
4) Post process output data to obtain thermal-hydraulic performance.

The detailed steps of approach B are shown in Figure 18 (b) and as follows:

1) Build first case and set parametric table in ANSYS ${ }^{\circledR}$ Workbench ${ }^{\mathrm{TM}}$;
2) Generate design of experiment (DOE) table for parameters;
3) Import DOE table to parametric table in ANSYS ${ }^{\circledR}$ Workbench ${ }^{\mathrm{TM}}$ and update the table to get output parameters;
4) Export parametric table and conduct post data process.

These two difference approaches both enable automatic calculation and allow for shape change. However, only approach A has the capability of topology change and allows for customized settings for geometry, meshing and CFD simulation. Approach A also has the advantage of fully parallel computation for every step: geometry generation, meshing and CFD simulation. Approach B only allows parallel computation for CFD simulation. And the parameter numbers of approach B cannot exceed the limit of 20, which is the largest number supported by parametric table in ANSYS ${ }^{\circledR}$ Workbench ${ }^{\text {TM }}$ 18.0. However, to use approach A, one needs to learn different programming languages including Python, JavaScript and Scheme. Considering Approach B is easy to learn and there is no topology change requirement for current study, Approach B was selected to be the PPCFD approach in current study.

(a) Approach A


Figure 18 Two PPCFD approaches in ANSYS ${ }^{\circledR}$ Workbench ${ }^{\text {TM }}$

### 2.3. Approximation Techniques and Optimization

### 2.3.1. Design of experiment

In many engineering problems, including current study, it is ideal to run simulations for each point in design domain to determine the relationship between input and output parameters. However, this is impossible because of the computational cost limitation. Hence, developing methods for efficiently selecting the experiments becomes important. Design of experiment (DOE) is a systematic approach to effectively sample the design space to achieve the optimal quality information of the relationship between input parameters and output responses. The quality of DOE plays a critical role in the accuracy of meta-model prediction. Different DOE generating methods are available in literature, such as random sampling, Latin hypercube sampling (LHS) (McKay et al., 2000), full-factorial designs (Box et al., 1986), maximum entropy sampling (MES) (Shewry and Wynn, 1987). In current study, Latin hypercube sampling (LHS) was used.

### 2.3.2. Kriging metamodeling

A meta-model or surrogate model, is a model of model. It is developed from the classical regression methodologies, which correlates data using least squares fitting methodologies. In current study, metamodeling was used to generate the airside and waterside heat transfer and friction correlations because it is hard to take into consideration of geometrical parameters using traditional regression methodologies. It is an approximation to system response constructed from sampling points, the design of experiments (DOE). Metamodeling is more applicable than classical regression methods for problems that the function form is not known a priori. Various metamodeling approaches are available, such as polynomial regression, spline regression, sparse grid, artificial neural network, kriging and hybrid model (Simpson et al., 2001), etc.

Kriging is an advanced interpolation technique that predicts the response of unknown design based on its linear distance from known design points and responses through a stochastic process, such as Gaussian process. Kriging technique provides superior performance for nonlinear problems and shows a higher degree of flexibility, and is recommended when design space has 50 or less variables (Wang et al., 2007). Kriging technique can fit higher order variations of the output parameters and auto refine the model by adding refinement points so that it can provide an improved response quality. Hence, in current study, Kriging method was used for metamodeling.

The accuracy of metamodel was evaluated using the Metamodel Acceptability Score (MAS) (Hamad, 2006). MAS was calculated using following equations.

$$
\begin{align*}
& e_{i}=\frac{\left|\hat{y}_{i}-y_{i}\right|}{y_{i}}  \tag{65}\\
& j_{i}=\left\{\begin{array}{l}
1, \text { if } e_{i}<e_{\max } \\
0, \text { if } e_{i}>e_{\max }
\end{array}\right.  \tag{66}\\
& M A S=\frac{\sum_{i=1}^{N} j_{i}}{N} \times 100  \tag{67}\\
& M A S \geq 1-e_{\max } \tag{68}
\end{align*}
$$

Where $y_{i}$ is true value, $\widehat{y_{i}}$ is predicted value by metamodel, $e_{i}$ is error, $e_{\max }$ is maximum tolerance error user defined (usually $5 \%$ to $10 \%$ ) and $N$ is number of test samples. The metamodel is accurate is Equation (68) is satisfied.

### 2.3.3.1. Modifications for current study

As computational power increases, it is possible to study enhanced surface performance improvement, flow arrangement, and circuitry using accurate segmented HX simulation tools, such as CoilDesigner ${ }^{\circledR}$ (Jiang et al., 2006). However, this requires the input of heat transfer and pressure drop of the new surface. This could be heat transfer and pressure drop correlations or data from CFD simulations or experiments. If the correlations or data come from CFD simulation of enhanced HX segment performance, then the method is regarded as multi-scale HX simulation.

Multi-scale simulation enables efficient integration of the enhanced HX segment performance prediction using CFD simulations with overall HX performance prediction using segmented $\varepsilon$-NTU method, which provides significant computational savings (Abdelaziz, 2009).

However, as far as the author knows, in all available applications of multi-scale HX simulations (Abdelaziz, 2009, Khaled et al. 2010, Bacellar et al. 2016), only airside was simulated using CFD while in-tube side was modeled using existing heat transfer and pressure drop correlations because the in-tube side geometry is round smooth tube, which has been comprehensively studied already. In current study, not only airside but also in-tube side geometries are new, thus the first improvement of current study is to apply multi-scale HX simulation on both air and in-tube side.

The second modification of current study is that it accounts for physical properties of working fluids so the metamodeling results could be used at any conditions. In previous studies, the airside CFD simulation was conducted at a specific working condition and later was used at different working conditions for optimization. This is acceptable due to the small variation of air
physical properties at different working conditions. In current study, working fluid at liquid side is water, of which the physical properties changes cannot be neglected. So, waterside's metamodel accounts for physical properties by using Reynolds number as an extra input in addition to geometry parameters.

### 2.3.3.2. bBTHX solver

There are many mathematical models and simulations tools that have been developed for design and rating of heat exchangers. However, none of them are applicable to current heat exchanger design due to its unique geometry. So, a customized heat exchanger solver needs to be developed. The calculation segment is shown in Figure 19 and the flow chart is shown in Figure 20. The definition of tube per row $\left(\mathrm{N}_{\mathrm{tpr}}\right)$, row number $\left(\mathrm{N}_{\mathrm{r}}\right)$ and water segment number $\left(\mathrm{N}_{\mathrm{ws}}\right)$ are shown in Figure 21.


Figure 19 Segment of bBTHX solver


Figure 20 Flow chart of bBTHX solver


Figure 21 Definitions of $\mathbf{N}_{\mathrm{tpr}}, \mathbf{N}_{\mathrm{r}}$ and $\mathbf{N}_{\mathrm{ws}}$
2.3.3.3. Approximation assisted optimization

The key procedure of multi-scale HX optimization is Approximation Assisted Optimization (AAO). The detailed steps applied in AAO are as follows:

1) Airside Design of Experiments (DoE) and PPCFD;
2) Waterside Design of Experiments (DoE) and PPCFD;
3) Airside meta-model building and verification;
4) Waterside meta-model building and verification;
5) Segmented HX solver;
6) Multi-objective optimization.

The optimization framework is illustrated in Figure 22.

Multi-objective optimization (MOO) problem is an optimization problem that involves multiple objective functions, as formulated in Equation (69). The solution of multi-objective optimization is usually a set of tradeoff designs called Pareto front.

$$
\begin{array}{cll}
\underset{\underline{x}}{\operatorname{minimize}} & f_{i}(\underset{\sim}{x}) & i=1, \ldots, M \\
\text { subject to: } & g_{j}(\underset{\sim}{x}) \leq 0 & j=1, \ldots, J \\
& h_{k}(\underset{\sim}{x})=0 & k=1, \ldots, K  \tag{69}\\
& {\underset{\text { lower }}{ } \leq \underset{\sim}{x} \leq{\underset{\sim}{u p p e r}}_{x}^{x}}^{\text {and }}
\end{array}
$$

Equation (69) describes an objective optimization problem with design variable vector $\underset{\sim}{x}$ and objective $f_{i}$. There are $M$ objectives in total and $f_{i}$ refers to the $i^{\text {th }}$ objective. There are $\boldsymbol{J}$ inequality constraints and $g_{j}(\underset{\sim}{x})$ refers to the $j^{\text {th }}$ inequality constraint. There are also $K$ equality
constraints and $h_{k}(\underset{\sim}{x})$ refers to the $k^{\text {th }}$ equality constraint. Lower bound of variable is ${\underset{\sim}{l}}_{\text {lower }}$ and upper bound is $\underset{\sim}{x}{ }_{u p p e r}$.

The problem above represented is solved using Multi-Objective Genetic Algorithms (MOGA) (Deb, 2001) in current study.


Figure 22 AAO flow chart

## Chapter 3: Experimental Results and Discussions

### 3.1. Dimensions of State-of-The-Art Heat Exchangers

3.1.1. Round bare tube heat exchanger (BTHX)


## Figure 23 Picture of BTHX (OD=0.8 mm)

Round bare tube heat exchanger with small diameter had been proposed by Bacellar et al. (2014) and numerically investigated. However, no experimental work has been done for such heat exchangers. A prototype was manufactured using stainless steel bare tubes with outer diameter equals 0.8 mm , as shown in Figure 23. The dimension is $152 \times 150 \times 5 \mathrm{~mm}$.
3.1.2. sBTHX Shape optimized bare tube heat exchanger (sBTHX)

Shape optimized bare tube heat exchanger (sBTHX) with small diameter had been proposed by Bacellar et al. (2016) and numerically investigated. However, no experimental work has been done for such heat exchangers. A prototype was manufactured using 3D printing with air side hydraulic diameter equals 1.5 mm , as shown in Figure 24. The dimension is $100 \times 100 \times 18 \mathrm{~mm}$. And the material is titanium.


Figure 24 Pictures of sBTHX ( $\mathrm{D}_{\mathrm{h}}=1.5 \mathrm{~mm}$ ) (a) overview, (b) tube pattern and (c) tube shape
3.1.3. Mini-channel heat exchanger (MCHX)


Figure 25 Picture of MCHX
A micro-channel heat exchanger was manufactured with dimension of $210 \times 120 \times 16 \mathrm{~mm}$, as shown in Figure 25. It has the same air frontal area as the bare tube heat exchanger. Louvered fin pitch is 20 . The material is aluminum.

Table 10 Dimensions of BTHX, sBTHX and MCHX

| Type | Frontal <br> area <br> $\left[\mathrm{m}^{2}\right]$ | Water <br> cross <br> section <br> area <br> $\left[\mathrm{m}^{2}\right]$ | Air heat <br> transfer <br> area <br> $\left[\mathrm{m}^{2}\right]$ | Water <br> heat <br> transfer <br> area <br> $\left[\mathrm{m}^{2}\right]$ | Envelop <br> volume <br> $\left[\mathrm{m}^{3}\right]$ | Material <br> volume <br> $\left[\mathrm{m}^{3}\right]$ | Material <br> mass <br> $[\mathrm{kg}]$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| BTHX | 0.0228 | 0.0001 | 0.1826 | 0.1364 | 0.0001 | 0.000013 | 0.1269 |
| sBTHX | 0.0100 | 0.0002 | 0.2178 | 0.1100 | 0.0002 | 0.000020 | 0.1663 |
| MCHX | 0.0247 | 0.0002 | 0.5525 | 0.0735 | 0.0004 | 0.000060 | 0.1628 |



Figure 26 Comparison of dimensions of BTHX, sBTHX and MCHX

The dimensions of BTHX and sBTHX and MCHX are shown in Table 11 and normalized values are shown in Figure 26. The air frontal areas of BTHX and MCHX are similar. Due to the addition of fins, the air side heat transfer area of MCHX is about three times that of BTHC and sBTHX. BTHX and sBTHX have two thirds smaller volume and material volume than MCHX.

### 3.1.4. Slit fin-and-tube heat exchangers with tube diameter of 5 and 4 mm

Two slit fin-and-tube heat exchangers (Figure 27 and Figure 28) were manufactured by a company and the dimensions of these two heat exchangers are summarized in Table 11.


Figure 27 Picture of slit fin-and-tube heat exchanger ( $\mathrm{OD}=\mathbf{5} \mathbf{~ m m}$ )


Figure 28 Picture of slit fin-and-tube heat exchanger ( $\mathrm{OD}=\mathbf{4} \mathbf{~ m m}$ )
Table 11 Dimensions of 5 and 4 mm slit fin-and-tube heat exchanger

|  | Primary Heat <br> Transfer Area <br> $\left[\mathrm{m}^{2}\right]$ | Secondary Heat <br> Transfer Area <br> $\left[\mathrm{m}^{2}\right]$ | Total Air Side Heat <br> Transfer Area <br> $\left[\mathrm{m}^{2}\right]$ | Fin Material <br> Volume <br> $\left[\mathrm{m}^{3}\right]$ | Tube Material <br> Volume <br> $\left[\mathrm{m}^{3}\right]$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 5 mm | 0.12 | 2.08 | 2.19 | 0.00011 | 0.000020 |
| 4 mm | 0.12 | 1.83 | 1.94 | 0.00010 | 0.000020 |
|  | Tube Material <br> Mass <br> $[\mathrm{kg}]$ | Coil Length <br> $[\mathrm{m}]$ | Coil Depth <br> $[\mathrm{m}]$ | Coil Height <br> $[\mathrm{m}]$ | Frontal Area <br> $\left[\mathrm{m}^{2}\right]$ |
| 5 mm | 0.21 | 0.39 | 0.011 | 0.38 | 0.148 |
| 4 mm | 0.21 | 0.42 | 0.009 | 0.36 | 0.152 |

### 3.2. Experimental Test Using Air and Water Under Dry Condition

3.2.1. Test matrix for BTHX, sBTHX and MCHX

BTHX, sBTHX and MCHX are rated for about 1 kW , thus were tested using the same test matrix to make a fair comparison. Three different water flow rates and three different air flow rates were tested, as shown in Table 12. This test matrix was for comparing the performance of all three heat exchangers at dry condition.

Table 12 Test matrix for BTHX, sBTHX and MCHX

| Test | Surface | Inlet Air <br> Temperature <br> $\left[{ }^{\circ} \mathrm{C}\right]$ | Inlet Air <br> RH <br> $[\%]$ | Inlet Water <br> Temperature <br> $\left[{ }^{\circ} \mathrm{C}\right]$ | Air Flow <br> Rate <br> $\left[\mathrm{m}^{3} / \mathrm{s}\right]$ | Water <br> Flow Rate <br> $[\mathrm{g} / \mathrm{s}]$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | Dry | 35 | 40 | 60 | 0.03 | 30 |
| 2 | Dry | 35 | 40 | 60 | 0.03 | 50 |
| 3 | Dry | 35 | 40 | 60 | 0.03 | 70 |
| 4 | Dry | 35 | 40 | 60 | 0.05 | 30 |
| 5 | Dry | 35 | 40 | 60 | 0.05 | 50 |
| 6 | Dry | 35 | 40 | 60 | 0.05 | 70 |
| 7 | Dry | 35 | 40 | 60 | 0.07 | 30 |
| 8 | Dry | 35 | 40 | 60 | 0.07 | 50 |
| 9 | Dry | 35 | 40 | 60 | 0.07 | 70 |

3.2.2. Test matrix for 4 and 5 mm slit fin-and-tube heat exchanger

Test matrix for 4 and 5 mm slit fin-and-tube heat exchanger is shown in Table 13.
Table 13 Test matrix for 4 and 5 mm slit fin-and-tube heat exchanger

| Test | Surface | Inlet Air <br> Temperature <br> $\left[{ }^{\circ} \mathrm{C}\right]$ | Inlet <br> Air <br> RH <br> $[\%]$ | Inlet Water <br> Temperature <br> $\left[{ }^{\circ} \mathrm{C}\right]$ | Air <br> Velocity <br> $[\mathrm{m} / \mathrm{s}]$ | Air <br> Flow <br> Rate <br> $\left[\mathrm{m}^{3} / \mathrm{s}\right]$ | Water <br> Velocity <br> $[\mathrm{m} / \mathrm{s}]$ | Water <br> Flow <br> Rate <br> $[\mathrm{g} / \mathrm{s}]$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | Dry | 35 | 40 | 60 | $1^{*}$ | 0.15 | 1.0 | 30 |
| 2 | Dry | 35 | 40 | 60 | 1 | 0.15 | 2.0 | 50 |
| 3 | Dry | 35 | 40 | 60 | 1 | 0.15 | 3.0 | 70 |
| 4 | Dry | 35 | 40 | 60 | 2.5 | 0.375 | 1.0 | 30 |
| 5 | Dry | 35 | 40 | 60 | 2.5 | 0.375 | 2.0 | 50 |
| 6 | Dry | 35 | 40 | 60 | 2.5 | 0.375 | 3.0 | 70 |
| 7 | Dry | 35 | 40 | 60 | $4 *$ | 0.6 | 1.0 | 30 |
| 8 | Dry | 35 | 40 | 60 | 4 | 0.6 | 2.0 | 50 |
| 9 | Dry | 35 | 40 | 60 | 4 | 0.6 | 3.0 | 70 |

3.2.3. Energy balance

The energy balance of all tests for each heat exchanger are all within $\pm 5 \%$, as shown in Figure 29, Figure 30 and Figure 31.


Figure 29 Energy balance of BTHX


Figure 30 Energy balance of sBTHX


## Figure 31 Energy balance of MCHX

### 3.2.4. Test results of BTHX

Figure 32 presents the heat exchanger capacity of BTHX, and airside pressure drop is shown in Figure 33. The airside heat transfer coefficient was calculated using Wilson plot method (Figure 34) and the results are shown in Figure 35. Chilton-Colburn j and ffactor were calculated and plotted in Figure 36. The j and f factor correlations were developed using power law, as shown in Equation (70) and (71), as well as in Figure 37. A comparison of AHTC prediction of current correlation against existing correlations in literature further reveals the necessity of developing new correlations through experimental data, as shown in Figure 38.

$$
\begin{align*}
& j=0.6499 \operatorname{Re}_{D_{h}}{ }^{-0.496}, 175 \leq \operatorname{Re}_{D_{h}} \leq 400  \tag{70}\\
& f=1.0114 \operatorname{Re}_{D_{h}}{ }^{-0.378}, 175 \leq \operatorname{Re}_{D_{h}} \leq 400 \tag{71}
\end{align*}
$$



Figure 32 Heat exchanger capacity of BTHX


Figure 34 Airside Wilson plot of BTHX


Figure $\mathbf{3 6} \mathbf{j}$ and f factor of BTHX


Figure 33 Airside pressure drop of BTHX


Figure 35 Airside heat transfer coefficient of BTHX


Figure $37 \mathbf{j}$ and $\mathbf{f}$ factor power laws of BTHX


Figure 38 Comparison of correlations with experimental data for BTHX
3.2.5. Test results of sBTHX

Figure 39 presents the energy balance results, all of which were within $\pm 4.9 \%$. Figure 40 presents the HX average capacity and airside pressure drop (ADP) is shown in Figure 41. Figure 42 presents the airside heat transfer coefficient using Wilson plot method (Wilson, 1915). AHTC is calculated for each airside velocity. Fitting equations for each velocity are shown on Figure 43, with y equals $\mathrm{R}_{o v}$ and x equals $1 / \mathrm{V}_{w}^{n}$. Chilton-Colburn j and f factor are shown in Figure 44 . Note that the Reynolds number was calculated based on maximum velocity and airside hydraulic diameter due to the irregular tube shape. Hydraulic diameter is defined as:

$$
\begin{equation*}
D_{h}=\frac{4 L_{s} A_{\min }}{A_{o}} \tag{72}
\end{equation*}
$$

where $L_{s}$ is air flow pass depth and $A_{\text {min }}$ is air free flow area.

There is no available correlation in literature for j and f factor for such geometry. Thus, new power law correlations of $j$ and $f$ factor against Reynolds number are developed, as shown in Figure 44. It should be noticed that Equations (73) and (74) could only be used for this certain heat exchanger geometry within the operation condition range as specified.

$$
\begin{align*}
& j=1.1583 \cdot \operatorname{Re}_{D_{h}}^{-0.623}\left(562 \leq \operatorname{Re}_{D_{h}} \leq 1266, \operatorname{Pr} \approx 0.74, d r y\right)  \tag{73}\\
& f=1.3349 \cdot \operatorname{Re}_{D_{h}}^{-0.365}\left(562 \leq \operatorname{Re}_{D_{h}} \leq 1266, \operatorname{Pr} \approx 0.74, d r y\right) \tag{74}
\end{align*}
$$

To examine the prediction of these new correlations and further explain the necessity of developing them, a comparison of new correlations against existing correlations for round bare tube bundle are presented in Figure 46 and Figure 47 . When applying these equations, an equivalent diameter was used as outer diameter to ensure the minimum free flow area is the same. For AHTC prediction, discrepancy between experimental data and the new correlation prediction is within $\pm 2 \%$. Zukauskas (1972) correlation gives a discrepancy of $\pm 15 \%$. Predictions of Khan et al. (2006) correlation and Bacellar et al. (2016a) correlation fall beyond $\pm 15 \%$. Even though Bacellar et al. (2016a) correlation is for bare tube bundle with small outer diameter of $0.5-2 \mathrm{~mm}$ (outer diameter of current design is 1.1 mm ), the difference of tube shapes causes the inaccuracy of prediction. We can also conclude comparing with round tube, the new geometry has lower AHTC (though air side total heat transfer area is larger). It was already found that Zukauskas (1972) correlation over predicted ADP (Bacellar et al., 2016a) when using for small diameter tubes. So, for ADP prediction, only Bacellar et al. (2016a) correlation was used. The discrepancy between the prediction of new power law correlation and experimental data is within $\pm 2 \%$ while the maximum deviation of bare tube correlation prediction is about $\pm 25 \%$. Thus, comparing with round tube, the new tube shape has lower ADP as well.


Figure 39 Energy balance of sBTHX


Figure 41 Air side pressure drop of sBTHX


Figure 43 Wilson plot of sBTHX


Figure 40 Heat exchanger average capacity of SBTHX


Figure 42 Air side heat transfer coefficient of sBTHX


Figure $44 \boldsymbol{j}$ and $\boldsymbol{f}$ factor of sBTHX


Figure $45 \boldsymbol{j}$ and $\boldsymbol{f}$ factor correlation (sBTHX)


Figure 46 Comparison of AHTC prediction of new correlation against existing correlations (sBTHX)


Figure 47 Comparison of ADP prediction of new correlation against existing correlations (sBTHX)

### 3.2.6. Test results of MCHX

HX average capacity is shown in Figure 48 and airside pressure drop (ADP) is shown in Figure 49. Test results are compared with simulation data from a software CoilDesigner ${ }^{\circledR}$ which is based on finite element heat exchanger model. The correlations used for airside are from Wang et al. (1999) while correlations used for waterside are from Dittus and Boelter (1930). Difference between simulated capacity and experimental data is within 6\% (Figure 50) and maximum airside pressure drop discrepancy is $20 \%$ (Figure 51). Wilson plot graph is in Figure 52 and heat transfer
coefficient calculated using Wilson plot method is shown in Figure 53. The j and f factors are plotted in Figure 54.


Figure 48 Heat exchanger capacity of MCHX


Figure 50 Simulation and experiments data comparison: heat exchanger capacity (MCHX)


Figure 52 Wilson plot of MCHX


Figure 49 Airside pressure drop of MCHX


Figure 51 Simulation and experiments data comparison: airside pressure drop (MCHX)


Figure 53 Airside heat transfer coefficient of MCHX


Figure $54 \boldsymbol{j}$ and $\boldsymbol{f}$ factor of MCHX

### 3.2.7. Comparison of BTHX, sBTHX and MCHX

Capacity comparison: Heat exchanger capacities of three heat exchangers are shown in Figure 55, Figure 56 and Figure 57. The gradient of heat exchanger capacity over air velocity, which is the slope of the trend line, decreases as the air flow rate increases, and increases as water flow increases at a certain air velocity. This is because when the air flow rate increases or the water flow decreases, the portion of air side thermal resistance decreases, so the influence of air velocity on capacity diminishes, and vice versa. The gradient of heat exchanger capacity over water velocity, which could be seen by comparing the discrepancy of the trend lines, decreases as water flow rate increases, and increases as air flow rate increases. Similar reason could be used to explain this, which is that when water flow rate increases or air flow rate decreases, the portion of water side thermal resistance decreases, then the influence of water flow rate on capacity reduces. It should be noted that BTHX has the largest capacity among these three and it has larger benefit at lower water flow rate. This is meaningful at partial load condition.

Air side pressure drop comparison: The air side pressure drop is shown in Figure 58. The heat exchangers' air-side pressure drop increases non-linearly with the increase of air velocity and the slope increases as air velocity increases and this is because higher air velocity increases
frictional losses. The change in water flow rate causes the air density to change, resulting in a different air pressure drop, but the change is not significant. sBTHX has the largest air side pressure drop, followed by BTHX and MCHX. The reason is the face area of sBTHX is half of that of MCHX and BTHX and the air velocity at the same air side flow rate are higher. To make a fair comparison, the air side pressure drop is also plotted against air side velocity, as shown in Figure 60. By using the optimized tube shape design (sBTHX), the air pressure drop is lower than bare tube design (BTHX).

Water side pressure drop comparison: The water side pressure drop is summarized in Figure 59. The mini-channel heat exchanger has the largest water side pressure drop due to smallest water cross section area and largest water velocity.

Air side heat transfer coefficient: Figure 61 shows the air side heat transfer coefficient of three heat exchanges. It shows that BTHX has the largest heat transfer coefficient.


Figure 55 Capacity comparison of BTHX, sBTHX and MCHX at $W F R=30 \mathrm{~g} / \mathrm{s}$


Figure 56 Capacity comparison of BTHX, sBTHX and MCHX at $\mathbf{W F R}=50 \mathrm{~g} / \mathrm{s}$


Figure 57 Capacity comparison of BTHX, sBTHX and MCHX at WFR=70 g/s


Figure 59 Water side pressure drop comparison of BTHX, sBTHX and MCHX


Figure 58 Air side pressure drop comparison of BTHX, sBTHX and MCHX


Figure 60 Air side pressure drop comparison of BTHX, sBTHX and MCHX


Figure 61 Air side heat transfer coefficient comparison of BTHX, sBTHX and MCHX
3.2.8. Test results of 5 and 4 mm slit fin-and-tube heat exchanger

Energy balances of 5 and 4 mm fin-and-tube heat exchanger test results are shown in Figure 62 and Figure 63. They are all within $\pm 5 \%$.



Thermal performance: From heat transfer point of view, there is not much difference between 4 and 5 mm slit fin-and-tube heat exchangers. The capacities are shown in Figure 66. The heat transfer coefficients which were calculated based on Wilson plot are shown in Figure 64 and Figure 65, separately. The heat transfer areas for 4 and 5 mm heat exchanger are 19.4 and $2.19 \mathrm{~m}^{2}$, respectively. Thus, the UA values of 4 and 5 mm heat exchanger are not much different, as shown in Figure 68.

Hydraulic performance: The air side pressure drops of 4 and 5 mm heat exchanger are shown in Figure 67. Water side pressure drops are shown in Figure 69. The reason that 4 mm heat exchanger has lower air pressure drop is due to smaller fin number. The higher water pressure drop for 4 mm heat exchanger is due to longer circuit length.

The 4 and 5 mm coils' j and f factors are compared with test data from literature (Wang et al., 2001) for traditional slit fin-and-tube heat exchangers with larger diameter ( 7.6 mm ). As shown in Figure 70 and Figure 71, j and f factors for both 4 and 5 mm coil are larger than 7.6 mm diameter heat exchanger. The j and f factors calculated using correlation from Wang et al. (2001) are also shown in these two graphs.


### 3.3. Experimental Test Using Air and Water Under Dehumidifying Condition

### 3.3.1. Test matrix for BTHX

Among three 1 kW heat exchangers, BTHX was tested under wet condition for both vertical (Figure 72) and horizontal tube orientation (Figure 73), and the test matrix is shown in Table 14. Three different air flow rates, three different water flow rates and three different inlet air relative humidity were tested.

Table 14 Wet condition test matrix for BTHX

| Test | Inlet Air Temperature $\left[{ }^{\circ} \mathrm{C}\right]$ | $\begin{gathered} \text { Inlet Air } \\ \text { RH } \\ {[\%]} \\ \hline \end{gathered}$ | Inlet Water Temperature $\left[{ }^{\circ} \mathrm{C}\right]$ | $\begin{gathered} \text { Air Flow } \\ \text { Rate } \\ {\left[\mathrm{m}^{3} / \mathrm{s}\right]} \\ \hline \end{gathered}$ | Water Flow Rate [g/s] |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 26.7 | 35 | 7.2 | 0.03 | 20 |
| 2 | 26.7 | 35 | 7.2 | 0.03 | 35 |
| 3 | 26.7 | 35 | 7.2 | 0.03 | 50 |
| 4 | 26.7 | 35 | 7.2 | 0.06 | 20 |
| 5 | 26.7 | 35 | 7.2 | 0.06 | 35 |
| 6 | 26.7 | 35 | 7.2 | 0.06 | 50 |
| 7 | 26.7 | 35 | 7.2 | 0.09 | 20 |
| 8 | 26.7 | 35 | 7.2 | 0.09 | 35 |
| 9 | 26.7 | 35 | 7.2 | 0.09 | 50 |
| 10 | 26.7 | 50 | 7.2 | 0.03 | 20 |
| 11 | 26.7 | 50 | 7.2 | 0.03 | 35 |
| 12 | 26.7 | 50 | 7.2 | 0.03 | 50 |
| 13 | 26.7 | 50 | 7.2 | 0.06 | 20 |
| 14 | 26.7 | 50 | 7.2 | 0.06 | 35 |
| 15 | 26.7 | 50 | 7.2 | 0.06 | 50 |
| 16 | 26.7 | 50 | 7.2 | 0.09 | 20 |
| 17 | 26.7 | 50 | 7.2 | 0.09 | 35 |
| 18 | 26.7 | 50 | 7.2 | 0.09 | 50 |
| 19 | 26.7 | 70 | 7.2 | 0.03 | 20 |
| 20 | 26.7 | 70 | 7.2 | 0.03 | 35 |
| 21 | 26.7 | 70 | 7.2 | 0.03 | 50 |
| 22 | 26.7 | 70 | 7.2 | 0.06 | 20 |
| 23 | 26.7 | 70 | 7.2 | 0.06 | 35 |
| 24 | 26.7 | 70 | 7.2 | 0.06 | 50 |
| 25 | 26.7 | 70 | 7.2 | 0.09 | 20 |
| 26 | 26.7 | 70 | 7.2 | 0.09 | 35 |
| 27 | 26.7 | 70 | 7.2 | 0.09 | 50 |



### 3.3.2. Test matrix of sBTHX

The test matrix of sBTHX under wet condition is summarized in Table 15.
Table 15 Test matrix for sBTHX

| Test | Inlet Air Temperature | Inlet Air <br> RH | Inlet Water Temperature | Air Velocity | Water Mass <br> Flow Rate |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | $\left[{ }^{\circ} \mathrm{C}\right]$ | $[\%]$ | $\left[{ }^{\circ} \mathrm{C}\right]$ | $[\mathrm{m} / \mathrm{s}]$ | $[\mathrm{g} / \mathrm{s}]$ |
| 1 | 26.7 | 50 | 12.0 | 0.03 | 20 |
| 2 | 26.7 | 50 | 12.0 | 0.03 | 35 |
| 3 | 26.7 | 50 | 12.0 | 0.03 | 50 |
| 4 | 26.7 | 50 | 12.0 | 0.06 | 20 |
| 5 | 26.7 | 50 | 12.0 | 0.06 | 35 |
| 6 | 26.7 | 50 | 12.0 | 0.06 | 50 |
| 7 | 26.7 | 50 | 12.0 | 0.09 | 20 |
| 8 | 26.7 | 50 | 12.0 | 0.09 | 35 |
| 9 | 26.7 | 50 | 12.0 | 0.09 | 50 |
| 10 | 26.7 | 70 | 12.0 | 0.03 | 20 |
| 11 | 26.7 | 70 | 12.0 | 0.03 | 35 |
| 12 | 26.7 | 70 | 12.0 | 0.03 | 50 |
| 13 | 26.7 | 70 | 12.0 | 0.06 | 20 |
| 14 | 26.7 | 70 | 12.0 | 0.06 | 35 |
| 15 | 26.7 | 70 | 12.0 | 0.06 | 50 |
| 16 | 26.7 | 70 | 12.0 | 0.09 | 20 |
| 17 | 26.7 | 70 | 12.0 | 0.09 | 35 |
| 18 | 26.7 | 70 |  | 0.09 | 50 |

3.3.3. Test matrix of 4 and 5 mm slit fin-and-tube heat exchanger

Both 4 and 5 mm slit fin-and-tube heat exchangers are tested under wet condition. The test matrix is shown in Table 16.

Table 16 Test matrix for 5 and 4 mm slit fin-and-tube heat exchanger

| Test | Inlet Air Temperature <br> $\left[{ }^{\circ} \mathrm{C}\right]$ | Inlet Air <br> RH <br> $[\%]$ | Inlet Water Temperature <br> $\left[{ }^{\circ} \mathrm{C}\right]$ | Air Velocity <br> $[\mathrm{m} / \mathrm{s}]$ | Water Velocity <br> $[\mathrm{m} / \mathrm{s}]$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 26.7 | 50 | 7.2 | 1 | 1.0 |
| 2 | 26.7 | 50 | 7.2 | 1 | 2.0 |
| 3 | 26.7 | 50 | 7.2 | 1 | 3.0 |
| 4 | 26.7 | 50 | 7.2 | 2.5 | 1.0 |
| 5 | 26.7 | 50 | 7.2 | 2.5 | 2.0 |
| 6 | 26.7 | 50 | 7.2 | 2.5 | 3.0 |
| 7 | 26.7 | 50 | 7.2 | 4 | 1.0 |
| 8 | 26.7 | 50 | 7.2 | 4 | 2.0 |
| 9 | 26.7 | 50 | 7.2 | 4 | 3.0 |

3.3.4. Test results for BTHX

This heat exchanger prototype was tested under both vertical and horizonal orientations, as summarized in Table 1. The inlet air conditions were fixed dry bulb temperature of $26.7^{\circ} \mathrm{C}$ with various relative humidity of $35 \%, 50 \%$ and $70 \%$. The inlet air frontal velocity varied at 3,6 and 9 $\mathrm{m} / \mathrm{s}$. The inlet water temperature was $12^{\circ} \mathrm{C}$ and the water mass flow rates were 20,35 and $50 \mathrm{~g} / \mathrm{s}$, respectively. For the horizontal orientation test, only conditions of the smallest and largest air velocity were tested.

The results from the wet test conditions are summarized in Figure 74 through Figure 83, where vertical orientation is on the left and horizontal orientation is on the right. Energy balance (EB) for all data points are within $\pm 5 \%$, as shown in Figure 74 and Figure 75. Here the effects of inlet air humidity, air flow rate and water flow rate and orientation on heat exchanger capacity, sensible capacity, latent capacity and airside pressure drop (ADP) are discussed.
3.3.4.1. Effect of inlet air relative humidity (RH)

When inlet air relative humidity $(\mathrm{RH})$ is $35 \%$, the heat exchanger is at dry condition, meaning there is only sensible cooling. As inlet air relative humidity increases from $35 \%$ to $70 \%$, overall heat transfer capacity increases (Figure 76) for the vertical tube orientation tests. However, when air RH changes from $35 \%$ to $50 \%$, heat exchanger capacity decreases slightly at horizontal orientation (Figure 77). Increased RH also leads to lower sensible capacity (SC), as shown in Figure 78 and Figure 79. This is because higher inlet air humidity leads to additional condensing water accumulation on the heat exchanger surface, which reduces dry surface area and restrains sensible capacity transfer. Accordingly, latent capacity increases (Figure 80 and Figure 81) as inlet RH increases. In terms of airside pressure drop, larger inlet air humidity results in larger airside pressure drop due to the bridging effect formed by retained condensate water between the tubes (Figure 82 and Figure 83).

### 3.3.4.2. Effect of inlet air flow rate (AFR)

As air flow rate (AFR) increases, total capacity (Figure 76 and Figure 77), sensible capacity (Figure 78 and Figure 79) and airside pressure drop (Figure 82 and Figure 83) all increase while latent capacity either increases or decreases (Figure 80 and Figure 81). The change of latent capacity is also affected by other factors such as inlet air humidity, water flow rate, condensate removal, heat exchanger orientation and heat exchanger geometry. For this HX, the latent capacity either decreases or increases depending on test condition (Figure 80 and Figure 81). Generally, as air flow rate increases, total capacity increases, causing water outlet temperature to increase, resulting in higher average wall temperature. The latent capacity transfer is expected to decrease due to increased wall surface temperature. However, larger air flow rate also means more moisture in the air stream, which produces more condensation. Besides, there are also other factors need to
be considered such as the ease of condensate water removal. Thus, whether latent capacity increases or decreases depends on which factor is dominant. Here explains the effect of AFR under vertical orientation. The most crucial factor under vertical orientation is the inlet air RH. When inlet air RH is low, the surface is partially wet and sensible capacity transfer is the dominant factor, leading to latent capacity decrease. This phenomenon was observed in the experiment, as shown in Figure 84 (a) and (b). Compare the results at $\mathrm{AFR}=0.06$ and $0.09 \mathrm{~m}^{3} / \mathrm{s}$, it can be noticed that as air flow rate increases, wet surface area becomes smaller and dry surface area becomes larger, especially on the top of the heat exchanger. When inlet air RH is high, the surface is fully wet, the extra moisture in the air becomes the dominant factor, causing latent capacity to increase. Figure 84 (c) and (d) show that both surfaces are fully wet. Next, the effect of AFR under horizontal orientation will be explained. Here, the most crucial factor is removal of condensate water. At horizontal orientation, it is hard to remove the condensate water at low flow rate due to the orientation, as shown in Figure 85 (a). Instead of flowing along the tubes as in the cases of vertical orientation, condensate water just accumulates in between the tubes and forms a water bridge if the air flow is not strong enough. At higher velocity as shown in Figure 85 (b), the condensate water is blown out by the incoming air flow. Water splashes in the downstream of air flow, leaving water marks on the wind tunnel duct wall. For pressure drop, higher air flow rate results in higher ADP, which is expected (Figure 82 and Figure 83). One interesting phenomenon is under vertical orientation, airside pressure drop for $\mathrm{RH}=35 \%$ and $70 \%$ is not much affected by waterside flow rate, but airside pressure drop for $\mathrm{RH}=50 \%$ increases as water flow rate increases. This is because for each heat exchanger geometry, there is a certain maximum amount of water retention corresponding to a specific condition. When RH is $50 \%$, the amount of retained water has not reached its maximum yet, thus the increase of waterside flow rate will cause latent capacity
increase, enhancing the bridging effect. When RH is $70 \%$, the amount of retained water has reached maximum, and consequently the extra condensing water flows down the tubes.

### 3.3.4.3. Effect of inlet water flow rate (WFR)

As water flow rate increases, total capacity (Figure 76 and Figure 77), sensible capacity (Figure 78 and Figure 79) and latent capacity (Figure 80 and Figure 81) all increase. Water flow rate increase has a negligible effect on airside pressure drop when inlet air RH=35\% and 70\% but it leads to airside pressure drop increase when $\mathrm{RH}=50 \%$. This has been discussed previously.

### 3.3.4.4. Effect of heat exchanger tube orientation

When there is no water condensate, there is no difference regarding total capacity, sensible capacity, latent capacity and airside pressure drop between heat exchanger operating under either orientation. Under the wet conditions, compared with heat exchanger under vertical tube orientation, the horizontal tube orientation tests show smaller total capacity (Figure 76 and Figure 77), sensible capacity (Figure 78 and Figure 79) and latent capacity (Figure 80 and Figure 81). However, the airside pressure drops (Figure 82 and Figure 83) under horizontal tube orientation are larger than those under vertical tube orientation due to bridging effect. Thus, it is recommended to operate this heat exchanger under vertical orientation if there is condensation.


Figure 74 Energy balance (BTHX, vertical)


Figure 76 Heat exchanger capacity (BTHX, vertical)


Figure 78 Sensible capacity (BTHX, vertical)


Figure 75 Energy balance (BTHX, horizontal)


Figure 77 Heat exchanger capacity (BTHX, horizontal)


Figure 79 Sensible capacity (BTHX, horizontal)


Figure 80 Latent capacity (BTHX, vertical)


Figure 82 Airside pressure drop (BTHX, vertical)


Figure 81 Latent capacity (BTHX, horizontal)


Figure 83 Airside pressure drop (BTHX, horizontal)


Figure 84 Effect of air flow rate on condensation at $\mathrm{RH}=50 \%$ (a)(b) and $\mathbf{R H}=\mathbf{7 0 \%}$ (c)(d) under vertical orientation (BTHX)


Figure 85 Effect of air flow rate on condensation at $\mathbf{R H}=\mathbf{7 0 \%}$ under horizontal orientation (BTHX)

### 3.3.4.5. $\quad \mathrm{j}, \mathrm{j}_{\mathrm{m}}$ and f factors

Here the $\mathrm{j}, \mathrm{j}_{\mathrm{m}}$ and f factor results are discussed only under the preferred vertical tube orientation. As it was already demonstrated, existing $j$ and $f$ factor correlations for bare tube bundles under dry condition in literature are not applicable for such small diameter tubes (Bacellar et al., 2016a) and there are no available wet condition correlations, so that new correlations are developed.

Figure 86 presents the variation of f factor on the effect of inlet air RH and Reynolds number. Friction increases as inlet air RH increases from $35 \%$ to $70 \%$ due to the bridging effect of condensate water. Correlations of f factor were developed as power law based on Reynolds number, as shown in Equation (75), (76) and (77). Maximum deviation for $f$ factor is within $\pm 2 \%$ under dry condition and $\pm 15 \%$ under wet condition (Figure 87).

$$
\begin{align*}
& f=1.705 \cdot \operatorname{Re}_{D_{o}}^{-0.45}\left(200 \leq \operatorname{Re}_{D_{o}} \leq 600, \operatorname{Pr} \approx 0.74, d r y\right)  \tag{75}\\
& f=209.33 \cdot \operatorname{Re}_{D_{o}}^{-1.157}\left(200 \leq \operatorname{Re}_{D_{o}} \leq 600, \operatorname{Pr} \approx 0.74, R H=50 \%\right) \tag{76}
\end{align*}
$$

$$
\begin{equation*}
f=38.67 \cdot \operatorname{Re}_{D_{o}}^{-0.837}\left(200 \leq \operatorname{Re}_{D_{o}} \leq 600, \operatorname{Pr} \approx 0.74, R H=70 \%\right) \tag{77}
\end{equation*}
$$

Figure 88 and Figure 90 present the variation of j and $\mathrm{j}_{\mathrm{m}}$ factor by the effect of inlet air RH and Reynolds number. The variation of j and $\mathrm{j}_{\mathrm{m}}$ factor by the relative humidity effect of inflow air is not sensitive. This is consistent with observations by Wang et al. (2000) and Phan et al. (2011). Correlations of j and $\mathrm{j}_{\mathrm{m}}$ factor are developed as power law based on Reynolds number, as shown in Equations (78) and (79). Maximum deviations for j and $\mathrm{j}_{\mathrm{m}}$ factor are within $\pm 10 \%$ (Figure 89 and Figure 91).

$$
\begin{equation*}
j=1.004 \cdot \operatorname{Re}_{D_{o}}^{-0.559}\left(200 \leq \operatorname{Re}_{D_{o}} \leq 600, \operatorname{Pr} \approx 0.74\right) \tag{78}
\end{equation*}
$$

$$
\begin{equation*}
j_{m}=0.6978 \cdot \operatorname{Re}_{D_{o}}^{-0.487}\left(200 \leq \operatorname{Re}_{D_{o}} \leq 600, \operatorname{Pr} \approx 0.74, \text { wet }\right) \tag{79}
\end{equation*}
$$



Figure $86 f$ factor (BTHX, wet, vertical)


Figure 87 Prediction of $\boldsymbol{f}$ factor correlation (BTHX, wet, vertical)


Figure $88 \boldsymbol{j}$ factor (BTHX, wet, vertical)


Figure $90 \boldsymbol{j}_{\boldsymbol{m}}$ factor (BTHX, wet, vertical)

Figure 89 Prediction of $\boldsymbol{j}$ factor correlation (BTHX, wet, vertical)


Figure 91 Prediction of $\boldsymbol{j}_{\boldsymbol{m}}$ factor correlation (BTHX, wet, vertical)

### 3.3.5. Test results for sBTHX

Wet condition test results are summarized in Figure 92 through Figure 96, with results for low inlet air relative humidity $(\mathrm{RH}=50 \%)$ on the left and high inlet air relative humidity $(\mathrm{RH}=70 \%)$ on the right. Figure 92 shows the energy balance of all wet condition tests are within $\pm 4.3 \%$. Here the effects of inlet air humidity, air flow rate and water flow rate on heat exchanger capacity, sensible capacity, latent capacity, airside pressure drop are discussed respectively. ChiltonColburn j , Chilton-Colburn $\mathrm{j}_{\mathrm{m}}$ and f factor are shown in Figure 97.


Figure 92 Energy balance for (a) $\mathrm{RH}=50 \%$ and (b) $\mathbf{R H}=70 \%$ (sBTHX, wet)

(a)

(b)

Figure 93 Heat exchanger average capacity for (a) $\mathbf{R H}=\mathbf{5 0 \%}$ and (b) $\mathrm{RH}=\mathbf{7 0 \%}$ (sBTHX, wet)

(a)

(b)

Figure 94 Sensible capacity for (a) $\mathrm{RH}=50 \%$ and (b) $\mathrm{RH}=\mathbf{7 0 \%}$ (sBTHX, wet)


Figure 95. Latent capacity for (a) $\mathbf{R H}=\mathbf{5 0 \%}$ and (b) $\mathbf{R H}=\mathbf{7 0 \%}$ (sBTHX, wet)

(a)

(b)

Figure 96. Air side pressure drop for (a) $\mathbf{R H}=\mathbf{5 0 \%}$ and (b) $\mathrm{RH}=\mathbf{7 0 \%}$ (sBTHX, wet)


Figure 97. $\mathbf{j}$ and f factor for (a) $\mathrm{RH}=\mathbf{5 0 \%}$ and (b) $\mathrm{RH}=\mathbf{7 0 \%}$ (sBTHX, wet)

### 3.3.5.1. Effect of inlet air relative humidity (RH)

From heat transfer point of view, larger inlet air humidity leads to larger latent capacity (LC, comparing Figure 94 (a) and (b)), lower sensible capacity (SC, comparing Figure 95 (a) and (b)) and larger overall heat exchanger capacity ( $\dot{Q}$, comparing Figure 93 (a) and (b)). This is because higher inlet air humidity leads to additional condensing water accumulation on the heat exchanger surface, which reduces dry surface area and restrains sensible capacity transfer. From airside pressure drop point of view, larger inlet air humidity results in larger airside pressure drop (comparing Figure 96 (a) and (b)). This difference is within $11 \%$, which is not a significant penalty for the trade-off in latent capacity improvement. Comparison with dry condition pressure drop is discussed later.

### 3.3.5.2. Effect of inlet air flow rate (AFR)

As air flow rate increases, total capacity (Figure 93), sensible capacity (Figure 94) and airside pressure drop (Figure 96) increase. However, latent capacity either increases (Figure 95 (b)) or decreases (Figure 95 (a)). It is also affected by other factors like inlet air humidity and heat exchanger geometry. For this HX, latent capacity decreases (Figure 95 (a)) for the 50\% RH test condition and increases for the $70 \%$ RH (Figure 95 (b)) test condition. Typically, as air flow rate increases, total capacity increases, causing waterside outlet temperature increase, resulting in higher average wall temperature. Therefore, the latent capacity transfer is expected to decrease due to increased wall surface temperature. However, larger air flow rate also means more moisture in the airstream, which will produce more condensation. This conflict indicates that whether latent capacity increases or decreases depends on which factor is dominant. When inlet air RH is low, the surface is partially wet, thus sensible capacity transfer dominates. So, the surface temperature increase is the dominant factor compared with airstream moisture increase, leading to latent
capacity decrease. This phenomenon is observed in the experiment, as shown in Figure 98. Dark area on the surface is wet surface and light area is dry surface. Comparing AFR=0.03, 0.06 and $0.09 \mathrm{~m}^{3} / \mathrm{s}$, as air flow rate increases, wet surface area is smaller and dry surface area becomes larger. When inlet air humidity ratio is high, the surface is fully wet, the extra moisture in the air becomes the dominant factor, causing latent capacity to increase.

For pressure drop, higher air flow rate results in higher ADP, as expected (Figure 96).


Figure 98 Effect of air flow rate on condensation at $\mathrm{RH}=\mathbf{5 0 \%}$ (sBTHX, wet)
3.3.5.3. Effect of inlet water flow rate (WFR)

As waterside flow rate increases, sensible capacity (Figure 94), latent capacity (Figure 95)
and total capacity (Figure 93) all increase. The effect of water flow rate on airside pressure drop is negligible (Figure 96). Figure 99 presents the condensate level of three different water flow rates at inlet $\mathrm{RH}=50 \%$. Only $\mathrm{RH}=50 \%$ pictures are selected because it is hard to recognize the difference on fully wet surface at inlet $\mathrm{RH}=70 \%$.


Figure 99 Effect of water flow rate on condensation (sBTHX, wet)

### 3.3.5.4. Comparison of dry and wet condition

Comparison of dry and wet condition airside heat transfer coefficient, airside pressure drop, and $j$ and $f$ factors are summarized in Figure 100, Figure 101 and Figure 102, respectively. AHTC is sensible capacity transfer coefficient. Dry condition's sensible capacity transfer coefficient is higher than that of wet condition at the same air side velocity. Figure 101 shows the pressure drop variation due to different inlet air humidity levels. Airside pressure drop under $\mathrm{RH}=70 \%$ is $0 \sim 11 \%$ larger than that of $\mathrm{RH}=50 \%$, and is $0 \sim 15 \%$ larger compared with that of dry condition. And the pressure drop penalty is more obvious at low air velocity ( $<5 \mathrm{~m} / \mathrm{s}$ ). The most important reason of
low airside pressure drop penalty is the header shape. As shown in Figure 24, the header of heat exchanger is diamond shape so that condensate water flows away easily along the header instead of accumulates at the bottom of the heat exchanger. This reduces the water bridging effect between tubes, which was observed in most fin-and-tube heat exchangers. Another reason why there is almost no pressure drop penalty at high air velocity (>5 m/s) is the blow out effect. Figure 104 shows water was blew out by air from backside of the heat exchanger. Two arcs formed by water were marked on the graph.

Figure 102 shows j and f factors on the effect of inlet relative humidity for dry and wet conditions. As Reynolds number increases, j and f factors decrease. Addition of latent capacity decreases j factor. As inlet relative humidity increases from $50 \%$ to $70 \%, \mathrm{j}$ factor decreases and f factor increases slightly. However, the variation of $j$ and $f$ factors on the relative humidity effect of inflow air is not sensitive. The present results are similar to that of Fu et al. (1995), Wang et al. (2000) and Phan et al. (2011).

Figure 103 presents the variation of $\mathrm{j}_{\mathrm{m}}$ factor on the effect of inlet relative humidity. As inlet relative humidity increases from $50 \%$ to $70 \%, \mathrm{j}_{\mathrm{m}}$ factor slightly decreases. The slight degradation of mass transfer performance may due to the condensate retention phenomenon between tubes. This is consistent with observations by Phan et al. (2011).

Correlations for $\mathrm{j}, \mathrm{j}_{\mathrm{m}}$ and f factor were developed for wet condition test shown in Figure 105 as power law based on Reynolds number in Equation (80), (81) and (82). Maximum deviations are within $\pm 7 \%$ of $\mathrm{j}_{\mathrm{m}}$ and f factor prediction and are within $\pm 10 \%$ of j factor.

$$
\begin{equation*}
j=0.6678 \cdot \operatorname{Re}_{D_{D_{h}}}^{-0.64}\left(315 \leq \operatorname{Re}_{D_{h}} \leq 1080, \operatorname{Pr} \approx 0.74, \text { wet }\right) \tag{80}
\end{equation*}
$$

$$
\begin{equation*}
j_{m}=0.6853 \cdot \operatorname{Re}_{D_{h}}^{-0.599}\left(315 \leq \operatorname{Re}_{D_{h}} \leq 1080, \operatorname{Pr} \approx 0.74, \text { wet }\right) \tag{81}
\end{equation*}
$$

$f=6.2974 \cdot \operatorname{Re}_{D_{h}}^{-0.015}\left(315 \leq \operatorname{Re}_{D_{h}} \leq 1080, \operatorname{Pr} \approx 0.74, w e t\right)$


Figure 100 Comparison of AHTC under dry and wet condition (sBTHX)


Figure 102 Comparison of $\mathbf{j}$ and $\mathbf{f}$ factor under dry and wet condition (sBTHX)


Figure 101 Comparison of ADP under dry and wet condition (sBTHX)


Figure 103 Comparison of $\mathbf{j}_{\mathrm{m}}$ factor at different RH level (sBTHX)


Figure 104 Water blow out effect for sBTHX ( $\mathrm{RH}=\mathbf{7 0 \%}, \mathrm{AFR}=\mathbf{0 . 0 9} \mathrm{m}^{3} / \mathrm{s}, \mathrm{WFR}=50 \mathrm{~g} / \mathrm{s}$ )


Figure 105 Wet condition $j$, $j$ m and factor correlation (sBTHX)


Figure 106 Prediction of wet condition $j$ factor correlation (sBTHX)

3.3.6. Test results of 5 and 4 mm slit fin-and-tube heat exchanger

Energy balance for both heat exchangers is within $\pm 5 \%$, as shown in Figure 109 and Figure 110 separately. The total capacities of both heat exchanger are shown in Figure 111 and Figure 112. The total capacities of 4 mm heat exchanger are slightly higher than that of 5 mm coil. The air side pressure drops are shown in Figure 113 and Figure 114. The air side pressure increases both as air and water flow rate increases. This is due to the blockage caused by condensation water. The air side pressure drop is slightly lower than that of 5 mm coil. Sensible capacities are shown in Figure 115 and Figure 116 and latent capacities are shown in Figure 117 and Figure 118. The sensible capacities of both heat exchanger are similar, but latent capacities for 4 mm coil are higher than those of 5 mmm coil. $\mathrm{j}, \mathrm{j}_{\mathrm{m}}$ and f factors are shown in Figure 119 and Figure 120, respectively.


Figure 109 Energy balance for $5 \mathbf{~ m m}$ coil under wet condition


Figure 111 Capacity for $5 \mathbf{m m}$ coil under wet condition


Figure 113 Air side pressure drop for $5 \mathbf{~ m m}$ coil under wet condition


Figure 110 Energy balance for 4 mm coil under wet condition


Figure 112 Capacity for $\mathbf{4 m m}$ coil under wet condition


Figure 114 Air side pressure drop for 4 mm coil under wet condition


Figure 115 Sensible capacity for $5 \mathbf{m m}$ coil under wet condition


Figure 117 Latent capacity for $5 \mathbf{m m}$ coil under wet condition


Figure $\mathbf{1 1 9} \mathbf{j}$, $\mathbf{j m}$ and $\mathbf{f}$ factor for $\mathbf{5 m m}$ coil under wet condition


Figure 116 Sensible capacity for $4 \mathbf{m m}$ coil under wet condition


Figure 118 Latent capacity for $\mathbf{4 m m}$ coil under wet condition


Figure $\mathbf{1 2 0} \mathbf{j}$, $\mathbf{j m}_{\mathrm{m}}$ and $\mathbf{f}$ factor for $\mathbf{4 m m}$ coil under wet condition

### 3.3.7. Comparison of all heat exchangers

AHTC and ADP/Depth for all baseline heat exchangers are plotted in Figure 121. BTHX has the largest AHTC as well as ADP/Depth. There are three reasons for this:

The first is for different heat exchangers, the mass flux based on free flow area is different. As shown in Figure 122, the mass flux for BTHX is the highest. However, in Figure 123, even if we plot the AHTC and ADP/Depth over mass flux, values of BTHX are still on the top.

The second reason is boundary layer interruption and redevelopment. From heat transfer point of view, the highest Nusselt number appears at the beginning of boundary layer. For different types of fins, their design principle is also boundary layer interruption and redevelopment, as shown in Figure 124. In bare tube designs, for each tube, the boundary layer redevelops, as shown in Figure 125. However, for fins, especially louvered fins and slit fins, the louver and slit size are also small, this cannot sufficiently explain the phenomenon.

The third reason is about the shape. Round tube has intrinsic advantage over flat plate regarding heat transfer. Flat plate and round tube were simulated using the same condition. Inlet air velocity is $2 \mathrm{~m} / \mathrm{s}$ and temperature is 300 K and the wall temperature is 350 K . The temperature contours are as shown in Figure 126 and Figure 127. From Figure 128 and Figure 129, flat plate has better hydraulic performance while round tube has better heat transfer performance.


Figure 121 Air side heat transfer coefficient and pressure drop per depth for baseline heat exchangers


Figure 122 G $_{\text {max }}$ for all baseline heat exchangers


Figure 123 Air side heat transfer coefficient and pressure drop per depth over $\mathbf{G}_{\text {max }}$ for baseline heat exchangers.


Figure 124 Boundary layer disruption and attachment regions on different fin types: a) Plain, b) Louver and c) Slit. (Bacellar, 2016)


Figure 125 Boundary layer disruption and attachment regions on a round tube bundle (Bacellar, 2016)


Figure 126 Flat plate temperature contour plot (CFD results)


Figure 127 Round tube temperature contour plot (CFD results)


Figure 128 AHTC of flat plate and round tube (CFD results)


Figure 129 Skin friction coefficient of flat plate and round tube (CFD results)

## Chapter 4: Bifurcated Bare Tube Heat Exchanger (bBTHX) Design

### 4.1. Design Concept

The innovative design is inspired by human lung and blood vein structure. If directly copying the nature, the design would be hierarchical bifurcated tubes, such as fractal channels for electronic cooling. However, the biggest disadvantage of using multi-level hierarchical bifurcated structure is the low utility of space. So, for fractal channels, the round disc shape (Pence, 2002) or sandwich structure (Chen and Cheng, 2002) are common designs to better utilize space. However, in the application, such designs would dramatically increase the difficulty of header design due to irregular header shape and increase the difficulty of manufacturing due to the small branch tube diameter so that it might not be able to be 3D printed. It may also increase the difficulty of heat pump/air conditioner system geometry design if the new heat exchanger is used as evaporator or condenser because of the irregular geometry. Thus, in current study, instead of using multi-level hierarchical bifurcated tubes, only two levels of tubes are used.

This novel heat exchanger consists of two levels of tubes, the main tubes and the branch tubes. Main tubes are all vertical tubes as shown in Figure 130 of which outer diameters are symbolled as $D_{1}$ while branch tubes are labeled as $D_{2}$. Longitudinal tube pitch $\left(P_{l}\right)$ is defined as the center distance of two adjacent main tubes. The transversal tube pitch $\left(P_{t}\right)$ is then defined as the central distance of two adjacent rows, as shown in Figure 131. Bifurcation angle $(\theta)$ is the angle between branch tube and the center line. Centerlines of all tubes generate a honeycomb structure consisting of multiple hexagons in this example, indicated by dotted line. Figure 132 shows the heat exchanger shape with header and the flow directions of two fluids on both sides. Tubes can be either staggered or in-line in the air flow direction. For staggered configuration, two
different patterns are studied in current research, as shown in Figure 133 (a) and (b). Darker tubes are first row and lighter tubes are second row. The difference between these two configurations is that the bypass flow, which flows through the honeycomb area formed by first row tube, hits the main tube of second row in pattern 1 while it hits the part of main tube and bifurcated tubes of second row in pattern 2 . Computational domain is shown in the red box.


Figure 130 bBTHX-tube structure


Figure 131 Two rows of bBTHX in staggered pattern


Figure 132 bBTHX schematic (staggered) and simulation domain


Figure 133 Two staggered patterns of bBTHX: pattern 1 (a) and pattern 2 (b)

### 4.2. Airside First Order Analysis

Airside heat transfer area is one of the key parameters to enhance airside heat transfer. Before further investigation of the new heat exchanger, first order analysis was conducted to evaluate airside heat transfer area by comparing the compactness and material utilization of current design and baseline heat exchangers. Compactness is defined as airside total heat transfer area over heat exchanger envelope volume and material utilization is defined as airside total heat transfer area over heat exchanger material volume.

The assumptions for first order analysis include: (1) same liquid side cross section area; (2) same tube pitch ratio; (3) same tube bank number; (4) same tube thickness ratio; (4) constant inlet air velocity, and (5) FPI of finned heat exchangers are all 30. Figure 134 and Figure 135 show the results of compactness and material utilization, respectively. It can be seen that at large diameter range ( $>1 \mathrm{~mm}$ ), mini-channel heat exchanger and plain fin-and-tube heat exchanger have larger compactness and material utilizations due to additional secondary surface area. However, as diameter becomes smaller, the advantage of secondary surface area diminishes. Compactness and
material utilization of round bare tube heat exchanger with diameter less than 1 mm are similar to that of finned heat exchangers. bBTHX even has larger compactness and material utilization than finned heat exchangers. Thus, current study focuses on small diameter bifurcated bare tube HX.

Another advantage of small diameter is the reduced internal volume, which means less refrigerant charge if applied in air conditioner and heat pump systems, leading to reduced system weight and less environmental impact.


Figure 134 First order analysis: compactness


Figure 135 First order analysis: material utilization

### 4.3. Parametric Study

The airside parameters studied are summarized in Table 17 and the waterside parameters studied are summarized in Table 18. Diameter ratio (DR) is defined as the ratio of $D_{1}$ over $D_{2}$ and
length ratio (LR) is defined as the ratio of length of main tube $\left(L_{1}\right)$ over length of branch tube $\left(L_{2}\right)$. DR was fixed to be 0.7 to maintain the mass flux constant inside tubes in current study. To compare the heat exchanger's performance, the main tube's diameter $\left(D_{1}\right)$ was kept the same as the diameter of bare tube heat exchanger $(D)$ in comparison.

Table 17 Airside Parametric Study Parameters for BTHX and bBTHX

| Types | Parameters | Units | BTHX | bBTHX |
| :---: | :---: | :---: | :---: | :---: |
| Constants | $P_{l}$ | $[\mathrm{~mm}]$ | $1.5 D$ | $1.5 D_{1}$ |
|  | $P_{t}$ | $[\mathrm{~mm}]$ | $1.5 D$ | $1.5 D_{1}$ |
|  | DR | - | - | 0.7 |
|  | $D$ or $D_{1}$ | $[\mathrm{~mm}]$ | $0.8,2,3,4$ | $0.8,2,3,4$ |
|  | $V_{a}$ | $[\mathrm{~m} / \mathrm{s}]$ | $0.5,2,3.5,5$ | $0.5,2,3.5,5$ |
|  |  | $[\mathrm{deg}]$ | - | $10,35,60$ |
|  | LR | - | - | $0.5,1.5,2.5$ |
|  | Pattern | - | - | 1,2 |
|  |  |  |  |  |

Table 18 Waterside Parametric Study Parameters for BTHX and bBTHX

| Types | Parameters | Units | BTHX | bBTHX |
| :---: | :---: | :---: | :---: | :---: |
| Constants | $P_{l}$ | $[\mathrm{~mm}]$ | - | $1.5 D_{1}$ |
|  | DR | - | - | 0.7 |
| Variables | $D$ or $D_{1}$ | $[\mathrm{~mm}]$ | $0.8,2,3,4$ | $0.8,2,3,4$ |
|  | $V_{w}$ | $[\mathrm{~m} / \mathrm{s}]$ | $0.1,0.3,0.6,1$ | $0.1,0.3,0.6,1$ |
|  | $\theta$ | $[\mathrm{deg}]$ | - | $10,35,60$ |
|  |  | - | - | $0.5,1.5,2.5$ |

### 4.4. Experimental Validation of Airside Hydraulic Performance

The airside CFD model was validated against experimental data measured from a fourbank HX with 0.8 mm diameter, 1.19 mm longitudinal pitch and 1.24 mm transverse pitch. Working fluids are air and water. The test facility and measurements followed the ASHRAE standards (ASHRAE, 1987, 2000). The average uncertainty in the capacity measurements was $2.71 \%$, whereas the pressure drop was $2.91 \%$. Deviations for heat transfer coefficient between experimental data and simulation data range from $1 \%$ (at higher air side velocity) to $15 \%$ (at lower
air side velocity). Over-prediction of pressure drop was observed, and addition of a correction factor of 0.61 brought the maximum deviation down to $7 \%$. Details could be found in Baceller et al., 2016.

One bBTHX sample was 3D printed by Connex 3 Systems Objet500 using VeroWhite filament, as shown in Figure 136. There are 13 tubes per row and 7 rows in total. Diameter for first layer tube is 2 mm , diameter ratio is 0.7 , longitudinal pitch is 3 mm , transverse pitch is 3 mm and length ratio is 1.732 . Due to leakage issue, this sample was only tested to validate the air side pressure drop for now and a new leak tight prototype is on the way to validate heat transfer. Similar to bare tube heat exchanger, consistent over-prediction of pressure drop is also observed. Experimental validation shows a good agreement of less than 3\% after applying a correction factor of 0.51 , as shown in Figure 137. The potential reasons for deviation include but are not limited to experimental uncertainties, model uncertainties and manufacturing uncertainties. Note that in the following session, data shown are all simulation data, without correction since, the observed pressure drop is related to lots of factors including material type, surface roughness, manufacturing method, etc.


Figure 136 bBTHX sample


Figure 137 Experimental validation for bBTHX air side pressure drop

### 4.5. Airside Simulation Results and Discussion

### 4.5.1. Heat transfer area

The comparison of all bBTHX configurations studied is summarized in Figure 138. As the bifurcation angle increases or the length ratio decreases, the bifurcation number will have to increase for a certain heat exchanger length, resulting in a larger heat transfer area improvement. Diameter has no effect on heat transfer area improvement as long as the diameter ratio is fixed. The difference between pattern 1 and pattern 2 is more obvious at larger bifurcation angle. It can be noticed that for geometry with $\theta=10^{\circ}$, there is nearly no heat transfer area improvement, and even degradation. Theoretically, tubes with bifurcations should have larger total area than bare tubes with same diameter because the area of two branch tubes with small diameter are larger than tube with larger diameter. However, when the bifurcation angle is very small (for example, $10^{\circ}$ ), the merging area of secondary tube balances out the extra area that created by secondary tubes, resulting in similar or even smaller heat transfer area than baseline, as illustrated in Figure 139.


Figure 138 Air-side heat transfer area improvement compared with BTHX


Figure 139 bBTHX tube configuration with $\theta=10^{\circ}, \mathbf{L R}=\mathbf{2} .5$, pattern 1

### 4.5.2. Free flow area

The decrease in free flow area results in stronger flow acceleration, leading to larger mass flux, thus larger heat transfer coefficient. So, this factor needs to be considered separately in order to make a fair comparison of heat transfer coefficient later. The free flow area percentage reduction is plotted in Figure 140. It shows the same trend as that of heat transfer area improvement. Free flow area decreases as bifurcation angle increases or length ratio decreases.


Figure 140 Air-side free flow area decrease compared with BTHX
4.5.3. Air-side heat transfer coefficient (AHTC)

Since it will be overwhelming to plot all data in a single plot, representative points are selected to illustrate the influence of each parameter. The results are summarized in Figure 141 through Figure 144. AHTC of bBTHX is higher than that of BTHX at higher air velocity (2, 3.5 and $5 \mathrm{~m} / \mathrm{s}$ ) and might be lower when air velocity is low ( $0.5 \mathrm{~m} / \mathrm{s}$ ).

Effect of air velocity: The effect of air velocity on AHTC is shown in Figure 141. As air velocity increases, the AHTC increases non-linearly for all cases.


Figure 141 Effect of velocity and diameter on AHTC ( $\theta=10^{\circ}, \mathrm{LR}=0.5$, Pattern 1)
Effect of diameter: The effect of diameter on AHTC is also shown in Figure 141. AHTC increases as diameter decreases. This can be explained by using definition of AHTC and Nusselt number. The relationship between AHTC and diameter as shown in Equation (86) can be derived by using the definition of Nusselt number and Reynolds number as shown in Equation (83), (84) and (85).

$$
\begin{align*}
& N u=\frac{h D}{k}  \tag{83}\\
& N u=C_{0} \operatorname{Re}^{m} \operatorname{Pr}^{n}(m, n \in[0,1])  \tag{84}\\
& \operatorname{Re}=\frac{U D}{v}  \tag{85}\\
& h=\frac{U^{m}}{v^{m}} C_{0} k \operatorname{Pr}^{n} \frac{1}{D^{1-m}} \tag{86}
\end{align*}
$$

Effect of bifurcation angle: AHTCs of bBTHX with different bifurcation angles are shown in Figure 142. AHTC increases as bifurcation angle increases. The main reason is that the increase of bifurcation angle leads to smaller free flow area, as shown in Figure 140, thus better flow mixing.


Figure 142 Effect of bifurcation angle on AHTC ( $\mathrm{d}=\mathbf{2} \mathbf{~ m m}, \mathrm{LR}=0.5$, Pattern 1)
Effect of length ratio: AHTCs of different bBTHX with different LR are shown in Figure 143. It can be found that AHTC increases as length ratio decreases. This can also be explained using the free flow area percentage decrease (Figure 140) described before. Designs with smaller LR have smaller free flow area. Another reason is with smaller length ratio, there are more bifurcation structures in unit volume, which enhance 3D flow mixing and result in higher AHTC.


Figure 143 Effect of LR on AHTC ( $\mathrm{d}=\mathbf{2} \mathbf{~ m m}, \boldsymbol{\theta}=60^{\circ}$, Pattern 2)

Effect of pattern: Two different patterns have been studied and the difference is shown in Figure 144. Pattern 2 has higher AHTC than pattern 1. The free flow area and heat transfer area for pattern 1 and 2 per unit envelope volume are similar thus the difference in AHTC must come from the pattern itself.


Figure 144 Effect of pattern on AHTC $\left(\mathrm{d}=\mathbf{2} \mathrm{mm}, \boldsymbol{\theta}=10^{\circ}, \mathrm{LR}=0.5\right)$
Take the configuration in Figure $144\left(\mathrm{~d}=2 \mathrm{~mm}, \theta=10^{\circ}, \mathrm{LR}=0.5\right)$ as an example, I plotted the contours of surface heat transfer coefficient at $5 \mathrm{~m} / \mathrm{s}$ to see the difference between two patterns, as shown in Figure 145. In the computational domain, two rows of tubes are calculated and the results are plotted separately. The difference mainly comes from the second-row tube. For pattern 1, there is certain area on each of secondary tube surface where the heat transfer coefficient is nearly zero due to the blockage of first row tube. For pattern 2, the effect of blockage is reduced by staggering the second-row tube to let the bifurcation area of second row tube be in the middle of the bypass area formed by first row tube, thus the average AHTC on secondary tube is higher than that of pattern 1 . Therefore, pattern 2 geometry has higher heat transfer coefficient.


AHTC Pattern 1 (left: second row; right: first and second row)
area


Pattern 2, second row


Pattern 2, first and second row

Figure 145 Surface heat transfer coefficient of pattern 1 and 2 $\left(\mathrm{d}=2 \mathrm{~mm}, \theta=10^{\circ}, \mathrm{LR}=0.5, \mathrm{Va}=5 \mathrm{~m} / \mathrm{s}\right)$
4.5.4. Overall conductance (hA)

To find out which geometry is better for heat exchanger design, the comparison of overall conductance (represented using hA value) per envelope volume, which is the product of heat transfer coefficient and heat transfer area, is necessary. The influence of air velocity, diameter, bifurcation angle, length ratio (LR) and different pattern on overall conductance are the same as that on AHTC. Thus, here this part is not repeated.


Figure 146 Effect of diameter on airside $h A$ ( $\theta=10^{\circ}, L R=0.5$, Pattern 1)


Figure 147 Effect of bifurcation angle on airside $h A$ ( $\mathrm{d}=2 \mathrm{~mm}, \mathrm{LR}=0.5$, Pattern 1)


Figure 148 Effect of $L R$ on airside $h A$ (d=2 mm, $\boldsymbol{\theta}=60^{\circ}$, Pattern 2)


Figure 149 Effect of pattern on airside $h A$ $\left(\mathrm{d}=\mathbf{2} \mathrm{mm}, \theta=10^{\circ}, \mathrm{LR}=0.5\right)$
4.5.5. Air-side pressure drop (ADP) per depth

First, the influence of air velocity, diameter, bifurcation angle, length ratio (LR) and different pattern on air side pressure drop (ADP) per depth are discussed. Again, representative data are summarized in Figure 150 through Figure 154.

Effect of air velocity: The effect of air velocity on ADP/Depth is shown in Figure 150. As air velocity increases, ADP/Depth increases non-linearly.

Effect of diameter: The effect of diameter on ADP/Depth is also shown in Figure 150. It is easy to understand that ADP/Depth increases as diameter decreases.


Figure 150 Effect of diameter on ADP/Depth $\left(\boldsymbol{\theta}=\mathbf{1 0}^{\circ}, \mathrm{LR}=\mathbf{0} .5\right.$, Pattern 1)

Effect of bifurcation angle: ADP/Depth values of bBTHX with different bifurcation angles are shown in Figure 151. Increase of bifurcation angle is related to decrease of free flow area, larger mass flux, thus larger ADP/Depth.


Figure 151 Effect of bifurcation angle on ADP/Depth ( $\mathrm{d}=\mathbf{2} \mathbf{~ m m}, \mathrm{LR}=\mathbf{0 . 5}$, Pattern 1)
Effect of pattern: This was discussed in detail in AHTC comparison. It is easy to tell from Figure 153 that pattern 2 has more flow bypass area as compared to pattern 1 thus ADP is smaller for pattern 2.


Figure 152 Effect of pattern on ADP/Depth


Pattern 2
Figure 153 Geometry of bBTHX at $\boldsymbol{\theta}=\mathbf{1 0}^{\circ}, \mathrm{LR}=\mathbf{2 . 5}$, Pattern 2
Effect of length ratio: ADP/Depth of different bBTHX with different LR are shown in Figure 154 and Figure 155. The difference between the two graphs is the bifurcation angle. Bifurcation angle is $60^{\circ}$ in Figure 154 and $10^{\circ}$ in Figure 155. As length ratio increases, ADP/Depth decreases. This is expected considering that as length ratio increases, free flow area increases, thus ADP/Depth decreases. However, it should be noticed that in Figure 155, as length ratio increases, the ADP/Depth of bBTHX is smaller than BTHX.


Figure 154 Effect of LR on ADP/Depth ( $\mathrm{d}=\mathbf{2} \mathbf{~ m m}, \boldsymbol{\theta}=\mathbf{6 0}^{\circ}$, Pattern 2)


Figure 155 Effect of LR on ADP/Depth ( $\mathrm{d}=\mathbf{2} \mathbf{~ m m}, \boldsymbol{\theta}=\mathbf{1 0}^{\circ}$, Pattern 2)
4.5.6. Summary of airside simulation results

Based on previous discussion we conclude that the bBTHX geometry has larger heat transfer area, AHTC and overall conductance than BTHX thus it has immense potential to be applied in air-cooled heat exchanger field. Table 19 summarizes the influence of air velocity,
diameter, bifurcation angle, length ratio (LR) and different pattern on AHTC, ADP and overall conductance.

Table 19 Summary of airside parametric study

|  | Air velocity | Tube diameter | Bifurcation angle | Length <br> ratio | Pattern |
| :---: | :---: | :---: | :---: | :---: | :---: |
| AHTC $\uparrow$ | $\uparrow$ | $\downarrow$ | $\uparrow$ | $\downarrow$ | 2 |
| ADP $\downarrow$ | $\downarrow$ | $\uparrow$ | $\downarrow$ | $\uparrow$ | 2 |
| $\mathrm{hA} \uparrow$ | $\uparrow$ | $\downarrow$ | $\uparrow$ | $\downarrow$ | 2 |

Note: $\uparrow$ : increase; $\downarrow$ : decrease

### 4.6. Waterside Simulation Results and Discussion

### 4.6.1. Heat transfer area

Since the bBTHX is a bare tube heat exchanger, the waterside heat transfer area is close to airside heat transfer area, thus the analysis of heat transfer area is the same as in Chapter 4.5.1.

### 4.6.2. Heat transfer coefficient

This section discusses about the effect of different parameters on heat transfer coefficient.

Effect of water velocity: The effect of water velocity on WHTC is shown in Figure 156. As the water velocity increases, the waterside heat transfer coefficient (WHTC) increases for all cases. And for all cases, the bBTHX has a larger WHTC than BTHX. The advantage of WHTC improvement is more obvious at the laminar flow cases for BTHX. This is because for bBTHX, even when Reynolds number is low, the addition of bifurcation will cause boundary layer reinitiation, secondary flow and flow mixing, which all contribute to a higher local heat transfer coefficient.

Effect of diameter: The effect of diameter on WHTC is also shown in Figure 156. The WHTC increases as diameter decreases. This can be explained by using the definitions of WHTC and Nusselt number. This analysis is the same as that of airside, shown in Equation (83) through (86).


## Figure 156 Effect of velocity and diameter on WHTC ( $\theta=10^{\circ}$, LR=0.5)

Effect of bifurcation angle: The WHTCs of bBTHX with different bifurcation angles are shown in Figure 157. WHTC increases as bifurcation angle increases. As bifurcation angle increase, the velocity vector change of original flow is larger, causing more flow mixing. This is consistent with findings in fractal channels by Wang et al. (2007).


Figure 157 Effect of bifurcation angle on WHTC ( $\mathbf{d}=2 \mathbf{~ m m}, ~ L R=1.5$ )
Effect of length ratio: the WHTCs of different bBTHX with different LR are shown in Figure 158. It can be found that the WHTC increases as LR decreases. This is because the WHTC is area-averaged HTC, designs with smaller LR have more bifurcations in unit length, resulting in larger overall WHTC.


Figure 158 Effect of LR on WHTC $\left(\mathrm{d}=2 \mathbf{~ m m}, \boldsymbol{\theta}=10^{\circ}\right)$

### 4.6.3. Overall conductance

To find out which geometry is better for heat exchanger design, the comparison of overall conductance per envelope volume, which is the product of heat transfer coefficient and heat transfer area, is necessary. Here the area A is calculated based on unit length. The influences of velocity, diameter, bifurcation angle, and length ratio (LR) on overall conductance are the same as that on AHTC. Thus, here we do not repeat it.


Figure 159 Effect of velocity and diameter on $h A\left(d=\mathbf{2 m m}, \boldsymbol{\theta}=10^{\circ}\right)$


Figure 160 Effect of bifurcation angle on $h A(d=2 \mathrm{~mm}, L R=1.5)$


Figure 161 Effect of LR on hA
$\left(\mathrm{d}=\mathbf{2} \mathbf{~ m m}, \theta=10^{\circ}\right)$
4.6.4. Water-side pressure drop (WDP) per length

This section first discusses about the influences of water velocity, diameter, bifurcation angle and length ratio (LR) on water side pressure drop (WDP) per length. The definition of length is shown in Figure 162.


Figure 162 bBTHX computational domain

Effect of water velocity: The effect of water velocity on WDP/Depth is shown in Figure 163. As water velocity increases, WDP/Depth increases non-linearly.

Effect of diameter: The effect of diameter on WDP/Depth is also shown in Figure 163. It is easy to understand that WDP/Depth increases as diameter decreases.


## Figure 163 Effect of velocity and diameter on WDP/Length ( $\theta=10^{\circ}, \mathrm{LR}=\mathbf{0 . 5}$ )

Effect of bifurcation angle: WDP/Depth of bBTHX with different bifurcation angles are shown in Figure 164. Increase of bifurcation angle is related to better flow mixing and more secondary flow motions, leading to larger ADP/Depth. This is explained in detail later.


Figure 164 Effect of bifurcation angle on WDP/Length ( $\mathrm{d}=\mathbf{2} \mathbf{~ m m}, \mathrm{LR}=\mathbf{1 . 5}$ )
Effect of length ratio: WDP/Depth of bBTHX with different length ratios are shown in Figure 165. Designs with smaller LR have more bifurcations in unit length, resulting in larger pressure drop per depth.


Figure 165 Effect of length ratio on WDP/Length ( $\mathrm{d}=\mathbf{2} \mathbf{~ m m , ~} \boldsymbol{\theta}=10^{\circ}$ )

### 4.6.5. j and f factors

$j$ and $f$ factors are calculated using Equation (23) and Equation (24), respectively. Here, f factor refers to Fanning friction factor. Figure 166 summarizes the results of j factors. From this graph, it can be found that j factor increases as Reynolds number decreases for both BTHX and bBTHX. And j factor of bBTHX increases as LR decreases or bifurcation angle increases. This is consistent with the findings in Chapter 4.5.2. The two points highlighted inside the red circle are in turbulence regime.


Figure $166 \boldsymbol{j}$ factor of waterside parametric study results for bBTHX

Figure 167 summarizes the results of f factors. From this graph, we can find that f factor increases as Reynolds number decreases for both BTHX and bBTHX. And f factor of bBTHX increases as LR decreases or bifurcation angle increases. This is consistent with the findings in Chapter 4.5.4.


## Figure $167 \boldsymbol{f}$ factor of waterside parametric study results for bBTHX

Figure 168 summarizes the results of $\mathrm{j} / \mathrm{f}$ value. From this graph, it can be found that for cases that bifurcation angle is $35^{\circ}, \mathrm{j} / \mathrm{f}$ values are similar when LR equals $0.5,1.5$ and 2.5. For cases that bifurcation angle is $10^{\circ}$ and $60^{\circ}, \mathrm{j} / \mathrm{f}$ values increase as LR increases. The $\mathrm{j} / \mathrm{f}$ value of most cases of bBTHX are higher than those of BTHX. For cases that $\mathrm{LR}=0.5$ and $\theta=60^{\circ}, \mathrm{j} / \mathrm{f}$ value is lower than that of BTHX. One reason is flow separation, which generates large form drag. This is explained in detail in Chapter 4.8. Again, $\mathrm{j} / \mathrm{f}$ value of BTHX in turbulent regime is much higher than that in laminar regime. However, for such heat exchangers, the operating conditions are mostly in laminar regime.


Figure $168 \boldsymbol{j} / f$ value of waterside parametric study results for bBTHX
Another lesson learned here is that j and f factor can be described as the functions of Reynolds number and geometry parameters. And this is the theoretical foundation of building metamodel. In metamodeling, Nusselt number and f factor are used. Nusselt number is expressed as the function of Reynolds number, Prandtl number and geometry parameters while $f$ factor is expressed as the function of Reynolds number and geometry parameters.

### 4.6.6. Summary of waterside simulation results

Based on previous discussion it is found that the bBTHX geometry has larger heat transfer area, WHTC and overall conductance than BTHX thus it has immense potential to be applied in air-cooled heat exchanger field. Table 19 summarized the influence of water velocity, diameter, bifurcation angle and length ratio (LR) on WHTC, WDP/Length and overall conductance.

Table 20 Summary of waterside parametric study

|  | Water velocity | Tube diameter | Bifurcation angle | Length <br> ratio |
| :---: | :---: | :---: | :---: | :---: |
| WHTC $\uparrow$ | $\uparrow$ | $\downarrow$ | $\uparrow$ | $\downarrow$ |
| WDP/Length $\downarrow$ | $\downarrow$ | $\uparrow$ | $\downarrow$ | $\uparrow$ |
| hA $\uparrow$ | $\uparrow$ | $\downarrow$ | $\uparrow$ | $\downarrow$ |

Note: $\uparrow$ : increase; $\downarrow$ : decrease

In real heat exchanger design, what engineers care most is the heat transfer capacity per power input. Here the overall conductance and WDP/Length of bBTHX and BTHX are compared. From Figure 169, bBTHX has higher overall conductance. But BTHX can have considerably high overall conductance when it transients to turbulent flow.


Figure 169 Overall conductance and WDP/Length comparison

### 4.7. Airside Thermal and Hydraulic Mechanisms

4.7.1. Heat Transfer Mechanism

There are three main reasons for AHTC improvement. To illustrate this, I use the data for 0.8 mm since it was validated experimentally.

Reason 1: bBTHX has larger mass flux based on free flow area (Figure 170). And it can also be noticed that even with the same mass flux, bBTHX still has a higher AHTC (Figure 171). This is due to reasons 2 and 3 .


Reason 2: Branch tube of bBTHX has smaller diameter, and it can be seen from Figure 172 that smaller diameter bare tube with same pitch value has larger AHTC.


Figure 172 AHTC of BTHX with different diameter

Reason 3: 3D boundary layer development. This will be explained in detail in the following hydraulic performance discussion. Briefly, 3D boundary layer formed around bifurcation enhances flow mixing, resulting in higher heat transfer coefficient. This phenomenon could not be observed in BTHX due to 2D flow assumption. Thus factors, for example, smaller length ratio, that can lead to more bifurcation structures in a certain control volume have positive impact on heat transfer coefficient improvement.

### 4.7.2. Flow friction mechanism

Here explains the reason why bBTHX could have lower pressure drop for some designs. Still, the results of 0.8 mm BTHX and bBTHX are used here to explain the reasons because they were validated experimentally. Figure 173 shows ADP/Depth of bBTHX is about $4 \sim 12 \%$ lower than that of BTHX at air velocity of 3.5 and $5 \mathrm{~m} / \mathrm{s}$.


Figure 173 ADP/Depth of bBTHX and BTHX $\left(\mathrm{d}=\mathbf{0 . 8} \mathbf{~ m m}, \boldsymbol{\theta}=10^{\circ}, \mathrm{LR}=\mathbf{2 . 5}\right.$, Pattern 2)

Reason 1: For designs with bifurcation equals $10^{\circ}$ and LR equals 2.5, the increase of heat transfer area and decrease of free flow area are the smallest among all configurations.

Reason 2: There is flow bypass in bBTHX designs (as shown in Figure 174) and at larger length ratio, the effect of bypass is more obvious. From Figure 174, it can be noticed that there is a sharp decrease of pressure drop for bBTHX where there is flow bypass at the bifurcation.


Figure 174 Flow bypass at bifurcation ( $\mathrm{d}=\mathbf{0 . 8} \mathrm{mm}, \boldsymbol{\theta}=10^{\circ}, \mathrm{LR}=\mathbf{2 . 5}$, Pattern 2)

Reason 3: Lower flow rate at bare tube region. For bBTHX, it can be divided into two regions, one is bare tube region which consists of staggered vertical tubes, and this configuration is the same as BTHX. The other region is 3D flow region, where the bifurcations is. This is shown in Figure 174. bBTHX has smaller velocity magnitude in that region, as shown in Figure 175, the cross section selected are shown in Figure 176. Note that the computational domain for BTHX is 2D, here a 3D graph is used to make it easier for readers to understand.


Figure 175 Radial velocity magnitude of BTHX and bBTHX

(a)

$z=0 \mathrm{~mm} \quad z=10 \mathrm{~mm}$

## (b)

Figure 176 Cross section of BTHX (a) and bBTHX (b) ( $\mathrm{z}=0 \mathbf{~ m m}$ and $\mathrm{z}=10 \mathrm{~mm}$ )

Reason 4: 3D flow at bifurcation. In 3D region of bBTHX, there is $z$ direction velocity, as shown in Figure 177. This 3D flow phenomenon occurs at the bifurcation only and is the key mechanism causing the improvement of AHTC and decrease of ADP/Depth for bBTHX simultaneously.


Figure 177 Airside velocity in z direction for bBTHX
To understand the influence of z direction velocity, the drag coefficient of sphere and cylinder are shown in Figure 178. Solid line is for sphere and dotted line is for cylinder. At low Reynolds number, the cylinder has lower drag coefficient but at larger Reynolds number, the sphere has lower pressure drop. In this study, the Reynolds number range is from 23 to 233. Plotting the Nusselt number for cylinder and sphere, it can be found the sphere's Nusselt number is always larger than cylinder, as shown in Figure 179. In current design, the bifurcation enables 3D flow just like the sphere does.


Figure 178 Drag coefficient for cylinder and sphere in free flow (Incropera, 2007)


Figure 179 Nusselt number of cylinder and sphere in free flow (Correlations from Incropera, 2007)

### 4.7.3. Summary of airside mechanisms

Novel bifurcated bare tube heat exchanger had $15 \%$ higher air-side heat transfer coefficient and 4~12\% lower air-side pressure drop than baseline bare tube heat exchanger with same diameter $(0.8 \mathrm{~mm})$, frontal area, volume at a certain air velocity range $(3.5 \sim 5 \mathrm{~m} / \mathrm{s})$. All the mechanism and their influences on AHTC and ADP/Depth are summarized in Table 21. The key mechanism that enables some designs in the parametric study to have higher heat transfer coefficient and smaller pressure drop is 3D flow enabled by bifurcation in bBTHX design.

Table 21 Summary of all airside mechanisms

|  | AHTC | ADP/Depth |
| :---: | :---: | :---: |
| Larger mass flux | + | + |
| Branch tube (smaller OD) | + | + |
| Flow bypass | - | - |
| Lower flow rate at bare tube region | - | - |
| 3D flow | + | - |

### 4.8. Waterside thermal and hydraulic mechanisms

In this section, the thermal and hydraulic mechanisms of waterside are discussed. Figure 180 shows the velocity contour of BTHX and bBTHX, the color map scales are the same and only fully developed regions are presented. It can be found that due to the addition of bifurcation, the flow in bBTHX is quite different from that of BTHX.

### 4.8.1. Smaller branch tubes

The diameter of branch tubes is 0.7 of the main tube diameter. As discussed before, $h \propto \frac{1}{D^{1-m}}$, where $0 \leq \mathrm{m} \leq 1$, thus smaller tubes will result in higher heat transfer coefficient. Similarly, the relationship between pressure drop and diameter can be derived. Equation (87), (88) and (89) show smaller diameter tubes results in larger pressure drop.

$$
\begin{equation*}
\Delta P=f \frac{l}{D} \frac{\rho V^{2}}{2} \tag{87}
\end{equation*}
$$

$$
\begin{equation*}
f \propto \frac{1}{\operatorname{Re}^{m}}, \quad 0<m \leq 1 \tag{88}
\end{equation*}
$$

$$
\begin{equation*}
\Delta P / l \propto \frac{1}{\operatorname{Re}^{m}} \frac{l}{D} \frac{\rho V^{2}}{2} \propto \frac{1}{\left(\frac{V D}{v}\right)^{m}} \frac{l}{D} \frac{\rho V^{2}}{2} \propto \frac{1}{D^{m+1}}, \quad 0<m \leq 1 \tag{89}
\end{equation*}
$$

In current design, addition of small diameter tube improves local heat transfer coefficient and leads to higher pressure drop. This is illustrated in the following section.

### 4.8.2. Boundary layer interruption and redevelopment

Figure 180 presents the velocity contour of middle plane in BTHX and bBTHX. The definition of middle plane is also shown. Figure 180 shows at the bifurcation, flow is interrupted and boundary layer redevelops at the bifurcation. Figure 181 shows the velocity vector. Flow is brought to rest at the forward stagnation point, with an accompanying rise in pressure. The kinetic pressure converts to static pressure. The boundary layer redevelopment boosts local heat transfer, but contributes to larger pressure drop as well.

Pressure along the flow direction of bBTHX is plotted in Figure 182. Pressure of BTHX is also plotted as a comparison. bBTHX has a higher pressure drop compared with BTHX with the same inlet diameter. Static pressure is related to kinetic pressure; thus, velocity contour is also plotted in Figure 182. Static pressure of bBTHX spikes at the bifurcation. But right after it, there is a sudden drop due to velocity increase which is related to the surface shape change. Another reason that bBTHX has larger pressure drop is smaller branch tubes, as discussed before. Figure 182 shows the static pressure drops faster in branch tube region, and slower in main tube region, and the slope of main tube region is similar to bare tube with same diameter. But the slope of branch tube region is slightly higher than that of bare tube with same diameter due to the local loss of bifurcation and boundary layer redevelopment.

Local heat transfer coefficient also spikes at the bifurcation where the stagnation point is, which can be seen from Figure 183. After the short entry length, surface heat transfer of BTHX becomes very stable, which is expected because it is laminar flow and the Nusselt number is constant theoretically. Smaller diameter bare tube has a larger local heat transfer coefficient. So, for bBTHX, the heat transfer coefficient is higher in branch tube region. At large diameter tube region, the heat transfer coefficient of bBTHX is quite similar to that of BTHX.


Figure 180 Velocity contour of fully developed region of BTHX at mid-plane (a) $\left(\mathrm{d}=2 \mathrm{~mm}, \mathrm{~V}_{\mathrm{w}}=0.1 \mathrm{~m} / \mathrm{s}\right)$ and bBTHX (b) $\left(\mathrm{d}=2 \mathrm{~mm}, \theta=10^{\circ}, \mathrm{LR}=0.5, \mathrm{~V}_{\mathrm{w}}=0.1 \mathrm{~m} / \mathrm{s}\right)$


Figure 181 Velocity vector at bifurcation at mid-plane (color is velocity magnitude) $\left(\mathrm{d}=2 \mathrm{~mm}, \theta=10^{\circ}, \mathrm{LR}=0.5, \mathrm{~V}_{\mathrm{w}}=0.1 \mathrm{~m} / \mathrm{s}\right)$


Figure 182 Static pressure of BTHX ( $\mathrm{d}=\mathbf{2} \mathbf{~ m m}, \mathrm{V}_{\mathrm{w}}=\mathbf{0 . 1} \mathrm{m} / \mathrm{s}$ ), BTHX- $0.7 \mathrm{D}(\mathrm{d}=\mathbf{1 . 4} \mathbf{~ m m}$, $V_{w}=0.1 \mathrm{~m} / \mathrm{s}$ ) and bBTHX ( $\mathrm{d}=2 \mathrm{~mm}, \boldsymbol{\theta}=35^{\circ}, \mathrm{LR}=2.5, \mathrm{~V}_{\mathrm{w}}=0.1 \mathrm{~m} / \mathrm{s}$ ) at mid-plane


Figure 183 Surface heat transfer coefficient of BTHX ( $\mathrm{d}=2 \mathrm{~mm}, \mathrm{~V}_{\mathrm{w}}=\mathbf{0 . 1} \mathbf{~ m} / \mathrm{s}$ ), BTHX$0.7 \mathrm{D}\left(\mathrm{d}=1.4 \mathrm{~mm}, \mathrm{~V}_{\mathrm{w}}=0.1 \mathrm{~m} / \mathrm{s}\right)$ and $\mathrm{bBTHX}\left(\mathrm{d}=2 \mathrm{~mm}, \theta=35^{\circ}, L R=2.5, V_{w}=0.1 \mathrm{~m} / \mathrm{s}\right)$

### 4.8.3. Flow separation

Flow separation is not found in all cases, but in those with large bifurcation angle. As discussed before, boundary layer redevelops at the bifurcation. But as the boundary layer develops, there is a chance that the pressure gradient becomes adverse, causing flow deceleration and separation. Figure 184 shows the boundary layer detaches from the surface and a stable pair of wake is formed in the downstream region. The flow is separated but steady and the formation of the wake generates high form drag, in addition to friction drag caused by surface shear.


Figure 184 Velocity field of $\mathrm{bBTHX}\left(\mathrm{d}=4 \mathrm{~mm}, \mathrm{LR}=0.5, \theta=60^{\circ}, V_{w}=1 \mathrm{~m} / \mathrm{s}, \mathrm{Re}=3943\right.$ ) at mid-plane
Then here explains why cases with bifurcation angle equals $60^{\circ}$ has considerably larger f factor than others. The most important reason is flow separation. As length ratio increases, the f factor decreases. This is not only because larger length ratio results in less bifurcation per length, but also due to the smoother transition of angle, as shown in Figure 185. Comparing (a) (b) and (c), as LR increases, the geometry becomes more streamline, thus the flow separation point is further in the downstream, and the wake area generated is smaller. The difference between (b) and
(c) is smaller than that between (a) and (b). This explains why in Figure 167, f factor of cases with bifurcation angle $60^{\circ}$ and LR 0.5 is much larger than others.


Figure 185 Effect of $L R$ on flow separation
$\left(\mathrm{d}=\mathbf{4 \mathrm { mm }}, \boldsymbol{\theta}=60^{\circ}, V_{\mathrm{w}}=1 \mathrm{~m} / \mathrm{s}\right)$
4.8.4. Secondary flow

Secondary flow is also not found in all cases, but in those with large bifurcation angle. Figure 186 shows that there is secondary flow due to the shape change of the tube. In this example, bifurcation angle is large, so there is a separated region of flow near the inside of the tube bend. The swirling secondary flow occurs because of the imbalance of centripetal forces because of the curvature of the tube centerline (Munson et al., 2006). Secondary flow acts to replace the slowmoving fluid near the walls with faster-moving fluid, therefore generating large viscous friction at the wall. Secondary flow also enhances thermal mixing by bringing colder flow from inside to the wall, thus larger local Nusselt number (Senn and Poulikakos, 2004). Another influence on heat transfer is that secondary flow changes the location of the highest temperature, as shown in Figure 187.


Figure 186 Velocity field of Secondary flow ( $\mathrm{d}=2 \mathrm{~mm}, \mathrm{LR}=0.5, \theta=60^{\circ}, V_{w}=0.1 \mathrm{~m} / \mathrm{s}$ )


Figure 187 Temperature contour of geometry without (a) and with (b) secondary flow
4.8.5. Summary of waterside mechanisms

The heat transfer and pressure drop characteristics of bBTHX have several unique features due to addition of bifurcations.

- The branch tube with smaller diameter increases heat transfer coefficient as well as pressure drop.
- At bifurcation, flow splits into two streams and boundary layer interrupts and redevelops along the wall. This enhances heat transfer and increases pressure drop. Pressure drop and heat transfer coefficient spike at stagnation point.
- The shape of bifurcation along the waterflow direction is usually close to oval shape, causing flow separations when the angle is large. This results in heat transfer coefficient increase but accompanying with form drag.
- Secondary flow, found in geometries with large bifurcation angle, enhances flow mixing with accompanying larger pressure loss. Secondary flow also changes the location of the highest temperature flow.

Table 22 Summary of all waterside mechanisms

|  | WHTC | WDP/Depth |
| :---: | :---: | :---: |
| Branch tube (smaller ID) | + | + |
| Boundary layer redevelop | + | + |
| Flow separation | + | + |
| Secondary flow | + | + |

## Chapter 5: bBTHX Optimization

### 5.1. Airside Meta-Model

The input and output parameters of bBTHX airside meta-model is summarized in Table 23. Chilton - Colburn j and f factors are selected as the output parameters. As stated before, the accuracy of metamodel is evaluated using the Metamodel Acceptability Score (MAS) (Hamad, 2006). $100 \%$ of j and f factor predicted are within $15 \%$ deviation from CFD values, as shown in Figure 188.

Table 23 Summary of airside meta-model

| Input parameters (normalized) | $N_{\mathrm{r}}$ | $D_{1}$ | $P_{l} / D_{1}$ | $P_{t} / D_{1}$ | LR | $\theta$ | $V_{a}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Output parameters | j |  |  | f |  |  |  |
| Meta-model points | 783 |  |  |  |  |  |  |
| Random points for verification | 50 |  |  |  |  |  |  |
| Metamodel Acceptability | j |  |  | f |  |  |  |
| Score | 100\% within $15 \%$ |  |  | 100\% within 15\% |  |  |  |



Figure 188 Airside metamodel prediction of $\mathbf{j}$ factor (a) and $\mathbf{f}$ factor (b)

### 5.2. Waterside Meta-Model

The input and output parameters of bBTHX waterside meta-model are summarized in Table 24. Nusselt number and WDP/segment are selected as the output parameters and the definition of one segment is shown in Figure 189. It is the minimum segment during bBTHX modeling. As stated before, the accuracy of metamodel is evaluated using the Metamodel Acceptability Score (MAS) (Hamad, 2006). 100\% of Nusselt number predicted are within 8\% deviation from CFD Nusselt values. $96 \%$ of WDP/segment predicted by meta-model are within 10\% deviation from CFD results, as shown in Figure 190.

Table 24 Summary of waterside meta-model

| Input parameters <br> (normalized) | $D_{1}$ | $P_{l} / D_{1}$ | LR |
| :---: | :---: | :---: | :---: |
| Output parameters | Nu | $V_{w}$ |  |
| Meta-model points | 258 |  |  |
| WDP/segment |  |  |  |
| Random points for verification |  | 25 |  |
| Metamodel Acceptability Score | Nu | WDP/segment |  |
|  | $100 \%$ within $8 \%$ | $96 \%$ within $10 \%$ |  |

Figure 189 Definition of one segment on waterside


Figure 190 Waterside metamodel prediction of Nusselt number (a) and ADP/segment (b)

## 5.3. bBTHX Solver and Validation

### 5.3.1. bBTHX area correction

The heat transfer area calculated using geometry parameters approximates the area of bBTHX geometry model built in ANSYS ${ }^{\circledR}$ software. bBTHX heat transfer area of one segment calculated using geometry parameters is shown in Equation (90) and (91).

$$
\begin{align*}
& A_{\text {air }}=2 \pi D_{1, o} L_{1}+4 \pi D_{2, o} L_{2}  \tag{90}\\
& A_{\text {water }}=2 \pi D_{1, i} L_{1}+4 \pi D_{2, i} L_{2} \tag{91}
\end{align*}
$$

Where $D_{1, o}$ and $D_{2, o}$ are outer diameter of main tube and branch tube and $D_{1, i}$ and $D_{2, i}$ are inner diameter of main tube and branch tube. The other parameters are shown in Figure 191. From this graph, the heat transfer area calculated using the above two equations are heat transfer area of several cylinders built around the centerline of bBTHX which is different from the real geometry. Thus, correction is needed.


Figure 191 Heat transfer area calculation for bBTHX


Using heat transfer area output from CFD software FLUENT ${ }^{\circledR}$ as the real value, a correction correlation was developed, as shown in Figure 192 and Equation (92). And the maximum deviation of the predicted values is within $\pm 3 \%$, as shown in Figure 193.

$$
\begin{equation*}
A_{C F D}=-856.03 A_{e q u}^{3}+17.206 A_{e q u}^{2}+0.9162 A_{e q u} \tag{92}
\end{equation*}
$$

### 5.3.2. CFD Verification

The bBTHX solver is verified against simulation results. This simulation is done using air and water simultaneously. The tube wall is set to be aluminum with thickness of 0.3 mm .

The input parameters are:
Table 25 Inlet conditions of bBTHX solver verification case

|  | $\mathrm{T}_{\text {in }}$ <br> $[\mathrm{K}]$ | $\mathrm{P}_{\text {in }}$ <br> $[\mathrm{kPa}]$ | $\mathrm{V}_{\text {in }}$ <br> $[\mathrm{m} / \mathrm{s}]$ | h <br> $\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right]$ | $\Delta \mathrm{P}$ <br> $[\mathrm{Pa}]$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Air | 300 | 101.325 | 4.9 | 280.1 | 25.3 |
| Water | 360 | 101.325 | 0.1 | 4858 | 245.6 |

Table 26 Geometry parameters of bBTHX solver verification case

| $\begin{gathered} O D \\ {[\mathrm{~mm}]} \end{gathered}$ | Thickness [mm] | $\begin{gathered} I D \\ {[\mathrm{~mm}]} \end{gathered}$ | $\begin{gathered} P_{l} / O D \\ {[-]} \\ \hline \end{gathered}$ | $\begin{gathered} \theta \\ {[\mathrm{deg}]} \end{gathered}$ | $\begin{aligned} & \text { LR } \\ & {[-]} \end{aligned}$ | Tube per row [-] | Tube segment \# [-] | Row \# <br> [-] |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.7 | 0.3 | 1.1 | 2.38 | 47.58 | 2.3 | 10 | 2 | 1 |

The simulation domain is shown in Figure 194.


Figure 194 Single segment computational domain for bBTHX
Results of CFD and bBTHX solver are shown in Table 27. The percentage difference between heat exchanger capacities is $0.44 \%$. Thus, the solver is verified.

Table 27 Comparison of simulation results from CFD and solver for bBTHX

|  | $\mathrm{T}_{\text {air, out }}$ <br> $[\mathrm{K}]$ | $\mathrm{T}_{\text {water, out }}[\mathrm{K}]$ | $\Delta \mathrm{P}_{\text {air }}$ <br> $[\mathrm{Pa}]$ | $\Delta \mathrm{P}_{\text {water }}$ <br> $[\mathrm{Pa}]$ | Q <br> $[\mathrm{W}]$ | $\mathrm{Q}_{\text {diff }}$ <br> $[-]$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| CFD | 303.73 | 352.66 | 24.8 | 245.28 | 28.28 | $0.44 \%$ |
| Solver | 303.40 | 352.60 | 25.2 | 245.60 | 28.40 | 0 |

### 5.4. Design Problem (DP)

The application of the methodology is to design a single-phase air-to-water heat exchanger which can deliver the same capacity as baseline but with smaller envelope volume, smaller material volume and lower total pumping power. Since the material volume is usually directly related to envelope volume so that the two optimization objectives are total power and heat exchanger envelope volume. Two baseline heat exchangers are selected, a conventional louvered fin mini-channel heat exchanger (MCHX) and a bare tube heat exchanger (BTHX) with outer
diameter of 0.8 mm . These two heat exchangers are plotted in Figure 195. However, these two heat exchangers are not optimized heat exchangers.


Figure 195 Two baseline heat exchangers for optimization
5.4.1. DP I: Single-Phase Heat Exchanger

### 5.4.1.1. Baseline heat exchanger

The first application of the methodology is a single-phase air-to-water heat exchanger. A louvered mini channel radiator, as shown in Figure 25, was chosen to be baseline for comparison. This new heat exchanger is designed to deliver the same capacity (835W) for such application. This MCHX is a one slab heat exchanger with 11 tubes and 2 ports per tube. The airside hydraulic diameter is 1.94 mm . Other parameters are summarized in Table 28. The aspect ratio is 1.8 , but in current optimization, its reciprocal 0.56 is used. This MCHX has been tested under dry condition using air and water and design condition used here is one of the testing conditions. The MCHX's data can be found in appendix and the experimental data has been compared to predictions using correlations from literature (Wang et al., 1999). The maximum deviation for capacity and pressure drop are $\pm 6$ and $\pm 20 \%$ respectively (Figure 196).

Table 28 Baseline MCHX

| Metric | Unit | Value |
| :---: | :---: | :---: |
| Air flow rate | $\mathrm{m}^{3} / \mathrm{s}$ | 0.05 |
| Air inlet temperature | K | 308.15 |
| Water flow rate | $\mathrm{g} / \mathrm{s}$ | 50 |
| Water inlet temperature | K | 333.15 |
| Heat load | W | 835 |
| Air pressure drop | Pa | 32.8 |
| Water pressure drop | kPa | 11.0 |
| Air heat transfer coefficient | $\mathrm{W} / \mathrm{m}^{2} \cdot \mathrm{~K}$ | 158 |
| Airside heat transfer area | $\mathrm{m}^{2}$ | 0.5525 |
| Water heat transfer coefficient | $\mathrm{W} / \mathrm{m}^{2} \cdot \mathrm{~K}$ | 2007.4 |
| Waterside heat transfer area | $\mathrm{m}^{2}$ | 0.0735 |
| Total pumping power (air+water) | W | 2.2 |
| Volume | $\mathrm{cm}^{3}$ | 395 |
| Material volume | $\mathrm{cm}^{3}$ | 59.5 |
| Length | m | 0.23 |
| Height | m | 0.1175 |
| Depth | m | 0.016 |
| Airside frontal area | $\mathrm{m}^{2}$ | 0.0247 |
| Aspect ratio (length/height or | - | 1.8 or 0.56 |
| height/length) | - | 1 |
| Number of banks | - | 11 |
| Tube per bank | - | 20 |
| Fin per Inch (FPI) | - |  |




Figure 196 Deviation of capacity and airside pressure drop against experimental data for MCHX

### 5.4.1.2. Optimization problem

The design problem here is described as below. Two objectives are total power and heat exchanger volume. The constraints include (1) total heat exchanger capacity should be similar or larger than baseline; (2) total power is $30 \%$ less than baseline; (3) total heat exchanger volume is $30 \%$ less than baseline; and (4) aspect ratio (AR) is similar to baseline. The last three constraints are dismissed for the real calculation because these will cause the calculation time to be much longer. The $30 \%$ better points were selected manually after the Pareto curve is produced.
Optimization I:
min Power
$\min V_{H X}$
s.t.

$$
\begin{aligned}
& \dot{Q}_{\text {baseline }} \leq \dot{Q} \leq 1.1 \cdot \dot{Q}_{\text {baseline }} \\
& 0.9 \cdot A R_{\text {baseline }} \leq A R \leq 1.1 \cdot A R_{\text {baseline }} \\
& \hline \text { Power } \leq 0.7 \cdot \text { Power }_{\text {baseline }} \\
& V_{H X} \leq 0.7 \cdot V_{H X \_ \text {baseline }} \\
& A_{f r} \leq A_{\text {fr_baseline }} \\
& \hline
\end{aligned}
$$

5.4.1.3. Optimization results and discussion

Optimization results data can be found in the appendix. The Pareto front is plotted in Figure 197. Pareto front points are those have $30 \%$ lower total power and $30 \%$ smaller volume than baseline MCHX with smaller frontal area. Pareto front with no constrains are plotted on the graph as well. The following paragraphs discuss the difference between the Pareto front points.

Along the Pareto front line, the volume increases while power decreases. The outer diameter of the tube increase (Figure 198) is the main reason of volume increase. Transverse pitch ( $\mathrm{Pt} / \mathrm{OD}$ ) reaches the minimum limit (1.5) and does not change while longitudinal pitch ( $\mathrm{Pl} / \mathrm{OD}$ ) stays constant at smaller volume and experiences a sudden increase. This change is accompanied
by the diameter and length ratio sudden changes (Figure 198). Pl/OD has a relationship with LR, as shown in Figure 199. As $\mathrm{Pl} / \mathrm{OD}$ increases, LR tends to decrease. This is because larger $\mathrm{Pl} / \mathrm{OD}$ results in lower airside heat transfer coefficient, thus LR needs to be smaller to increase the heat transfer coefficient, and vice versa.

Designs with larger volumes also have larger airside frontal area (Figure 197), airside heat transfer area (Figure 201), larger waterside heat transfer area (Figure 202) and larger material volume (Figure 205). Larger airside frontal area will cause the air velocity to decrease, thus airside heat transfer coefficient (Figure 201) and airside pumping power (Figure 203) decreases as the design volume increases.

As design diameter increases, the waterside velocity decreases, resulting in lower waterside heat transfer coefficient (Figure 202) and lower waterside pumping power (Figure 203). Airside total power percentage decreases as volume increases, as shown in Figure 204. Figure 205 presents the aspect ratio and material volume. The aspect ratio is controlled within the range of $0.49 \sim 0.62$ to maintain a similar value as baseline (0.56). Material volume increases as heat exchanger envelope volume increases.


Figure 197 Pareto front (DP I)


Figure 198 OD and LR of Pareto front points w/o constraints (DP I)


Figure 199 Pl and Pt of Pareto front points w/o constraints (DP I)


Figure 200 LR and PI/OD of Pareto front points w/o constraints (DP I)


Figure 201 AHTA and AHTC of Pareto front points w/o constraints (DP I)


Figure 202 WHTA and WHTC of Pareto front points w/o constraints (DP I)


Figure 203 Airside and waterside pumping power of Pareto front points w/o constraints (DP I)


Figure 204 Airside power percentage of Pareto front points w/o constraints (DP I)


Figure 205 Aspect ratio and material volume of Pareto front points w/o constraints (DP I)
5.4.1.4. Results verification

The optimization results were verified against CFD simulation results. Three points from Pareto front were selected and CFD runs were conducted. Table 29 shows the comparison of the airside heat transfer coefficient (AHTC), airside pressure drop (ADP), waterside heat transfer coefficient (WHTC) and waterside pressure drop (WDP). Note WDP means waterside pressure drop per simulation domain. The maximum percentage deviation of AHTC, ADP, WHTC and ADP is $6.4 \%$, which is acceptable. Thus, the Pareto front is verified.

Table 29 Optimization results verification against CFD simulation (DP I)

| Optimization results |  |  |  | CFD results |  |  |  | Percentage deviation |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\begin{gathered} \text { AHTC } \\ {\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right]} \end{gathered}$ | $\begin{gathered} \text { ADP } \\ {[\mathrm{Pa}]} \\ \hline \end{gathered}$ | $\begin{gathered} \text { WHTC } \\ {[\mathrm{W} / \mathrm{m} 2 \mathrm{~K}]} \end{gathered}$ | $\begin{gathered} \hline \text { WDP } \\ {[\mathrm{Pa}]} \\ \hline \end{gathered}$ | $\begin{gathered} \text { AHTC } \\ {\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right]} \end{gathered}$ | $\begin{gathered} \hline \mathrm{ADP} \\ {[\mathrm{~Pa}]} \\ \hline \end{gathered}$ | $\begin{gathered} \hline \text { WHTC } \\ {\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right]} \\ \hline \end{gathered}$ | $\begin{aligned} & \hline \text { WDP } \\ & {[\mathrm{Pa}]} \\ & \hline \end{aligned}$ | AHTC <br> [\%] | $\begin{gathered} \hline \text { ADP } \\ {[\%]} \\ \hline \end{gathered}$ | WHTC <br> [\%] | $\begin{gathered} \hline \text { WDP } \\ {[\%]} \\ \hline \end{gathered}$ |
| 386.38 | 20.91 | 14346.31 | 649.91 | 401.19 | 22.04 | 13538.14 | 622.92 | 3.8\% | 5.4\% | -5.6\% | -4.2\% |
| 370.00 | 18.79 | 13665.71 | 586.06 | 384.30 | 19.99 | 13740.29 | 618.43 | 3.9\% | 6.4\% | 0.5\% | 5.5\% |
| 351.90 | 17.46 | 12772.66 | 467.64 | 361.44 | 17.81 | 13028.65 | 467.81 | 2.7\% | 2.0\% | 2.0\% | 0.04\% |

### 5.4.1.5. Selected optimal design and baseline comparison:

One of the verified optimal designs is compared with baseline heat exchanger. Table 30 summarized all parameter values and percentage differences. This optimal design has $38 \%$ lower
total pumping power, $83 \%$ smaller volume and $87 \%$ smaller material volume. In following paragraphs, each parameter will be examined separately.

Higher airside heat transfer coefficient: bBTHX has 134\% higher airside heat transfer coefficient than baseline. Mechanisms of enhancing heat transfer were discussed in previous chapters, including larger mass flux, branch tube with smaller diameter and 3D flow caused by the addition of bifurcation. Another reason is the air frontal velocity of bBTHX ( $2.2 \mathrm{~m} / \mathrm{s}$ ) is slightly higher than that of MCHX $(2.02 \mathrm{~m} / \mathrm{s})$.

Lower airside pressure drop: Despites the reasons discussed that can reduce the airside pressure drop, the most important reason for a lower pressure drop is the reduction in depth. bBTHX's depth is $75 \%$ smaller than that of MCHX. The ADP/Depth value of bBTHX is higher than MCHX.

Higher waterside heat transfer coefficient: bBTHX has five times higher waterside heat transfer coefficient than baseline. The reasons were discussed in Chapter 4.8, i.e. smaller OD branch tube, boundary layer redevelopment and flow separation.

Lower waterside pressure drop: Waterside pressure drop of bBTHX is $25 \%$ lower than baseline. The reason is that it has smaller water flow path. Tube length of bBTHX is 11.24 mm while that of MCHX is 23 mm . bBTHX has larger WDP/Length value.

Lower total pumping power: bBTHX has $38 \%$ lower total pumping power, which is mainly due to the reduction on airside pressure drop.

Smaller volume and material volume: bBTHX has about $83 \%$ smaller volume and $87 \%$ smaller material volume than baseline.

Table 30 Comparison of selected optimal design and baseline MCHX (DP I)

| Metric | Unit | MCHX (experimental data) | bBTHX (simulation data) | \%diff |
| :---: | :---: | :---: | :---: | :---: |
| Air flow rate | $\mathrm{m}^{3} / \mathrm{s}$ | 0.05 | 0.05 |  |
| Air inlet temperature | K | 308.15 | 308.15 |  |
| Water flow rate | g/s | 50 | 50 |  |
| Water inlet temperature | K | 333.15 | 333.15 |  |
| Heat load | W | 835 | 845.9 | 1.3\% |
| Air pressure drop | Pa | 32.8 | 18.8 | -42.7\% |
| Water pressure drop | kPa | 11.0 | 8.3 | -24.5\% |
| Air heat transfer coefficient | $\mathrm{W} / \mathrm{m}^{2} \cdot \mathrm{~K}$ | 158 | 370 | 134.2\% |
| Airside heat transfer area | $\mathrm{m}^{2}$ | 0.5525 | 0.1639 | -70.3\% |
| Water heat transfer coefficient | $\mathrm{W} / \mathrm{m}^{2} \cdot \mathrm{~K}$ | 2007.4 | 13665 | 580.7\% |
| Waterside heat transfer area | $\mathrm{m}^{2}$ | 0.0735 | 0.13115 | 78.4\% |
| Total pumping power (air+water) | W | 2.2 | 1.36 | -38.2\% |
| Volume | $\mathrm{cm}^{3}$ | 395 | 67.71 | -82.9\% |
| Material volume | $\mathrm{cm}^{3}$ | 59.5 | 7.96 | -86.6\% |
| Length | M | 0.23 | 0.2022 | -12.1\% |
| Height | M | 0.1175 | 0.1124 | -4.3\% |
| Depth | M | 0.016 | 0.004 | -75.0\% |
| Airside frontal area | $\mathrm{m}^{2}$ | 0.0247 | 0.022744 | -7.9\% |
| Aspect ratio (length/height or height/length) | - | 1.8 or 0.56 |  |  |
| Number of banks | - | 1 | 4 |  |
| Tube per bank | - | 11 | 200 |  |
| Fin per Inch (FPI) | - | 20 | - |  |
| $\mathrm{Pl} / \mathrm{OD}$ | - | - | 1.8739 |  |
| Pt/OD | - | - | 1.506 |  |
| LR | - | - | 6.609 |  |
| $\theta$ | $\bigcirc$ | - | 62.79 |  |
| effectiveness |  | 0.565 | 0.572 |  |

5.4.2. DP II: Single-Phase Heat Exchanger of diameter 0.8 mm

### 5.4.2.1. Baseline heat exchanger

The second application of the methodology is also a single-phase air-to-water heat exchanger. And the bare tube heat exchanger (BTHX) with diameter of 0.8 mm is used as baseline for comparison. To make a fair comparison, this bifurcated bare tube heat exchanger's diameter is controlled to be 0.8 mm and the tube bank number is controlled to be four, which are the same as those of bare tube heat exchanger. The experimental data of BTHX can be found in the appendix and the key parameters are summarized in Table 31.

Table 31 Baseline BTHX

| Metric | Unit | BTHX |
| :---: | :---: | :---: |
| Air flow rate | $\mathrm{m}^{3} / \mathrm{s}$ | 0.05 |
| Air inlet temperature | K | 308.15 |
| Water flow rate | $\mathrm{g} / \mathrm{s}$ | 50 |
| Water inlet temperature | K | 333.15 |
| Heat load | W | 849 |
| Air pressure drop | Pa | 55.9 |
| Water pressure drop | kPa | 6.8 |
| Air heat transfer coefficient | $\mathrm{W} / \mathrm{m}^{2} \cdot \mathrm{~K}$ | 343 |
| Airside heat transfer area | $\mathrm{m}^{2}$ | 0.183 |
| Water heat transfer coefficient | $\mathrm{W} / \mathrm{m}^{2} \cdot \mathrm{~K}$ | 9196 |
| Waterside heat transfer area | $\mathrm{m}^{2}$ | 0.137 |
| Total pumping power (air+water) | W | 3.145 |
| Volume | $\mathrm{cm}^{3}$ | 109 |
| Material volume | $\mathrm{cm}^{3}$ | 12.81 |
| Length | m | 0.152 |
| Height | m | 0.1498 |
| Depth | m | 0.0048 |
| Airside frontal area | $\mathrm{m}^{2}$ | 0.0228 |
| Aspect ratio (length/height) | - | 1.0 |
| Number of banks | - | 4 |
| Tube per bank | - | 121 |

### 5.4.2.2. Optimization problem

The design optimization problem here is described as below. Two objectives are total power and heat exchanger volume. The constraints include (1) total heat exchanger capacity should be similar or larger than baseline; (2) total power is $10 \%$ less than baseline; (3) total heat exchanger volume is $10 \%$ less than baseline; (4) aspect ratio (AR) is similar to baseline; (5) outside diameter of first level tube is the same as baseline; and (6) tube bank number is the same as baseline. The last three constraints are dismissed for the real calculation because these will cause the calculation time to be much longer. The $10 \%$ better points were selected manually after the Pareto curve is produced.

$$
\begin{aligned}
& \text { Optimization II: } \\
& \quad \min \text { Power } \\
& \quad \min V_{H X} \\
& \text { s.t. } \\
& \quad \dot{Q}_{\text {baseline }} \leq \dot{Q} \leq 1.1 \cdot \dot{Q}_{\text {baseline }} \\
& 0.9 \cdot A R_{\text {baseline }} \leq A R \leq 1.1 \cdot A R_{\text {baseline }} \\
& O D=O D_{\text {baseline }} \\
& N_{\text {tubebank }}=N_{\text {tubebank,baseline }} \\
& \hline \begin{array}{l}
\text { Power } \leq 0.9 \cdot \text { Power }_{\text {baseline }} \\
V_{H X} \leq 0.9 \cdot V_{H X} \\
A_{f r} \leq A_{f r-\text { baseline }} \\
\hline
\end{array}
\end{aligned}
$$

5.4.2.3. Optimization results and discussion

Optimization results data can be found in the appendix. The Pareto front is plotted in Figure 206. Pareto front points are those have $10 \%$ lower total power and $10 \%$ smaller volume than baseline BTHX. They also have smaller frontal area than baseline. Pareto front with no constrains are plotted on the graph as well.

As shown in Figure 207, the transverse pitch reaches the minimum limit (1.5) and does not change while longitudinal pitch increases, resulting in larger airside frontal area (Figure 206), airside heat transfer area (Figure 208), larger volume and larger material volume (Figure 213). Larger airside frontal area will cause the air velocity to decrease, thus airside heat transfer coefficient decreases. Figure 210 shows bifurcation angle $\theta$ and length ratio are related to each other. This is because larger length ratio and smaller bifurcation both leads to smaller airside heat transfer coefficient, so to maintain similar heat transfer capacity, larger length ratio is always accompanied by a larger bifurcation angle. Again, larger volume designs have larger airside frontal area, thus lower frontal velocity and smaller airside pressure drop. So, the airside pumping power decreases as the design volume increases (Figure 211).

The waterside heat transfer area also increases as heat exchanger volume increases while waterside heat transfer coefficient does not change much, as shown in Figure 209. Since the airside thermal resistance is dominant thus the change in waterside thermal resistance has very limited effect on the overall heat transfer capacity.

As for power consumption, the airside fan power decreases as heat exchanger volume increases. But larger airside frontal area also means longer water flow path, so waterside pumping power increases (Figure 211). Airside total power percentage decreases as volume increases, as shown in Figure 212. Figure 213 presents the aspect ratio and material volume. The aspect ratio is controlled within the range of $0.9 \sim 1.1$ to maintain a similar value as baseline (1.0). Material volume increases as heat exchanger envelope volume increases.


Figure 206 Pareto front (DP II)


Figure 207 Pl and Pt of Pareto front points w/o constraints (DP II)


Figure 208 AHTA and AHTC of Pareto front points w/o constraints (DP II)


Figure 209 WHTA and WHTC of Pareto front points w/o constraints (DP II)


Figure 210 Bifurcation angle and length ratio of Pareto front points w/o constraints (DP II)


Figure 211 Airside and waterside pumping power of Pareto front points w/o constraints (DP II)


Figure 212 Airside power percentage of Pareto front points w/o constraints (DP II)


Figure 213 Aspect ratio and material volume of Pareto front points w/o constraints (DP II)

### 5.4.2.4. Results verification

The optimization results were verified against CFD simulation results. Three points from Pareto front were selected and CFD runs were conducted. Table 32 compares airside heat transfer coefficient (AHTC), airside pressure drop (ADP), waterside heat transfer coefficient (WHTC) and
waterside pressure drop (WDP). The maximum percentage deviation of AHTC, ADP, WHTC and ADP is $5.3 \%$, which is acceptable. Thus, the Pareto front is verified.

Table 32 Optimization results verification against CFD simulation (DP II)

| Optimization results |  |  |  | CFD results |  |  |  | Percentage deviation |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\begin{gathered} \hline \text { AHTC } \\ {\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right]} \end{gathered}$ | $\begin{aligned} & \mathrm{ADP} \\ & {[\mathrm{~Pa}]} \\ & \hline \end{aligned}$ | $\begin{gathered} \hline \text { WHTC } \\ {[\mathrm{W} / \mathrm{m} 2 \mathrm{~K}]} \end{gathered}$ | $\begin{aligned} & \text { WDP } \\ & {[\mathrm{Pa}]} \\ & \hline \end{aligned}$ | $\begin{gathered} \text { AHTC } \\ {\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right]} \end{gathered}$ | $\begin{aligned} & \mathrm{ADP} \\ & {[\mathrm{~Pa}]} \end{aligned}$ | $\begin{gathered} \text { WHTC } \\ {\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right]} \end{gathered}$ | $\begin{gathered} \hline \text { WDP } \\ {[\mathrm{Pa}]} \\ \hline \end{gathered}$ | $\begin{gathered} \text { AHTC } \\ {[\%]} \\ \hline \end{gathered}$ | $\begin{gathered} \mathrm{ADP} \\ {[\%]} \\ \hline \end{gathered}$ | $\begin{gathered} \hline \text { WHTC } \\ {[\%]} \\ \hline \end{gathered}$ | $\begin{gathered} \text { WDP } \\ {[\%]} \\ \hline \end{gathered}$ |
| 403.88 | 40.60 | 9486.7 | 464.68 | 407.80 | 41.96 | 9100.7 | 439.87 | 1.0\% | 3.3\% | -4.1\% | -5.3\% |
| 454.94 | 65.23 | 9722.4 | 444.73 | 226.93 | 6.07 | 5780.6 | 456.41 | 2.3\% | -1.8\% | -1.7\% | -0.7\% |
| 221.76 | 6.19 | 5878.3 | 459.51 | 461.64 | 67.15 | 10128.9 | 430.99 | 1.5\% | 2.9\% | 4.2\% | -3.1\% |

5.4.2.5. Selected optimal design and baseline comparison

Here we select one of the verified optimal design and compare it with baseline heat exchanger.

Table 33 summarized all parameter values and percentage differences. This optimal design has $28 \%$ lower total pumping power, $11 \%$ smaller volume and $10 \%$ smaller material volume. Discussions of each parameter are as follows.

Higher airside heat transfer coefficient: bBTHX has 12\% higher airside heat transfer coefficient than baseline. bBTHX has less tube number per bank and smaller frontal area, thus inlet air velocity is higher for bBTHX, which will lead to higher airside heat transfer coefficient. The addition of bifurcation increases flow mixing and enhance heat transfer. Branch tube with smaller diameter is also beneficial for heat transfer enhancement.

Lower airside pressure drop: bBTHX has $27 \%$ lower airside pressure drop. As discussed in Chapter 4.7, flow bypass, smaller flow rate at bare tube region and 3D flow are the main mechanisms that contributes to lower airside pressure drop. And 3D flow is the only mechanism that contributes to both heat transfer enhancing and pressure drop reduction. In current case, the Reynolds number is 113 . From Figure 178, the drag coefficient is about one for smooth sphere and is about two for smooth cylinder, meaning the drag force per length of smooth sphere is $50 \%$ lower
than smooth cylinder with the same diameter. From Figure 179, Nusselt number is about 5.5 for cylinder and 6.9 for sphere, meaning that heat transfer coefficient of sphere is $25 \%$ higher than that of cylinder of the same diameter. Another reason is the bBTHX has a larger longitudinal pitch (1.64) than baseline (1.5) and a large length ratio (7.18) which are both beneficial for lower pressure drop. This indicates that though addition of bifurcation usually increases airside pressure drop, the introduction of 3D flow will result in lower airside pressure drop than bare tube bundles with proper design.

Higher waterside heat transfer coefficient: bBTHX has 84\% higher waterside heat transfer coefficient than baseline. Besides the three reasons as discussed in Chapter 4.8, i.e. smaller OD branch tube, boundary layer redevelopment and flow separation, the most important reason is in current case, it is turbulent flow due to the existence of bifurcation. But for baseline, it is laminar flow and the heat transfer coefficient is calculated using laminar flow heat transfer coefficient correlation.

Lower/Higher waterside pressure drop: As for waterside pressure drop, note that the experimental waterside pressure drop is 6.8 kPa . However, in experiment, the pressure drop in the heat exchanger header is also measured as well as all connection fittings and tubes in between the different pressure transducer. If using laminar f factor equation ( $\mathrm{f}=64 / \mathrm{Re}$ ), then the waterside pressure drop of baseline drops to 2.4 kPa . And bBTHX has two times the waterside pressure drop which is consistent with the simulation results. The total power of baseline drops to 2.915 W if theoretical waterside pressure drop is used. However, since airside pressure drop dominates, the bBTHX still has $65 \%$ lower total pumping power than baseline.

Lower total pumping power: bBTHX has $28 \%$ lower total pumping power. This is mainly due to the reduction on airside pressure drop.

Smaller volume and material volume: bBTHX has about $10 \%$ smaller volume and material volume than baseline.

Table 33 Comparison of selected optimal design and baseline BTHX (DP II)

| Metric | Unit | BTHX (experimental data) | bBTHX (simulation data) | \% diff |
| :---: | :---: | :---: | :---: | :---: |
| Air flow rate | $\mathrm{m}^{3} / \mathrm{s}$ | 0.05 | 0.05 | - |
| Air inlet temperature | K | 308.15 | 308.15 | - |
| Water flow rate | g/s | 50 | 50 | - |
| Water inlet temperature | K | 333.15 | 333.15 | - |
| Heat load | W | 849 | 865.7 | 2\% |
| Air pressure drop | Pa | 55.9 | 40.6 | -27\% |
| Water pressure drop | kPa | $\begin{gathered} 6.8 \\ (2.4 \mathrm{kPa}, \\ \text { laminar) } \\ \hline \end{gathered}$ | 4.7 | -45\% |
| Air heat transfer coefficient | $\mathrm{W} / \mathrm{m}^{2} \cdot \mathrm{~K}$ | 359 | 403.4 | 12\% |
| Airside heat transfer area | $\mathrm{m}^{2}$ | 0.183 | 0.16 | -13\% |
| Water heat transfer coefficient | $\mathrm{W} / \mathrm{m}^{2} \cdot \mathrm{~K}$ | 4825 | 8892 | 84\% |
| Waterside heat transfer area | $\mathrm{m}^{2}$ | 0.137 | 0.128 | -7\% |
| Total pumping power (air+water) | W | 3.15 | 2.27 | -28\% |
| Volume | $\mathrm{cm}^{3}$ | 109 | 96.8 | -11\% |
| Material volume | $\mathrm{cm}^{3}$ | 12.81 | 11.54 | -10\% |
| Length | m | 0.152 | 0.141 | -7\% |
| Height | m | 0.1498 | 0.143 | -5\% |
| Depth | m | 0.0048 | 0.0048 | - |
| Airside frontal area | $\mathrm{m}^{2}$ | 0.0228 | 0.0202 | -11\% |
| Number of banks | - | 4 | 4 | - |
| Tube per bank | - | 121 | 109 | -10\% |
| $\mathrm{Pl} / \mathrm{OD}$ | - | 1.5 | 1.64 | 9.3\% |
| Pt/OD | - | 1.49 | 1.5 | 0.7\% |
| LR | - | - | 7.18 | - |
| $\theta$ | 。 | - | 46.92 | - |
| effectiveness |  | 0.574 | 0.585 |  |

## Chapter 6: bBTHX Application

In this chapter, two applications of the new heat exchanger are discussed. The first one is car radiator, and the second is indoor coil for hybrid variable refrigerant flow (VRF) system. The advantages of utilizing it as car radiator and indoor coil for hybrid variable VRF system include smaller volume, less material volume and inner volume and lower pumping power.

### 6.1. Car Radiator

### 6.1.1. Baseline heat exchanger

Current design can be applied as car radiator due to the advantages of smaller volume, less material volume and, inner volume and lower pumping power when delivering the same capacity. A louvered fin radiator is used as baseline for comparison. This car radiator has been tested under dry condition using air and water and here the design condition is one of the testing condition. The experimental data can be found in appendix and the experimental data has been compared to predictions using correlations from literature (Wang et al., 1999), as shown in Figure 215. Other parameters are summarized in Table 34.


Figure 214 Louvered fin car radiator


Figure 215 Louvered fin car radiator experimental data and correlation comparison
Table 34 Baseline car radiator

| Metric | Unit | Value |
| :---: | :---: | :---: |
| Air flow rate | $\mathrm{m}^{3} / \mathrm{s}$ | 0.772 |
| Air inlet temperature | K | 308.15 |
| Water flow rate | $\mathrm{g} / \mathrm{s}$ | 279.2 |
| Water inlet temperature | K | 333.15 |
| Heat load | W | 10477.2 |
| Air pressure drop | Pa | 111 |
| Water pressure drop | kPa | 10.6 |
| Air heat transfer coefficient | $\mathrm{W} / \mathrm{m}^{2} \cdot \mathrm{~K}$ | 148 |
| Airside heat transfer area | $\mathrm{m}^{2}$ | 9.3469 |
| Water heat transfer coefficient | $\mathrm{W} / \mathrm{m}^{2} \cdot \mathrm{~K}$ | 1666 |
| Waterside heat transfer area | $\mathrm{m}^{2}$ | 1.1463 |


| Total pumping power (air+water) | W | 88.74 |
| :---: | :---: | :---: |
| Volume | $\mathrm{cm}^{3}$ | 6716.25 |
| Material volume | $\mathrm{cm}^{3}$ | 1220 |
| Internal volume | $\mathrm{cm}^{3}$ | 753 |
| Length | m | 0.457 |
| Height | m | 0.4211 |
| Depth | m | 0.0349 |
| Airside frontal area | $\mathrm{m}^{2}$ | 0.1925 |
| Aspect ratio (length/height or |  |  |
| height/length) |  |  |

### 6.1.2. Design optimization

Two objectives are minimizing total power and heat exchanger volume. The constraints include (1) total heat exchanger capacity should be similar or larger than baseline; (2) total power is $30 \%$ less than baseline; (3) total heat exchanger volume is $30 \%$ less than baseline; (4) aspect ratio (AR) is similar to baseline and (5) frontal area is similar to baseline.

> Optimization:
> $\quad \min$ Power
> $\quad \min V_{H X}$
> s.t.
> $\quad 0.9 \cdot \dot{Q}_{\text {baseline }} \leq \dot{Q} \leq 1.1 \cdot \dot{Q}_{\text {baseline }}$
> $\quad$ Power $\leq 0.7 \cdot$ Power $_{\text {baseline }}$
> $V_{H X} \leq 0.7 \cdot V_{H X}$ _baseline
> $0.9 \cdot A R_{\text {baseline }} \leq A R \leq 1.1 \cdot A R_{\text {baseline }}$
> $0.9 \cdot A_{f r, b a s e l i n e} \leq A_{f r} \leq 1.1 \cdot A_{f r, \text { baseline }}$
6.1.3. Optimization results and discussion

The optimization results in shown in Figure 216. One of the designs was selected to be the final design and the comparison results are summarized in Table 35. The bBTHX car radiator has $30 \%$ less total pumping power, $68.1 \%$ less heat exchanger envelope volume and $87.1 \%$ less
material volume, which meet the design goal. Another advantage of bBTHX is that the water mass of the heat exchanger is $66.7 \%$ less than baseline because it has $66.7 \%$ less internal volume. For a car radiator, reduction in total mass of heat exchanger can contribute to car overall efficiency. The CFD verification results is shown in Table 36.


Figure 216 Optimization results of car radiator design

Table 35 Comparison of bBTHX car radiator design and baseline

| Metric | Unit | Baseline <br> (experimental <br> data) | bBTHX <br> (simulation <br> data) | \%diff |
| :---: | :---: | :---: | :---: | :---: |
| Air flow rate | $\mathrm{m}^{3} / \mathrm{s}$ | 0.772 | 0.772 | - |
| Air inlet temperature | K | 308.15 | 308.15 | - |
| Water flow rate | $\mathrm{g} / \mathrm{s}$ | 279.2 | 279.2 | - |
| Water inlet temperature | K | 333.15 | 333.15 | - |
| Heat load | W | 10477.2 | 10428.7 | $-0.5 \%$ |
| Air pressure drop | Pa | 111 | 71.1 | $\mathbf{- 3 5 . 9 \%}$ |
| Water pressure drop | kPa | 10.6 | 15.4 | $\mathbf{4 5 . 3 \%}$ |
| Air heat transfer coefficient | $\mathrm{W} / \mathrm{m}^{2} \cdot \mathrm{~K}$ | 148 | 388.5 | $\mathbf{1 6 2 . 5 \%}$ |
| Airside heat transfer area | $\mathrm{m}^{2}$ | 9.3469 | 2.0819 | $\mathbf{- 7 7 . 7 \%}$ |
| Water heat transfer coefficient | $\mathrm{W} / \mathrm{m}^{2} \cdot \mathrm{~K}$ | 1666 | 8816 | $\mathbf{4 2 9 . 2 \%}$ |
| Waterside heat transfer area | $\mathrm{m}^{2}$ | 1.1463 | 1.5612 | $\mathbf{3 6 . 2 \%}$ |
| Total pumping power (air+water) | W | 88.74 | 62.13 | $\mathbf{- 3 0 . 0 \%}$ |
| Volume | $\mathrm{cm}^{3}$ | 6716.25 | 2140.6 | $\mathbf{- 6 8 . 1 \%}$ |
| Material volume | $\mathrm{cm}^{3}$ | 1220 | 157.58 | $\mathbf{- 8 7 . 1 \%}$ |
| Internal volume | $\mathrm{cm}^{3}$ | 753 | 250.56 | $\mathbf{- 6 6 . 7 \%}$ |
| Length | m | 0.457 | 0.4412 | $-3.5 \%$ |
| Height | m | 0.4211 | 0.4457 | $5.8 \%$ |
| Depth | m | 0.0349 | 0.012 | $-65.6 \%$ |
| Airside frontal area | $\mathrm{m}^{2}$ | 0.1925 | 0.196 | $1.8 \%$ |
| Aspect ratio (length/height or | - | 1.09 | 1.01 | $-7.3 \%$ |
| height/length) | - | 2 | 9 |  |
| Number of banks | - | - | 42 | 186 |
| Tube per bank | - | 15.3 | - |  |
| Fin per Inch (FPI) |  |  |  |  |

Table 36 Design point verification against CFD simulation for car radiator design

| Optimization results |  |  |  | CFD results |  |  |  | Percentage deviation |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\begin{gathered} \hline \text { AHTC } \\ {\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right]} \end{gathered}$ | $\begin{gathered} \text { ADP } \\ {[\mathrm{Pa}]} \\ \hline \end{gathered}$ | $\begin{gathered} \text { WHTC } \\ {[\mathrm{W} / \mathrm{m} 2 \mathrm{~K}]} \\ \hline \end{gathered}$ | $\begin{gathered} \hline \text { WDP } \\ {[\mathrm{Pa}]} \\ \hline \end{gathered}$ | $\begin{gathered} \text { AHTC } \\ {\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right]} \\ \hline \end{gathered}$ | $\begin{gathered} \mathrm{ADP} \\ {[\mathrm{~Pa}]} \\ \hline \end{gathered}$ | $\begin{gathered} \text { WHTC } \\ {\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right]} \\ \hline \end{gathered}$ | $\begin{gathered} \hline \mathrm{WDP} \\ {[\mathrm{~Pa}]} \\ \hline \end{gathered}$ | $\begin{gathered} \hline \text { AHTC } \\ {[\%]} \\ \hline \end{gathered}$ | $\begin{gathered} \text { ADP } \\ {[\%]} \\ \hline \end{gathered}$ | $\begin{gathered} \text { WHTC } \\ {[\%]} \\ \hline \end{gathered}$ | $\begin{gathered} \hline \text { WDP } \\ {[\%]} \\ \hline \end{gathered}$ |
| 388.50 | 71.11 | 8816.0 | 901.06 | 370.53 | 73.09 | 8684.3 | 887.63 | -4.63\% | 2.78\% | -1.49\% | -1.49\% |

### 6.2. Indoor Coil of Hybrid Variable Refrigerant Flow (HVRF) System

Besides being applied as car radiator, the bBTHX can also be applied as indoor coil in hybrid VRF (HVRF) system. VRF system is multiple-unit split-type system. Compared with traditional central air conditioning unit, VRF system can achieve higher efficiency by modulating
the flow of refrigerant according to individual area cooling and heating load. Heat recovery VRF system can also save compressor work by internal heat recovery.

Traditional VRF system is shown in Figure 217 and hybrid VRF system is shown in Figure 218. Instead of using refrigerant inside home, hybrid VRF system utilizes water (or other coolant) coils as indoor unit. Compared with traditional VRF system, there is less concern of flammability of refrigerant, giving a wider selection of refrigerants. Fin-and-tube heat exchangers are usually used as indoor coils to exchange heat between coolant and indoor air. Current bifurcated bare tube heat exchanger design can be produced using 3D printing technology, thus there is a possibility of applying bBTHX as indoor unit for hybrid VRF system. Offsite and onsite 3D printing manufacturing could be both possible in the future.


Figure 217 Traditional VRF System (with Heat Recovery)


Figure 218 Hybrid VRF System

In this chapter, system performance of HVRF is numerically simulated using VapCyc ${ }^{\circledR}$ (Winkler et al., 2008) software. For comparison, traditional VRF system is chosen as baseline. The refrigerant investigated are R410A, R290 (Propane) and R600a (Isobutane). R410A is a widely used refrigerant in market while R290 and R600a are flammable refrigerants which are not applicable in traditional refrigerant system.
6.2.1. Traditional R410A VRF system design (baseline)

Traditional VRF system is evaluated using R410A as refrigerant. The piping design restrictions shown in Figure 219 are used. It is from Mitsubishi HVRF system data book (Mitsubishi, 2012). The schematic of traditional VRF system is shown in Figure 220. Due to AHRI VRF testing standard (AHRI, 2010), the minimum indoor unit quantity is two, thus in current study, two indoor units are designed. Length is 110 m ( 360 feet) for refrigerant pipe A and 60 m (196 feet) for pipe B. Diameter of pipe A is 9.5 mm ( 0.375 inches) and diameter of pipe $B$ is 16 mm (0.625 inches). Both indoor and outdoor units are traditional fin-and-tube heat exchangers. The simulation condition is dry test condition for air condition mode from AHRI standard (AHRI, 2008). Ambient dry/wet bulb temperature is $35 / 23.9^{\circ} \mathrm{C}$ and indoor air dry/wet bulb temperature is $26.7 / 19.4^{\circ} \mathrm{C}$. The vapor compression cycle is simulated using steady state vapor compression simulation tool VapCyc ${ }^{\circledR}$ (Winkler et al., 2008).


| Piping length limitation |  |  | (m [ft.]) |
| :--- | :--- | :--- | :--- |
| Item | Piping in the figure | Max. length | Max. equivalent |
| Distance between OU and HBC | A | $110\left[360^{\prime}\right]$ | $130\left[426^{\prime}\right]$ |
| Farthest IU from HBC controller | $\mathrm{B}+\mathrm{d}$ | $60\left[196^{\prime}\right]$ | $70\left[229^{\prime}\right]$ |
| Height between OU and IU (OU above IU) | H | $50\left[164^{\prime}\right]$ | - |
| Height between OU and IU (OU under IU) | $\mathrm{H}^{\prime}$ | $40\left[131^{\prime}\right]$ | - |
| Height between IU and HBC | h1 | $15\left[49^{\prime}\right]$ | - |
| Height between IU and IU | h2 | $15\left[49^{\prime}\right]$ | - |

OU : Outdoor Unit ; IU : Indoor Unit ; HBC : HBC controller

Figure 219 VRF piping design restrictions (from Mitsubishi data book)


Figure 220 Schematic of traditional VRF system

Convergence criteria for high pressure side is subcooling temperature equals 5.5 K and convergence criteria for low pressure side is suction super heat temperature equals 5 K . Power input to the fan motors are 36 W and 18 W for $1 \mathrm{~m}^{3} / \mathrm{s}$ airflow rate for evaporator and condenser, respectively. The air flow rate for evaporator is $0.283 \mathrm{~m}^{3} / \mathrm{s}$ each, and the air flow rate for condenser is $0.693 \mathrm{~m}^{3} / \mathrm{s}$ total. Figure 221 shows the baseline traditional R410A VRF system schematic in

VapCyc ${ }^{\circledR}$ and the corresponding properties of each state point are summarized in Table 37. The system evaluation results are shown in Table 38. COP is defined as the ratio of capacity over total power.


Figure 221 Baseline traditional R410A VRF system schematic
Table 37 Simulation results for baseline R410A VRF system

| Junction | T <br> $[\mathrm{K}]$ | P <br> $[\mathrm{kPa}]$ | x | H <br> $[\mathrm{kJ} / \mathrm{kg}]$ | Delta $\mathrm{T}_{\text {sat }}$ <br> $[\mathrm{K}]$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 279.9 | 843.8 | 1.1 | 428.6 | 5.0 |
| 2 | 356.9 | 2775.3 | 1.1 | 480.5 | 38.0 |
| 3 | 311.0 | 2630.8 | -1.0 | 263.3 | -5.5 |
| 4 | 279.3 | 969.7 | 0.2 | 263.3 | 0.0 |
| 5 | 279.3 | 969.7 | 0.2 | 263.3 | 0.0 |
| 6 | 282.5 | 945.3 | 1.1 | 428.6 | 4.0 |
| 7 | 282.5 | 945.3 | 1.1 | 428.6 | 4.0 |
| 8 | 311.1 | 2700.3 | -1.0 | 263.3 | -6.6 |
| 9 | 311.1 | 2700.3 | -1.0 | 263.3 | -6.6 |
| 10 | 279.3 | 969.7 | 0.2 | 263.3 | 0.0 |
| 11 | 279.3 | 969.7 | 0.2 | 263.3 | 0.0 |

Table 38 Baseline R410A VRF system performance
$\left.\begin{array}{|c|c|c|c|c|c|c|}\hline \text { COP } & \text { Capacity } & \text { P_comp } & \text { P_eva,fan } & \text { P_con,fan } & \text { P_total } & \begin{array}{c}\text { Refrigerant } \\ \text { charge } \\ {[-]}\end{array} \\ {[\mathrm{W}]} & {[\mathrm{W}]} & {[\mathrm{W}]} & {[\mathrm{W}]} & {[\mathrm{W}]} & {[\mathrm{kg}]}\end{array}\right]$
6.2.2. Hybrid R410A VRF system design

### 6.2.2.1. Refrigerant loop design

The piping design restriction of HVRF system is also from Figure 219. The schematic of traditional VRF system is shown in Figure 222. Two indoor units are designed with the height of 15 m (49 feet). Pipe A is refrigerant loop and the length is 110 m ( 360 feet). Pipe B is water pipe and the length is 60 m (196 feet) and the heights of two indoor units are both 15 m ( 49 feet). The diameter of pipe B is 22 mm ( 0.875 inches) (Mitsubishi, 2012). Outdoor unit is the same traditional fin-and-tube heat exchanger as what was used for baseline. Indoor unit is bBTHX. The design of bBTHX is discussed later. Outdoor unit and indoor unit exchange heat through a plate heat exchanger. The simulation condition is dry test condition for air condition mode from AHRI standard (AHRI, 2008). Ambient dry/wet bulb temperature is $35 / 23.9^{\circ} \mathrm{C}$ and indoor air dry/wet bulb temperature is $26.7 / 19.4^{\circ} \mathrm{C}$. The vapor compression cycle is simulated using steady state vapor compression simulation tool VapCyc ${ }^{\circledR}$ (Winkler et al., 2008).


Figure 222 Schematic of HVRF system
Convergence criteria for high pressure side is subcooling temperature equals 5.5 K and convergence criteria for low pressure side is suction super heat temperature equals 5 K . Power input to the fan motor is 18 W for $1 \mathrm{~m}^{3} / \mathrm{s}$ airflow rate for condenser. Figure 221 shows the R410A

HVRF system schematic in VapCyc ${ }^{\circledR}$ and the corresponding properties of each state point are summarized in Table 39 and Table 40.


Figure 223 R410A HVRF system schematic
Table 39 Simulation results for R410A HVRF system

| Junction | $\mathrm{T}[\mathrm{K}]$ | $\mathrm{P}[\mathrm{kPa}]$ | x | $\mathrm{h}[\mathrm{kJ} / \mathrm{kg}]$ | Delta $\mathrm{T}_{\text {sat }}$ <br> $[\mathrm{K}]$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 278.2 | 798.3 | 1.1 | 428.1 | 5.0 |
| 2 | 358.6 | 2775.0 | 1.1 | 482.5 | 39.7 |
| 3 | 311.4 | 2652.1 | -1.0 | 263.9 | -5.5 |
| 4 | 275.8 | 870.8 | 0.3 | 263.9 | 0.0 |
| 5 | 275.8 | 870.8 | 0.3 | 263.9 | 0.0 |
| 6 | 280.1 | 870.8 | 1.1 | 428.1 | 4.2 |
| 7 | 280.1 | 870.8 | 1.1 | 428.1 | 4.2 |
| 8 | 311.4 | 2698.3 | -1.0 | 263.9 | -6.2 |
| 9 | 311.4 | 2698.3 | -1.0 | 263.9 | -6.2 |

Table 40 R410A HVRF system performance

| Capacity [W] | $\begin{gathered} \text { P_comp }^{[W]} \end{gathered}$ | $\mathbf{P}_{-} \text {con,fan }$ <br> [W] | Refrigerant charge [kg] |
| :---: | :---: | :---: | :---: |
| 6862.17 | 2277.11 | 12.48 | 10.51 |

### 6.2.2.2. Indoor unit coil design

Indoor unit of HVRF system is a fan coil unit which consists of air-to-water heat exchanger, fan and motor, drain pan and drainage pump. Different indoor units can be used for HVRF system,
such as ducted or ceiling fan coil unit, floor mounted fan coil unit and cassettes fan coil unit. Among these three types, ducted or ceiling fan coil unit is the most compact design thus was selected as the indoor unit type for current study. The picture of such kind of indoor coil unit is shown in Figure 224 and the flow direction is shown in Figure 225. The model number is PEFY-WP32VMA-E. The specifications are summarized in Table 41 and Figure 226.


Figure 224 Indoor coil unit (Mitsubishi, 2012)
(1) In case of rear inlet
(2) In case of bottom inlet


Figure 225 Airflow direction of indoor coil unit (Mitsubishi, 2012)

Table 41 Specifications of indoor unit (Mitsubishi, 2012)

| Model |  | PEFY-WP32VMA-E |
| :---: | :---: | :---: |
| Power source | - | 1-phase 220-230-240V $50 / 60 \mathrm{~Hz}$ |
| Capacity | kW | 3.6 |
| External finish | - | Galvanized steel plate |
| External dimension <br> $\mathrm{H} \times \mathrm{W} \times \mathrm{D}$ | mm | $250 \times 900 \times 732$ |
| Net Weight | kg | 26 |
| Heat exchanger | - | Aluminum cross fin and copper <br> tube |
| Water volume | L | 1.0 |
| Fan Air flow rate | $\mathrm{m}^{3} / \mathrm{s}$ | $0.20-0.242-0.283$ |
| Motor output | W | 85 |



| Model | A | B | C | D | E | F | G | (1)Water pipe <br> (From HBC unit) | (2)Water pipe <br> (To HBC unit) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| PEFY-WP20VMA-E | 700 | 754 | 800 | 660 | 7 | 600 | 658 |  |  |
| PEFY-WP25, 32VMA-E | 900 | 954 | 1000 | 860 | 9 | 800 | 858 |  |  |
| RE3/4 screw |  |  |  |  |  |  |  |  |  |

Figure 226 Dimensions of indoor coil unit
(Mitsubishi, 2012)

The fin-and-tube heat exchanger is placed diagonally inside the indoor coil unit. The frontal area $(\mathrm{HxW})$ limit for the heat exchanger is approximately $250 \times 900 \mathrm{~mm}$. Here we design two heat exchangers, one is a 5 mm slit fin-and-tube heat exchanger with enhanced micro fin tube on waterside and the other is bBTHX. Here are the assumptions and design conditions:

- Inlet air and water temperatures are 26.7 and $7^{\circ} \mathrm{C}$.
- Inlet air flow rate is $0.283 \mathrm{~m}^{3} / \mathrm{s}$ and inlet water flow rate is $0.163 \mathrm{~kg} / \mathrm{s}$.
- Design capacity is 3413 W .
- Dimension limitation is $250 \times 900 \times 100 \mathrm{~mm}$.


## Indoor unit A (Baseline): $\mathbf{5} \mathbf{~ m m}$ slit fin-and-tube heat exchanger

This heat exchanger is designed based the 5 mm slit fin-and-tube heat exchanger that has been tested and the correlations used are experimentally validated. Data and validation can be found in appendix. The tube internal surface is micro fin, but the specification data cannot be shown due to confidentiality. The heat exchanger specifications are shown in Table 42. Note fan power is the product of ADP and air volume flow rate over efficiency.

Table 42 Design of indoor unit A-5 mm slit fin-and-tube HX (simulation data)

| Tube per bank | - | 42 | Capacity | W | 3464 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Tube bank \# | - | 1 | T_water_out | K | 285.2 |
| Circuit \# | - | 7 | T_air_out | K | 289.3 |
| Tube length | m | 0.25 | ADP | Pa | 11.79 |
| Tube OD | mm | 5 | WDP | kPa | 25.75 |
| Tube ID | mm | 4.6 | AHTA | $\mathrm{m}^{2}$ | 4.14 |
| Tube spacing | mm | 21 | WHTA | $\mathrm{m}^{2}$ | 0.15 |
| FPI | - | 22 | fin effectiveness |  | 0.796 |
| H $\times \mathrm{W} \times \mathrm{D}$ | mm | $250 \times 882 \times 11.4$ | AHTC | $\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}$ | 121 |
| A_fr | $\mathrm{m}^{2}$ | 0.221 | WHTC | $\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}$ | 10521 |
| Water volume | L | 0.174 | Tube material volume | $\mathrm{cm}^{3}$ | 31 |
| Volume | $\mathrm{cm}^{3}$ | 2513.7 | Total material volume | $\mathrm{cm}^{3}$ | 249 |
| Fin material <br> volume | $\mathrm{cm}^{3}$ | 218 | Total power $(100 \%$ <br> efficiency $)$ | W | 7.53 |
|  |  |  | Fan efficiency | - | 0.6 |
|  |  | Fan Power | W | 5.5 |  |

## Indoor unit B: bBTHX

The second indoor unit is bBTHX. An optimization problem is defined to find the proper design.

## bBTHX Optimization:

The design optimization problem here is described as below. Two objectives are total power and heat exchanger volume. The constraints include (1) total heat exchanger capacity should be similar or larger than baseline; (2) total power is $30 \%$ less than baseline; (3) total heat exchanger volume is $30 \%$ less than baseline; (4) aspect ratio (AR) is similar to baseline; and (5) frontal area is similar to baseline. Note here the power means the sum of product of pressure drop and volume flow rate of water and air and the efficiency is assumed to be one.

$$
\begin{aligned}
& \text { Optimization: } \\
& \quad \min \text { Power } \\
& \quad \min V_{H X} \\
& \text { s.t. } \\
& \quad 0.9 \cdot \dot{Q}_{\text {baseline }} \leq \dot{Q} \leq 1.1 \cdot \dot{Q}_{\text {baseline }} \\
& \quad \text { Power } \leq 0.7 \cdot \text { Power }_{\text {baseline }} \\
& V_{H X} \leq 0.7 \cdot V_{H X} \text { _baseline } \\
& 0.9 \cdot A R_{\text {baseline }} \leq A R \leq 1.1 \cdot A R_{\text {baseline }} \\
& 0.9 \cdot A_{f r, \text { baseline }} \leq A_{f r} \leq 1.1 \cdot A_{f r, \text { baseline }}
\end{aligned}
$$

## Optimization results and discussion:

The optimization results are shown in Figure 227. One of the designs was selected to be the final design and the results are summarized in Table 43. Comparing the two distinctive designs, the advantages of applying bBTHX as an indoor coil for VRF system are clear. It has $75.4 \%$ less total material volume and $65.1 \%$ less envelope volume than baseline 5 mm fin-and-tube heat exchanger. bBTHX also has $37.9 \%$ less water volume, meaning that the fan coil unit has less water
mass during operation. The waterside heat transfer coefficient of bBTHX is $17.2 \%$ lower than that of 5 mm coil but it has much larger waterside heat transfer area and $62.7 \%$ waterside pressure drop. This means comparing to the method of using micro fins to enhance waterside heat transfer, using bifurcation would result in much lower pressure drop and slightly lower heat transfer. The CFD verification of the design point is summarized in Table 44.


Figure 227 Optimization results of bBTHX indoor unit coil

Table 43 Design of indoor unit B - bBTHX (simulation data)

| Tube per <br> bank | - | 548 | Capacity | W | 3449 | $-0.4 \%$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Tube bank \# | - | 5 | T_water_out $^{2}$ | K | 285.2 | $0.0 \%$ |
| Tube length | m | 0.25 | T_air_out | K | 289.6 | $0.1 \%$ |
| Tube OD | mm | 0.56 | ADP | Pa | 4.62 | $\mathbf{- 6 0 . 8 \%}$ |
| Tube ID | mm | 0.50 | WDP | kPa | 9.6 | $\mathbf{- 6 2 . 7 \%}$ |
| $\mathrm{Pl} / \mathrm{OD}$ | - | 2.95 | AHTA | $\mathrm{m}^{2}$ | 1.22 | $\mathbf{- 7 0 . 5 \%}$ |
| Pt/OD | - | 1.51 | WHTA | $\mathrm{m}^{2}$ | 0.98 | $\mathbf{5 5 3 . 3 \%}$ |
| LR |  | 5.48 | AHTC | $\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}$ | 262.6 | $\mathbf{1 1 7 . 0 \%}$ |
| $\theta$ |  | 19.7 | WHTC | $\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}$ | 8714.6 | $\mathbf{- 1 7 . 2 \%}$ |
| $\mathrm{H} \times \mathrm{W} \times \mathrm{D}$ | mm | $250 \times 900 \mathrm{x}$ <br> 3.9 | Total material <br> volume | $\mathrm{cm}^{3}$ | 61.2 | $\mathbf{- 7 5 . 4 \%}$ |
| A_fr | $\mathrm{m}^{2}$ | 0.225 | Water volume | L | 0.108 | $\mathbf{- 3 7 . 9 \%}$ |
|  |  |  | Volume | $\mathrm{cm}^{3}$ | 877.5 | $\mathbf{- 6 5 . 1 \%}$ |
|  |  |  | Total power $(100 \%$ <br> efficiency | W | 2.66 | $\mathbf{- 6 4 . 7 \%}$ |
|  |  | Fan efficiency | - | 0.6 |  |  |
|  |  | Fan Power | W | 2.18 | $\mathbf{- 6 0 . 4 \%}$ |  |

Table 44 Design point verification against CFD simulation for HVRF indoor coil

| Optimization results |  |  |  | CFD results |  |  |  | Percentage deviation |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\begin{gathered} \text { AHTC } \\ {\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right]} \end{gathered}$ | $\begin{aligned} & \mathrm{ADP} \\ & {[\mathrm{~Pa}]} \\ & \hline \end{aligned}$ | $\begin{gathered} \text { WHTC } \\ {[\mathrm{W} / \mathrm{m} 2 \mathrm{~K}]} \end{gathered}$ | $\begin{aligned} & \hline \text { WDP } \\ & {[\mathrm{Pa}]} \\ & \hline \end{aligned}$ | $\begin{gathered} \text { AHTC } \\ {\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right]} \end{gathered}$ | $\begin{gathered} \mathrm{ADP} \\ {[\mathrm{~Pa}]} \\ \hline \end{gathered}$ | $\begin{gathered} \text { WHTC } \\ {\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right]} \end{gathered}$ | $\begin{gathered} \hline \mathrm{WDP} \\ {[\mathrm{~Pa}]} \\ \hline \end{gathered}$ | $\begin{gathered} \text { AHTC } \\ {[\%]} \\ \hline \end{gathered}$ | $\begin{aligned} & \mathrm{ADP} \\ & {[\%]} \\ & \hline \end{aligned}$ | $\begin{gathered} \text { WHTC } \\ {[\%]} \\ \hline \end{gathered}$ | $\begin{gathered} \text { WDP } \\ {[\%]} \\ \hline \end{gathered}$ |
| 262.55 | 4.62 | 8714.6 | 1180.08 | 270.31 | 4.75 | 8284.8 | 1121.63 | 2.95\% | 2.86\% | -4.9\% | -5.0\% |

### 6.2.2.3. Water loop pressure drop calculation

The water loop friction needs to be evaluated to get the total power consumption for HVRF system. Total water loop pressure drop includes (1) pressure drop of plate heat exchanger, (2) pressure drop of water loop piping and (3) pressure drop of indoor unit. Pressure drop of plate heat exchanger is calculated using PHEsim ${ }^{\text {TM }}$, a steady state plate heat exchanger simulation tool. The results are summarized in Table 45

Table 45 Pressure drop of plate heat exchanger (waterside)

| T_water_in | T_water_out | Mass flow rate | Pressure drop |
| :---: | :---: | :---: | :---: |
| $[\mathrm{K}]$ | $[\mathrm{K}]$ | $[\mathrm{kg} / \mathrm{s}]$ | $[\mathrm{Pa}]$ |
| 285.15 | 280.2 | 0.326 | 1419.07 |

Pressure drop of water loop piping is calculated using Equation (37), which is developed by Petukhov (1970). There are two identical indoor unit with inner diameter of 20 mm , length of 392 feet and height of 49 feet. The tube is assumed to be smooth PVC pipe. The pressure drop of water loop is shown in Table 46.

Table 46 Pressure drop of water loop piping

| ID <br> $[\mathrm{mm}]$ | Length <br> $[\mathrm{m}]$ | DP_fric <br> $[\mathrm{kPa}]$ | DP_height <br> $[\mathrm{kPa}]$ | DP_total <br> $[\mathrm{kPa}]$ |
| :---: | :---: | :---: | :---: | :---: |
| 20 | 119.5 | 23.95 | 149.4 | 173.7 |

Pressure drop of indoor unit is calculated previously. Pressure drop is 25.75 kPa for indoor coil A and 9.6 kPa for indoor coil B. Pumping power of water loop with two different indoor units are summarized in Note that pumping power is defined as the product of water volume flow rate and waterside pressure drop (WDP) over efficiency.

Table 47 Pumping power of water loop
$\left.\begin{array}{|c|c|c|c|c|c|c|c|}\hline \begin{array}{c}\text { Indoor } \\ \text { unit }\end{array} & \text { Amount } & \begin{array}{c}\text { Water } \\ \text { flow } \\ \text { rate } \\ {[\mathrm{kg} / \mathrm{s}]}\end{array} & \text { DP_PHX } & \text { DP_Piping } & \begin{array}{c}\text { DP_Indoor } \\ \text { unit }\end{array} & \text { Efficiency } & \begin{array}{c}\text { Pumping } \\ \text { power } \\ {[\mathrm{kPa}]}\end{array} \\ \hline \text { [kPa] }\end{array}\right]$
6.2.2.4. Summary

The R410A HVRF system's performances with different indoor coil units are summarized in Table 48. For each system, there are two indoor units.

Table 48 R410A HVRF system performance with two different indoor units

| Indoor <br> unit | COP | Capacity | P_comp | P_con,fan $^{\text {P }}$ | P_waterloop | P_eva,fan | P_total | Refrigerant <br> charge |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | $[-]$ | $[\mathrm{W}]$ | $[\mathrm{W}]$ | $[\mathrm{W}]$ | $[\mathrm{W}]$ | $[\mathrm{W}]$ | $[\mathrm{W}]$ | $[\mathrm{kg}]$ |
| A | 2.90 | 6862.17 | 2277.11 | 12.48 | 65.48 | 11 | 2366.07 | 10.51 |
| B | 2.91 | 6862.17 | 2277.11 | 12.48 | 60.22 | 4.36 | 2354.17 | 10.51 |

6.2.3. Hybrid R290 and R600a VRF system

The biggest advantage of Hybrid VRF system is the possibility of utilizing flammable refrigerants. Two HVRF system are evaluated using R290 and R600a. The assumptions are the same as R410A HVRF system. The simulation results for R290 are summarized in Table 49 and Table 50.


Figure 228 R290 HVRF system schematic

Table 49 Simulation results for R290 HVRF system

| Junction | $\mathrm{T}[\mathrm{K}]$ | $\mathrm{P}[\mathrm{kPa}]$ | x | $\mathrm{h}[\mathrm{kJ} / \mathrm{kg}]$ | Delta $\mathrm{T}_{\text {sat }}$ <br> $[\mathrm{K}]$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 279.6 | 495.9 | 1.1 | 585.2 | 5.0 |
| 2 | 342.9 | 1601.0 | 1.1 | 671.9 | 22.8 |
| 3 | 311.8 | 1503.7 | -1.0 | 303.0 | -5.5 |
| 4 | 276.1 | 518.7 | 0.3 | 303.0 | 0.0 |
| 5 | 276.1 | 518.7 | 0.3 | 303.0 | 0.0 |
| 6 | 311.8 | 1528.8 | -1.0 | 303.0 | -6.2 |
| 7 | 311.8 | 1528.8 | -1.0 | 303.0 | -6.2 |
| 8 | 280.1 | 518.7 | 1.1 | 585.2 | 4.0 |
| 9 | 280.1 | 518.7 | 1.1 | 585.2 | 4.0 |

Table 50 R290 HVRF system performance

| Capacity <br> [W] | P_comp <br> [W] | P_con,fan <br> $[\mathrm{W}]$ | Refrigerant <br> charge <br> $[\mathrm{kg}]$ |
| :---: | :---: | :---: | :---: |
| 6869.13 | 2111.14 | 12.48 | 5.66 |

The simulation results for R600a are summarized in Figure 229, Table 51 and Table 52.


Figure 229 R600a HVRF system schematic
Table 51 Simulation results for R600a HVRF system

| Junction | $\mathrm{T}[\mathrm{K}]$ | $\mathrm{P}[\mathrm{kPa}]$ | x | $\mathrm{h}[\mathrm{kJ} / \mathrm{kg}]$ | Delta $\mathrm{T}_{\text {sat }}$ <br> $[\mathrm{K}]$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 280.1 | 167.9 | 1.1 | 565.2 | 5.0 |
| 2 | 337.8 | 696.2 | 1.1 | 650.0 | 13.9 |
| 3 | 309.6 | 559.3 | -1.0 | 287.4 | -5.5 |
| 4 | 275.1 | 167.9 | 0.2 | 287.4 | 0.0 |
| 5 | 275.1 | 167.9 | 0.2 | 287.4 | 0.0 |
| 6 | 309.6 | 559.3 | -1.0 | 287.4 | -5.5 |
| 7 | 309.6 | 559.3 | -1.0 | 287.4 | -5.5 |
| 8 | 280.1 | 167.9 | 1.1 | 565.2 | 5.0 |
| 9 | 280.1 | 167.9 | 1.1 | 565.2 | 5.0 |

Table 52 R600a HVRF system performance

| Capacity <br> [W] | $\mathbf{P}_{-}$comp <br> $[\mathrm{W}]$ | $\mathbf{P}_{-}$con,fan <br> $[\mathrm{W}]$ | Refrigerant <br> charge <br> $[\mathrm{kg}]$ |
| :---: | :---: | :---: | :---: |
| 6868.58 | 2096.61 | 12.48 | 5.86 |

### 6.2.4. System performance comparison

Performances of all systems are summarized in Table 53. All five different VRF systems have similar capacities and the R410A systems have similar COPs. The biggest advantage of HVRF system is the refrigerant charge reduction. bBTHX is proved to have the potential to be applied as the indoor unit of VRF system. Besides having smaller pressure drop on both air and water side, bBTHX also reduces the weight of the coil by reducing material volume and internal water volume.

Table 53 System performance comparison

| System | $\begin{gathered} \text { COP } \\ {[-]} \end{gathered}$ | Capacity <br> [W] | P_comp <br> [W] | $\mathbf{P}_{-} \text {con,fan }$ <br> [W] | P_waterloop <br> [W] | P_eva,fan <br> [W] | P_total <br> [W] | Refrigerant charge [kg] |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Baseline <br> R410A <br> VRF | 3.14 | 6869.04 | 2153.71 | 12.48 | - | 20.38 | 2186.57 | 17.83 |
| $\begin{gathered} \text { R410A } \\ \text { HVRF-A } \end{gathered}$ | 2.90 | 6862.17 | 2277.11 | 12.48 | 65.48 | 11 | 2366.07 | $\begin{gathered} 10.51 \\ (-41 \%) \\ \hline \end{gathered}$ |
| R410A HVRF-B | 2.91 | 6862.17 | 2277.11 | 12.48 | 60.22 | 4.36 | 2354.17 | $\begin{gathered} \hline 10.51 \\ (-41 \%) \end{gathered}$ |
| $\begin{gathered} \text { R290 } \\ \text { HVRF-B } \end{gathered}$ | 3.14 | 6869.13 | 2111.14 | 12.48 | 60.22 | 4.36 | 2188.2 | $\begin{gathered} 5.66 \\ (-68 \%) \end{gathered}$ |
| $\begin{gathered} \text { R600a } \\ \text { HVRF-B } \end{gathered}$ | 3.16 | 6868.58 | 2096.61 | 12.48 | 60.22 | 4.36 | 2173.67 | $\begin{gathered} 5.86 \\ (-67 \%) \end{gathered}$ |

## Chapter 7: Conclusions

This dissertation presents a novel compact air-to-fluid heat exchanger, bifurcated bare tube heat exchanger, that has the advantages of smaller volume, smaller material volume and lower pumping power. The key feature is the addition of bifurcation, which enhances airside heat transfer by creating 3D flow and improves waterside heat transfer by boundary layer interruption and redevelopment. This dissertation exploits the frontier of next generation compact heat exchanger design. This section summarizes the major contributions, publications and recommendations for future work.

### 7.1. Summary of Contributions

The contribution of the dissertation is broken down into three main parts and summarized as follows:
(1) Investigation of distinct types air-to-fluid heat exchanger performance:

- Experimentally investigated the thermal and hydraulic performance characteristics of slit fin-and-tube heat exchangers with diameter of 4 and 5 mm , louvered finned mini-channel heat exchanger, round bare tube heat exchanger ( $\mathrm{OD}=0.8 \mathrm{~mm}$ ) and shape optimized bare tube heat exchanger ( $\mathrm{D}_{\mathrm{h}}=0.87 \mathrm{~mm}$ ) under both dry and dehumidifying conditions.
- Developed empirical correlations for Chilton - Colburn jand factors based on experimental data for round bare tube and shape optimized bare tube heat exchangers.
- Analyzed the heat exchanger performance and gained the heat exchanger design fundamentals. The current airside heat transfer enhancement technology has the limitation of further improving heat transfer without pressure drop penalty.
(2) Novel heat exchanger design optimization:
- Reviewed nature inspired heat exchanger devices comprehensively and developed a nature-inspired heat exchanger design guideline.
- Developed two approaches of conducting Parallel Parameterized Computational Fluid Dynamics (PPCFD) technique using ANSYS ${ }^{\circledR}$ Workbench ${ }^{\text {TM }}$ and coupled it with Approximation Assisted Optimization (AAO) HX design framework.
- Invented, investigated, 3d printed, validated and optimized a novel heat exchanger:
- Invented a novel nature-inspired heat exchanger concept, bBTHX, that utilizes 3D flow at airside heat exchanger primary surface.
- Investigated numerically the fundamental heat transfer and flow friction mechanisms for both airside and waterside.
- Manufactured a prototype by 3D printing and experimentally validated the hydraulic performance of the novel heat exchanger numerical model.
- Optimized a single phase HX that has $38 \%$ lower total power and $83 \%$ smaller volume and $87 \%$ smaller material volume than those of louvered fin mini-channel heat exchanger. Compared to round bare tube heat exchanger, the optimal design has $28 \%$ lower total power and $11 \%$ smaller volume and $10 \%$ smaller material volume.
(3) Analysis of bBTHX applicability:
- Investigated the potential applications for bBTHX as car radiator and indoor coil for hybrid VRF system.
- Optimized a car radiator that has $30 \%$ lower total pumping power, $68 \%$ smaller heat exchanger envelope volume, $87 \%$ smaller material volume and $67 \%$ smaller water weight than louvered fin-and-tube radiator.
- Optimized an indoor unit for HVRF system that has: $40 \sim 70 \%$ smaller refrigerant charge than traditional VRF system. The advantage of utilizing bBTHX design is that it provides the potential of fast onsite indoor unit manufacturing using 3d printing technology to lower logistics cost.


### 7.2. Publications

Based on this work, five journal papers, seven conference papers and one invention record were published as follows:

Journal papers:

- Huang, Z., Li, Z., Hwang, Y. and Radermacher, R., Application of Entransy Dissipation Based Thermal Resistance to Design Optimization of a Novel Finless Evaporator. Sci China Tech Sci. (2016) 59: 1486. doi:10.1007/s11431-016-0312-3
- Huang, Z., Hwang, Y., Radermacher, R., Review of Nature-inspired Heat Exchanger Technology, International Journal of Refrigeration, (2017) 78: 1-17, ISSN 0140-7007, https://doi.org/10.1016/j.ijrefrig.2017.03.006.
- Huang, Z., Ling, J., Hwang, Y., Aute, V., Radermacher, R., Design and Numerical Parametric Study of Fractal Heat Exchanger, Science and Technology for the Built Environment. (2017), ISSN 2374-4731, DOI: 10.1080/23744731.2017.1335164
- Bacellar, D., Aute. V., Huang, Z., Radermacher, R., Airside Friction and Heat Transfer Characteristics for Staggered Tube Bundle in Crossflow Configuration with Diameters from 0.5 mm to 2.0 mm , International Journal of Heat and Mass Transfer, (2016) 98: 448-454, ISSN 0017-9310.
- Bacellar, D., Aute, V., Huang, Z., Radermacher, R., Design Optimization and Validation of High Performance Heat Exchangers using Multi-Scale Approximation Assisted Optimization and Additive Manufacturing, Science and Technology for the Built Environment. DOI: 10.1080/23744731.2017.1333877, DOI: 10.1080/23744731.2017.1333877


## Conference papers:

- Huang, Z., Hwang, Y., Aute, V., Radermacher, R., Review of Fractal Heat Exchangers, (2016), International Refrigeration and Air Conditioning Conference. Paper 1725
- Huang, Z., Ling, J., Hwang, Y., Aute, V., Radermacher, R, Design and Numerical Parametric Study of Fractal Heat Exchanger, (2016), International Refrigeration and Air Conditioning Conference. Paper 1723
- Huang, Z., Ling, J., Hwang, Y., Radermacher, R., Airside Thermal and Hydraulic Performance of a Bare Tube Heat Exchanger with Diameter of 0.8 mm under Dehumidifying Conditions, ASHRAE winter conference, Chicago, IL, Jan 20-24, 2018.
- Shabtay, Y., Huang, Z., Aute, V., Sharma, V., Radermacher, R., Manufacturing \& Testing of Air-to-refrigerant Heat Exchangers Based on 0.8mm Diameter Tubes, (2016), International Refrigeration and Air Conditioning Conference. Paper 1694
- Bacellar, D., Aute, V., Huang, Z., Radermacher, R., Novel Airside Heat Transfer Surface Designs Using an Integrated Multi-Scale Analysis with Topology and Shape Optimization, (2016), International Refrigeration and Air Conditioning Conference. Paper 1610
- Bacellar, D., Aute, V., Huang, Z. Radermacher, R., High Performance Gas-to-fluid Crossflow Heat Exchangers using Micro Tubes with Round and Novel Shapes, Thermal Fluids Analysis Workshop (TFAWS16) at NASA-Ames Research Center, August 01-04, 2016
- Bacellar, D., Huang, Z., Aute, V., Tancabel, J., Radermacher, R., Multi-scale Analysis, Shape Optimization and Experimental Validation of Novel Air-torefrigerant Heat Exchangers, 9th World Conference on Experimental Heat Transfer, Fluid Mechanics and Thermodynamics, Foz do Iguaçu, Brazil, June 11-15, 2017.

Invention record:

- Huang, Z., Bacellar, D., Ling, J., Hwang, Y., Aute, V., Hwang, Y., Radermacher, R., Finless air-to-refrigerant heat exchanger using novel micro fractal tube structure


### 7.3. Recommendations for Future Work

The focus of current dissertation is to enhance the heat transfer of air-to-fluid heat exchanger through novel geometry design to make it more compact and efficient. The key heat transfer mechanism utilized in current work is 3D flow caused by bifurcation. However, the bifurcation geometry is not the only one approach to generate 3D flow at primary heat transfer surface. Future work may be extended to investigate more novel geometries that can produce the same or better effect. Besides creating new geometries, other approaches, such as adding vortex
generators, novel fins should also be studied. Although it was found finless design with small diameter has an intrinsic advantage of improving heat transfer coefficient, it is still of significance to reconsider the design of secondary heat transfer area when cost and manufacturing easiness are taken into consideration.

In current multi-scale heat exchanger optimization design framework, the airside and waterside heat transfer and friction characteristics are simulated separately and then combined through a heat exchanger solver. It would be more meaningful to include both sides in the heat exchanger simulation domain in the future work. Current design framework is limited to geometry parameterization, which means geometry must be parameterized before being proceeded to numerical simulation and optimization. In the future, the design framework should be further improved and extended to have the capability of optimizing the heat exchanger through topology design without parameterization. And finally, this design framework should have the ability to learn from existing designs and generate innovative designs that meet the user requirements automatically.

In current study, the tube-side fluid used is water, however, two-phase flow simulation should be done in the future to widen the application of the new geometry. And the bifurcation geometry is not limited to hollow tubes, it can also be applied in other heat exchange device such as heat sink in electronic cooling to enhance heat transfer and reduce pumping power. This application has the advantage of easy manufacturing. There is a need to investigate more applications of current heat exchanger design. For instance, it can be applied as heat recovery heat exchanger in buildings, two-phase heat exchanger used as condenser/evaporator in heat pump/air conditioning systems, even air-to-air heat exchanger.

The main interest of developing the heat exchanger in current study is on the heat transfer improvement so that the stress analysis, vibration and noise, condensation, fouling/frosting and corrosion issues were not discussed thoroughly. In the future, such physics should be investigated to understand the new heat exchanger better. The consideration of these physics should also be added into the heat exchanger design framework.

One challenge in current study is mass production manufacturing method. The growing additive manufacturing technology makes it possible produce rapid prototypes for researchers. However, the tradeoff between the manufacturing cost, time, accuracy and reliability should be taken into consideration. Therefore, for mass production, the manufacturing method should be further investigated. And the manufacturing constraints should be added into the design framework. Another interesting research topic would be the onsite indoor unit 3D printing for HVRF system.

## Appendices

## Appendix A: Pictures of test facility



Figure A. 1 Test facility picture-test section


Figure A. 2 Test facility picture-flow rate measurement section


Figure A. 3 Test facility picture-water loop module


Figure A. 4 Test facility picture-refrigerant loop module

## Appendix B: HX blockage test pictures

Here is the summary of heat exchanger blockage tests. Warm water was drawn through the heat exchanger and infrared pictures are on the right side. If the temperature of the whole heat exchanger is even, then the heat exchanger is not blocked. If the temperature of the whole heat exchanger is not even, then the heat exchanger is blocked where the temperature is lower.

For all the heat exchangers tested, only 10 kW copper BTHX has blockage issue.

## 10kW-Copper BTHX blockage test



Figure B. 110 kW copper BTHX (front)


Figure B. 310 kW copper BTHX (back)


Figure B. 210 kW copper BTHX blockage test (front)


Figure B. 410 kW copper BTHX blockage test (back)

WTHX blockage test


Figure B. 5 WTHX (front)


Figure B. 7 WTHX (back)


Figure B. 6 WTHX blockage test (front)


Figure B. 8 WTHX blockage test (back)
sBTHX blockage test


Figure B. 9 sBTHX (front)


Figure B. 11 sBTHX (back)


Figure B. 10 sBTHX blockage test (front)


Figure B. 12 sBTHX blockage test (back)

BTHX blockage test


Figure B. 13 BTHX (front)


Figure B. 15 BTHX (back)


Figure B. 14 BTHX blockage test (front)


Figure B. 16 BTHX blockage test (back)

## BTHX-copper blockage test



Figure B. 18 BTHX-copper blockage test (front)

Figure B. 17 BTHX-copper (front)


Figure B. 20 BTHX-copper blockage test (front)

Figure B. 19 BTHX-copper (back)

## Appendix C: Heat exchanger test data

In this section, all heat exchanger test data are summarized.

## BTHX test data

Table C. 1 BTHX dry condition test data

| Case \# |  | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 35.15 | 34.93 | 35.04 | 34.93 | 34.94 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 51.64 | 50.28 | 49.21 | 48.20 | 47.14 |
| RH_air_in | [\%] | 0.31 | 0.31 | 0.32 | 0.33 | 0.34 |
| RH_air_out | [\%] | 0.14 | 0.15 | 0.16 | 0.17 | 0.19 |
| P_air_in | [ kPa ] | 100.86 | 100.88 | 100.94 | 101.06 | 101.10 |
| DP_air | [Pa] | 27.10 | 39.90 | 54.70 | 71.80 | 95.30 |
| uncertainty_DP | [Pa] | 1.10 | 1.30 | 1.50 | 1.80 | 2.00 |
| MFR_air | [g/s] | 36.21 | 46.39 | 56.45 | 66.84 | 79.09 |
| VFR_air | [m³/s] | 0.03 | 0.04 | 0.05 | 0.06 | 0.07 |
| Capacity_air | [W] | 601.63 | 717.20 | 805.79 | 893.61 | 971.92 |
| Capacity_air_uncertainty | [W] | 11.27 | 11.44 | 10.42 | 12.66 | 15.34 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 60.01 | 59.99 | 59.96 | 59.98 | 60.08 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 55.18 | 54.30 | 53.71 | 53.09 | 52.53 |
| P_water in | [kPa] | 98.50 | 98.53 | 98.70 | 98.81 | 98.85 |
| DP_water | [kPa] | 3.52 | 3.58 | 3.59 | 3.62 | 3.72 |
| MFR_water | [g/s] | 29.95 | 29.94 | 30.44 | 30.32 | 30.17 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 989.25 | 989.50 | 989.75 | 990.05 | 990.43 |
| Capacity_water | [W] | 605.40 | 712.40 | 795.20 | 873.20 | 952.60 |
| Capacity_water_uncertainty | [W] | 11.37 | 8.82 | 12.44 | 12.59 | 13.17 |
| Energy balance | [\%] | 0.62 | -0.67 | -1.32 | -2.31 | -2.01 |
| Average capacity | [W] | 603.52 | 714.80 | 800.49 | 883.40 | 962.26 |
| Capacity uncertainty | [W] | 8.00 | 7.22 | 8.11 | 8.93 | 10.11 |
| Case \# |  | 6 | 7 | 8 | 9 | 10 |
| T_air_in | [ ${ }^{\circ} \mathrm{C}$ ] | 35.18 | 35.03 | 35.10 | 34.93 | 35.11 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 52.22 | 51.01 | 49.87 | 48.73 | 47.94 |
| RH_air_in | [\%] | 30.4\% | 26.4\% | 26.5\% | 26.2\% | 29.0\% |
| RH_air_out | [\%] | 13.8\% | 12.4\% | 13.4\% | 14.2\% | 15.5\% |
| P_air_in | [kPa] | 100.83 | 101.39 | 101.43 | 101.48 | 101.46 |
| DP_air | [Pa] | 27.10 | 38.70 | 55.90 | 74.80 | 95.70 |
| uncertainty_DP | [Pa] | 1.10 | 1.30 | 1.50 | 1.80 | 2.00 |
| MFR_air | [g/s] | 36.22 | 45.57 | 57.32 | 68.79 | 36.22 |
| VFR_air | [m³/s] | 0.03 | 0.04 | 0.05 | 0.06 | 0.03 |
| Capacity_air | [W] | 621.63 | 733.29 | 852.62 | 956.35 | 1029.80 |
| Capacity_air_uncertainty | [W] | 11.93 | 12.42 | 12.93 | 13.08 | 17.08 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 59.97 | 59.95 | 60.01 | 59.99 | 59.94 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 56.98 | 56.46 | 56.00 | 55.52 | 55.13 |
| P_water in | [ kPa ] | 102.71 | 103.34 | 103.38 | 103.40 | 103.26 |
| DP_water | [ kPa ] | 7.36 | 6.75 | 6.75 | 6.71 | 7.31 |
| MFR_water | [g/s] | 50.18 | 50.26 | 50.30 | 50.28 | 50.13 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 988.44 | 988.72 | 988.79 | 988.80 | 989.19 |
| Capacity_water | [W] | 626.80 | 733.60 | 845.00 | 940.60 | 1010.00 |
| Capacity_water_uncertainty | [W] | 15.46 | 14.09 | 22.38 | 15.54 | 15.81 |
| Energy balance | [\%] | 0.83 | 0.04 | -0.90 | -1.66 | -1.94 |


| Average capacity | $[\mathrm{W}]$ | 624.21 | 733.44 | 848.81 | 948.47 | 1019.90 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| capacity uncertainty | $[\mathrm{W}]$ | 9.76 | 9.39 | 12.92 | 10.16 | 11.64 |
| Case\# |  | 11 | 12 | 13 | 14 | 15 |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 34.95 | 34.90 | 34.92 | 35.15 | 35.00 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 52.47 | 51.25 | 50.07 | 49.16 | 48.42 |
| RH_air_in | $[\%]$ | $30.8 \%$ | $26.4 \%$ | $26.6 \%$ | $26.4 \%$ | $29.2 \%$ |
| RH_air_out | $[\%]$ | $13.8 \%$ | $12.1 \%$ | $12.8 \%$ | $13.3 \%$ | $15.3 \%$ |
| P_air_in | $[\mathrm{kPa}]$ | 100.82 | 101.36 | 101.32 | 101.32 | 101.43 |
| DP_air | $[\mathrm{Pa}]$ | 27.10 | 38.70 | 56.10 | 75.30 | 93.50 |
| uncertainty_DP | $[\mathrm{Pa}]$ | 1.10 | 1.30 | 1.50 | 1.80 | 2.00 |
| MFR_air | $[\mathrm{g} / \mathrm{s}]$ | 36.21 | 45.56 | 57.42 | 68.75 | 78.73 |
| VFR_air | $\left[\mathrm{m}^{3} / \mathrm{s}\right]$ | 0.03 | 0.04 | 0.05 | 0.06 | 0.07 |
| Capacity_air | $[\mathrm{W}]$ | 639.18 | 750.42 | 876.39 | 970.04 | 1064.23 |
| Capacity_air_uncertainty | $[\mathrm{W}]$ | 11.41 | 10.95 | 11.70 | 13.25 | 19.37 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 60.01 | 59.91 | 59.95 | 60.00 | 60.01 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 57.81 | 57.33 | 56.97 | 56.71 | 56.46 |
| P_water in | $[\mathrm{kPa}]$ | 108.44 | 108.93 | 108.88 | 108.92 | 108.90 |
| DP_water | $[\mathrm{kPa}]$ | 12.59 | 11.40 | 11.37 | 11.41 | 11.68 |
| MFR_water | $[\mathrm{g} / \mathrm{s}]$ | 70.05 | 69.85 | 69.82 | 69.86 | 69.96 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 988.05 | 988.38 | 988.53 | 988.70 | 988.44 |
| Capacity_water | $[\mathrm{W}]$ | 646.50 | 752.70 | 871.20 | 961.20 | 1041.00 |
| Capacity_water_uncertainty | $[\mathrm{W}]$ | 23.95 | 21.50 | 23.89 | 23.92 | 31.14 |
| Energy balance | $[\%]$ | 1.14 | 0.30 | -0.59 | -0.91 | -2.21 |
| Average capacity | $[\mathrm{W}]$ | 642.84 | 751.56 | 873.80 | 965.62 | 1052.61 |
| capacity uncertainty | $[\mathrm{W}]$ | 13.26 | 12.06 | 13.30 | 13.67 | 18.34 |

Table C. 2 BTHX wet condition data (Vertical orientation, inlet air condition 1)

| Case\# |  | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.67 | 26.69 | 26.61 | 26.67 | 26.73 | 26.65 | 26.67 | 26.82 | 26.63 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 17.55 | 20.26 | 21.57 | 16.93 | 19.30 | 20.63 | 16.73 | 19.11 | 20.32 |
| RH_air_in | $[\%]$ | $13.7 \%$ | $14.1 \%$ | $15.9 \%$ | $14.0 \%$ | $14.3 \%$ | $21.9 \%$ | $13.4 \%$ | $15.0 \%$ | $24.3 \%$ |
| RH_air_out | $[\%]$ | $23.8 \%$ | $20.8 \%$ | $21.8 \%$ | $25.4 \%$ | $22.4 \%$ | $30.9 \%$ | $24.4 \%$ | $23.9 \%$ | $34.7 \%$ |
| P_air_in | $[\mathrm{kPa}]$ | 102.08 | 101.63 | 101.73 | 102.00 | 101.65 | 101.18 | 102.13 | 101.70 | 100.95 |
| DP_air | $[\mathrm{Pa}]$ | 29.00 | 84.30 | 155.50 | 28.90 | 83.80 | 152.40 | 29.10 | 83.30 | 150.40 |
| uncertainty_DP | $[\mathrm{Pa}]$ | 1.20 | 1.90 | 1.80 | 1.20 | 1.90 | 1.80 | 1.10 | 1.90 | 1.90 |
| correctionof DP | $[\mathrm{Pa}]$ | 0.00 | 0.00 | 62.35 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| MFR_air | $[\mathrm{g} / \mathrm{s}]$ | 37.79 | 74.68 | 110.70 | 37.80 | 74.51 | 109.00 | 37.83 | 74.27 | 108.10 |
| VFR_air | $\left[\mathrm{m}^{3} / \mathrm{s}\right]$ | 0.03 | 0.06 | 0.09 | 0.03 | 0.06 | 0.09 | 0.03 | 0.06 | 0.09 |
| Capacity_air | $[\mathrm{W}]$ | 346.70 | 483.56 | 561.94 | 370.50 | 557.13 | 660.96 | 378.35 | 576.61 | 686.71 |
| Capacity_air_uncertainty | $[\mathrm{W}]$ | 8.48 | 10.50 | 15.49 | 8.01 | 9.32 | 16.25 | 8.02 | 8.61 | 18.80 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 11.87 | 12.17 | 12.14 | 11.92 | 11.76 | 11.82 | 11.95 | 11.91 | 11.81 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 15.99 | 17.86 | 18.62 | 14.41 | 15.44 | 16.13 | 13.70 | 14.56 | 14.96 |
| P_water in | $[\mathrm{kPa}]$ | 100.25 | 99.55 | 99.71 | 103.91 | 103.36 | 102.77 | 108.46 | 107.86 | 107.04 |
| DP_water | $[\mathrm{kPa}]$ | 2.46 | 2.18 | 2.18 | 6.11 | 5.57 | 6.19 | 9.88 | 9.77 | 10.56 |
| MFR_water | $[\mathrm{g} / \mathrm{s}]$ | 20.11 | 20.34 | 20.70 | 34.89 | 35.34 | 35.18 | 50.16 | 50.27 | 50.32 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 996.94 | 997.16 | 997.54 | 996.76 | 996.85 | 996.36 | 996.75 | 996.86 | 995.93 |
| Capacity_water | $[\mathrm{W}]$ | 346.70 | 484.20 | 560.90 | 363.10 | 543.70 | 634.40 | 368.10 | 558.20 | 663.30 |
| Capacity_water_uncertainty | $[\mathrm{W}]$ | 7.08 | 16.21 | 16.77 | 9.76 | 7.56 | 7.92 | 12.02 | 12.37 | 15.26 |
| Energy balance | $[\%]$ | 0.00 | 0.13 | -0.19 | -2.02 | -2.44 | -4.10 | -2.75 | -3.24 | -3.47 |
| Average capacity | $[\mathrm{W}]$ | 346.70 | 483.88 | 561.42 | 366.80 | 550.41 | 647.68 | 373.23 | 567.40 | 675.00 |
| capacity uncertainty | $[\mathrm{W}]$ | 5.52 | 9.66 | 11.41 | 6.31 | 6.00 | 9.04 | 7.22 | 7.54 | 12.11 |

Table C. 3 BTHX Wet condition data (Vertical orientation, inlet air condition 2)

| Case \# |  | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.56 | 26.70 | 26.71 | 26.79 | 26.69 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 17.90 | 20.30 | 21.27 | 16.95 | 19.41 |
| RH_air_in | [\%] | 50.3\% | 51.2\% | 51.5\% | 50.6\% | 51.3\% |
| RH_air_out | [\%] | 81.8\% | 74.8\% | 71.6\% | 86.8\% | 77.6\% |
| P_air_in | [ kPa ] | 100.40 | 100.93 | 102.13 | 100.30 | 100.81 |
| DP_air | [Pa] | 88.50 | 150.80 | 214.90 | 98.90 | 165.50 |
| uncertainty_DP | [Pa] | 2.70 | 2.60 | 2.60 | 3.70 | 3.60 |
| SH | [W] | 351.20 | 480.30 | 592.20 | 384.90 | 540.60 |
| SH_un | [W] | 7.70 | 10.30 | 23.30 | 9.00 | 11.60 |
| LH | [W] | 20.10 | 16.90 | 12.90 | 33.30 | 42.50 |
| LH_un | [W] | 25.40 | 48.00 | 96.30 | 24.90 | 55.60 |
| MFR_air | [g/s] | 40.28 | 74.62 | 108.20 | 38.88 | 73.83 |
| VFR_air | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | 0.03 | 0.06 | 0.09 | 0.03 | 0.06 |
| Capacity_air | [W] | 369.42 | 492.45 | 599.75 | 416.40 | 578.44 |
| Capacity_air_uncertainty | [W] | 27.48 | 50.68 | 102.30 | 27.42 | 58.51 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 12.00 | 12.07 | 12.00 | 12.04 | 12.06 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 16.21 | 17.92 | 18.67 | 14.77 | 15.90 |
| P_water in | [ kPa ] | 98.22 | 98.76 | 99.98 | 101.83 | 102.26 |
| DP_water | [ kPa ] | 3.32 | 3.55 | 3.35 | 6.62 | 6.75 |
| MFR_water | [g/s] | 20.26 | 20.39 | 20.60 | 35.16 | 35.10 |
| Water density | [ $\left.\mathrm{kg} / \mathrm{m}^{3}\right]$ | 996.56 | 996.82 | 996.89 | 996.42 | 996.75 |
| Capacity_water | [W] | 356.90 | 498.50 | 574.80 | 402.20 | 563.50 |
| Capacity_water_uncertainty | [W] | 6.10 | 5.83 | 7.47 | 9.73 | 11.62 |
| Energy balance | [\%] | -3.45 | 1.22 | -4.25 | -3.47 | -2.62 |
| Average capacity | [W] | 363.16 | 495.47 | 587.28 | 409.30 | 570.97 |
| Capacity uncertainty | [W] | 14.075 | 25.51 | 51.29 | 14.55 | 29.83 |
| w_omega_in | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0110330 | 0.0112635 | 0.0112021 | 0.0112627 | 0.0112768 |
| w_omega_in_uncertainty | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0002166 | 0.0002958 | 0.0002980 | 0.0002911 | 0.0003897 |
| w_omega_out | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0108312 | 0.0111719 | 0.0111539 | 0.0109163 | 0.0110439 |
| w_omega_out_uncertainty | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 |
| condensate | [g/s] | 0.0081296 | 0.0068358 | 0.0052175 | 0.0134689 | 0.0171904 |
| condenate_uncertainty | [g/s] | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 |
| Case \# |  | 6 | 7 | 8 | 9 |  |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.77 | 26.77 | 26.74 | 26.76 |  |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 20.69 | 16.92 | 18.95 | 20.12 |  |
| RH_air_in | [\%] | 51.3\% | 50.7\% | 51.4\% | 51.0\% |  |
| RH_air_out | [\%] | 73.9\% | 87.3\% | 79.9\% | 75.7\% |  |
| P_air_in | [ kPa ] | 102.09 | 100.51 | 100.74 | 102.11 |  |
| DP_air | [Pa] | 234.00 | 104.70 | 180.00 | 243.10 |  |
| uncertainty_DP | [Pa] | 3.90 | 3.10 | 3.90 | 4.50 |  |
| SH | [W] | 672.20 | 379.90 | 576.00 | 721.80 |  |
| SH_un | [W] | 15.90 | 9.60 | 10.60 | 22.20 |  |
| LH | [W] | 22.20 | 59.00 | 48.80 | 30.90 |  |
| LH_un | [W] | 97.50 | 18.90 | 47.50 | 95.10 |  |
| MFR_air | [g/s] | 109.90 | 38.34 | 73.50 | 108.00 |  |
| VFR_air | [m³/s] | 0.09 | 0.03 | 0.06 | 0.09 |  |
| Capacity_air | [W] | 687.66 | 437.92 | 619.66 | 745.40 |  |
| Capacity_air_uncertainty | [W] | 102.00 | 22.30 | 50.17 | 100.70 |  |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 12.06 | 11.89 | 11.99 | 12.01 |  |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 16.60 | 13.87 | 14.93 | 15.41 |  |
| P_water in | [ kPa ] | 103.47 | 106.67 | 106.75 | 108.19 |  |


| DP_water | $[\mathrm{kPa}]$ | 6.69 | 10.58 | 10.76 | 10.86 |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| MFR_water | $[\mathrm{g} / \mathrm{s}]$ | 34.80 | 50.38 | 50.32 | 50.43 |  |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 996.83 | 996.34 | 996.67 | 996.85 |  |
| Capacity_water | $[\mathrm{W}]$ | 660.60 | 418.90 | 617.70 | 716.10 |  |
| Capacity_water_uncertainty | $[\mathrm{W}]$ | 10.64 | 12.97 | 14.43 | 19.68 |  |
| Energy balance | $[\%]$ | -4.01 | -4.44 | -0.32 | -4.01 |  |
| Average capacity | $[\mathrm{W}]$ | 674.13 | 428.41 | 618.68 | 730.75 |  |
| Capacity uncertainty | $[\mathrm{W}]$ | 51.28 | 12.90 | 26.10 | 51.30 |  |
| w_omega_in | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0112065 | 0.0112394 | 0.0113600 | 0.0111313 |  |
| w_omega_in_uncertainty | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0002963 | 0.0002895 | 0.0002925 | 0.0002910 |  |
| w_omega_out | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0111248 | 0.0106170 | 0.0110915 | 0.0110155 |  |
| w_omega_out_uncertainty | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 |  |
| condensate | $[\mathrm{g} / \mathrm{s}]$ | 0.0089795 | 0.0238604 | 0.0197373 | 0.0124978 |  |
| condenate_uncertainty | $[\mathrm{g} / \mathrm{s}]$ | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 |  |

Table C. 4 BTHX Wet condition data (Vertical orientation, inlet air condition 3)

| Case \# |  | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.68 | 26.70 | 26.77 | 26.68 | 26.74 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 18.25 | 20.64 | 21.68 | 17.40 | 19.81 |
| RH_air_in | [\%] | 70.6\% | 70.1\% | 71.1\% | 70.2\% | 70.4\% |
| RH_air_out | [\%] | 98.0\% | 93.4\% | 93.3\% | 97.2\% | 93.6\% |
| P_air_in | [ kPa ] | 100.47 | 100.86 | 100.76 | 100.49 | 100.88 |
| DP_air | [Pa] | 95.30 | 213.40 | 301.90 | 93.10 | 222.10 |
| uncertainty_DP | [Pa] | 3.10 | 4.00 | 1.20 | 3.40 | 4.50 |
| SH | [W] | 323.80 | 445.20 | 551.70 | 359.00 | 505.70 |
| SH_un | [W] | 8.50 | 8.80 | 14.00 | 8.30 | 10.70 |
| LH | [W] | 221.10 | 212.90 | 168.20 | 283.50 | 339.30 |
| LH_un | [W] | 36.70 | 47.70 | 72.70 | 32.00 | 60.90 |
| MFR_air | [g/s] | 38.15 | 72.99 | 107.60 | 38.42 | 72.54 |
| VFR_air | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | 0.03 | 0.06 | 0.09 | 0.03 | 0.06 |
| Capacity_air | [W] | 549.87 | 660.65 | 720.40 | 649.48 | 850.89 |
| Capacity_air_uncertainty | [W] | 39.57 | 50.21 | 76.86 | 35.17 | 64.04 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 11.90 | 11.81 | 11.98 | 11.87 | 11.74 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 18.24 | 19.62 | 20.47 | 16.09 | 17.45 |
| P_water in | [ kPa ] | 98.14 | 98.64 | 98.41 | 101.93 | 102.19 |
| DP_water | [kPa] | 3.55 | 2.26 | 2.66 | 6.57 | 5.77 |
| MFR_water | [g/s] | 20.18 | 20.16 | 20.59 | 35.19 | 35.15 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 996.80 | 996.70 | 996.91 | 996.58 | 996.74 |
| Capacity_water | [W] | 535.30 | 658.70 | 731.60 | 621.90 | 838.80 |
| Capacity_water_uncertainty | [W] | 8.21 | 8.54 | 7.22 | 9.72 | 10.45 |
| Energy balance | [\%] | -2.68 | -0.30 | 1.54 | -4.34 | -1.43 |
| Average capacity | [W] | 542.58 | 659.68 | 726.00 | 635.69 | 844.84 |
| Capacity uncertainty | [W] | 20.206 | 25.47 | 38.60 | 18.24 | 32.44 |
| w_omega_in | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0156909 | 0.0155322 | 0.0158450 | 0.0155908 | 0.0156254 |
| w_omega_in_uncertainty | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0002972 | 0.0002958 | 0.0002980 | 0.0002956 | 0.0003897 |
| w_omega_out | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0133470 | 0.0143527 | 0.0152128 | 0.0126067 | 0.0137340 |
| w_omega_out_uncertainty | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 |
| condensate | [g/s] | 0.0894168 | 0.0860932 | 0.0680283 | 0.1146492 | 0.1371981 |
| condenate_uncertainty | [g/s] | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 |
| Case \# |  | 6 | 7 | 8 | 9 |  |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.61 | 26.76 | 26.67 | 26.77 |  |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 20.73 | 16.78 | 19.40 | 20.31 |  |
| RH_air_in | [\%] | 71.4\% | 69.4\% | 70.2\% | 70.3\% |  |


| RH_air_out | $[\%]$ | $93.7 \%$ | $98.7 \%$ | $93.4 \%$ | $93.6 \%$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| P_air_in | $[\mathrm{kPa}]$ | 100.74 | 100.04 | 100.93 | 100.73 |  |
| DP_air | $[\mathrm{Pa}]$ | 304.80 | 86.00 | 223.80 | 307.50 |  |
| uncertainty_DP | $[\mathrm{Pa}]$ | 1.30 | 3.20 | 4.80 | 1.30 |  |
| SH | $[\mathrm{W}]$ | 633.10 | 366.70 | 530.90 | 694.00 |  |
| SH_un | $[\mathrm{W}]$ | 16.20 | 9.80 | 11.20 | 16.90 |  |
| LH | $[\mathrm{W}]$ | 337.80 | 283.60 | 383.30 | 406.10 |  |
| LH_un | $[\mathrm{W}]$ | 91.50 | 30.70 | 69.10 | 69.90 |  |
| MFR_air | $[\mathrm{g} / \mathrm{s}]$ | 107.00 | 36.51 | 72.56 | 106.80 |  |
| VFR_air | $\left[\mathrm{m}^{3} / \mathrm{s}\right]$ | 0.09 | 0.03 | 0.06 | 0.09 |  |
| Capacity_air | $[\mathrm{W}]$ | 976.61 | 656.52 | 921.30 | 1106.24 |  |
| Capacity_air_uncertainty | $[\mathrm{W}]$ | 96.64 | 34.94 | 72.48 | 75.21 |  |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 11.89 | 11.79 | 11.84 | 11.85 |  |
| T_water_out | $[\mathrm{C}]$ | 18.30 | 14.83 | 16.16 | 16.93 |  |
| P_water in | $[\mathrm{kPa}]$ | 101.96 | 106.03 | 106.76 | 106.42 |  |
| DP_water | $[\mathrm{kPa}]$ | 6.12 | 10.24 | 10.19 | 10.13 |  |
| MFR_water | $[\mathrm{g} / \mathrm{s}]$ | 35.19 | 50.18 | 50.22 | 50.02 |  |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 996.78 | 996.42 | 996.65 | 996.74 |  |
| Capacity_water | $[\mathrm{W}]$ | 944.20 | 639.10 | 907.80 | 1064.00 |  |
| Capacity_water_uncertainty | $[\mathrm{W}]$ | 11.76 | 16.88 | 19.96 | 14.74 |  |
| Energy balance | $[\%]$ | -3.37 | -2.69 | -1.48 | -3.89 |  |
| Average capacity | $[\mathrm{W}]$ | 960.40 | 647.81 | 914.55 | 1085.12 |  |
| Capacity uncertainty | $[\mathrm{W}]$ | 48.68 | 19.40 | 37.59 | 38.32 |  |
| W_omega_in | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0157473 | 0.0155629 | 0.0155203 | 0.0156499 |  |
| w_omega_in_uncertainty | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0002963 | 0.0002954 | 0.0002925 | 0.0002910 |  |
| w_omega_out | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0144706 | 0.0124219 | 0.0133841 | 0.0141122 |  |
| w_omega_out_uncertainty | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 |  |
| condensate | $[\mathrm{g} / \mathrm{s}]$ | 0.1366111 | 0.1146809 | 0.1550046 | 0.1642263 |  |
| condenate_uncertainty | $[\mathrm{g} / \mathrm{s}]$ | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 |  |
|  |  |  |  |  |  |  |

Table C. 5 BTHX Wet condition data (Horizontal orientation, inlet air condition 1)

| Case \# |  | 1 | 2 | 3 | 4 | 5 | 6 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.77 | 26.78 | 26.74 | 26.75 | 26.70 | 26.83 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 17.50 | 21.07 | 16.98 | 20.34 | 16.59 | 20.14 |
| RH_air_in | $[\%]$ | 0.19 | 0.21 | 0.19 | 0.21 | 0.19 | 0.21 |
| RH_air_out | $[\%]$ | 0.36 | 0.30 | 0.36 | 0.32 | 0.37 | 0.32 |
| P_air_in | $[\mathrm{kPa}]$ | 101.17 | 100.81 | 101.10 | 100.87 | 101.08 | 100.76 |
| DP_air | $[\mathrm{Pa}]$ | 27.20 | 151.70 | 28.70 | 151.50 | 28.40 | 151.60 |
| uncertainty_DP | $[\mathrm{Pa}]$ | 1.1 | 1.8 | 1.1 | 1.9 | 1.1 | 1.9 |
| MFR_air | $[\mathrm{g} / \mathrm{s}]$ | 37.70 | 110.20 | 38.84 | 110.10 | 38.61 | 110.20 |
| VFR_air | $\left[\mathrm{m}^{3} / \mathrm{s}\right]$ | 0.03 | 0.09 | 0.03 | 0.09 | 0.03 | 0.09 |
| Capacity_air | $[\mathrm{W}]$ | 344.36 | 580.08 | 368.62 | 657.72 | 385.04 | 695.42 |
| Capacity_air_uncertainty | $[\mathrm{W}]$ | 7.16 | 14.81 | 7.50 | 16.08 | 8.89 | 15.76 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 11.89 | 11.92 | 11.93 | 11.92 | 11.75 | 11.97 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 16.04 | 18.77 | 14.45 | 16.24 | 13.55 | 15.19 |
| P_water in | $[\mathrm{kPa}]$ | 100.76 | 100.12 | 105.99 | 105.83 | 113.07 | 112.67 |
| DP_water | $[\mathrm{kPa}]$ | 3.71 | 3.78 | 9.65 | 9.33 | 16.24 | 16.39 |
| MFR_water | $[\mathrm{g} / \mathrm{s}]$ | 19.79 | 19.92 | 34.99 | 35.25 | 50.10 | 49.79 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 996.80 | 997.14 | 996.71 | 997.07 | 996.63 | 996.70 |
| Capacity_water | $[\mathrm{W}]$ | 343.90 | 570.30 | 368.50 | 637.20 | 376.00 | 670.90 |
| Capacity_water_uncertainty | $[\mathrm{W}]$ | 7.14 | 9.14 | 8.02 | 11.53 | 14.20 | 17.38 |
| Energy balance | $[\%]$ | -0.13 | -1.70 | -0.03 | -3.17 | -2.38 | -3.59 |


| Average capacity | $[\mathrm{W}]$ | 344.1 | 575.2 | 368.6 | 647.5 | 380.52 | 683.16 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Capacity_uncertainty | $[\mathrm{W}]$ | 5.1 | 8.7 | 5.5 | 9.9 | 8.4 | 11.7 |

Table C. 6 BTHX Wet condition data (Horizontal orientation, inlet air condition 2)

| Case\# |  | 1 | 2 | 3 | 4 | 5 | 6 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.76 | 26.82 | 26.76 | 26.68 | 26.76 | 26.64 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 19.32 | 21.36 | 19.09 | 20.50 | 18.85 | 19.89 |
| RH_air_in | $[\%]$ | 0.51 | 0.52 | 0.51 | 0.51 | 0.51 | 0.52 |
| RH_air_out | $[\%]$ | 0.81 | 0.72 | 0.81 | 0.74 | 0.82 | 0.77 |
| P_air_in | $[\mathrm{kPa}]$ | 102.38 | 100.59 | 102.38 | 100.62 | 102.40 | 100.73 |
| DP_air | $[\mathrm{Pa}]$ | 123.20 | 253.80 | 133.40 | 263.00 | 135.90 | 279.90 |
| uncertainty_DP | $[\mathrm{Pa}]$ | 3.7 | 3.6 | 4.7 | 4.4 | 5 | 1.3 |
| MFR_air | $[\mathrm{g} / \mathrm{s}]$ | 39.99 | 108.10 | 38.77 | 105.80 | 38.55 | 105.80 |
| VFR_air | $\left[\mathrm{m}^{3} / \mathrm{s}\right]$ | 0.03 | 0.09 | 0.03 | 0.09 | 0.03 | 0.09 |
| Sensible capacity | $[\mathrm{W}]$ | 291.9 | 526.6 | 276.7 | 591.3 | 284.6 | 653 |
| Sensible capacity_uncertainty | $[\mathrm{W}]$ | 7.7 | 20 | 8.6 | 20.3 | 9.4 | 24.2 |
| Latent capacity | $[\mathrm{W}]$ | 0 | 0 | 0 | 11.2 | 5.4 | 10.8 |
| Latent capacity_uncertainty | $[\mathrm{W}]$ | 0.385 | 1 | 0.43 | 1.015 | 0.47 | 1.21 |
| Capacity_air | $[\mathrm{W}]$ | 287.38 | 512.51 | 272.60 | 595.49 | 287.17 | 656.55 |
| Capacity_air_uncertainty | $[\mathrm{W}]$ | 3.9 | 10.0 | 4.3 | 10.2 | 4.7 | 12.1 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 11.99 | 11.94 | 11.91 | 11.95 | 11.91 | 11.96 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 14.83 | 18.00 | 13.53 | 15.81 | 13.06 | 14.84 |
| P_water in | $[\mathrm{kPa}]$ | 102.16 | 100.59 | 107.63 | 105.93 | 115.38 | 113.15 |
| DP_water | $[\mathrm{kPa}]$ | 4.68 | 4.01 | 9.74 | 10.17 | 16.90 | 16.81 |
| MFR_water | $[\mathrm{g} / \mathrm{s}]$ | 19.96 | 19.80 | 35.05 | 34.98 | 50.92 | 50.17 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 996.60 | 997.02 | 996.45 | 996.94 | 996.36 | 996.86 |
| Capacity_water | $[\mathrm{W}]$ | 237.20 | 502.00 | 238.60 | 565.50 | 243.70 | 605.30 |
| Capacity_water_uncertainty | $[\mathrm{W}]$ | 6.56 | 9.14 | 16.95 | 12.67 | 27.12 | 26.17 |
| Energy balance | $[\%]$ | -4.00 | -2.07 | -3.00 | -5.17 | -2.50 | -4.00 |
| Average capacity | $[\mathrm{W}]$ | 262.29 | 507.26 | 255.6 | 580.5 | 265.4 | 630.93 |
| Capacity_uncertainty | $[\mathrm{W}]$ | 3.8 | 6.8 | 8.7 | 8.1 | 13.8 | 14.4 |

Table C. 7 BTHX Wet condition data (Horizontal orientation, inlet air condition 3)

| Case\# |  | 1 | 2 | 3 | 4 | 5 | 6 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.79 | 26.70 | 26.65 | 26.71 | 26.65 | 26.71 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 20.29 | 21.64 | 20.06 | 20.99 | 20.32 | 20.66 |
| RH_air_in | $[\%]$ | 0.70 | 0.70 | 0.70 | 0.71 | 0.70 | 0.71 |
| RH_air_out | $[\%]$ | 1.00 | 0.92 | 1.00 | 0.93 | 0.98 | 0.94 |
| P_air_in | $[\mathrm{kPa}]$ | 101.45 | 101.30 | 101.39 | 101.19 | 101.38 | 101.19 |
| DP_air | $[\mathrm{Pa}]$ | 161.60 | 315.14 | 166.60 | 334.95 | 151.50 | 338.05 |
| uncertainty_DP | $[\mathrm{Pa}]$ | 6.1 | 3.1 | 3.9 | 3.1 | 5 | 3.1 |
| MFR_air | $[\mathrm{g} / \mathrm{s}]$ | 37.65 | 108.40 | 37.26 | 109.30 | 37.13 | 108.70 |
| VFR_air | $\left[\mathrm{m}^{3} / \mathrm{s}\right]$ | 0.03 | 0.09 | 0.03 | 0.09 | 0.03 | 0.09 |
| Sensiblecapacity | $[\mathrm{W}]$ | 240.1 | 521.4 | 241.9 | 603.7 | 230.5 | 638.8 |
| Sensible capacity_uncertainty | $[\mathrm{W}]$ | 9.4 | 20.5 | 7.9 | 16.7 | 7.9 | 16.4 |
| Latent capacity | $[\mathrm{W}]$ | 50.8 | 125.5 | 58.5 | 243.3 | 66.9 | 315.3 |
| Latent capacity_uncertainty | $[\mathrm{W}]$ | 0.47 | 1.025 | 0.395 | 0.835 | 0.395 | 0.82 |
| Capacity_air | $[\mathrm{W}]$ | 290.72 | 648.19 | 300.69 | 850.83 | 298.10 | 959.80 |
| Capacity_air_uncertainty | $[\mathrm{W}]$ | 4.7 | 10.3 | 4.0 | 8.4 | 4.0 | 8.2 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 11.87 | 11.98 | 11.85 | 11.96 | 11.96 | 11.94 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 15.30 | 19.64 | 13.82 | 17.59 | 13.32 | 16.31 |


| P_water in | $[\mathrm{kPa}]$ | 102.07 | 101.96 | 106.69 | 105.98 | 113.77 | 113.21 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| DP_water | $[\mathrm{Pa}]$ | 4.17 | 4.09 | 10.42 | 9.83 | 16.98 | 16.06 |
| MFR_water | $[\mathrm{g} / \mathrm{s}]$ | 19.81 | 19.79 | 34.98 | 35.04 | 49.99 | 50.11 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 996.21 | 996.46 | 996.07 | 996.47 | 996.02 | 996.36 |
| Capacity_water | $[\mathrm{W}]$ | 284.30 | 633.90 | 287.40 | 826.40 | 285.20 | 917.60 |
| Capacity_water_uncertainty | $[\mathrm{W}]$ | 8.27 | 8.07 | 13.38 | 9.35 | 18.63 | 14.67 |
| Energy balance | $[\%]$ | -2.23 | -2.23 | -4.52 | -2.91 | -4.42 | -4.50 |
| Average capacity | $[\mathrm{W}]$ | 287.51 | 641.04 | 294.05 | 838.61 | 291.65 | 938.70 |
| Capacity_uncertainty | $[\mathrm{W}]$ | 4.8 | 6.5 | 7.0 | 6.3 | 9.5 | 8.4 |

## WTHX test data

WTHX (webbed tube heat exchanger) is one of the 1 kW heat exchanger prototypes that has been manufactured. Dimensions are shown below and the test data are summarized.


Figure C. 1 Geometry of Webbed Tube Heat Exchanger (WTHX)

Table C. 8 WTHX dimensions

| Air cross section area | $\left[\mathrm{m}^{2}\right]$ | 0.0097 |
| :---: | :--- | :--- |
| Water cross section area | $\left[\mathrm{m}^{2}\right]$ | 0.0001 |
| Air heat transfer area | $\left[\mathrm{m}^{2}\right]$ | 0.1647 |
| Water heat transfer area | $\left[\mathrm{m}^{2}\right]$ | 0.0759 |

Table C. 9 WTHX dry condition test data

| Case \# |  | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 34.97 | 34.79 | 34.84 | 34.98 | 34.94 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 41.14 | 41.27 | 41.42 | 41.52 | 41.46 |
| RH_air_in | [\%] | 35.8\% | $35.7 \%$ | 37.2\% | 35.8\% | 35.7\% |
| RH_air_out | [\%] | 25.2\% | 24.9\% | 25.8\% | 24.5\% | 24.6\% |
| P_air_in | [ kPa ] | 100.9 | 100.5 | 100.6 | 100.7 | 100.7 |
| DP_air | [Pa] | 275.5 | 201.3 | 141.7 | 94.7 | 58.6 |
| MFR_air | [g/s] | 78.89 | 67.34 | 56.38 | 45.96 | 35.57 |
| VFR_air | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | 0.0702 | 0.0601 | 0.0503 | 0.0406 | 0.0314 |
| Capacity_air | [W] | 579.5 | 516.9 | 443.7 | 370.1 | 290.0 |
| Capacity_air_uncertainty | [W] | 13.3 | 15.4 | 10.1 | 9.1 | 7.6 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 60.03 | 60.08 | 60.05 | 59.98 | 59.96 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 57.90 | 58.15 | 58.32 | 58.46 | 58.69 |
| P_water in | [ kPa ] | 98.3 | 97.8 | 97.8 | 98.0 | 98.0 |
| DP_water | [ kPa ] | 95.3 | 94.6 | 94.6 | 94.7 | 94.6 |
| MFR_water | [g/s] | 3.7 | 3.8 | 3.8 | 3.8 | 3.8 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 70.0 | 70.1 | 69.9 | 70.0 | 69.8 |
| Capacity_water | [W] | 988.1 | 987.9 | 987.8 | 987.9 | 987.7 |
| Capacity_water_uncertainty | [W] | 41.5 | 33.2 | 33.1 | 49.7 | 39.3 |
| Energy balance | [\%] | 7.4\% | 0.7\% | 1.9\% | 3.6\% | 5.4\% |
| Average capacity | [W] | 601.6 | 541.5 | 474.7 | 407.5 | 330.2 |
| Case \# |  | 6 | 7 | 8 | 9 | 10 |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 34.88 | 34.90 | 34.79 | 34.78 | 34.99 |


| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 40.90 | 41.11 | 41.23 | 41.14 | 41.43 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| RH_air_in | [\%] | 35.3\% | 36.0\% | 36.2\% | 37.4\% | 35.3\% |
| RH_air_out | [\%] | 24.7\% | 25.1\% | 25.0\% | 25.8\% | 24.7\% |
| P_air_in | [ kPa ] | 100.6 | 100.5 | 100.7 | 100.7 | 100.6 |
| DP_air | [Pa] | 204.6 | 141.2 | 95.0 | 58.4 | 204.6 |
| MFR_air | [g/s] | 67.84 | 56.39 | 45.84 | 35.55 | 67.84 |
| VFR_air | [ $\left.\mathrm{m}^{3} / \mathrm{s}\right]$ | 0.0605 | 0.0503 | 0.0404 | 0.0314 | 0.0605 |
| Capacity_air | [W] | 500.9 | 434.8 | 362.2 | 284.9 | 500.9 |
| Capacity_air_uncertainty | [W] | 14.4 | 8.5 | 8.9 | 6.4 | 14.4 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 59.85 | 59.97 | 60.00 | 59.94 | 59.85 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 57.33 | 57.76 | 58.04 | 58.33 | 57.33 |
| P_water in | [ kPa ] | 96.8 | 96.7 | 96.8 | 96.8 | 96.8 |
| DP_water | [ kPa ] | 3.2 | 3.4 | 3.4 | 3.4 | 3.2 |
| MFR_water | [g/s] | 49.6 | 50.7 | 50.6 | 50.2 | 49.6 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 988.3 | 988.1 | 988.3 | 987.9 | 988.3 |
| Capacity_water | [W] | 522.2 | 468.3 | 414.5 | 337.7 | 522.2 |
| Capacity_water_uncertainty | [W] | 27.9 | 31.5 | 34.4 | 29.7 | 27.9 |
| Energy balance | [\%] | -2.5\% | -1.6\% | -0.6\% | 0.8\% | -2.5\% |
| Average capacity | [W] | 511.6 | 451.6 | 388.3 | 311.3 | 511.6 |
| Case \# |  | 11 | 12 | 13 | 14 | 15 |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 35.13 | 34.97 | 34.81 | 35.06 | 34.95 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 40.84 | 40.94 | 40.95 | 41.36 | 41.33 |
| RH_air_in | [\%] | 35.5\% | 35.6\% | 33.3\% | 34.8\% | 35.2\% |
| RH_air_out | [\%] | 25.6\% | 25.5\% | 23.6\% | 24.6\% | 24.8\% |
| P_air_in | [ kPa ] | 100.7 | 100.7 | 100.6 | 100.6 | 100.6 |
| DP_air | [Pa] | 265.3 | 202.4 | 141.9 | 96.2 | 57.4 |
| uncertainty_DP | [Pa] | 3.7 | 3.0 | 2.4 | 1.9 | 1.5 |
| MFR_air | [g/s] | 77.43 | 67.48 | 56.49 | 46.23 | 35.15 |
| VFR_air | [m3/s] | 0.0690 | 0.0601 | 0.0502 | 0.0410 | 0.0311 |


| Capacity_air | $[\mathrm{W}]$ | 526.3 | 480.4 | 417.9 | 356.2 | 279.4 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Capacity_air_uncertainty | $[\mathrm{W}]$ | 12.9 | 16.8 | 9.2 | 7.7 | 7.1 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 60.07 | 60.05 | 60.01 | 60.01 | 60.03 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 55.95 | 56.27 | 56.61 | 57.05 | 57.63 |
| P_water in | $[\mathrm{kPa}]$ | 96.1 | 96.1 | 95.9 | 95.9 | 95.8 |
| DP_water | $[\mathrm{kPa}]$ | 2.9 | 2.9 | 3.0 | 3.0 | 2.9 |
| MFR_water | $[\mathrm{g} / \mathrm{s}]$ | 30.1 | 30.2 | 29.9 | 30.0 | 30.0 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 989.0 | 988.9 | 988.7 | 988.6 | 988.4 |
| Capacity_water | $[\mathrm{W}]$ | 518.0 | 476.6 | 424.6 | 370.9 | 301.4 |
| Capacity_water_uncertainty | $[\mathrm{W}]$ | 20.5 | 15.3 | 22.1 | 19.5 | 20.5 |
| Energy balance | $[\%]$ | $-5.8 \%$ | $-5.3 \%$ | $-5.1 \%$ | $-4.4 \%$ | $-3.6 \%$ |
| Average capacity | $[\mathrm{W}]$ | 522.1 | 478.5 | 421.3 | 363.5 | 290.4 |

## sBTHX test data

Table C. 10 sBTHX dry condition test data

| Case \# |  | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 34.95 | 35.05 | 34.97 | 34.94 | 34.95 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 49.25 | 48.21 | 46.58 | 45.47 | 44.80 |
| RH_air_in | $[\%]$ | $26.9 \%$ | $27.1 \%$ | $33.8 \%$ | $31.6 \%$ | $32.0 \%$ |
| RH_air_out | $[\%]$ | $14.9 \%$ | $15.8 \%$ | $19.4 \%$ | $19.5 \%$ | $20.0 \%$ |
| P_air_in | $[\mathrm{kPa}]$ | 101.62 | 101.34 | 101.19 | 101.45 | 101.14 |
| DP_air | $[\mathrm{Pa}]$ | 70.80 | 101.00 | 148.40 | 196.40 | 253.60 |
| uncertainty_DP | $[\mathrm{Pa}]$ | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| MFR_air | $[\mathrm{g} / \mathrm{s}]$ | 36.27 | 45.38 | 56.86 | 68.65 | 79.26 |
| VFR_air | $\left[\mathrm{m}^{3} / \mathrm{s}\right]$ | 0.03 | 0.04 | 0.05 | 0.06 | 0.07 |
| Capacity_air | $[\mathrm{W}]$ | 522.32 | 601.38 | 664.99 | 728.38 | 786.54 |
| Capacity_air_uncertainty | $[\mathrm{W}]$ | 9.15 | 10.15 | 19.48 | 11.89 | 12.8 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 60.03 | 59.98 | 59.86 | 59.97 | 59.97 |


| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 55.90 | 55.31 | 54.70 | 54.31 | 53.91 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| P_water in | [ kPa ] | 97.53 | 97.30 | 97.17 | 97.63 | 97.22 |
| DP_water | [ kPa ] | 3.01 | 3.85 | 2.89 | 3.16 | 3.1 |
| MFR_water | [g/s] | 30.02 | 29.81 | 30.20 | 30.19 | 30.21 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 988.63 | 988.80 | 988.92 | 989.43 | 989.36 |
| Capacity_water | [W] | 518.90 | 587.20 | 652.50 | 713.80 | 766.50 |
| Capacity_water_uncertainty | [W] | 12.22 | 9.47 | 18.17 | 16.46 | 13.25 |
| Energy balance | [\%] | -0.66 | -2.39 | -1.90 | -2.02 | -2.58 |
| Average capacity | [W] | 520.61 | 594.29 | 658.75 | 721.09 | 776.52 |
| capacity uncertainty | [W] | 7.63 | 6.94 | 13.32 | 10.15 | 9.21 |
| Case \# |  | 6 | 7 | 8 | 9 | 10 |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 34.92 | 34.99 | 35.06 | 34.87 | 34.95 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 50.09 | 48.88 | 47.63 | 46.01 | 45.78 |
| RH_air_in | [\%] | 26.9\% | 27.0\% | 35.9\% | 31.3\% | $32.4 \%$ |
| RH_air_out | [\%] | 14.0\% | 14.9\% | 20.3\% | 18.5\% | 19.9\% |
| P_air_in | [ kPa ] | 101.68 | 101.38 | 101.25 | 101.53 | 101.09 |
| DP_air | [Pa] | 71.20 | 99.30 | 141.20 | 202.70 | 243.00 |
| uncertainty_DP | [Pa] | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| MFR_air | [g/s] | 36.27 | 45.29 | 56.08 | 69.50 | 77.77 |
| VFR_air | [m³/s] | 0.03 | 0.04 | 0.05 | 0.06 | 0.07 |
| Capacity_air | [W] | 554.02 | 633.71 | 710.10 | 780.01 | 848.34 |
| Capacity_air_uncertainty | [W] | 9.51 | 9.83 | 11.37 | 10.39 | 13.20 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 59.94 | 59.94 | 59.88 | 59.87 | 59.90 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 57.23 | 56.85 | 56.54 | 56.15 | 55.96 |
| P_water in | [ kPa ] | 99.39 | 99.17 | 99.19 | 99.55 | 99.06 |
| DP_water | [ kPa ] | 4.93 | 4.75 | 4.82 | 5.15 | 5.17 |
| MFR_water | [g/s] | 50.01 | 49.67 | 50.07 | 50.25 | 49.94 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 988.09 | 988.25 | 988.22 | 988.58 | 988.63 |
| Capacity_water | [W] | 566.10 | 640.50 | 700.80 | 781.80 | 823.60 |


| Capacity_water_uncertainty | [W] | 18.82 | 23.51 | 22.00 | 14.15 | 15.42 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Energy balance | [\%] | 2.16 | 1.07 | -1.32 | 0.23 | -2.96 |
| Average capacity | [W] | 560.06 | 637.11 | 705.45 | 780.90 | 835.97 |
| capacity uncertainty | [W] | 10.54 | 12.74 | 12.38 | 8.78 | 10.15 |
| Case \# |  | 11 | 12 | 13 | 14 | 15 |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 35.03 | 34.97 | 35.05 | 35.05 | 35.17 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 50.40 | 49.14 | 47.60 | 46.94 | 46.15 |
| RH_air_in | [\%] | 27.5\% | 26.7\% | 27.1\% | 28.0\% | 29.5\% |
| RH_air_out | [\%] | 14.0\% | 14.2\% | 15.1\% | 16.3\% | 17.0\% |
| P_air_in | [ kPa ] | 101.72 | 101.46 | 101.31 | 101.21 | 101.08 |
| DP_air | [Pa] | 70.20 | 100.70 | 149.50 | 193.50 | 247.40 |
| uncertainty_DP | [Pa] | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| MFR_air | [g/s] | 36.35 | 45.33 | 58.24 | 67.48 | 78.75 |
| VFR_air | [m3/s] | 0.03 | 0.04 | 0.05 | 0.06 | 0.07 |
| Capacity_air | [W] | 562.81 | 646.88 | 735.95 | 807.78 | 871.01 |
| Capacity_air_uncertainty | [W] | 10.12 | 9.54 | 13.41 | 13.86 | 13.38 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 60.04 | 60.12 | 60.09 | 60.09 | 60.12 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 58.02 | 57.80 | 57.51 | 57.32 | 57.15 |
| P_water in | [ kPa ] | 101.55 | 101.51 | 101.25 | 101.18 | 101.17 |
| DP_water | [ kPa ] | 6.09 | 6.16 | 6.25 | 6.09 | 6.24 |
| MFR_water | [g/s] | 69.98 | 70.00 | 70.43 | 70.36 | 70.67 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 987.89 | 987.95 | 987.93 | 987.82 | 987.85 |
| Capacity_water | [W] | 591.20 | 678.20 | 760.00 | 815.10 | 876.80 |
| Capacity_water_uncertainty | [W] | 23.94 | 23.94 | 21.70 | 31.28 | 19.79 |
| Energy balance | [\%] | 4.92 | 4.73 | 3.21 | 0.90 | 0.66 |
| Average capacity | [W] | 577.00 | 662.54 | 747.98 | 811.44 | 873.91 |
| capacity uncertainty | [W] | 13.00 | 12.88 | 12.75 | 17.11 | 11.94 |

Wet condition data (Inlet air condition 1):

| Case \# |  | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.49 | 26.58 | 26.80 | 26.71 | 26.53 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 17.96 | 20.66 | 22.07 | 17.59 | 20.47 |
| RH_air_in | [\%] | 50.2\% | 49.0\% | 49.0\% | 49.9\% | 50.8\% |
| RH_air_out | [\%] | 82.8\% | 74.4\% | 68.9\% | 83.6\% | 74.4\% |
| P_air_in | [ kPa ] | 101.45 | 101.60 | 102.24 | 101.38 | 102.28 |
| DP_air | [Pa] | 46.60 | 142.20 | 248.40 | 47.00 | 141.20 |
| uncertainty_DP | [Pa] | 1.50 | 2.90 | 2.70 | 1.40 | 2.70 |
| SH | [W] | 231.45 | 358.02 | 422.17 | 245.14 | 363.64 |
| SH_un | [W] | 6.05 | 13.10 | 11.22 | 6.58 | 16.75 |
| LH | [W] | 28.06 | 0.00 | 0.00 | 33.71 | 26.24 |
| LH_un | [W] | 0.51 | 0.00 | 0.00 | 0.19 | 0.10 |
| MFR_air | [g/s] | 26.96 | 60.10 | 88.71 | 26.71 | 59.57 |
| VFR_air | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | 0.02 | 0.05 | 0.08 | 0.02 | 0.05 |
| Capacity_air | [W] | 259.50 | 358.02 | 422.17 | 278.85 | 389.88 |
| Capacity_air_uncertainty | [W] | 6.07 | 13.10 | 11.22 | 6.59 | 16.75 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 11.93 | 11.90 | 11.93 | 11.97 | 11.93 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 15.03 | 16.10 | 16.96 | 13.86 | 14.60 |
| P_water in | [ kPa ] | 98.75 | 98.25 | 98.82 | 100.24 | 100.94 |
| DP_water | [ kPa ] | 3.70 | 3.50 | 3.40 | 5.30 | 5.20 |
| MFR_water | [g/s] | 19.79 | 19.97 | 19.91 | 34.95 | 34.87 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 995.83 | 996.17 | 995.32 | 995.62 | 995.26 |
| Capacity_water | [W] | 257.00 | 350.70 | 418.50 | 276.90 | 389.60 |
| Capacity_water_uncertainty | [W] | 10.87 | 8.48 | 7.09 | 15.73 | 14.96 |
| Energy balance | [\%] | -0.97 | -2.07 | -0.87 | -0.70 | -0.07 |
| Average capacity | [W] | 258.25 | 354.36 | 420.34 | 277.88 | 389.74 |
| Capacity uncertainty | [W] | 6.225 | 7.80 | 6.64 | 8.53 | 11.23 |
| w_omega_in | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0108500 | 0.0106200 | 0.0106900 | 0.0109200 | 0.0109000 |
| w_omega_in_uncertainty | [ $\mathrm{kg} / \mathrm{kg}]$ | 0.0001423 | 0.0001395 | 0.0001403 | 0.0002143 | 0.0002136 |


| w_omega_out | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0104291 | 0.0106200 | 0.0106900 | 0.0104096 | 0.0107218 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| w_omega_out_uncertainty | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 |
| condensate | [g/s] | 0.0113465 | 0.0000000 | 0.0000000 | 0.0136325 | 0.0106129 |
| condenate_uncertainty | [g/s] | 0.0002061 | 0.0000000 | 0.0000000 | 0.0000767 | 0.0000401 |
| Case \# |  | 6 | 7 | 8 | 9 |  |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.78 | 26.67 | 26.78 | 26.70 |  |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 21.55 | 17.98 | 20.34 | 21.23 |  |
| RH_air_in | [\%] | 50.0\% | 50.0\% | 50.3\% | 51.2\% |  |
| RH_air_out | [\%] | 72.1\% | 83.7\% | 75.3\% | 74.4\% |  |
| P_air_in | [ kPa ] | 102.46 | 101.40 | 102.21 | 100.89 |  |
| DP_air | [Pa] | 257.10 | 48.00 | 145.50 | 265.30 |  |
| uncertainty_DP | [Pa] | 4.20 | 1.60 | 3.20 | 3.10 |  |
| SH | [W] | 476.75 | 231.57 | 383.60 | 486.61 |  |
| SH_un | [W] | 11.26 | 6.00 | 11.76 | 11.66 |  |
| LH | [W] | 14.30 | 44.47 | 43.01 | 15.40 |  |
| LH_un | [W] | 0.09 | 0.51 | 0.41 | 0.03 |  |
| MFR_air | [g/s] | 90.50 | 26.48 | 59.19 | 88.38 |  |
| VFR_air | [m³/s] | 0.08 | 0.02 | 0.05 | 0.08 |  |
| Capacity_air | [W] | 491.05 | 276.05 | 426.61 | 502.01 |  |
| Capacity_air_uncertainty | [W] | 11.26 | 6.02 | 11.77 | 11.66 |  |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 11.90 | 11.89 | 11.92 | 11.86 |  |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 15.11 | 13.24 | 13.90 | 14.26 |  |
| P_water in | [ kPa ] | 101.04 | 102.52 | 103.16 | 101.30 |  |
| DP_water | [ kPa ] | 5.10 | 7.10 | 7.10 | 6.60 |  |
| MFR_water | [g/s] | 34.97 | 49.91 | 50.03 | 50.00 |  |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 995.64 | 995.58 | 995.11 | 995.20 |  |
| Capacity_water | [W] | 470.20 | 281.30 | 415.90 | 502.20 |  |
| Capacity_water_uncertainty | [W] | 12.74 | 11.13 | 12.87 | 20.71 |  |
| Energy balance | [\%] | -4.34 | 1.89 | $-2.54$ | 0.04 |  |


| Average capacity | $[\mathrm{W}]$ | 480.63 | 278.67 | 421.26 | 502.10 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Capacity uncertainty | $[\mathrm{W}]$ | 8.50 | 6.33 | 8.72 | 11.88 |
| w_omega_in | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0108700 | 0.0109200 | 0.0109800 | 0.0112700 |
| w_omega_in_uncertainty | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0001424 | 0.0002856 | 0.0002151 | 0.0002206 |
| w_omega_out | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0108061 | 0.0102408 | 0.0106861 | 0.0111995 |
| w_omega_out_uncertainty | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 |
| condensate | $[\mathrm{g} / \mathrm{s}]$ | 0.0057837 | 0.0179856 | 0.0173951 | 0.0062280 |
| condenate_uncertainty | $[\mathrm{g} / \mathrm{s}]$ | 0.0000366 | 0.0002055 | 0.0001650 | 0.0000141 |

Table C. 11 sBTHX Wet condition data (Inlet air condition 2)

| Case \# |  | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.60 | 26.80 | 26.74 | 26.53 | 26.63 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 19.54 | 21.76 | 22.58 | 18.86 | 21.06 |
| RH_air_in | [\%] | 71.0\% | 70.6\% | 71.7\% | 70.4\% | 70.4\% |
| RH_air_out | [\%] | 92.2\% | 88.8\% | 87.9\% | 91.0\% | 88.6\% |
| P_air_in | [ kPa ] | 101.39 | 101.47 | 101.86 | 101.38 | 101.53 |
| DP_air | [Pa] | 50.80 | 155.70 | 263.80 | 51.10 | 156.80 |
| uncertainty_DP | [Pa] | 1.60 | 3.00 | 2.90 | 1.60 | 3.00 |
| SH | [W] | 189.49 | 300.33 | 372.27 | 205.11 | 332.46 |
| SH_un | [W] | 5.75 | 12.06 | 15.68 | 5.09 | 16.50 |
| LH | [W] | 179.58 | 207.00 | 217.36 | 202.94 | 257.69 |
| LH_un | [W] | 0.00 | 0.10 | 0.82 | 0.68 | 1.48 |
| MFR_air | [g/s] | 26.68 | 59.25 | 88.84 | 26.57 | 59.32 |
| VFR_air | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | 0.02 | 0.05 | 0.08 | 0.02 | 0.05 |
| Capacity_air | [W] | 369.07 | 507.33 | 589.63 | 408.05 | 590.15 |
| Capacity_air_uncertainty | [W] | 5.75 | 12.06 | 15.70 | 5.14 | 16.57 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 12.00 | 12.10 | 11.99 | 11.95 | 11.92 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 16.60 | 18.05 | 18.80 | 14.81 | 15.87 |
| P_water in | [ kPa ] | 98.52 | 98.30 | 98.44 | 100.14 | 100.26 |


| DP_water | [ kPa ] | 3.50 | 3.50 | 3.50 | 5.30 | 5.30 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| MFR_water | [g/s] | 19.92 | 19.96 | 20.10 | 34.99 | 35.06 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 995.83 | 995.91 | 996.00 | 995.68 | 995.81 |
| Capacity_water | [W] | 383.40 | 497.40 | 572.40 | 418.30 | 579.50 |
| Capacity_water_uncertainty | [W] | 9.50 | 12.98 | 10.97 | 12.07 | 13.41 |
| Energy balance | [\%] | 3.81 | -1.98 | -2.97 | 2.48 | -1.82 |
| Average capacity | [W] | 376.23 | 502.36 | 581.02 | 413.18 | 584.82 |
| Capacity uncertainty | [W] | 5.555 | 8.86 | 9.58 | 6.56 | 10.66 |
| w_omega_in | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0155400 | 0.0156300 | 0.0157600 | 0.0153500 | 0.0154300 |
| w_omega_in_uncertainty | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0003924 | 0.0002958 | 0.0002980 | 0.0002911 | 0.0003897 |
| w_omega_out | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0128176 | 0.0142168 | 0.0147704 | 0.0122609 | 0.0136732 |
| w_omega_out_uncertainty | [ $\mathrm{kg} / \mathrm{kg}]$ | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 |
| condensate | [g/s] | 0.0726334 | 0.0837309 | 0.0879125 | 0.0820765 | 0.1042150 |
| condenate_uncertainty | [g/s] | 0.8108259 | 0.0966548 | 0.8189490 | 0.6820596 | 1.4835172 |
| Case \# |  | 6 | 7 | 8 | 9 |  |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.60 | 26.50 | 26.51 | 26.80 |  |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 22.04 | 18.58 | 20.72 | 21.77 |  |
| RH_air_in | [\%] | 71.8\% | 70.1\% | 70.9\% | 69.7\% |  |
| RH_air_out | [\%] | 88.6\% | 90.7\% | 88.9\% | 87.7\% |  |
| P_air_in | [ kPa ] | 101.93 | 101.36 | 101.54 | 101.96 |  |
| DP_air | [Pa] | 279.10 | 51.00 | 158.80 | 279.90 |  |
| uncertainty_DP | [Pa] | 1.90 | 1.60 | 3.00 | 0.70 |  |
| SH | [W] | 418.88 | 210.96 | 344.82 | 459.09 |  |
| SH_un | [W] | 19.89 | 6.28 | 11.82 | 18.63 |  |
| LH | [W] | 301.76 | 218.87 | 294.30 | 314.11 |  |
| LH_un | [W] | 0.19 | 0.67 | 0.48 | 0.96 |  |
| MFR_air | [g/s] | 91.32 | 26.45 | 59.13 | 90.59 |  |
| VFR_air | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | 0.08 | 0.02 | 0.05 | 0.08 |  |
| Capacity_air | [W] | 720.64 | 429.82 | 639.12 | 773.20 |  |


| Capacity_air_uncertainty | [W] | 19.89 | 6.32 | 11.83 | 18.65 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 11.77 | 11.93 | 11.92 | 11.90 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 16.60 | 13.98 | 14.84 | 15.46 |
| P_water in | [ kPa ] | 100.62 | 102.49 | 102.71 | 103.06 |
| DP_water | [ kPa ] | 5.20 | 7.20 | 7.20 | 7.20 |
| MFR_water | [g/s] | 34.85 | 50.51 | 50.47 | 50.53 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 995.97 | 995.57 | 995.75 | 996.04 |
| Capacity_water | [W] | 703.30 | 434.70 | 618.10 | 754.10 |
| Capacity_water_uncertainty | [W] | 13.28 | 16.44 | 19.02 | 17.10 |
| Energy balance | [\%] | -2.43 | 1.13 | -3.34 | -2.50 |
| Average capacity | [W] | 711.97 | 432.26 | 628.61 | 763.65 |
| Capacity uncertainty | [W] | 11.96 | 8.81 | 11.20 | 12.65 |
| w_omega_in | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0156600 | 0.0152500 | 0.0154300 | 0.0153500 |
| w_omega_in_uncertainty | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0002963 | 0.0002895 | 0.0002925 | 0.0002910 |
| w_omega_out | [ $\mathrm{kg} / \mathrm{kg}]$ | 0.0143238 | 0.0119035 | 0.0134171 | 0.0139477 |
| w_omega_out_uncertainty | [ $\mathrm{kg} / \mathrm{kg}$ ] | 0.0000000 | 0.0000000 | 0.0000000 | 0.0000000 |
| condensate | [g/s] | 0.1220219 | 0.0885162 | 0.1190218 | 0.1270322 |
| condenate_uncertainty | [g/s] | 0.1879926 | 0.6655276 | 0.4795712 | 0.9630440 |

## 1 kW-MCHX test data

Table C. 121 kW - MCHX dry condition test data

| Case\# |  | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 35.02 | 34.94 | 34.89 | 34.92 | 34.88 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 50.42 | 47.50 | 45.74 | 51.81 | 49.25 |
| RH_air_in | $[\%]$ | $37.1 \%$ | $36.8 \%$ | $39.5 \%$ | $37.8 \%$ | $35.8 \%$ |
| RH_air_out | $[\%]$ | $16.0 \%$ | $17.4 \%$ | $20.3 \%$ | $15.3 \%$ | $16.2 \%$ |
| P_air_in | $[\mathrm{kPa}]$ | 101.70 | 101.30 | 101.40 | 101.70 | 101.40 |
| DP_air | $[\mathrm{Pa}]$ | 17.60 | 32.50 | 52.00 | 17.50 | 32.80 |
| uncertainty_DP | $[\mathrm{Pa}]$ | 1.00 | 1.20 | 1.70 | 1.00 | 1.30 |


| MFR_air | [ $\mathrm{g} / \mathrm{s}$ ] | 37.43 | 57.05 | 78.21 | 37.30 | 57.18 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| VFR_air | [m³/s] | 0.03 | 0.05 | 0.07 | 0.03 | 0.05 |
| Capacity_air | [W] | 580.50 | 722.00 | 854.50 | 634.70 | 827.70 |
| Capacity_air_uncertainty | [W] | 11.50 | 12.70 | 12.30 | 11.40 | 13.20 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 60.02 | 59.96 | 59.97 | 60.01 | 59.96 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 55.54 | 54.19 | 53.13 | 57.08 | 55.99 |
| P_water in | [ kPa ] | 100.10 | 99.70 | 99.70 | 106.40 | 106.10 |
| DP_water | [ kPa ] | 5.20 | 5.30 | 5.20 | 11.00 | 11.00 |
| MFR_water | [ $\mathrm{g} / \mathrm{s}$ ] | 30.10 | 30.40 | 30.20 | 50.10 | 49.90 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 988.90 | 989.10 | 989.70 | 988.20 | 988.40 |
| Capacity_water | [W] | 567.20 | 736.60 | 872.00 | 608.30 | 835.40 |
| Capacity_water_uncertainty | [W] | 10.90 | 12.60 | 9.60 | 21.50 | 25.60 |
| Energy balance | [\%] | -2.32 | 2.00 | 2.03 | -4.25 | 0.93 |
| Average capacity | [W] | 573.85 | 729.30 | 863.25 | 621.50 | 831.55 |
| capacity uncertainty | [W] | 7.92 | 8.94 | 7.80 | 12.17 | 14.40 |
| Case \# |  | 6 | 7 | 8 | 9 |  |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 35.05 | 34.94 | 34.94 | 35.03 |  |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 47.09 | 52.73 | 49.88 | 48.41 |  |
| RH_air_in | [\%] | 34.8\% | 37.7\% | 37.7\% | 36.9\% |  |
| RH_air_out | [\%] | 16.7\% | 14.7\% | 15.8\% | 16.8\% |  |
| P_air_in | [ kPa ] | 101.70 | 101.70 | 101.80 | 101.80 |  |
| DP_air | [Pa] | 53.80 | 17.70 | 32.90 | 53.70 |  |
| uncertainty_DP | [Pa] | 1.70 | 1.00 | 1.20 | 1.80 |  |
| MFR_air | [g/s] | 79.89 | 37.56 | 56.80 | 79.04 |  |
| VFR_air | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | 0.07 | 0.03 | 0.05 | 0.07 |  |
| Capacity_air | [W] | 969.10 | 673.10 | 854.60 | 1065.10 |  |
| Capacity_air_uncertainty | [W] | 17.50 | 12.90 | 12.90 | 15.90 |  |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 59.94 | 60.10 | 60.04 | 60.03 |  |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 55.16 | 57.90 | 57.07 | 56.38 |  |


| P_water in | $[\mathrm{kPa}]$ | 106.70 | 115.90 | 116.00 | 116.00 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| DP_water | $[\mathrm{kPa}]$ | 11.20 | 19.70 | 19.70 | 19.60 |
| MFR_water | $[\mathrm{g} / \mathrm{s}]$ | 50.00 | 70.40 | 70.60 | 70.30 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 989.20 | 987.80 | 988.10 | 988.40 |
| Capacity_water | $[\mathrm{W}]$ | 983.80 | 647.50 | 856.60 | 1059.00 |
| Capacity_water_uncertainty | $[\mathrm{W}]$ | 22.30 | 24.10 | 24.20 | 24.10 |
| Energy balance | $[\%]$ | 1.51 | -3.88 | 0.23 | -0.57 |
| Average capacity | $[\mathrm{W}]$ | 976.45 | 660.30 | 855.60 | 1062.05 |
| capacity uncertainty | $[\mathrm{W}]$ | 14.17 | 13.67 | 13.71 | 14.44 |

## 10 kW -Copper BTHX geometry and test data

Though the 10 kW copper BTHX has blockage issues, it is still tested under dry condition.
The dimensions and test results are shown below.


Figure C. 2 10kW-copper BTHX geometry
Table C. 13 10kW-copper BTHX dimensions

| Material | Copper |  |
| :---: | :---: | :---: |
| Geometry | Length x width x depth | $444 \times 530 \times 71 \mathrm{~mm}$ |
|  | Frontal area | $0.235 \mathrm{~m}^{2}$ |
|  | Air heat transfer area | $2.515 \mathrm{~m}^{2}$ |
|  | Water heat transfer area | $2.012 \mathrm{~m}^{2}$ |

Table C. 1410 kW -copper BTHX test matrix

| Air | Inlet air temperature | $30.0 \pm 0.3$ | ${ }^{\circ} \mathrm{C}$ |
| :---: | :---: | :---: | :---: |
|  | Inlet air RH | $6 \pm 1$ | $\%$ |
|  | Air flow rate | $0.16 \pm 0.0016$ | $\mathrm{~m}^{3} / \mathrm{s}$ |
|  |  | $0.31 \pm 0.0031$ | $\mathrm{~m}^{3} / \mathrm{s}$ |
|  |  | $0.47 \pm 0.0047$ | $\mathrm{~m}^{3} / \mathrm{s}$ |
| Inlet water <br> temperature | $55.0 \pm 0.6$ | ${ }^{\circ} \mathrm{C}$ |  |
|  |  | Water mass flow rate | $71 \pm 0.71$ |
|  |  | $95 \pm 0.95$ | $\mathrm{~g} / \mathrm{s}$ |
|  |  | $118 \pm 1.18$ | $\mathrm{~g} / \mathrm{s}$ |
|  |  | $141 \pm 1.41$ | $\mathrm{~g} / \mathrm{s}$ |
|  |  | $165 \pm 1.65$ | $\mathrm{~g} / \mathrm{s}$ |



Figure C. 3 Capacity of 10 kW -copper BTHX


Figure C. 4 Energy balance of 10 kW -copper BTHX


Figure C. 5 Airside pressure drop of 10 kW -copper BTHX
Table C. 1510 kW -copper BTHX dry condition test data

| Case\# |  | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 30.10 | 30.20 | 30.10 | 30.10 | 30.20 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 43.77 | 44.63 | 45.16 | 45.47 | 45.79 |
| RH_air_in | $[\%]$ | $7.1 \%$ | $7.3 \%$ | $7.6 \%$ | $8.0 \%$ | $6.8 \%$ |


| RH_air_out | [\%] | 7.3\% | 7.3\% | 7.4\% | 7.5\% | 3.3\% |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| P_air_in | [ kPa ] | 101.6 | 101.5 | 101.5 | 101.5 | 101.4 |
| DP_air | [Pa] | 10.1 | 10.1 | 10.1 | 10.1 | 10.1 |
| uncertainty_DP | [Pa] | 0.7 | 0.7 | 0.7 | 0.7 | 0.7 |
| MFR_air | [g/s] | 208.30 | 207.40 | 207.30 | 207.50 | 207.50 |
| VFR_air | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | 0.1857 | 0.1856 | 0.1857 | 0.1863 | 0.1863 |
| Capacity_air | [W] | 2867.2 | 3013.8 | 3143.7 | 3212.5 | 3257.7 |
| Capacity_air_uncertainty | [W] | 41.5 | 42.5 | 39.2 | 44.0 | 44.2 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 54.98 | 54.98 | 55.01 | 54.94 | 55.01 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 45.29 | 47.36 | 48.67 | 49.54 | 50.31 |
| P_water in | [ kPa ] | 114.5 | 118.4 | 122.6 | 127.4 | 132.7 |
| DP_water | [ kPa ] | 10.0 | 10.0 | 10.0 | 10.0 | 10.0 |
| MFR_water | [g/s] | 70.6 | 94.5 | 117.8 | 141.4 | 164.9 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 992.4 | 991.9 | 991.4 | 991.1 | 990.9 |
| Capacity_water | [W] | 2861.0 | 3008.0 | 3122.0 | 3191.0 | 3240.0 |
| Capacity_water_uncertainty | [W] | 22.3 | 33.7 | 42.1 | 49.0 | 50.0 |
| Energy balance | [\%] | (0.22) | (0.19) | (0.69) | (0.67) | (0.54) |
| Average capacity | [W] | 2864.1 | 3010.9 | 3132.8 | 3201.8 | 3248.9 |
| Capacity uncertainty | [W] | 23.5 | 27.1 | 28.7 | 32.9 | 33.4 |
| Case \# |  | 6 | 7 | 8 | 9 | 10 |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 30.10 | 30.00 | 30.10 | 30.00 | 30.20 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 40.27 | 41.07 | 41.78 | 42.22 | 42.53 |
| RH_air_in | [\%] | 6.5\% | 6.5\% | 6.5\% | 6.5\% | 6.6\% |
| RH_air_out | [\%] | 7.3\% | 3.7\% | 6.0\% | 3.6\% | 4.8\% |
| P_air_in | [ kPa ] | 101.8 | 101.3 | 102.0 | 101.4 | 102.1 |
| DP_air | [Pa] | 24.2 | 24.0 | 24.2 | 24.1 | 24.2 |
| uncertainty_DP | [Pa] | 0.8 | 0.8 | 0.8 | 0.8 | 0.8 |
| MFR_air | [g/s] | 362.60 | 359.30 | 361.70 | 359.20 | 360.90 |
| VFR_air | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | 0.3199 | 0.3193 | 0.3198 | 0.3202 | 0.3197 |


| Capacity_air | [W] | 3713.4 | 4005.6 | 4253.9 | 4418.3 | 4480.6 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Capacity_air_uncertainty | [W] | 59.6 | 60.6 | 61.8 | 59.4 | 62.7 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 54.89 | 54.95 | 54.97 | 55.10 | 54.97 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 42.08 | 44.57 | 46.19 | 47.52 | 48.40 |
| P_water in | [ kPa ] | 114.9 | 118.5 | 123.3 | 127.7 | 133.9 |
| DP_water | [ kPa ] | 10.0 | 10.0 | 10.0 | 10.0 | 10.0 |
| MFR_water | [g/s] | 70.6 | 94.5 | 117.4 | 141.2 | 166.4 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 993.3 | 992.5 | 992.4 | 991.8 | 991.9 |
| Capacity_water | [W] | 3779.0 | 4099.0 | 4311.0 | 4475.0 | 4575.0 |
| Capacity_water_uncertainty | [W] | 28.3 | 30.4 | 36.6 | 48.6 | 57.6 |
| Energy balance | [\%] | 1.75 | 2.30 | 1.33 | 1.27 | 2.08 |
| Average capacity | [W] | 3746.2 | 4052.3 | 4282.5 | 4446.7 | 4527.8 |
| Capacity uncertainty | [W] | 33.0 | 33.9 | 35.9 | 38.4 | 42.6 |
| Case \# |  | 11 | 12 | 13 | 14 | 15 |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 30.00 | 30.00 | 30.10 | 30.00 | 30.00 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 37.81 | 38.62 | 39.27 | 39.56 | 40.00 |
| RH_air_in | [\%] | 5.2\% | 6.4\% | 5.1\% | 6.4\% | 5.2\% |
| RH_air_out | [\%] | 7.1\% | 4.2\% | 6.3\% | 4.1\% | 5.5\% |
| P_air_in | [ kPa ] | 102.4 | 101.3 | 102.4 | 101.3 | 102.3 |
| DP_air | [Pa] | 47.1 | 47.1 | 47.0 | 47.2 | 47.0 |
| uncertainty_DP | [Pa] | 0.9 | 0.9 | 0.9 | 0.9 | 0.9 |
| MFR_air | [g/s] | 548.80 | 545.20 | 547.60 | 544.70 | 547.00 |
| VFR_air | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | 0.4786 | 0.4814 | 0.4795 | 0.4819 | 0.4800 |
| Capacity_air | [W] | 4314.2 | 4731.0 | 5057.2 | 5245.6 | 5509.3 |
| Capacity_air_uncertainty | [W] | 87.4 | 94.1 | 90.5 | 97.5 | 99.5 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 55.08 | 54.77 | 54.95 | 54.72 | 54.82 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 39.99 | 42.49 | 44.38 | 45.60 | 46.76 |
| P_water in | [ kPa ] | 115.7 | 118.8 | 124.0 | 127.9 | 134.7 |
| DP_water | [ kPa ] | 10.0 | 10.0 | 10.0 | 10.0 | 10.0 |
| 236 |  |  |  |  |  |  |


| MFR_water | $[\mathrm{g} / \mathrm{s}]$ | 70.9 | 94.7 | 117.9 | 141.0 | 166.3 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 993.6 | 993.0 | 992.6 | 992.3 | 991.9 |
| Capacity_water | $[\mathrm{W}]$ | 4478.0 | 4859.0 | 5210.0 | 5377.0 | 5608.0 |
| Capacity_water_uncertainty | $[\mathrm{W}]$ | 33.4 | 36.0 | 40.9 | 65.2 | 62.1 |
| Energy balance | $[\%]$ | 3.73 | 2.67 | 2.98 | 2.47 | 1.78 |
| Average capacity | $[\mathrm{W}]$ | 4396.1 | 4795.0 | 5133.6 | 5311.3 | 5558.6 |
| Capacity uncertainty | $[\mathrm{W}]$ | 46.8 | 50.4 | 49.6 | 58.6 | 58.6 |

## 10 kW -louvered fin radiator test data

This radiator is tested to compare with the 10 kW copper BTHX. The dimensions and test results are shown below. However, since the 10 kW copper BTHX has blockage issues, there is no comparison between these two heat exchangers.


Figure C. 610 kW-louvered fin radiator
Table C. 1610 kW -louvered fin radiator dimensions

| Material | Aluminum |  |
| :---: | :---: | :---: |
| Geometry | Length x width x depth | $457 \times 419 \times 35 \mathrm{~mm}$ |
|  | Frontal area | $0.192 \mathrm{~m}^{2}$ |
|  | Air heat transfer area | $7.844 \mathrm{~m}^{2}$ |
|  | Water heat transfer area | $1.147 \mathrm{~m}^{2}$ |

Table C. 1710 kW -louvered fin radiator test matrix

| Air | Inlet air temperature | $35.0 \pm 0.3$ | ${ }^{\circ} \mathrm{C}$ |
| :---: | :---: | :---: | :---: |
|  | Inlet air RH | $35 \pm 1$ | $\%$ |
|  | Air flow rate | $0.192 \pm 0.00192$ | $\mathrm{~m}^{3} / \mathrm{s}$ |
|  |  | $0.479 \pm 0.00479$ | $\mathrm{~m}^{3} / \mathrm{s}$ |
|  |  | $0.766 \pm 0.00766$ | $\mathrm{~m}^{3} / \mathrm{s}$ |
| Water | Inlet water <br> temperature | $60.0 \pm 0.6$ | ${ }^{\circ} \mathrm{C}$ |
|  | Water mass flow |  |  |
|  | Wate | $161 \pm 1.61$ | $\mathrm{~g} / \mathrm{s}$ |
|  |  | $279 \pm 2.79$ | $\mathrm{~g} / \mathrm{s}$ |



Figure C. 7 Energy balance of 10 kW -louvered fin radiator


Figure C. 8 Capacity of 10 kW -louvered fin radiator


Figure C. $9 \eta_{0} h_{0}$ of 10 kW -louvered fin radiator


Figure C. 10 Airside pressure drop of 10 kW -louvered fin radiator
Table C. 1810 kW -louvered fin radiator dry condition test data

| Case \# |  | 1 | 2 | 3 | 4 | 5 | 6 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 35.10 | 35.02 | 35.02 | 35.12 | 35.12 | 35.03 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 54.04 | 49.02 | 45.54 | 54.78 | 49.98 | 46.68 |
| RH_air_in | $[\%]$ | $34.0 \%$ | $38.3 \%$ | $37.8 \%$ | $33.1 \%$ | $34.4 \%$ | $37.8 \%$ |
| RH_air_out | $[\%]$ | $13.6 \%$ | $18.8 \%$ | $21.8 \%$ | $12.9 \%$ | $16.6 \%$ | $20.7 \%$ |
| P_air_in | $[\mathrm{kPa}]$ | 101.27 | 100.84 | 100.86 | 101.23 | 101.30 | 100.88 |
| DP_air | $[\mathrm{Pa}]$ | 10.85 | 51.97 | 110.19 | 10.79 | 53.41 | 110.55 |
| uncertainty_DP | $[\mathrm{Pa}]$ | 0.77 | 1.01 | 1.28 | 0.76 | 1.04 | 1.34 |
| MFR_air | $[\mathrm{g} / \mathrm{s}]$ | 206.13 | 514.72 | 842.95 | 205.58 | 524.72 | 842.61 |
| VFR_air | $[\mathrm{m} / \mathrm{s}]$ | 0.19 | 0.47 | 0.77 | 0.19 | 0.48 | 0.77 |
| Capacity_air | $[\mathrm{W}]$ | 3933.62 | 7256.42 | 8928.25 | 4071.29 | 7850.90 | 9886.64 |
| Capacity_air_uncertainty | $[\mathrm{W}]$ | 50.93 | 109.10 | 123.80 | 50.43 | 121.40 | 138.30 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 59.90 | 59.91 | 59.91 | 60.00 | 60.03 | 59.94 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 53.90 | 49.12 | 46.81 | 55.48 | 51.49 | 49.45 |
| P_water in | $[\mathrm{kPa}]$ | 114.98 | 112.58 | 112.57 | 122.77 | 121.12 | 121.11 |
| DP_water | $[\mathrm{kPa}]$ | 7064.67 | 6948.75 | 6905.76 | 8891.39 | 8856.38 | 8821.11 |


| MFR_water | [g/s] | 160.41 | 160.18 | 159.53 | 220.03 | 220.64 | 220.91 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 988.36 | 990.16 | 990.87 | 987.34 | 989.75 | 990.10 |
| Capacity_water | [W] | 4024.98 | 7234.03 | 8738.80 | 4140.85 | 7842.69 | 9607.21 |
| Capacity_water_uncertainty | [W] | 66.65 | 59.28 | 64.58 | 113.90 | 77.38 | 92.15 |
| Energy balance | [\%] | 2.30 | -0.31 | -2.14 | 1.69 | -0.10 | -2.87 |
| Average capacity | [W] | 3979.30 | 7245.23 | 8833.52 | 4106.07 | 7846.80 | 9746.93 |
| capacity uncertainty | [W] | 41.94 | 62.08 | 69.82 | 62.28 | 71.98 | 83.09 |
| Case \# |  | 7 | 8 | 9 | 10 | 11 | 12 |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 35.10 | 35.01 | 35.03 | 35.09 | 35.13 | 34.96 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 55.33 | 50.79 | 47.45 | 55.55 | 51.31 | 47.95 |
| RH_air_in | [\%] | 31.3\% | 33.1\% | 35.9\% | 30.9\% | 33.5\% | 33.9\% |
| RH_air_out | [\%] | 12.2\% | 15.1\% | 19.2\% | 12.1\% | 15.0\% | 17.8\% |
| P_air_in | [ kPa ] | 101.20 | 101.32 | 100.94 | 101.17 | 101.31 | 101.02 |
| DP_air | [Pa] | 10.75 | 53.79 | 111.03 | 10.74 | 54.31 | 111.34 |
| uncertainty_DP | [Pa] | 0.76 | 0.97 | 1.29 | 0.76 | 0.96 | 1.27 |
| MFR_air | [g/s] | 205.33 | 527.10 | 844.53 | 205.09 | 529.94 | 844.74 |
| VFR_air | [m³/s] | 0.19 | 0.48 | 0.77 | 0.19 | 0.49 | 0.77 |
| Capacity_air | [W] | 4183.32 | 8378.80 | 10564.07 | 4227.02 | 8638.21 | 11058.59 |
| Capacity_air_uncertainty | [W] | 50.50 | 116.90 | 124.10 | 50.61 | 113.60 | 126.70 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 60.07 | 60.04 | 59.92 | 59.98 | 59.96 | 59.90 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 56.38 | 52.87 | 51.01 | 56.83 | 53.72 | 52.14 |
| P_water in | [ kPa ] | 133.04 | 131.56 | 131.68 | 143.87 | 142.26 | 144.17 |
| DP_water | [ kPa ] | 11154.61 | 11285.66 | 11269.44 | 13799.13 | 13866.52 | 14206.59 |
| MFR_water | [g/s] | 279.62 | 280.24 | 279.18 | 331.96 | 332.29 | 336.32 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 986.63 | 988.77 | 989.81 | 985.21 | 987.75 | 989.59 |
| Capacity_water | [W] | 4326.86 | 8320.68 | 10390.34 | 4442.63 | 8754.52 | 10970.03 |
| Capacity_water_uncertainty | [W] | 86.42 | 115.70 | 101.30 | 102.50 | 109.20 | 136.90 |
| Energy balance | [\%] | 3.37 | -0.70 | -1.66 | 4.97 | 1.34 | -0.80 |
| Average capacity | [W] | 4255.09 | 8349.74 | 10477.21 | 4334.82 | 8696.36 | 11014.31 |


| Capacity uncertainty | $[\mathrm{W}]$ | 50.05 | 82.24 | 80.10 | 57.16 | 78.79 | 93.27 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |

$4 \& 5 \mathrm{~mm}$ slit fin-and-tube HX test matrix (air-water wet condition, condenser and evaporator condition)

Table C. 194 and 5 mm slit fin-and-tube HX wet condition test matrix

| Test | Surface | Inlet Air <br> Temperature | Inlet Air RH | Inlet Water <br> Temperature | Air Velocity | Air Flow Rate | Water Flow Rate |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | $\left[{ }^{\circ} \mathrm{C}\right]$ | [\%] | $\left[{ }^{\circ} \mathrm{C}\right]$ | [m/s] | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | [g/s] |
| 1 | Dry | 26.7 | 50 | 7.2 | 1 | 0.15 | 50 |
| 2 | Dry | 26.7 | 50 | 7.2 | 1 | 0.15 | 100 |
| 3 | Dry | 26.7 | 50 | 7.2 | 1 | 0.15 | 150 |
| 4 | Dry | 26.7 | 50 | 7.2 | 2.5 | 0.375 | 50 |
| 5 | Dry | 26.7 | 50 | 7.2 | 2.5 | 0.375 | 100 |
| 6 | Dry | 26.7 | 50 | 7.2 | 2.5 | 0.375 | 150 |
| 7 | Dry | 26.7 | 50 | 7.2 | 4 | 0.6 | 50 |
| 8 | Dry | 26.7 | 50 | 7.2 | 4 | 0.6 | 100 |
| 9 | Dry | 26.7 | 50 | 7.2 | 4 | 0.6 | 150 |

## 5mm slit fin-and-tube HX test data

Table C. 205 mm slit fin-and-tube HX condenser test data (air and R410A)


|  | Ref-side | 8.4 | $\pm$ | 0.1 | 2.9 | $\pm$ | 8.2 | 1518.073233 |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4 | Air-Side | 440.6 | $\pm$ | 3.0 | 27.3 | $\pm$ | 0.9 | 845.1 | $\pm$ | 63.2 | 189.8 | 1035 | $\pm$ | 63.17 | 6.10\% |
|  | Ref-side | 7.2 | $\pm$ | 0.1 | 2.9 | $\pm$ | 6.7 | 1329.585558 |  |  |  |  |  |  |  |
| 5 | Air-Side | 440.2 | $\pm$ | 3.0 | 27.3 | $\pm$ | 0.9 | 622.4 | $\pm$ | 63.0 | 182.8 | 805.2 | $\pm$ | 63.0 | 7.83\% |
|  | Ref-side | 7.2 | $\pm$ | 0.1 | 2.4 | $\pm$ | 5.3 | 1123.213095 |  |  |  |  |  |  |  |

Table C. 215 mm slit fin-and-tube HX evaporator test (air and R410A)

|  |  | Tin |  |  | Tout |  |  | Pin_gage |  |  | Pout_gage |  |  | T_in_sat |  |  | T_out_sat |  |  | Qualityin |  |  | RHin |  |  | RHout |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | $\left[{ }^{\circ} \mathrm{C}\right]$ |  |  | $\left[{ }^{\circ} \mathrm{C}\right]$ |  |  | [ kPa ] |  |  | [ kPa ] |  |  | $\left[{ }^{\circ} \mathrm{C}\right]$ |  |  | $\left[{ }^{\circ} \mathrm{C}\right]$ |  |  | [-] |  |  | [\%] |  |  | [\%] |  |  |
| 1 | Air | 26.80 | $\pm$ | 0.10 | 23.30 | $\pm$ | 0.10 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 50.9\% | $\pm$ | 1.7\% | 60.0\% | $\pm$ | 1.6\% |
|  | Ref | 9.17 | $\pm$ | 0.40 | 22.22 | $\pm$ | 0.29 | 962.5 | $\pm$ | 15.8 | 951.6 | $\pm$ | 14.4 | 9.4 | $\pm$ | 0.5 | 9.0 | $\pm$ | 0.5 | 0.28 | $\pm$ | 0.05 |  | - |  |  | - |  |
| 2 | Air | 26.70 | $\pm$ | 0.10 | 23.60 | $\pm$ | 0.20 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 70.5\% | $\pm$ | 1.4\% | 79.3\% | $\pm$ | 1.8\% |
|  | Ref | 9.70 | $\pm$ | 0.51 | 22.32 | $\pm$ | 0.40 | 979.9 | $\pm$ | 20.1 | 967.9 | $\pm$ | 18.3 | 9.9 | $\pm$ | 0.6 | 9.5 | $\pm$ | 0.6 | 0.26 | $\pm$ | 0.05 |  | - |  |  | - |  |
| 3 | Air | 26.80 | $\pm$ | 0.10 | 23.10 | $\pm$ | 0.10 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 50.3\% | $\pm$ | 1.3\% | 61.1\% | $\pm$ | 1.6\% |
|  | Ref | 10.32 | $\pm$ | 0.30 | 21.23 | $\pm$ | 0.33 | 998.7 | $\pm$ | 12.0 | 985.6 | $\pm$ | 11.3 | 10.5 | $\pm$ | 0.4 | 10.1 | $\pm$ | 0.4 | 0.35 | $\pm$ | 0.02 |  | - |  |  | - |  |
| 4 | Air | 26.90 | $\pm$ | 0.10 | 23.90 | $\pm$ | 0.10 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 70.2\% | $\pm$ | 2.2\% | 79.3\% | $\pm$ | 2.0\% |
|  | Ref | 10.24 | $\pm$ | 0.44 | 22.64 | $\pm$ | 0.30 | 996.2 | $\pm$ | 17.8 | 982.8 | $\pm$ | 16.2 | 10.4 | $\pm$ | 0.5 | 10.0 | $\pm$ | 0.5 | 0.36 | $\pm$ | 0.03 |  | - |  |  | - |  |
| 5 | Air | 26.80 | $\pm$ | 0.10 | 22.10 | $\pm$ | 0.10 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 50.3\% | $\pm$ | 1.3\% | 64.1\% | $\pm$ | 1.7\% |
|  | Ref | 10.61 | $\pm$ | 0.12 | 16.62 | $\pm$ | 0.70 | 1008.6 | $\pm$ | 5.4 | 991.9 | $\pm$ | 5.3 | 10.8 | $\pm$ | 0.2 | 10.3 | $\pm$ | 0.2 | 0.27 | $\pm$ | 0.01 |  | - |  |  | - |  |
| 6 | Air | 26.80 | $\pm$ | 0.10 | 22.90 | $\pm$ | 0.20 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 67.6\% | $\pm$ | 3.4\% | 79.2\% | $\pm$ | 3.1\% |
|  | Ref | 10.65 | $\pm$ | 0.29 | 20.63 | $\pm$ | 0.47 | 1010.4 | $\pm$ | 12.4 | 993.1 | $\pm$ | 11.4 | 10.8 | $\pm$ | 0.4 | 10.3 | $\pm$ | 0.4 | 0.27 | $\pm$ | 0.01 |  |  |  |  |  |  |
| 7 | Air | 26.80 | $\pm$ | 0.10 | 22.60 | $\pm$ | 0.10 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 49.6\% | $\pm$ | 1.3\% | 62.2\% | $\pm$ | 1.6\% |
|  | Ref | 10.64 | $\pm$ | 0.10 | 17.92 | $\pm$ | 0.60 | 1009.0 | $\pm$ | 4.4 | 992.6 | $\pm$ | 4.4 | 10.8 | $\pm$ | 0.1 | 10.3 | $\pm$ | 0.1 | 0.37 | $\pm$ | 0.01 |  |  |  |  |  |  |
| 8 | Air | 26.80 | $\pm$ | 0.10 | 23.30 | $\pm$ | 0.30 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 67.2\% | $\pm$ | 5.0\% | 78.3\% | $\pm$ | 4.6\% |
|  | Ref | 10.65 | $\pm$ | 0.22 | 21.49 | $\pm$ | 0.64 | 1009.7 | $\pm$ | 8.2 | 992.3 | $\pm$ | 7.8 | 10.8 | $\pm$ | 0.3 | 10.3 | $\pm$ | 0.2 | 0.37 | $\pm$ | 0.02 |  |  |  |  |  |  |
| Test No. |  | Volume Flow Rate |  |  | Mass Flow Rate |  |  | Air/Ref Pressure Drop |  |  | Capacity |  |  |  |  | $\begin{aligned} & \text { eat } \\ & \text { oss } \end{aligned}$ | SH/LH |  |  |  | Adjusted Capacity |  |  | Uncertainty |  |  | Energy Balance |  |


| 1 |  | [m3/s] |  |  | [ $\mathrm{g} / \mathrm{s}$ ] |  |  | [ Pa ] / [kPa] |  |  | [W] |  |  | [W]$-53.7$ | [W]/[W] |  |  | [W] |  |  | [\%] <br> 4.24\% | [\%]$1.0 \%$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Air | 0.387 | $\pm$ | 1E-04 | 461.9 | $\pm$ | 3.4 | 31.1 | $\pm$ | 0.9 | 1657.0 | $\pm$ | 68.1 |  | 1603.3 | $\pm$ | 68.1 | 1603.3 | $\pm$ | 68.1 |  |  |
|  | Ref | 2.581404735 |  |  | 9.7 | $\pm$ | 1.9 | 10.9 | $\pm$ | 21.3 | 1619.0 | $\pm$ | 425.7 | - | 0.0 | $\pm$ | 0.0 | 1619.0 | $\pm$ | 425.7 | 26.29\% |  |
| 2 | Air | 0.383 | $\pm$ | 2E-04 | 454.9 | $\pm$ | 3.0 | 34.4 | $\pm$ | 1.0 | 1456.0 | $\pm$ | 105.5 | -46.9 | 1409.1 | $\pm$ | 105.5 | 1736.1 | $\pm$ | 105.5 | 6.08\% |  |
|  | Ref | 2.550591377 |  |  | 10.4 | $\pm$ | 1.9 | 12.0 | $\pm$ | 27.2 | 1755.0 | $\pm$ | 415.9 | - | 327.0 | $\pm$ | 1.2 | 1755.0 | $\pm$ | 415.9 | 23.70\% |  |
| 3 | Air | 0.384 | $\pm$ | 2E-04 | 456.7 | $\pm$ | 3.4 | 31.2 | $\pm$ | 0.9 | 1732.0 | $\pm$ | 67.5 | -47.2 | 1684.8 | $\pm$ | 67.5 | 1694.4 | $\pm$ | 67.5 | 3.98\% |  |
|  | Ref | 2.554993286 |  |  | 11.2 | $\pm$ | 0.7 | 13.1 | $\pm$ | 16.5 | 1658.0 | $\pm$ | 160.8 | - | 9.6 | $\pm$ | 0.2 | 1658.0 | $\pm$ | 160.8 | 9.70\% |  |
| 4 | Air | 0.384 | $\pm$ | 1E-04 | 455.3 | $\pm$ | 3.0 | 34.2 | $\pm$ | 1.0 | 1411.0 | $\pm$ | 67.2 | -39.2 | 1371.8 | $\pm$ | 67.2 | 1662.6 | $\pm$ | 67.2 | 4.04\% |  |
|  | Ref | 2.558137506 |  |  | 11.0 | $\pm$ | 0.9 | 13.4 | $\pm$ | 24.1 | 1623.0 | $\pm$ | 212.8 | - | 290.8 | $\pm$ | 2.9 | 1623.0 | $\pm$ | 212.8 | 13.11\% |  |
| 5 | Air | 0.381 | $\pm$ | 3E-04 | 454.8 | $\pm$ | 3.9 | 31.4 | $\pm$ | 0.9 | 2191.0 | $\pm$ | 68.5 | -64.5 | 2126.5 | $\pm$ | 68.5 | 2133.8 | $\pm$ | 68.5 | $3.21 \%$ |  |
|  | Ref | 2.538014497 |  |  | 13.4 | $\pm$ | 0.4 | 16.7 | $\pm$ | 7.5 | 2138.0 | $\pm$ | 95.7 | - | 7.3 | $\pm$ | 0.1 | 2138.0 | $\pm$ | 95.7 | 4.47\% |  |
| 6 | Air | 0.38 | $\pm$ | 2E-04 | 450.7 | $\pm$ | 3.1 | 35.7 | $\pm$ | 1.1 | 1813.0 | $\pm$ | 104.7 | -49.4 | 1763.6 | $\pm$ | 104.7 | 2238.5 | $\pm$ | 105.0 | 4.69\% |  |
|  | Ref | 2.532354901 |  |  | 13.4 | $\pm$ | 0.6 | 17.3 | $\pm$ | 16.9 | 2185.0 | $\pm$ | 128.1 | - | 474.8 | $\pm$ | 7.9 | 2185.0 | $\pm$ | 128.1 | 5.86\% |  |
| 7 | Air | 0.38 | $\pm$ | 3E-04 | 452.0 | $\pm$ | 3.1 | 31.6 | $\pm$ | 0.9 | 1945.0 | $\pm$ | 66.8 | -64.6 | 1880.4 | $\pm$ | 66.8 | 1888.9 | $\pm$ | 66.8 | 3.54\% |  |
|  | Ref | 2.534555855 |  |  | 13.0 | $\pm$ | 0.4 | 16.4 | $\pm$ | 6.3 | 1827.0 | $\pm$ | 93.9 | - | 8.5 | $\pm$ | 0.3 | 1827.0 | $\pm$ | 93.9 | 5.14\% |  |
| 8 | Air | 0.381 | $\pm$ | 3E-04 | 451.2 | $\pm$ | 3.1 | 35.3 | $\pm$ | 1.1 | 1629.0 | $\pm$ | 147.6 | -47.3 | 1581.7 | $\pm$ | 147.6 | 1925.8 | $\pm$ | 147.9 | 7.68\% |  |
|  | Ref | 2.538643341 |  |  | 13.1 | $\pm$ | 0.5 | 17.4 | $\pm$ | 11.3 | 1879.0 | $\pm$ | 120.3 |  | 344.1 | $\pm$ | 9.8 | 1879.0 | $\pm$ | 120.3 | 6.40\% |  |

Table C. 225 mm slit fin-and-tube HX dry condition (air and water)

| Case \# |  | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 35.10 | 35.00 | 35.00 | 35.00 | 35.10 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 46.56 | 41.17 | 39.34 | 49.82 | 43.96 |
| RH_air_in | [\%] | 29.4\% | 26.3\% | 26.1\% | 29.7\% | 28.6\% |
| RH_air_out | [\%] | 17.0\% | 19.0\% | 20.6\% | 14.9\% | 18.6\% |
| P_air_in | [ kPa ] | 102.24 | 102.13 | 102.13 | 102.23 | 101.54 |
| DP_air | [Pa] | 6.30 | 26.10 | 55.80 | 6.40 | 25.80 |
| uncertainty_DP | [Pa] | 0.70 | 0.90 | 1.50 | 0.70 | 0.90 |
| MFR_air | [g/s] | 176.20 | 431.50 | 679.40 | 175.40 | 423.30 |
| VFR_air | [m³/s] | 0.16 | 0.38 | 0.60 | 0.16 | 0.38 |
| Capacity_air | [W] | 2034.56 | 2679.98 | 2967.05 | 2619.09 | 3775.76 |
| Capacity_air_uncertainty | [W] | 33.96 | 100.20 | 103.80 | 38.66 | 65.22 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 60.01 | 60.01 | 60.01 | 59.92 | 59.88 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 50.06 | 46.88 | 45.51 | 53.62 | 50.79 |
| P_water in | [ kPa ] | 118.20 | 118.10 | 118.19 | 160.31 | 159.36 |
| DP_water | [ kPa ] | 16090.00 | 15140.00 | 15930.00 | 56200.00 | 56800.00 |
| MFR_water | [g/s] | 50.29 | 50.35 | 50.25 | 100.20 | 100.40 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 990.47 | 991.30 | 991.72 | 989.25 | 990.25 |
| Capacity_water | [W] | 2094.00 | 2764.00 | 3047.00 | 2640.00 | 3818.00 |


| Capacity_water_uncertainty | [W] | 28.12 | 23.03 | 18.41 | 41.50 | 35.20 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Energy balance | [\%] | 2.88 | 3.09 | 2.66 | 0.80 | 1.11 |
| Average capacity | [W] | 2064.28 | 2721.99 | 3007.02 | 2629.54 | 3796.88 |
| Capacity uncertainty | [W] | 22.05 | 51.41 | 52.71 | 28.36 | 37.06 |
| Case \# |  | 6 | 7 | 8 | 9 |  |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 35.00 | 35.00 | 35.10 | 35.10 |  |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 41.46 | 51.02 | 44.91 | 42.54 |  |
| RH_air_in | [\%] | 26.1\% | 29.9\% | 30.1\% | 25.8\% |  |
| RH_air_out | [\%] | 18.5\% | 14.2\% | 18.3\% | 17.5\% |  |
| P_air_in | [ kPa ] | 102.20 | 102.22 | 102.12 | 102.26 |  |
| DP_air | [Pa] | 56.30 | 6.40 | 25.80 | 56.50 |  |
| uncertainty_DP | [Pa] | 1.10 | 0.70 | 0.90 | 1.10 |  |
| MFR_air | [g/s] | 680.50 | 175.60 | 424.70 | 680.80 |  |
| VFR_air | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | 0.60 | 0.16 | 0.38 | 0.61 |  |
| Capacity_air | [W] | 4428.96 | 2833.74 | 4198.14 | 5099.22 |  |
| Capacity_air_uncertainty | [W] | 105.40 | 40.43 | 67.79 | 107.60 |  |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 59.95 | 60.04 | 60.02 | 60.05 |  |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 49.31 | 55.52 | 53.24 | 52.00 |  |
| P_water in | [ kPa ] | 159.97 | 225.72 | 226.50 | 226.50 |  |
| DP_water | [Pa] | 55680 | 119300 | 119800 | 119800 |  |


| MFR_water | $[\mathrm{g} / \mathrm{s}]$ | 99.87 | 149.90 | 150.40 | 150.00 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 990.62 | 988.64 | 989.45 | 989.91 |
| Capacity_water | $[\mathrm{W}]$ | 4442.00 | 2831.00 | 4265.00 | 5050.00 |
| Capacity_water_uncertainty | $[\mathrm{W}]$ | 34.95 | 56.78 | 57.56 | 57.85 |
| Energy balance | $[\%]$ | 0.29 | -0.10 | 1.58 | -0.97 |
| Average capacity | $[\mathrm{W}]$ | 4435.48 | 2832.37 | 4231.57 | 5074.61 |
| Capacity uncertainty | $[\mathrm{W}]$ | 55.52 | 34.85 | 44.47 | 61.08 |

Table C. 235 mm slit fin-and-tube HX wet condition test data (air and water)

| Case \# |  | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.70 | 26.70 | 26.70 | 26.80 | 26.70 |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 19.39 | 22.78 | 24.04 | 16.93 | 20.76 |
| RH_air_in | [\%] | 50.3\% | 51.2\% | 50.6\% | 50.9\% | 51.2\% |
| RH_air_out | [\%] | 75.3\% | 63.1\% | 57.9\% | 83.6\% | 69.9\% |
| P_air_in | [ kPa ] | 101.42 | 101.19 | 100.03 | 101.51 | 101.11 |
| DP_air | [Pa] | 7.00 | 27.90 | 57.60 | 8.50 | 29.60 |
| uncertainty_DP | [Pa] | 0.70 | 0.90 | 1.20 | 0.70 | 0.90 |
| SH | [W] | 1299.31 | 1744.03 | 1860.76 | 1790.39 | 2639.40 |
| SH_un | [W] | 33.07 | 51.77 | 75.05 | 26.17 | 54.42 |
| LH | [W] | 0.00 | 0.00 | 3.20 | 144.56 | 1.11 |
| LH_un | [W] | 0.00 | 0.00 | 0.06 | 0.26 | 0.05 |


| MFR_air | [g/s] | 176.70 | 441.80 | 695.90 | 180.20 | 441.10 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| VFR_air | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | 0.15 | 0.37 | 0.60 | 0.15 | 0.37 |
| Capacity_air | [W] | 1299.31 | 1744.03 | 1863.96 | 1934.95 | 2640.51 |
| Capacity_air_uncertainty | [W] | 33.07 | 51.77 | 75.05 | 26.17 | 54.42 |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 7.16 | 7.17 | 7.13 | 7.19 | 7.11 |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 13.45 | 15.12 | 15.78 | 11.90 | 13.29 |
| P_water in | [ kPa ] | 119.01 | 118.70 | 117.19 | 165.11 | 163.19 |
| DP_water | [ kPa ] | 16270.00 | 16420.00 | 15650.00 | 59940.00 | 57960.00 |
| MFR_water | [g/s] | 50.11 | 50.00 | 49.96 | 100.10 | 100.00 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 996.33 | 996.30 | 996.41 | 995.92 | 996.09 |
| Capacity_water | [W] | 1321.00 | 1664.00 | 1810.00 | 1978.00 | 2592.00 |
| Capacity_water_uncertainty | [W] | 18.23 | 15.50 | 19.26 | 37.05 | 37.34 |
| Energy balance | [\%] | 1.66 | -4.70 | -2.94 | 2.20 | -1.85 |
| Average capacity | [W] | 1310.16 | 1704.01 | 1836.98 | 1956.48 | 2616.25 |
| Capacity uncertainty | [W] | 18.881 | 27.02 | 38.74 | 22.68 | 33.00 |
| w_omega_in | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0109900 | 0.0112300 | 0.0112200 | 0.0112000 | 0.0112400 |
| w_omega_in_uncertainty | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0002156 | 0.0001468 | 0.0002932 | 0.0002192 | 0.0001470 |
| w_omega_out | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0109900 | 0.0112300 | 0.0112200 | 0.0108700 | 0.0112400 |
| w_omega_out_uncertainty | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0002156 | 0.0001468 | 0.0002932 | 0.0002192 | 0.0001470 |
| condensate | [g/s] | 0.0000000 | 0.0000000 | 0.0012986 | 0.0586831 | 0.0004491 |


| condenate_uncertainty | [g/s] | 0.0000000 | 0.0000000 | 0.0000247 | 0.0001050 | 0.0000203 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Case \# |  | 6 | 7 | 8 | 9 |  |
| T_air_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 26.70 | 26.70 | 26.90 | 26.70 |  |
| T_air_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 22.44 | 15.97 | 19.92 | 21.47 |  |
| RH_air_in | [\%] | 50.3\% | 50.3\% | 50.6\% | 50.9\% |  |
| RH_air_out | [\%] | 62.6\% | 82.5\% | 72.5\% | 66.6\% |  |
| P_air_in | [ kPa ] | 101.39 | 101.37 | 100.88 | 100.94 |  |
| DP_air | [Pa] | 61.80 | 8.60 | 32.70 | 70.90 |  |
| uncertainty_DP | [Pa] | 1.50 | 0.70 | 1.30 | 1.20 |  |
| SH | [W] | 3041.82 | 1958.73 | 3126.66 | 3728.43 |  |
| SH_un | [W] | 86.77 | 32.88 | 82.30 | 83.50 |  |
| LH | [W] | 0.00 | 332.10 | 100.09 | 35.84 |  |
| LH_un | [W] | 0.00 | 0.16 | 0.24 | 0.04 |  |
| MFR_air | [g/s] | 708.70 | 181.30 | 444.80 | 708.70 |  |
| VFR_air | [ $\mathrm{m}^{3} / \mathrm{s}$ ] | 0.60 | 0.15 | 0.37 | 0.60 |  |
| Capacity_air | [W] | 3041.82 | 2290.83 | 3226.75 | 3764.12 |  |
| Capacity_air_uncertainty | [W] | 86.77 | 32.88 | 82.30 | 83.50 |  |
| T_water_in | $\left[{ }^{\circ} \mathrm{C}\right]$ | 7.31 | 7.32 | 7.16 | 7.17 |  |
| T_water_out | $\left[{ }^{\circ} \mathrm{C}\right]$ | 14.31 | 11.05 | 12.27 | 13.03 |  |
| P_water in | [ kPa ] | 176.13 | 239.20 | 239.94 | 240.85 |  |


| DP_water | $[\mathrm{Pa}]$ | 59750 | 129800 | 126100 | 125900 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| MFR_water | $[\mathrm{g} / \mathrm{s}]$ | 100.20 | 150.40 | 149.70 | 149.90 |
| Water density | $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$ | 996.23 | 995.41 | 995.38 | 995.48 |
| Capacity_water | $[\mathrm{W}]$ | 2939.00 | 2353.00 | 3204.00 | 3679.00 |
| Capacity_water_uncertainty | $[\mathrm{W}]$ | 76.71 | 54.74 | 87.41 | 85.67 |
| Energy balance | $[\%]$ | -3.44 | 2.68 | -0.71 | -2.29 |
| Average capacity | $[\mathrm{W}]$ | 2990.41 | 2321.92 | 3215.38 | 3721.56 |
| Capacity uncertainty | $[\mathrm{W}]$ | 57.91 | 31.93 | 60.03 | 59.82 |
| w_omega_in | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0109900 | 0.0109900 | 0.0112700 | 0.0111900 |
| w_omega_in_uncertainty | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0002156 | 0.0002156 | 0.0002206 | 0.0001464 |
| w_omega_out | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0109900 | 0.0102500 | 0.0111800 | 0.0111700 |
| w_omega_out_uncertainty | $[\mathrm{kg} / \mathrm{kg}]$ | 0.0002156 | 0.0002159 | 0.0002206 | 0.0001464 |
| condensate | $[\mathrm{g} / \mathrm{s}]$ | 0.0000370 | 0.1347732 | 0.0407849 | 0.0145460 |
| [g/s] | 0.0000523 | 0.0000636 | 0.0000980 | 0.0000166 |  |
| condenate_uncertainty |  |  |  |  |  |

## 5mm slit fin-and-tube HX airside heat transfer and pressure drop correlation validation

Correlations to be validated is from Sarpotdar et al. (2016). The following two graphs shows the prediction of the correlations against experiments. It can be seen that the heat transfer prediction is acceptable but the maximum pressure drop prediction deviation
is about $35 \%$, thus correction factor is necessary to correct the correlation. Three correction factors 1.1, 1.35 and 1.55 are applied for three different inlet air velocity $1.05,2.55$ and $3.99 \mathrm{~m} / \mathrm{s}$, respectively. The corrected results are shown in the third graph.


Figure C. 115 mm slit fin-and-tube HX capacity validation


Figure C. 125 mm slit fin-and-tube HX ADP validation


Figure C. $13 \mathbf{5 m m}$ slit fin-and-tube HX corrected ADP

## 4 mm slit fin-and-tube HX test data

Table C. 244 mm slit fin-and-tube HX condenser test data (air and R410A)



|  | Ref | 8.4 | $\pm$ | 0.1 | 2.9 | $\pm$ | 4.9 | 1534.0 | $\pm$ | 14.1 |  | 1534.0 | $\pm$ | 14.1 | 0.92\% |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 5 | Air | 451.6 | $\pm$ | 3.0 | 21.4 | $\pm$ | 0.8 | 1232.0 | $\pm$ | 65.1 | 81.9 | 1313.9 | $\pm$ | 65.1 | 4.95\% | 3.1\% |
|  | Ref | 5.8 | $\pm$ | 0.1 | -1.3 | $\pm$ | 5.5 | 1274.0 | $\pm$ | 15.4 |  | 1274.0 | $\pm$ | 15.4 | 1.21\% |  |
| 6 | Air | 450.9 | $\pm$ | 3.0 | 21.3 | $\pm$ | 0.8 | 1185.0 | $\pm$ | 64.9 | 79.7 | 1264.7 | $\pm$ | 64.9 | 5.13\% | 4.3\% |
|  | Ref | 5.9 | $\pm$ | 0.1 | 0.1 | $\pm$ | 6.7 | 1212.0 | $\pm$ | 16.4 |  | 1212.0 | $\pm$ | 16.4 | 1.36\% |  |
| 7 | Air | 451.0 | $\pm$ | 3.0 | 21.3 | $\pm$ | 0.8 | 1094.0 | $\pm$ | 64.9 | 74.7 | 1168.7 | $\pm$ | 64.9 | 5.55\% | 5.1\% |
|  | Ref | 6.0 | $\pm$ | 0.1 | 0.9 | $\pm$ | 8.4 | 1111.0 | $\pm$ | 15.9 |  | 1111.0 | $\pm$ | 15.9 | 1.43\% |  |

Table C. 254 mm slit fin-and-tube HX evaporator test data (air and R410A)

| Test No. |  | Tin |  |  | Tout |  |  | Pin_gage |  |  | Pout_gage |  |  | T_in_sat |  |  | T_out_sat |  |  | Qualityin |  |  | RHin |  |  | RHout |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | [ ${ }^{\text {C }}$ ] |  |  | [ ${ }^{\text {C }}$ ] |  |  | [ kPa ] |  |  | [ kPa ] |  |  | $\left[{ }^{\circ} \mathrm{C}\right]$ |  |  | [ ${ }^{\text {C }}$ ] |  |  | [-] |  |  | [\%] |  |  | [\%] |  |  |
| 1 | Air | 26.70 | $\pm$ | 0.10 | 23.30 | $\pm$ | 0.10 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 50.9\% | $\pm$ | 1.7\% | 59.6\% | $\pm$ | 1.6\% |
|  | Ref | 10.15 | $\pm$ | 0.33 | 16.30 | $\pm$ | 1.50 | 992.7 | $\pm$ | 13.2 | 974.7 | $\pm$ | 11.7 | 10.3 | $\pm$ | 0.4 | 9.8 | $\pm$ | 0.4 | 0.27 | $\pm$ | 0.03 |  |  |  |  |  |  |
| 2 | Air | 26.80 | $\pm$ | 0.10 | 24.10 | $\pm$ | 0.20 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 68.8\% | $\pm$ | 3.8\% | 75.0\% | $\pm$ | 3.8\% |
|  | Ref | 10.28 | $\pm$ | 0.40 | 21.97 | $\pm$ | 0.58 | 996.6 | $\pm$ | 16.4 | 978.7 | $\pm$ | 14.3 | 10.4 | $\pm$ | 0.5 | 9.9 | $\pm$ | 0.4 | 0.28 | $\pm$ | 0.05 |  |  |  |  |  |  |
| 3 | Air | 26.60 | $\pm$ | 0.10 | 23.60 | $\pm$ | 0.00 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 51.2\% | $\pm$ | 1.3\% | 59.3\% | $\pm$ | 1.5\% |
|  | Ref | 10.75 | $\pm$ | 1.49 | 13.64 | $\pm$ | 0.26 | 1010.0 | $\pm$ | 10.4 | 992.6 | $\pm$ | 9.2 | 10.8 | $\pm$ | 1.5 | 13.6 | $\pm$ | 0.3 | 0.36 | $\pm$ | 0.03 |  |  |  |  |  |  |
| 4 | Air | 26.80 | $\pm$ | 0.10 | 24.30 | $\pm$ | 0.20 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 68.8\% | $\pm$ | 3.8\% | 75.5\% | $\pm$ | 3.9\% |
|  | Ref | 10.58 | $\pm$ | 0.37 | 22.05 | $\pm$ | 0.80 | 1004.7 | $\pm$ | 14.8 | 986.4 | $\pm$ | 13.0 | 10.7 | $\pm$ | 0.5 | 10.1 | $\pm$ | 0.4 | 0.38 | $\pm$ | 0.04 |  |  |  |  |  |  |
| 5 | Air | 26.70 | $\pm$ | 0.10 | 22.40 | $\pm$ | 0.10 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 49.6\% | $\pm$ | 1.3\% | 60.9\% | $\pm$ | 1.2\% |
|  | Ref | 11.57 | $\pm$ | 0.22 | 11.71 | $\pm$ | 0.60 | 1047.6 | $\pm$ | 9.2 | 1018.8 | $\pm$ | 7.6 | 12.0 | $\pm$ | 0.3 | 11.1 | $\pm$ | 0.2 | 0.26 | $\pm$ | 0.02 |  |  |  |  |  |  |
| 6 | Air | 26.80 | $\pm$ | 0.10 | 23.40 | $\pm$ | 0.10 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 68.8\% | $\pm$ | 1.7\% | 78.3\% | $\pm$ | 2.0\% |
|  | Ref | 11.47 | $\pm$ | 0.29 | 15.69 | $\pm$ | 1.31 | 1044.3 | $\pm$ | 12.7 | 1016.2 | $\pm$ | 11.2 | 11.9 | $\pm$ | 0.4 | 11.0 | $\pm$ | 0.3 | 0.28 | $\pm$ | 0.02 |  |  |  |  |  |  |
| 7 | Air | 26.80 | $\pm$ | 0.10 | 23.00 | $\pm$ | 0.10 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 51.9\% | $\pm$ | 0.7\% | 63.5\% | $\pm$ | 0.9\% |
|  | Ref | 11.67 | $\pm$ | 0.10 | 11.15 | * | 0.09 | 1048.4 | $\pm$ | 4.2 | 1019.2 | $\pm$ | 3.8 | 12.0 | $\pm$ | 0.1 | 11.1 | $\pm$ | 0.1 | 0.34 | $\pm$ | 0.02 |  |  |  |  |  |  |
| 8 | Air | 26.80 | $\pm$ | 0.10 | 23.70 | $\pm$ | 0.10 |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  | $\pm$ |  |  |  |  | 69.3\% | $\pm$ | 1.3\% | 77.8\% | $\pm$ | 1.5\% |
|  | Ref | 11.79 | $\pm$ | 0.09 | 16.37 | $\pm$ | 0.76 | 1052.3 | $\pm$ | 5.5 | 1022.1 | $\pm$ | 5.6 | 12.1 | $\pm$ | 0.2 | 11.2 | $\pm$ | 0.2 | 0.37 | $\pm$ | 0.02 |  |  |  |  |  |  |
| Test No. |  | Volume Flow Rate |  |  |  |  | Mass Flow Rate |  |  | Air/Ref Pressure Drop |  |  | Capacity |  |  |  | Heat <br> Loss | SH/LH |  |  |  | Adjusted Capacity |  |  |  | Uncertainty |  | Energy Balance |
|  |  | [m3/s] |  |  |  |  | [g/s] |  |  | [Pa]/ [kPa] |  |  | [W] |  |  |  | [W] | [W]/[W] |  |  |  | [W] |  |  |  | [\%] |  | [\%] |


| 1 | Air | 0.38458997 | $\pm$ | 0.00071028 | 458.7 | $\pm$ | 3.0 | 25.2 | $\pm$ | 0.9 | 1598.0 | $\pm$ | 67.3 | -65.5 | 1532.5 | $\pm$ | 67.3 | 1532.5 | $\pm$ | 68.0 | 4.44\% |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Ref | 2.562224992 |  |  | 9.9 | $\pm$ | 1.2 | 18.0 | $\pm$ | 17.6 | 1577.0 | $\pm$ | 256.9 |  | 0.0 | $\pm$ | 9.9 | 1577.0 | $\pm$ | 256.9 | 16.29\% |  |
|  | Air | 0.38350449 | $\pm$ | 0.00027231 | 455.9 | $\pm$ | 3.4 | 30.7 | $\pm$ | 1.4 | 1270.0 | $\pm$ | 105.6 | -47.2 | 1222.8 | $\pm$ | 105.6 | 1442.6 | $\pm$ | 105.6 | 7.32\% |  |
|  | Ref | 2.554993286 |  |  | 9.5 | $\pm$ | 1.5 | 17.9 | $\pm$ | 21.8 | 1547.0 | $\pm$ | 336.1 |  | 219.8 | $\pm$ | 2.9 | 1547.0 | $\pm$ | 336.1 | 21.73\% |  |
|  | Air | 0.38633618 | $\pm$ | 0.00070179 | 461.6 | $\pm$ | 3.4 | 25.5 | $\pm$ | 0.8 | 1419.0 | $\pm$ | 48.5 | -56.0 | 1363.0 | $\pm$ | 48.5 | 1395.6 | $\pm$ | 48.5 | 3.48\% |  |
|  | Ref | 2.573858607 |  |  | 9.5 | $\pm$ | 0.8 | 17.8 | $\pm$ | 13.9 | 1352.0 | $\pm$ | 162.9 |  | 32.6 | $\pm$ | 2.1 | 1352.0 | $\pm$ | 162.9 | 12.05\% |  |
|  | Air | 0.3846 | $\pm$ | 0.0004 | 457.1 | $\pm$ | 3.0 | 30.9 | $\pm$ | 1.1 | 1179.0 | $\pm$ | 105.8 | -42.8 | 1136.2 | $\pm$ | 105.8 | 1359.7 | $\pm$ | 105.8 | 7.78\% |  |
|  | Ref | 2.562291805 |  |  | 9.6 | $\pm$ | 1.1 | 18.3 | $\pm$ | 19.6 | 1366.0 | $\pm$ | 244.8 | - | 223.6 | $\pm$ | 2.1 | 1366.0 | $\pm$ | 244.8 | 17.92\% |  |
| 5 | Air | 0.38062561 | $\pm$ | 0.00028161 | 457.8 | $\pm$ | 3.5 | 22.2 | $\pm$ | 0.8 | 2016.0 | $\pm$ | 68.0 | -58.8 | 1957.2 | $\pm$ | 68.0 | 1957.2 | $\pm$ | 68.0 | 3.48\% |  |
|  | Ref | 2.535813543 |  |  | 13.4 | $\pm$ | 0.9 | 28.8 | $\pm$ | 12.0 | 2056.0 | $\pm$ | 199.8 |  | 0.0 | $\pm$ | 0.0 | 2056.0 | $\pm$ | 199.8 | 9.72\% |  |
| 6 | Air | 0.37954013 | $\pm$ | 0.00061825 | 453.1 | $\pm$ | 3.1 | 34.3 | $\pm$ | 1.2 | 1590.0 | $\pm$ | 67.0 | -48.0 | 1542.0 | $\pm$ | 67.0 | 2022.1 | $\pm$ | 68.0 | 3.36\% |  |
|  | Ref | 2.528581837 |  |  | 13.2 | $\pm$ | 0.6 | 28.1 | $\pm$ | 17.0 | 2046.0 | $\pm$ | 141.8 |  | 480.1 | $\pm$ | 11.8 | 2046.0 | $\pm$ | 141.8 | 6.93\% |  |
| 7 | Air | 0.3825606 | $\pm$ | $4.4387 \mathrm{E}-05$ | 457.1 | $\pm$ | 3.0 | 27.0 | $\pm$ | 0.9 | 1781.0 | $\pm$ | 67.3 | -56.3 | 1724.7 | $\pm$ | 67.3 | 1738.8 | $\pm$ | 67.3 | 3.87\% |  |
|  | Ref | 2.548704845 |  |  | 13.4 | $\pm$ | 0.5 | 29.2 | $\pm$ | 5.7 | 1833.0 | $\pm$ | 1067.0 |  | 14.1 | $\pm$ | 0.4 | 1833.0 | $\pm$ | 1067.0 | 58.21\% |  |
| 8 | Air | 0.38048403 | $\pm$ | 0.00036043 | 453.7 | $\pm$ | 3.1 | 34.0 | $\pm$ | 0.9 | 1451.0 | $\pm$ | 66.9 | -41.2 | 1409.8 | $\pm$ | 66.9 | 1856.7 | $\pm$ | 67.6 | 3.64\% |  |
|  | Ref | 2.534870277 |  |  | 13.5 | $\pm$ | 0.5 | 30.2 | $\pm$ | 7.9 | 1842.0 | $\pm$ | 114.0 |  | 446.9 | $\pm$ | 9.2 | 1842.0 | $\pm$ | 114.0 | 6.19\% |  |

## $4 \& 5 \mathrm{~mm}$ slit fin-and-tube HX test pictures



Figure C. 145 mm coil wet condition test picture


Figure C. 15 VFR_air= $\mathbf{3 7 5} \mathbf{c f m}$, MFR_water= $\mathbf{1 5 0 g} / \mathrm{s}$


Figure C. 16 VFR_air= $\mathbf{3 7 5} \mathbf{~ c f m , ~ M F R \_ w a t e r = ~} \mathbf{1 0 0 g} / \mathrm{s}$

Table D. 1 Airside GCI calculation data

|  | Grid elements |  |  | Grid Ratio |  | HHTCTC |  |  | DP |  |  | HTC |  | ADP |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Grid 1 | Grid 2 | Grid 3 | r21 | r32 | HTC 1 | HTC 2 | HTC 3 | DP 1 | DP 2 | DP 3 | GCI 21 | GCI 32 | GCI 21 | GCI 32 |
| InC1 | 2531096 | 813194 | 266860 | 1.46006 | 1.44978 | 840.913 | 839.848 | 795.695 | 719.788 | 712.334 | 708.7279 | 0.14\% | 6.29\% | 1.16\% | 0.58\% |
| InC2 | 1186452 | 330672 | 71996 | 1.53091 | 1.66225 | 923.183 | 921.545 | 907.427 | 787.866 | 772.257 | 758.6759 | 0.17\% | 1.10\% | 1.88\% | 1.27\% |
| InC3 | 5831038 | 1981929 | 508845 | 1.43291 | 1.57338 | 805.149 | 804.489 | 746.098 | 438.922 | 434.661 | 439.0497 | 0.10\% | 6.63\% | 1.16\% | 0.85\% |
| InC4 | 971569 | 342006 | 106026 | 1.41627 | 1.47754 | 805.199 | 807.28 | 781.209 | 413.371 | 408.15 | 417.6288 | 0.32\% | 3.53\% | 1.59\% | 2.40\% |
| InC5 | 2800144 | 912378 | 285698 | 1.45323 | 1.47261 | 739.092 | 739.478 | 711.539 | 730.801 | 716.211 | 660.5554 | 0.06\% | 4.20\% | 2.29\% | 9.01\% |
| InC6 | 651685 | 216051 | 75448 | 1.44486 | 1.42005 | 786.484 | 789.996 | 792.284 | 800.883 | 778.908 | 779.7605 | 1.15\% | 0.79\% | 7.29\% | 0.30\% |
| InC7 | 6521896 | 2142880 | 557278 | 1.44919 | 1.56666 | 718.093 | 718.296 | 704.308 | 461.026 | 450.347 | 430.6212 | 0.03\% | 1.71\% | 2.69\% | 3.94\% |
| InC8 | 1070725 | 367742 | 110469 | 1.42795 | 1.49314 | 724.448 | 726.35 | 717.93 | 441.682 | 431.027 | 427.5232 | 0.32\% | 1.19\% | 2.97\% | 0.83\% |
| InC9 | 6640449 | 2114484 | 565489 | 1.46442 | 1.55213 | 553.326 | 557.84 | 553.582 | 95.5166 | 93.726 | 93.15457 | 4.81\% | 3.91\% | 11.37\% | 3.12\% |
| InC10 | 1269633 | 405021 | 132656 | 1.46353 | 1.45072 | 601.431 | 604.051 | 603.712 | 90.0634 | 88.6098 | 87.03794 | 0.47\% | 0.06\% | 1.80\% | 2.04\% |
| InC11 | $1.6 \mathrm{E}+07$ | 6445999 | 2975095 | 1.353 | 1.29399 | 528.644 | 526.952 | 533.626 | 66.6594 | 66.624 | 65.56895 | 0.48\% | 2.32\% | 0.08\% | 2.98\% |
| InC12 | 1763522 | 560612 | 166770 | 1.46523 | 1.49801 | 533.606 | 537.977 | 540.255 | 61.2543 | 60.3861 | 60.97759 | 1.15\% | 0.55\% | 2.04\% | 1.27\% |
| InC13 | 7162884 | 2260219 | 592917 | 1.46886 | 1.56214 | 538.024 | 541.922 | 537.9 | 87.1149 | 85.2671 | 84.29393 | 4.24\% | 3.74\% | 12.78\% | 5.78\% |
| InC14 | 1215879 | 370046 | 142156 | 1.48666 | 1.37561 | 572.469 | 576.79 | 580.99 | 88.5862 | 87.1142 | 85.86484 | 4.27\% | 5.23\% | 9.63\% | 10.52\% |
| InC15 | $1.9 \mathrm{E}+07$ | 6750233 | 2728367 | 1.40135 | 1.3525 | 510.09 | 512.632 | 516.052 | 59.084 | 58.8197 | 58.05584 | 1.89\% | 2.87\% | 1.71\% | 5.69\% |


| InC16 | 2035375 | 624974 | 183639 | 1.48227 | 1.50418 | 513.207 | 517.862 | 519.304 | 55.0834 | 54.1299 | 54.65422 | $0.94 \%$ | $0.27 \%$ | $1.84 \%$ | $0.95 \%$ |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :--- | :--- | :--- | :--- | :--- | :--- |
| InC17 | 1497415 | 397580 | 218225 | 1.55587 | 1.22135 | 389.479 | 385.235 | 380.414 | 320.162 | 323.045 | 330.5782 | $5.57 \%$ | $15.06 \%$ | $4.51 \%$ | $27.09 \%$ |
| InC18 | 1209160 | 334331 | 66568 | 1.53498 | 1.71253 | 307.854 | 304.998 | 301.007 | 498.943 | 487.207 | 473.6736 | $2.35 \%$ | $2.51 \%$ | $6.05 \%$ | $5.41 \%$ |
| InC19 | 5813430 | 1522941 | 388518 | 1.56285 | 1.57673 | 344.294 | 293.703 | 286.044 | 306.088 | 292.857 | 287.7062 | $14.93 \%$ | $2.25 \%$ | $3.92 \%$ | $1.51 \%$ |
| InC20 | 967656 | 344812 | 126149 | 1.41052 | 1.39819 | 258.89 | 260.951 | 257.367 | 278.774 | 271.481 | 274.5281 | $1.49 \%$ | $2.71 \%$ | $5.06 \%$ | $2.16 \%$ |
| InC21 | 2802689 | 911053 | 281307 | 1.45438 | 1.47952 | 326.62 | 307.232 | 311.789 | 513.446 | 501.211 | 513.6211 | $7.07 \%$ | $1.54 \%$ | $2.74 \%$ | $2.54 \%$ |
| InC22 | 654076 | 217564 | 68818 | 1.44327 | 1.46767 | 286.065 | 285.66 | 279.903 | 520.235 | 504.106 | 502.9947 | $0.16 \%$ | $2.23 \%$ | $3.69 \%$ | $0.24 \%$ |
| InC23 | 6303207 | 2042656 | 530093 | 1.45587 | 1.56777 | 254.592 | 256.02 | 246.757 | 307.61 | 301.613 | 309.4478 | $0.62 \%$ | $3.22 \%$ | $2.22 \%$ | $2.17 \%$ |
| InC24 | 1067456 | 370805 | 106442 | 1.42255 | 1.51592 | 249.698 | 248.354 | 243.848 | 285.338 | 278.336 | 290.6868 | $0.66 \%$ | $1.78 \%$ | $3.07 \%$ | $4.09 \%$ |
| InC25 | 4696822 | 1233009 | 288087 | 1.56175 | 1.62361 | 364.276 | 357.626 | 351.202 | 72.7596 | 72.0456 | 70.75495 | $9.31 \%$ | $8.34 \%$ | $4.96 \%$ | $8.32 \%$ |
| InC26 | 1257391 | 409346 | 131846 | 1.45365 | 1.45884 | 204.445 | 203.872 | 202.296 | 67.8354 | 65.0434 | 61.31714 | $0.32 \%$ | $0.86 \%$ | $4.82 \%$ | $6.73 \%$ |
| InC27 | $1.6 \mathrm{E}+07$ | 4976319 | 1380017 | 1.47492 | 1.53347 | 368.637 | 370.537 | 344.147 | 82.3682 | 81.0235 | 78.68922 | $0.55 \%$ | $7.09 \%$ | $1.76 \%$ | $2.74 \%$ |
| InC28 | 1303620 | 345241 | 92114 | 1.55719 | 1.55334 | 215.009 | 203.709 | 199.672 | 30.2587 | 30.6907 | 34.74369 | $4.87 \%$ | $1.79 \%$ | $1.24 \%$ | $10.32 \%$ |
| InC29 | 7360612 | 2257636 | 619188 | 1.48281 | 1.53914 | 185.376 | 184.914 | 183.581 | 63.7856 | 60.1065 | 60.08887 | $0.26 \%$ | $0.66 \%$ | $6.38 \%$ | $0.03 \%$ |
| InC30 | 1210873 | 371373 | 138424 | 1.48285 | 1.38952 | 198.501 | 198.368 | 197.125 | 60.194 | 57.5606 | 55.60576 | $0.07 \%$ | $0.85 \%$ | $4.77 \%$ | $4.72 \%$ |
| InC31 | $1.9 \mathrm{E}+07$ | $1.1 \mathrm{E}+07$ | 5234983 | 1.18495 | 1.28697 | 176.461 | 176.494 | 176.701 | 42.6811 | 42.0311 | 39.82754 | $0.06 \%$ | $0.22 \%$ | $4.78 \%$ | $10.54 \%$ |
| InC32 | 2027063 | 623853 | 177840 | 1.48114 | 1.51944 | 178.102 | 178.24 | 175.348 | 39.4855 | 37.6538 | 36.65677 | $0.08 \%$ | $1.58 \%$ | $5.09 \%$ | $2.60 \%$ |

Table D. 2 Waterside GCI calculation data

|  | Grid |  |  | Grid Ratio |  | HTC |  |  | DP |  |  | HTC |  | ADP |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Grid 1 | Grid 2 | Grid 3 | r21 | r32 | HTC 1 | HTC 2 | HTC 3 | DP 1 | DP 2 | DP 3 | GCI 21 | GCI 32 | GCI 21 | GCI 32 |
| InC1 | 9386491 | 2391209 | 186964 | 1.57747 | 2.33859 | 6390.86 | 6131.88 | 6011.89 | 4362.631 | 4337.163 | 4258.9923 | 3.55\% | 0.56\% | 0.49\% | 0.51\% |
| InC2 | 918190 | 134588 | 97254 | 1.8966 | 1.11438 | 56486.1 | 57737 | 57205.2 | 4284.45 | 4193.39 | 4117.724 | 1.04\% | 4.80\% | 1.05\% | 9.50\% |
| InC3 | 3277121 | 1069894 | 149400 | 1.45228 | 1.92751 | 11173.6 | 10873.8 | 11377.1 | 23876.2 | 23477.8 | 23487.74 | 10.83\% | 8.83\% | 6.66\% | 0.08\% |
| InC 4 | 1062084 | 311250 | 107146 | 1.50551 | 1.42685 | 25860.3 | 25739.7 | 25530.8 | 5742.6 | 5618.93 | 5615.023 | 2.58\% | 5.26\% | 12.12\% | 0.45\% |
| $\mathrm{InC5}$ | 2238913 | 766666 | 159130 | 1.42937 | 1.68895 | 11223.2 | 11001.9 | 10965.2 | 15132.9 | 14933.8 | 14531.8 | 2.41\% | 0.23\% | 1.60\% | 1.87\% |
| InC6 | 287294 | 271485 | 217697 | 1.01905 | 1.07638 | 33218.8 | 33227.3 | 33385.5 | 3417.78 | 3414.82 | 3424.177 | 0.84\% | 3.73\% | 2.82\% | 2.15\% |
| InC7 | 2516418 | 896761 | 156693 | 1.41049 | 1.78872 | 10757.6 | 10464.3 | 11143.3 | 23494.9 | 23110.5 | 23440.38 | 7.15\% | 7.91\% | 4.24\% | 1.83\% |
| InC8 | 2652950 | 88578 | 667167 | 3.10552 | 0.51015 | 22531.6 | 22365.1 | 22607.9 | 9389.43 | 9053.31 | 9366.006 | 1.08\% | 4.35\% | 5.38\% | 13.51\% |
| InC9 | 1164051 | 374619 | 86774 | 1.45924 | 1.6283 | 5701.29 | 5576.57 | 5646.49 | 1370.03 | 1377.35 | 1303.053 | 2.48\% | 0.94\% | 0.59\% | 4.32\% |
| InC10 | 892930 | 226542 | 105339 | 1.57963 | 1.29078 | 25157.4 | 24036.1 | 23148.3 | 3868.99 | 3901.04 | 3937.132 | 3.90\% | 7.20\% | 0.69\% | 1.72\% |
| InC11 | 3826770 | 1125341 | 168300 | 1.50377 | 1.88394 | 5388.74 | 5195.58 | 5757.086 | 4591.27 | 4803.49 | 4838.318 | 5.79\% | 8.15\% | 6.88\% | 0.60\% |
| InC12 | 1141300 | 285227 | 137992 | 1.58758 | 1.27383 | 11466.2 | 11229.7 | 11119.3 | 4273.16 | 4233.35 | 3908.09 | 1.73\% | 1.99\% | 0.77\% | 16.71\% |
| InC13 | $1.5 \mathrm{E}+07$ | $1.4 \mathrm{E}+07$ | 8684757 | 1.01699 | 1.17626 | 6170 | 6168.59 | 6272.87 | 4364.82 | 4355.99 | 4192.583 | 0.84\% | 5.42\% | 7.40\% | 12.70\% |
| InC14 | 2892046 | 534579 | 200781 | 1.75549 | 1.386 | 13347.4 | 13542.7 | 12828.9 | 1371.87 | 1421.36 | 1539.285 | 1.10\% | 9.22\% | 2.66\% | 12.69\% |
| InC15 | 5260164 | 1847026 | 82385 | 1.41745 | 2.81973 | 6410.15 | 5754.88 | 9534.84 | 13337.1 | 13489.3 | 28739.92 | 14.10\% | 7.13\% | 1.40\% | 9.54\% |
| InC16 | 1833962 | 438964 | 104716 | 1.6106 | 1.6124 | 9139.83 | 8599.88 | 8931.22 | 2316.39 | 2382.86 | 2362.27 | 4.92\% | 2.90\% | 2.19\% | 0.68\% |

## Appendix E: Meta-model data

Table E. 1 Airside meta-model data

|  | Tube \# | D0 | $\mathrm{Pl} / \mathrm{D} 0$ | Pt/D0 | LR=L0/L1 | sita [deg] | air face velocity | j | f |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 0.008008 | 0.866667 | 0.969357 | 0.836333 | 0.158719 | 0.717750 | 0.135556 | 0.035890 | 0.835242 |
| 2 | 0.022022 | 0.562222 | 0.255890 | 0.535310 | 0.014757 | 0.538500 | 0.297778 | 0.014284 | 0.167818 |
| 3 | 0.023023 | 0.533333 | 0.416656 | 0.050745 | 0.200820 | 0.652625 | 0.011111 | 0.031217 | 0.264411 |
| 4 | 0.024024 | 0.148889 | 0.983322 | 0.168833 | 0.157757 | 0.351375 | 0.393333 | 0.034483 | 0.427176 |
| 5 | 0.034034 | 0.511111 | 0.544054 | 0.200042 | 0.194737 | 0.670625 | 0.402222 | 0.020778 | 0.232554 |
| 6 | 0.037037 | 0.851111 | 0.655208 | 0.541125 | 0.050043 | 0.189250 | 0.213333 | 0.028463 | 0.600552 |
| 7 | 0.042042 | 0.515556 | 0.714337 | 0.801557 | 0.529762 | 0.311250 | 0.155556 | 0.028376 | 0.335064 |
| 8 | 0.050050 | 0.164444 | 0.607798 | 0.934978 | 0.064815 | 0.770750 | 0.348889 | 0.023004 | 0.197962 |
| 9 | 0.051051 | 0.351111 | 0.163247 | 0.646935 | 0.607843 | 0.728750 | 0.013333 | 0.033143 | 0.222025 |
| 10 | 0.052052 | 0.146667 | 0.762200 | 0.434535 | 0.598485 | 0.701750 | 0.433333 | 0.029024 | 0.297190 |
| 11 | 0.041008 | 0.624444 | 0.008734 | 0.584982 | 0.955409 | 0.354625 | 0.788889 | 0.012470 | 0.172575 |
| 12 | 0.041208 | 0.033333 | 0.961599 | 0.926478 | 0.619048 | 0.668375 | 0.382222 | 0.045531 | 0.359882 |
| 13 | 0.042609 | 0.864444 | 0.800935 | 0.018355 | 0.577139 | 0.189250 | 0.006667 | 0.036245 | 0.433153 |
| 14 | 0.042809 | 0.066667 | 0.067884 | 0.670064 | 0.524775 | 0.676875 | 0.522222 | 0.019803 | 0.163549 |
| 15 | 0.043009 | 0.411111 | 0.276674 | 0.856379 | 0.545739 | 0.660375 | 0.115556 | 0.024862 | 0.221944 |
| 16 | 0.044009 | 0.111111 | 0.203957 | 0.791473 | 0.629310 | 0.098250 | 0.722222 | 0.022512 | 0.255326 |
| 17 | 0.044409 | 0.377778 | 0.633236 | 0.976729 | 0.703901 | 0.842625 | 0.220000 | 0.024675 | 0.243116 |
| 18 | 0.044809 | 0.662222 | 0.050316 | 0.841007 | 0.909420 | 0.983750 | 0.120000 | 0.014639 | 0.111271 |
| 19 | 0.045009 | 0.753333 | 0.706570 | 0.811217 | 0.607323 | 0.820000 | 0.526667 | 0.020739 | 0.408872 |
| 20 | 0.046209 | 0.944444 | 0.283232 | 0.212608 | 0.120833 | 0.841000 | 0.108889 | 0.012467 | 0.098307 |
| 21 | 0.008002 | 0.286667 | 0.752977 | 0.313185 | 0.658065 | 0.074250 | 0.593333 | 0.028713 | 0.291930 |
| 22 | 0.008202 | 0.604444 | 0.289192 | 0.977640 | 0.522242 | 0.358250 | 0.091111 | 0.031315 | 0.275098 |
| 23 | 0.008402 | 0.055556 | 0.036308 | 0.601331 | 0.450000 | 0.950000 | 0.255556 | 0.020450 | 0.094152 |
| 24 | 0.008802 | 0.455556 | 0.515931 | 0.614539 | 0.800388 | 0.072375 | 0.817778 | 0.019194 | 0.220610 |
| 25 | 0.009002 | 0.211111 | 0.037754 | 0.430126 | 0.341488 | 0.034250 | 0.231111 | 0.030155 | 0.268173 |


| 26 | 0.009402 | 0.635556 | 0.729994 | 0.809513 | 0.780303 | 0.477250 | 0.644444 | 0.018891 | 0.213257 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 27 | 0.009602 | 0.317778 | 0.665134 | 0.862737 | 0.880229 | 0.301875 | 0.233333 | 0.038221 | 0.342848 |
| 28 | 0.009802 | 0.206667 | 0.210256 | 0.840339 | 0.759662 | 0.804750 | 0.731111 | 0.018585 | 0.163469 |
| 29 | 0.010002 | 0.651111 | 0.361392 | 0.037289 | 0.541667 | 0.288625 | 0.415556 | 0.020034 | 0.226066 |
| 30 | 0.010202 | 0.526667 | 0.345861 | 0.947462 | 0.954710 | 0.227250 | 0.500000 | 0.020374 | 0.218216 |
| 31 | 0.010402 | 0.024444 | 0.196613 | 0.989875 | 0.686782 | 0.797500 | 0.360000 | 0.038993 | 0.283436 |
| 32 | 0.010602 | 0.837778 | 0.553603 | 0.188026 | 0.964721 | 0.028250 | 0.717778 | 0.015748 | 0.200469 |
| 33 | 0.010802 | 0.317778 | 0.636460 | 0.340052 | 0.237203 | 0.093875 | 0.104444 | 0.048602 | 0.447859 |
| 34 | 0.011002 | 0.068889 | 0.920716 | 0.469112 | 0.713312 | 0.206000 | 0.675556 | 0.041802 | 0.385319 |
| 35 | 0.011202 | 0.917778 | 0.122649 | 0.518207 | 0.832650 | 0.601125 | 0.893333 | 0.009781 | 0.130820 |
| 36 | 0.011402 | 0.231111 | 0.360580 | 0.577013 | 0.254464 | 0.493500 | 0.535556 | 0.023760 | 0.222423 |
| 37 | 0.011602 | 0.288889 | 0.851952 | 0.651227 | 0.641473 | 0.882000 | 0.086667 | 0.047015 | 0.344744 |
| 38 | 0.011802 | 0.484444 | 0.620354 | 0.167920 | 0.662381 | 0.793375 | 0.468889 | 0.021665 | 0.205453 |
| 39 | 0.012002 | 0.920000 | 0.010594 | 0.502494 | 0.734934 | 0.982375 | 0.346667 | 0.008273 | 0.067540 |
| 40 | 0.012202 | 0.066667 | 0.397725 | 0.486314 | 0.922101 | 0.275875 | 0.588889 | 0.038938 | 0.328778 |
| 41 | 0.012402 | 0.860000 | 0.877373 | 0.237991 | 0.459541 | 0.210250 | 0.611111 | 0.018376 | 0.230171 |
| 42 | 0.012803 | 0.877778 | 0.915791 | 0.866192 | 0.987730 | 0.925375 | 0.300000 | 0.021444 | 0.199080 |
| 43 | 0.013003 | 0.620000 | 0.504400 | 0.411864 | 0.499597 | 0.735125 | 0.717778 | 0.014990 | 0.161217 |
| 44 | 0.013203 | 0.306667 | 0.746197 | 0.695740 | 0.071029 | 0.631125 | 0.933333 | 0.018385 | 0.198648 |
| 45 | 0.013603 | 0.122222 | 0.965812 | 0.982215 | 0.036822 | 0.739750 | 0.300000 | 0.038452 | 0.309184 |
| 46 | 0.013803 | 0.097778 | 0.594247 | 0.725123 | 0.101852 | 0.488125 | 0.166667 | 0.051040 | 0.411322 |
| 47 | 0.014003 | 0.277778 | 0.457627 | 0.717860 | 0.191581 | 0.927000 | 0.886667 | 0.013197 | 0.104026 |
| 48 | 0.014203 | 0.277778 | 0.153674 | 0.338558 | 0.095118 | 0.559250 | 0.746667 | 0.012598 | 0.117607 |
| 49 | 0.014603 | 0.044444 | 0.953441 | 0.358137 | 0.647436 | 0.967375 | 0.351111 | 0.049561 | 0.346678 |
| 50 | 0.015003 | 0.126667 | 0.007246 | 0.007045 | 0.127637 | 0.320625 | 0.333333 | 0.022666 | 0.179442 |
| 51 | 0.015203 | 0.393333 | 0.727577 | 0.678475 | 0.325848 | 0.710500 | 0.948889 | 0.017185 | 0.183289 |
| 52 | 0.015603 | 0.177778 | 0.906881 | 0.024718 | 0.433642 | 0.676375 | 0.464444 | 0.035757 | 0.337841 |
| 53 | 0.015803 | 0.573333 | 0.846319 | 0.857321 | 0.771870 | 0.871625 | 0.680000 | 0.018112 | 0.179511 |
| 54 | 0.016603 | 0.482222 | 0.568581 | 0.105987 | 0.154122 | 0.660500 | 0.931111 | 0.014488 | 0.171530 |


| 55 | 0.016803 | 0.335556 | 0.039060 | 0.597861 | 0.758333 | 0.584125 | 0.077778 | 0.031667 | 0.234151 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 56 | 0.017003 | 0.566667 | 0.642672 | 0.347367 | 0.759477 | 0.047250 | 0.148889 | 0.035645 | 0.331938 |
| 57 | 0.017203 | 0.235556 | 0.622520 | 0.930593 | 0.344502 | 0.926750 | 0.353333 | 0.026285 | 0.196683 |
| 58 | -0.046731 | 0.515556 | 0.957833 | 0.233154 | 0.445152 | 0.548500 | 0.255556 | 0.028464 | 0.292511 |
| 59 | -0.045730 | 0.206667 | 0.976329 | 0.519096 | 0.762897 | 0.427375 | 0.935556 | 0.024751 | 0.259128 |
| 60 | -0.043728 | 0.506667 | 0.558675 | 0.474477 | 0.343063 | 0.630625 | 0.180000 | 0.026555 | 0.256704 |
| 61 | -0.042727 | 0.646667 | 0.930880 | 0.642599 | 0.933486 | 0.551500 | 0.017778 | 0.043269 | 0.365969 |
| 62 | -0.041726 | 0.148889 | 0.335670 | 0.532063 | 0.314000 | 0.286250 | 0.784444 | 0.022796 | 0.258965 |
| 63 | -0.039724 | 0.077778 | 0.091033 | 0.108984 | 0.638889 | 0.626625 | 0.935556 | 0.018422 | 0.186939 |
| 64 | -0.038723 | 0.382222 | 0.240125 | 0.380026 | 0.221235 | 0.333375 | 0.542222 | 0.016912 | 0.219831 |
| 65 | -0.021706 | 0.040000 | 0.443826 | 0.398769 | 0.899123 | 0.297250 | 0.184444 | 0.056176 | 0.411714 |
| 66 | -0.016701 | 0.288889 | 0.712140 | 0.750121 | 0.501589 | 0.167125 | 0.208889 | 0.031445 | 0.341533 |
| 67 | -0.015700 | 0.071111 | 0.126027 | 0.771053 | 0.982510 | 0.253250 | 0.906667 | 0.021213 | 0.242979 |
| 68 | -0.014699 | 0.206667 | 0.845170 | 0.104503 | 0.036195 | 0.992000 | 0.242222 | 0.025106 | 0.132763 |
| 69 | -0.012697 | 0.908889 | 0.031061 | 0.783565 | 0.583333 | 0.707750 | 0.611111 | 0.010420 | 0.154849 |
| 70 | -0.006691 | 0.815556 | 0.237585 | 0.022309 | 0.467874 | 0.364375 | 0.360000 | 0.023216 | 0.497174 |
| 71 | -0.001686 | 0.691111 | 0.332635 | 0.977942 | 0.099794 | 0.268250 | 0.002222 | 0.029597 | 0.303308 |
| 72 | -0.052527 | 0.266667 | 0.588262 | 0.740087 | 0.134428 | 0.532250 | 0.977778 | 0.017583 | 0.212972 |
| 73 | -0.052327 | 0.488889 | 0.380393 | 0.332904 | 0.179858 | 0.345625 | 0.982222 | 0.013221 | 0.217599 |
| 74 | -0.052127 | 0.806667 | 0.829758 | 0.553455 | 0.084758 | 0.403500 | 0.006667 | 0.038752 | 0.378707 |
| 75 | -0.051727 | 0.326667 | 0.471017 | 0.278068 | 0.425439 | 0.836750 | 0.997778 | 0.014263 | 0.155475 |
| 76 | -0.051527 | 0.351111 | 0.946686 | 0.672213 | 0.584773 | 0.176875 | 0.766667 | 0.022891 | 0.249103 |
| 77 | -0.051326 | 0.315556 | 0.200053 | 0.205448 | 0.922101 | 0.816000 | 0.535556 | 0.017286 | 0.166983 |
| 78 | -0.051126 | 0.280000 | 0.727253 | 0.235093 | 0.332339 | 0.134625 | 0.428889 | 0.029858 | 0.319486 |
| 79 | -0.050926 | 0.826667 | 0.399525 | 0.974824 | 0.838394 | 0.735750 | 0.657778 | 0.012734 | 0.158465 |
| 80 | -0.050726 | 0.371111 | 0.647665 | 0.003057 | 0.620098 | 0.950375 | 0.262222 | 0.025721 | 0.220245 |
| 81 | -0.050326 | 0.695556 | 0.807020 | 0.599280 | 0.781994 | 0.096000 | 0.175556 | 0.029547 | 0.290018 |
| 82 | -0.050126 | 0.051111 | 0.633667 | 0.282684 | 0.271739 | 0.911625 | 0.580000 | 0.027902 | 0.210588 |
| 83 | -0.049926 | 0.020000 | 0.534343 | 0.617340 | 0.530142 | 0.520500 | 0.993333 | 0.031120 | 0.290700 |


| 84 | -0.049726 | 0.686667 | 0.169938 | 0.100518 | 0.310041 | 0.917125 | 0.175556 | 0.011980 | 0.086655 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 85 | -0.049526 | 0.766667 | 0.619915 | 0.379898 | 0.119707 | 0.225000 | 0.968889 | 0.013141 | 0.223436 |
| 86 | -0.049326 | 0.295556 | 0.582519 | 0.419245 | 0.971311 | 0.728375 | 0.704444 | 0.020966 | 0.214164 |
| 87 | -0.048926 | 0.260000 | 0.420615 | 0.257711 | 0.315000 | 0.716750 | 0.755556 | 0.017145 | 0.183030 |
| 88 | -0.048526 | 0.762222 | 0.614624 | 0.735811 | 0.174567 | 0.521500 | 0.944444 | 0.012230 | 0.189229 |
| 89 | 0.068068 | 0.060000 | 0.593403 | 0.328508 | 0.164352 | 0.434375 | 0.508889 | 0.033093 | 0.345918 |
| 90 | 0.069069 | 0.304444 | 0.955996 | 0.265880 | 0.776848 | 0.146125 | 0.071111 | 0.048936 | 0.471479 |
| 91 | 0.070070 | 0.202222 | 0.979853 | 0.780667 | 0.303459 | 0.944000 | 0.706667 | 0.021106 | 0.226001 |
| 92 | 0.074074 | 0.062222 | 0.052080 | 0.158088 | 0.390625 | 0.853875 | 0.566667 | 0.014296 | 0.092095 |
| 93 | 0.078078 | 0.331111 | 0.198795 | 0.405104 | 0.093750 | 0.796750 | 0.064444 | 0.018651 | 0.105527 |
| 94 | 0.079079 | 0.644444 | 0.066936 | 0.992881 | 0.405754 | 0.613625 | 0.786667 | 0.010343 | 0.155209 |
| 95 | 0.092092 | 0.002222 | 0.068371 | 0.584414 | 0.706284 | 0.184125 | 0.388889 | 0.038159 | 0.292244 |
| 96 | 0.094094 | 0.242222 | 0.063123 | 0.683779 | 0.272727 | 0.856875 | 0.246667 | 0.011372 | 0.065362 |
| 97 | 0.095095 | 0.208889 | 0.733465 | 0.483966 | 0.464080 | 0.612625 | 0.791111 | 0.023234 | 0.302654 |
| 98 | 0.097097 | 0.433333 | 0.735862 | 0.756848 | 0.716117 | 0.705750 | 0.040000 | 0.039721 | 0.324364 |
| 99 | 0.101101 | 0.500000 | 0.952354 | 0.841334 | 0.091592 | 0.753750 | 0.455556 | 0.019350 | 0.241314 |
| 100 | 0.102102 | 0.053333 | 0.839860 | 0.186358 | 0.371795 | 0.961000 | 0.235556 | 0.044963 | 0.302679 |
| 101 | 0.105105 | 0.104444 | 0.273159 | 0.622575 | 0.597953 | 0.611625 | 0.446667 | 0.026358 | 0.246368 |
| 102 | 0.052811 | 0.284444 | 0.574972 | 0.111369 | 0.473389 | 0.265250 | 0.731111 | 0.024573 | 0.276880 |
| 103 | 0.053011 | 0.497778 | 0.082609 | 0.519410 | 0.168651 | 0.454875 | 0.471111 | 0.012926 | 0.138976 |
| 104 | 0.053211 | 0.773333 | 0.509279 | 0.739783 | 0.663450 | 0.591750 | 0.946667 | 0.013325 | 0.175257 |
| 105 | 0.053411 | 0.040000 | 0.029296 | 0.247662 | 0.490741 | 0.849000 | 0.013333 | 0.050058 | 0.266471 |
| 106 | 0.053611 | 0.062222 | 0.788318 | 0.209968 | 0.175926 | 0.859000 | 0.311111 | 0.039958 | 0.280492 |
| 107 | 0.053811 | 0.177778 | 0.717906 | 0.192025 | 0.442460 | 0.992375 | 0.977778 | 0.019985 | 0.163051 |
| 108 | 0.054211 | 0.357778 | 0.330628 | 0.081447 | 0.128788 | 0.598750 | 0.695556 | 0.015107 | 0.152343 |
| 109 | 0.054411 | 0.973333 | 0.431311 | 0.027905 | 0.855357 | 0.806750 | 0.353333 | 0.017466 | 0.181302 |
| 110 | 0.054611 | 0.731111 | 0.265106 | 0.737820 | 0.570175 | 0.117875 | 0.691111 | 0.014928 | 0.202525 |
| 111 | 0.054811 | 0.617778 | 0.253388 | 0.642958 | 0.602543 | 0.011250 | 0.180000 | 0.027717 | 0.278669 |
| 112 | 0.055211 | 0.268889 | 0.764330 | 0.028427 | 0.223621 | 0.621875 | 0.875556 | 0.021174 | 0.229642 |


| 113 | 0.058012 | 0.644444 | 0.902605 | 0.416940 | 0.528220 | 0.871375 | 0.180000 | 0.028419 | 0.255468 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 114 | 0.058212 | 0.815556 | 0.226520 | 0.924416 | 0.003652 | 0.560125 | 0.100000 | 0.016075 | 0.150803 |
| 115 | 0.058412 | 0.375556 | 0.814917 | 0.982254 | 0.671053 | 0.931000 | 0.757778 | 0.019764 | 0.187292 |
| 116 | 0.058812 | 0.566667 | 0.576827 | 0.548445 | 0.134780 | 0.381125 | 0.395556 | 0.022265 | 0.252230 |
| 117 | 0.059012 | 0.873333 | 0.340685 | 0.387744 | 0.684080 | 0.641500 | 0.900000 | 0.011868 | 0.154728 |
| 118 | 0.059412 | 0.586667 | 0.754678 | 0.300555 | 0.226763 | 0.988625 | 0.213333 | 0.021812 | 0.166492 |
| 119 | 0.059812 | 0.426667 | 0.468609 | 0.438658 | 0.823506 | 0.093000 | 0.964444 | 0.018085 | 0.220522 |
| 120 | 0.060412 | 0.635556 | 0.411362 | 0.458607 | 0.847733 | 0.176000 | 0.524444 | 0.019560 | 0.225528 |
| 121 | 0.060612 | 0.586667 | 0.867174 | 0.095585 | 0.882427 | 0.457500 | 0.315556 | 0.027902 | 0.303479 |
| 122 | 0.060812 | 0.360000 | 0.485213 | 0.833378 | 0.471557 | 0.500875 | 0.393333 | 0.026008 | 0.247834 |
| 123 | 0.061012 | 0.744444 | 0.732171 | 0.727166 | 0.457612 | 0.686375 | 0.095556 | 0.031797 | 0.283122 |
| 124 | 0.061412 | 0.480000 | 0.863518 | 0.622240 | 0.195445 | 0.361625 | 0.824444 | 0.019958 | 0.256731 |
| 125 | 0.061612 | 0.326667 | 0.764362 | 0.849696 | 0.493590 | 0.764125 | 0.424444 | 0.026968 | 0.243732 |
| 126 | 0.061812 | 0.513333 | 0.537508 | 0.220556 | 0.023990 | 0.411250 | 0.860000 | 0.015837 | 0.221112 |
| 127 | 0.062012 | 0.977778 | 0.291767 | 0.377539 | 0.568519 | 0.453750 | 0.088889 | 0.025609 | 0.231223 |
| 128 | 0.062212 | 0.533333 | 0.937107 | 0.459231 | 0.942569 | 0.044000 | 0.926667 | 0.019379 | 0.240755 |
| 129 | 0.146146 | 0.115556 | 0.546539 | 0.793305 | 0.040323 | 0.837875 | 0.780000 | 0.016223 | 0.140742 |
| 130 | 0.148148 | 0.048889 | 0.142283 | 0.436872 | 0.939038 | 0.087125 | 0.066667 | 0.070031 | 0.442732 |
| 131 | 0.152152 | 0.568889 | 0.443886 | 0.565129 | 0.993827 | 0.219250 | 0.120000 | 0.028936 | 0.314396 |
| 132 | 0.154154 | 0.071111 | 0.487161 | 0.286004 | 0.567901 | 0.658625 | 0.706667 | 0.027631 | 0.290666 |
| 133 | 0.156156 | 0.235556 | 0.969826 | 0.829411 | 0.533936 | 0.139125 | 0.175556 | 0.040803 | 0.406783 |
| 134 | 0.105421 | 0.011111 | 0.035752 | 0.526012 | 0.448925 | 0.491125 | 0.264444 | 0.035099 | 0.235436 |
| 135 | 0.105621 | 0.100000 | 0.233249 | 0.410995 | 0.414319 | 0.416875 | 0.260000 | 0.033337 | 0.282083 |
| 136 | 0.106221 | 0.213333 | 0.314884 | 0.988415 | 0.688406 | 0.581375 | 0.384444 | 0.023985 | 0.243869 |
| 137 | 0.106421 | 0.315556 | 0.217290 | 0.742619 | 0.073970 | 0.917625 | 0.262222 | 0.008302 | 0.038085 |
| 138 | 0.106821 | 0.115556 | 0.369235 | 0.608308 | 0.935185 | 0.911750 | 0.768889 | 0.021510 | 0.206564 |
| 139 | 0.109222 | 0.364444 | 0.784861 | 0.014166 | 0.960884 | 0.302500 | 0.820000 | 0.025240 | 0.431991 |
| 140 | 0.110422 | 0.791111 | 0.977944 | 0.022151 | 0.296899 | 0.492750 | 0.035556 | 0.035493 | 0.455846 |
| 141 | 0.111022 | 0.457778 | 0.490443 | 0.038345 | 0.814706 | 0.364250 | 0.208889 | 0.029070 | 0.362154 |


| 142 | 0.111622 | 0.200000 | 0.410369 | 0.649201 | 0.472222 | 0.826625 | 0.480000 | 0.021119 | 0.205518 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 143 | 0.112623 | 0.884444 | 0.021951 | 0.065983 | 0.054630 | 0.847000 | 0.106667 | 0.000547 | 0.000233 |
| 144 | 0.113023 | 0.140000 | 0.563543 | 0.306279 | 0.605000 | 0.457875 | 0.835556 | 0.024984 | 0.320574 |
| 145 | 0.113223 | 0.037778 | 0.307167 | 0.466671 | 0.993386 | 0.126625 | 0.317778 | 0.044493 | 0.358006 |
| 146 | 0.114623 | 0.171111 | 0.992855 | 0.480493 | 0.492938 | 0.609875 | 0.284444 | 0.036665 | 0.369472 |
| 147 | 0.115223 | 0.075556 | 0.079083 | 0.062727 | 0.260057 | 0.150875 | 0.153333 | 0.039503 | 0.336174 |
| 148 | 0.115423 | 0.020000 | 0.764764 | 0.466639 | 0.054348 | 0.365250 | 0.277778 | 0.055262 | 0.474856 |
| 149 | 0.116223 | 0.146667 | 0.992413 | 0.718224 | 0.621930 | 0.277625 | 0.868889 | 0.027014 | 0.374380 |
| 150 | 0.117023 | 0.104444 | 0.664256 | 0.145973 | 0.283129 | 0.070875 | 0.562222 | 0.033338 | 0.429983 |
| 151 | 0.117223 | 0.351111 | 0.422789 | 0.963117 | 0.895358 | 0.317500 | 0.308889 | 0.025146 | 0.281431 |
| 152 | 0.117423 | 0.431111 | 0.080569 | 0.267460 | 0.733498 | 0.922750 | 0.393333 | 0.013693 | 0.120289 |
| 153 | 0.118224 | 0.577778 | 0.086488 | 0.703997 | 0.638889 | 0.846375 | 0.240000 | 0.013968 | 0.123725 |
| 154 | 0.118824 | 0.031111 | 0.819745 | 0.915765 | 0.477891 | 0.733125 | 0.824444 | 0.030257 | 0.291090 |
| 155 | 0.119224 | 0.855556 | 0.584030 | 0.042962 | 0.010305 | 0.487750 | 0.220000 | 0.020662 | 0.320641 |
| 156 | 0.119424 | 0.466667 | 0.492916 | 0.903443 | 0.639937 | 0.768125 | 0.260000 | 0.021794 | 0.232574 |
| 157 | 0.119624 | 0.093333 | 0.971581 | 0.603478 | 0.521386 | 0.401500 | 0.748889 | 0.030414 | 0.379736 |
| 158 | 0.120224 | 0.064444 | 0.157568 | 0.955302 | 0.073529 | 0.989375 | 0.828889 | 0.003264 | 0.006157 |
| 159 | 0.120824 | 0.175556 | 0.493627 | 0.664071 | 0.083333 | 0.294625 | 0.148889 | 0.038647 | 0.352536 |
| 160 | 0.121024 | 0.175556 | 0.804679 | 0.118390 | 0.468519 | 0.899375 | 0.395556 | 0.028640 | 0.292838 |
| 161 | 0.121224 | 0.362222 | 0.668978 | 0.039518 | 0.166058 | 0.889125 | 0.011111 | 0.039253 | 0.268657 |
| 162 | 0.121624 | 0.244444 | 0.030677 | 0.747817 | 0.015537 | 0.193000 | 0.126667 | 0.021480 | 0.199416 |
| 163 | 0.123625 | 0.133333 | 0.293217 | 0.539161 | 0.898148 | 0.934375 | 0.382222 | 0.025341 | 0.208183 |
| 164 | 0.123825 | 0.566667 | 0.730555 | 0.024373 | 0.012483 | 0.603500 | 0.960000 | 0.016836 | 0.293585 |
| 165 | 0.124225 | 0.017778 | 0.165665 | 0.306172 | 0.685897 | 0.881375 | 0.357778 | 0.032051 | 0.208914 |
| 166 | 0.124625 | 0.173333 | 0.026127 | 0.868010 | 0.728311 | 0.183375 | 0.255556 | 0.025982 | 0.242825 |
| 167 | 0.125025 | 0.560000 | 0.204000 | 0.793192 | 0.685146 | 0.042000 | 0.582222 | 0.017526 | 0.272774 |
| 168 | 0.125425 | 0.691111 | 0.508224 | 0.913815 | 0.904047 | 0.340250 | 0.202222 | 0.023252 | 0.284726 |
| 169 | 0.127626 | 0.437778 | 0.856418 | 0.564595 | 0.030702 | 0.256625 | 0.573333 | 0.023905 | 0.387605 |
| 170 | 0.127826 | 0.055556 | 0.188144 | 0.959171 | 0.127451 | 0.905375 | 0.886667 | 0.009815 | 0.053908 |


| 171 | 0.128026 | 0.940000 | 0.446247 | 0.232528 | 0.143894 | 0.568500 | 0.822222 | 0.014502 | 0.301706 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 172 | 0.128426 | 0.264444 | 0.165838 | 0.288000 | 0.088442 | 0.143375 | 0.680000 | 0.018692 | 0.244273 |
| 173 | 0.198198 | 0.102222 | 0.481648 | 0.624710 | 0.227169 | 0.003000 | 0.293333 | 0.038382 | 0.373624 |
| 174 | 0.199199 | 0.162222 | 0.319183 | 0.383996 | 0.040870 | 0.223250 | 0.788889 | 0.021106 | 0.289665 |
| 175 | 0.206206 | 0.831111 | 0.264889 | 0.848299 | 0.810099 | 0.543500 | 0.262222 | 0.019091 | 0.270723 |
| 176 | 0.210210 | 0.542222 | 0.193778 | 0.513254 | 0.289506 | 0.898875 | 0.453333 | 0.010796 | 0.098122 |
| 177 | 0.170234 | 0.453333 | 0.197667 | 0.156590 | 0.882924 | 0.492125 | 0.811111 | 0.017258 | 0.286636 |
| 178 | 0.174035 | 0.268889 | 0.795825 | 0.973925 | 0.883947 | 0.501500 | 0.926667 | 0.021788 | 0.308944 |
| 179 | 0.174235 | 0.137778 | 0.518313 | 0.991378 | 0.129487 | 0.941750 | 0.528889 | 0.017081 | 0.123707 |
| 180 | 0.174435 | 0.566667 | 0.397400 | 0.942268 | 0.011947 | 0.142875 | 0.164444 | 0.023258 | 0.298723 |
| 181 | 0.175035 | 0.960000 | 0.266953 | 0.356421 | 0.769540 | 0.589750 | 0.013333 | 0.026919 | 0.262460 |
| 182 | 0.175235 | 0.195556 | 0.709521 | 0.181127 | 0.756024 | 0.339250 | 0.706667 | 0.027882 | 0.392168 |
| 183 | 0.176035 | 0.084444 | 0.267875 | 0.963716 | 0.850000 | 0.657750 | 0.011111 | 0.072207 | 0.403775 |
| 184 | 0.176435 | 0.020000 | 0.902425 | 0.675482 | 0.875000 | 0.614750 | 0.906667 | 0.033342 | 0.346014 |
| 185 | 0.176635 | 0.040000 | 0.967860 | 0.913761 | 0.148455 | 0.180000 | 0.797778 | 0.034194 | 0.407142 |
| 186 | 0.176835 | 0.244444 | 0.521831 | 0.133644 | 0.852657 | 0.457250 | 0.473333 | 0.027328 | 0.342553 |
| 187 | 0.178436 | 0.540000 | 0.906620 | 0.555754 | 0.868421 | 0.410250 | 0.108889 | 0.033498 | 0.376280 |
| 188 | 0.180236 | 0.066667 | 0.206230 | 0.675409 | 0.263514 | 0.881000 | 0.795556 | 0.017125 | 0.114556 |
| 189 | 0.180436 | 0.900000 | 0.928453 | 0.224831 | 0.531604 | 0.140625 | 0.822222 | 0.016854 | 0.256689 |
| 190 | 0.180636 | 0.733333 | 0.508137 | 0.065167 | 0.423554 | 0.722500 | 0.591111 | 0.015979 | 0.186784 |
| 191 | 0.180836 | 0.215556 | 0.115271 | 0.062144 | 0.292328 | 0.890750 | 0.453333 | 0.013887 | 0.077826 |
| 192 | 0.181036 | 0.068889 | 0.066535 | 0.387377 | 0.882576 | 0.534875 | 0.122222 | 0.056029 | 0.385397 |
| 193 | 0.181236 | 0.086667 | 0.766057 | 0.254935 | 0.588710 | 0.853125 | 0.417778 | 0.040771 | 0.331983 |
| 194 | 0.181436 | 0.646667 | 0.579896 | 0.028705 | 0.290340 | 0.438125 | 0.960000 | 0.015636 | 0.238943 |
| 195 | 0.181636 | 0.340000 | 0.006534 | 0.132802 | 0.982932 | 0.777750 | 0.164444 | 0.025173 | 0.166730 |
| 196 | 0.181836 | 0.768889 | 0.521016 | 0.781288 | 0.404487 | 0.690500 | 0.675556 | 0.014504 | 0.176190 |
| 197 | 0.182036 | 0.882222 | 0.492430 | 0.510633 | 0.376029 | 0.570375 | 0.753333 | 0.013614 | 0.183918 |
| 198 | 0.182436 | 0.711111 | 0.505659 | 0.698564 | 0.755236 | 0.016000 | 0.677778 | 0.017557 | 0.229424 |
| 199 | 0.182637 | 0.831111 | 0.681461 | 0.262798 | 0.838130 | 0.199625 | 0.646667 | 0.018042 | 0.241521 |


| 200 | 0.182837 | 0.622222 | 0.783863 | 0.450878 | 0.758638 | 0.466125 | 0.706667 | 0.019424 | 0.240306 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 201 | 0.183037 | 0.546667 | 0.252806 | 0.065984 | 0.130952 | 0.686750 | 0.722222 | 0.011261 | 0.115804 |
| 202 | 0.183237 | 0.277778 | 0.612127 | 0.902290 | 0.989411 | 0.384875 | 0.708889 | 0.025548 | 0.262163 |
| 203 | 0.183437 | 0.513333 | 0.219228 | 0.166105 | 0.286285 | 0.290875 | 0.802222 | 0.014872 | 0.194327 |
| 204 | 0.183637 | 0.348889 | 0.884038 | 0.853473 | 0.608918 | 0.435500 | 0.691111 | 0.025049 | 0.274253 |
| 205 | 0.183837 | 0.060000 | 0.275258 | 0.720810 | 0.162281 | 0.924750 | 0.573333 | 0.017766 | 0.098304 |
| 206 | 0.184037 | 0.262222 | 0.315577 | 0.157764 | 0.681944 | 0.485875 | 0.593333 | 0.024404 | 0.234541 |
| 207 | 0.184437 | 0.673333 | 0.987694 | 0.055344 | 0.945152 | 0.854625 | 0.495556 | 0.021796 | 0.257814 |
| 208 | 0.185037 | 0.977778 | 0.546302 | 0.094839 | 0.131492 | 0.607875 | 0.506667 | 0.014695 | 0.189816 |
| 209 | 0.185237 | 0.637778 | 0.106164 | 0.717791 | 0.138112 | 0.908750 | 0.664444 | 0.004103 | 0.021944 |
| 210 | 0.186237 | 0.086667 | 0.435829 | 0.464282 | 0.955882 | 0.165625 | 0.462222 | 0.043790 | 0.367314 |
| 211 | 0.186437 | 0.691111 | 0.280738 | 0.883979 | 0.426083 | 0.914750 | 0.488889 | 0.012311 | 0.114622 |
| 212 | 0.186637 | 0.220000 | 0.069716 | 0.024638 | 0.784314 | 0.716375 | 0.711111 | 0.017663 | 0.148245 |
| 213 | 0.187037 | 0.393333 | 0.429964 | 0.649297 | 0.848101 | 0.566875 | 0.140000 | 0.038143 | 0.317801 |
| 214 | 0.187237 | 0.235556 | 0.336677 | 0.945856 | 0.420068 | 0.596125 | 0.337778 | 0.028570 | 0.253913 |
| 215 | 0.188038 | 0.986667 | 0.053968 | 0.779470 | 0.137640 | 0.353875 | 0.313333 | 0.011162 | 0.140198 |
| 216 | 0.188238 | 0.513333 | 0.992138 | 0.895893 | 0.039831 | 0.504500 | 0.055556 | 0.044895 | 0.423843 |
| 217 | 0.188838 | 0.366667 | 0.281986 | 0.869080 | 0.461111 | 0.944000 | 0.915556 | 0.012181 | 0.110692 |
| 218 | 0.189038 | 0.431111 | 0.754726 | 0.914574 | 0.474044 | 0.334250 | 0.288889 | 0.031918 | 0.318368 |
| 219 | 0.189238 | 0.742222 | 0.973835 | 0.553683 | 0.918750 | 0.504125 | 0.497778 | 0.021663 | 0.255882 |
| 220 | 0.189638 | 0.728889 | 0.755016 | 0.705638 | 0.003623 | 0.933000 | 0.024444 | 0.024190 | 0.150466 |
| 221 | 0.190238 | 0.357778 | 0.248831 | 0.932245 | 0.173077 | 0.556875 | 0.877778 | 0.013504 | 0.189568 |
| 222 | 0.190638 | 0.242222 | 0.858106 | 0.610279 | 0.909292 | 0.928625 | 0.557778 | 0.028531 | 0.311451 |
| 223 | 0.191038 | 0.800000 | 0.661859 | 0.969507 | 0.330513 | 0.584750 | 0.188889 | 0.025165 | 0.308037 |
| 224 | 0.191238 | 0.411111 | 0.625522 | 0.282748 | 0.983333 | 0.938375 | 0.762222 | 0.019926 | 0.275976 |
| 225 | 0.191438 | 0.006667 | 0.973384 | 0.414248 | 0.637597 | 0.772125 | 0.726667 | 0.045247 | 0.459219 |
| 226 | 0.191638 | 0.364444 | 0.157359 | 0.423764 | 0.787634 | 0.984375 | 0.173333 | 0.024723 | 0.195819 |
| 227 | 0.191838 | 0.891111 | 0.205483 | 0.314199 | 0.154762 | 0.393875 | 0.417778 | 0.013729 | 0.209439 |
| 228 | 0.192038 | 0.477778 | 0.740887 | 0.824949 | 0.932870 | 0.070625 | 0.328889 | 0.030323 | 0.368775 |


| 229 | 0.192238 | 0.466667 | 0.124620 | 0.998545 | 0.057163 | 0.748500 | 0.573333 | 0.006451 | 0.057956 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 230 | 0.192639 | 0.397778 | 0.721412 | 0.987814 | 0.097756 | 0.197875 | 0.775556 | 0.021620 | 0.335034 |
| 231 | 0.192839 | 0.386667 | 0.607335 | 0.727554 | 0.845716 | 0.347125 | 0.068889 | 0.051090 | 0.578960 |
| 232 | 0.193239 | 0.342222 | 0.250521 | 0.094081 | 0.262027 | 0.929375 | 0.411111 | 0.014931 | 0.116438 |
| 233 | 0.193439 | 0.293333 | 0.601050 | 0.209021 | 0.382132 | 0.941375 | 0.468889 | 0.023063 | 0.230509 |
| 234 | 0.193839 | 0.086667 | 0.583722 | 0.673609 | 0.297222 | 0.715500 | 0.077778 | 0.065364 | 0.532095 |
| 235 | 0.219219 | 0.655556 | 0.216961 | 0.390382 | 0.249077 | 0.260250 | 0.037778 | 0.035262 | 0.325299 |
| 236 | 0.221221 | 0.155556 | 0.101447 | 0.427283 | 0.241870 | 0.403375 | 0.448889 | 0.024169 | 0.206632 |
| 237 | 0.222222 | 0.402222 | 0.168299 | 0.214271 | 0.988562 | 0.999000 | 0.626667 | 0.015482 | 0.134388 |
| 238 | 0.223223 | 0.253333 | 0.138889 | 0.782391 | 0.990964 | 0.656625 | 0.182222 | 0.033250 | 0.267097 |
| 239 | 0.224224 | 0.546667 | 0.641835 | 0.719016 | 0.325472 | 0.677625 | 0.537778 | 0.019520 | 0.213394 |
| 240 | 0.225225 | 0.184444 | 0.372164 | 0.837220 | 0.268116 | 0.542500 | 0.742222 | 0.022264 | 0.220208 |
| 241 | 0.227227 | 0.486667 | 0.185810 | 0.766186 | 0.655568 | 0.133125 | 0.020000 | 0.045229 | 0.388031 |
| 242 | 0.228228 | 0.002222 | 0.860242 | 0.253815 | 0.397059 | 0.489500 | 0.548889 | 0.054096 | 0.482855 |
| 243 | 0.229229 | 0.282222 | 0.584844 | 0.589863 | 0.670910 | 0.092125 | 0.633333 | 0.027223 | 0.290461 |
| 244 | 0.230230 | 0.455556 | 0.431702 | 0.720449 | 0.757317 | 0.461500 | 0.328889 | 0.026568 | 0.258226 |
| 245 | 0.231231 | 0.124444 | 0.529934 | 0.523910 | 0.863757 | 0.870875 | 0.980000 | 0.025064 | 0.226601 |
| 246 | 0.233233 | 0.171111 | 0.560399 | 0.869867 | 0.636640 | 0.156125 | 0.124444 | 0.058583 | 0.482313 |
| 247 | 0.234234 | 0.995556 | 0.107548 | 0.680348 | 0.543899 | 0.781750 | 0.164444 | 0.014477 | 0.131521 |
| 248 | 0.235235 | 0.340000 | 0.668926 | 0.029805 | 0.236111 | 0.867875 | 0.342222 | 0.025347 | 0.221939 |
| 249 | 0.236236 | 0.493333 | 0.073580 | 0.272502 | 0.801146 | 0.148125 | 0.411111 | 0.019701 | 0.213448 |
| 250 | 0.237237 | 0.833333 | 0.033580 | 0.852833 | 0.347368 | 0.689750 | 0.477778 | 0.008381 | 0.087269 |
| 251 | 0.238238 | 0.302222 | 0.059696 | 0.701905 | 0.035211 | 0.328375 | 0.271111 | 0.018281 | 0.174430 |
| 252 | 0.240240 | 0.975556 | 0.884613 | 0.133572 | 0.532018 | 0.269250 | 0.171111 | 0.028714 | 0.338887 |
| 253 | 0.241241 | 0.028889 | 0.642283 | 0.356712 | 0.769841 | 0.775750 | 0.142222 | 0.077608 | 0.514939 |
| 254 | 0.242242 | 0.237778 | 0.325070 | 0.779122 | 0.076765 | 0.076125 | 0.788889 | 0.020948 | 0.260657 |
| 255 | 0.244244 | 0.142222 | 0.934652 | 0.468182 | 0.799383 | 0.562625 | 0.255556 | 0.050073 | 0.433447 |
| 256 | 0.245245 | 0.831111 | 0.727046 | 0.681966 | 0.847113 | 0.514500 | 0.757778 | 0.016340 | 0.214841 |
| 257 | 0.246246 | 0.482222 | 0.417038 | 0.066534 | 0.931507 | 0.885875 | 0.375556 | 0.022970 | 0.212957 |


| 258 | 0.247247 | 0.508889 | -0.000614 | 0.217471 | 0.483918 | 0.766750 | 0.640000 | 0.008923 | 0.081520 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 259 | 0.248248 | 0.651111 | 0.449574 | 0.356537 | 0.138889 | 0.785750 | 0.126667 | 0.022733 | 0.180367 |
| 260 | 0.249249 | 0.613333 | 0.388926 | 0.115547 | 0.754318 | 0.708750 | 0.662222 | 0.016570 | 0.191156 |
| 261 | 0.250250 | 0.782222 | 0.744838 | 0.009642 | 0.574040 | 0.905875 | 0.906667 | 0.014251 | 0.187900 |
| 262 | 0.251251 | 0.553333 | 0.427901 | 0.651198 | 0.602028 | 0.159125 | 0.922222 | 0.016416 | 0.222724 |
| 263 | 0.252252 | 0.431111 | 0.391212 | 0.207920 | 0.915072 | 0.407375 | 0.106667 | 0.041096 | 0.337750 |
| 264 | 0.254254 | 0.160000 | 0.638605 | 0.972135 | 0.178082 | 0.322375 | 0.315556 | 0.041316 | 0.396013 |
| 265 | 0.255255 | 0.993333 | 0.156848 | 0.915059 | 0.263228 | 0.477500 | 0.195556 | 0.015651 | 0.170209 |
| 266 | 0.257257 | 0.660000 | 0.864459 | 0.274364 | 0.608724 | 0.829875 | 0.708889 | 0.017683 | 0.210114 |
| 267 | 0.258258 | 0.397778 | 0.400104 | 0.510348 | 0.429431 | 0.128125 | 0.128889 | 0.039944 | 0.361569 |
| 268 | 0.259259 | 0.606667 | 0.720176 | 0.119464 | 0.163171 | 0.084125 | 0.895556 | 0.018318 | 0.283275 |
| 269 | 0.261261 | 0.235556 | 0.772135 | 0.073082 | 0.785826 | 0.875875 | 0.844444 | 0.024285 | 0.254487 |
| 270 | 0.262262 | 0.373333 | 0.341214 | 0.717396 | 0.016667 | 0.850875 | 0.306667 | 0.011907 | 0.073554 |
| 271 | 0.210642 | 0.226667 | 0.683234 | 0.692613 | 0.988095 | 0.672875 | 0.128889 | 0.051600 | 0.410693 |
| 272 | 0.210842 | 0.435556 | 0.713217 | 0.919530 | 0.261561 | 0.762750 | 0.733333 | 0.018107 | 0.194775 |
| 273 | 0.211042 | 0.082222 | 0.965912 | 0.397969 | 0.179952 | 0.801750 | 0.895556 | 0.029409 | 0.269285 |
| 274 | 0.211242 | 0.700000 | 0.841723 | 0.921094 | 0.284038 | 0.722125 | 0.840000 | 0.015279 | 0.189923 |
| 275 | 0.211442 | 0.560000 | 0.580638 | 0.278707 | 0.215313 | 0.755375 | 0.697778 | 0.015722 | 0.174714 |
| 276 | 0.211642 | 0.100000 | 0.313910 | 0.260142 | 0.198357 | 0.178625 | 0.524444 | 0.034501 | 0.322750 |
| 277 | 0.211842 | 0.615556 | 0.553301 | 0.714703 | 0.844884 | 0.818125 | 0.200000 | 0.027351 | 0.245005 |
| 278 | 0.212042 | 0.282222 | 0.172349 | 0.184322 | 0.776389 | 0.030625 | 0.740000 | 0.021476 | 0.231779 |
| 279 | 0.212242 | 0.088889 | 0.151411 | 0.243163 | 0.466049 | 0.510125 | 0.542222 | 0.029877 | 0.233346 |
| 280 | 0.212442 | 0.195556 | 0.789298 | 0.864128 | 0.717120 | 0.423875 | 0.371111 | 0.039586 | 0.365135 |
| 281 | 0.213243 | 0.451111 | 0.358587 | 0.109770 | 0.531457 | 0.253625 | 0.522222 | 0.022427 | 0.251003 |
| 282 | 0.213443 | 0.526667 | 0.253746 | 0.055881 | 0.038440 | 0.275000 | 0.275556 | 0.021804 | 0.239513 |
| 283 | 0.213643 | 0.975556 | 0.041254 | 0.433875 | 0.399916 | 0.858375 | 0.271111 | 0.008540 | 0.066902 |
| 284 | 0.214243 | 0.242222 | 0.149360 | 0.027334 | 0.514678 | 0.078000 | 0.475556 | 0.026466 | 0.245484 |
| 285 | 0.214443 | 0.068889 | 0.476036 | 0.448472 | 0.112434 | 0.500750 | 0.448889 | 0.038898 | 0.344409 |
| 286 | 0.214643 | 0.926667 | 0.276693 | 0.145401 | 0.826634 | 0.095875 | 0.268889 | 0.021017 | 0.237887 |


| 287 | 0.214843 | 0.486667 | 0.422157 | 0.851065 | 0.665535 | 0.443875 | 0.915556 | 0.016502 | 0.206722 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 288 | 0.215043 | 0.073333 | 0.343815 | 0.519697 | 0.221491 | 0.357500 | 0.660000 | 0.032471 | 0.295600 |
| 289 | 0.215243 | 0.917778 | 0.352867 | 0.913509 | 0.153997 | 0.315250 | 0.966667 | 0.011617 | 0.184381 |
| 290 | 0.215643 | 0.460000 | 0.289457 | 0.381162 | 0.188328 | 0.216500 | 0.511111 | 0.019949 | 0.235513 |
| 291 | 0.215843 | 0.540000 | 0.827551 | 0.481105 | 0.997126 | 0.566375 | 0.631111 | 0.021864 | 0.251217 |
| 292 | 0.216043 | 0.466667 | 0.630307 | 0.427268 | 0.452273 | 0.111875 | 0.395556 | 0.027542 | 0.303034 |
| 293 | 0.216243 | 1.000000 | 0.492292 | 0.092428 | 0.675970 | 0.182875 | 0.944444 | 0.014188 | 0.221285 |
| 294 | 0.216443 | 0.773333 | 0.142546 | 0.745563 | 0.109010 | 0.052000 | 0.053333 | 0.026140 | 0.272564 |
| 295 | 0.216643 | 0.162222 | 0.467874 | 0.165223 | 0.852941 | 0.987375 | 0.826667 | 0.023593 | 0.202374 |
| 296 | 0.216843 | 0.808889 | 0.963521 | 0.803711 | 0.974858 | 0.379625 | 0.715556 | 0.018241 | 0.237171 |
| 297 | 0.271271 | 0.031111 | 0.081473 | 0.069468 | 0.551802 | 0.490500 | 0.664444 | 0.025424 | 0.245749 |
| 298 | 0.272272 | 0.068889 | 0.163838 | 0.234585 | 0.876984 | 0.642625 | 0.353333 | 0.031906 | 0.272527 |
| 299 | 0.274274 | 0.795556 | 0.342698 | 0.970492 | 0.120011 | 0.458500 | 0.415556 | 0.015239 | 0.220293 |
| 300 | 0.275275 | 0.082222 | 0.624820 | 0.266877 | 0.932471 | 0.277250 | 0.022222 | 0.095834 | 0.632987 |
| 301 | 0.276276 | 0.722222 | 0.694677 | 0.149459 | 0.477969 | 0.760750 | 0.140000 | 0.025918 | 0.313858 |
| 302 | 0.277277 | 0.060000 | 0.133871 | 0.957146 | 0.735294 | 0.882875 | 0.077778 | 0.045570 | 0.242027 |
| 303 | 0.280280 | 0.835556 | 0.012242 | 0.198286 | 0.401341 | 0.795750 | 0.151111 | 0.010471 | 0.082497 |
| 304 | 0.281281 | 0.528889 | 0.403614 | 0.651722 | 0.014192 | 0.455500 | 0.051111 | 0.028014 | 0.274785 |
| 305 | 0.283283 | 0.084444 | 0.624628 | 0.650606 | 0.747685 | 0.537500 | 0.471111 | 0.034575 | 0.337771 |
| 306 | 0.287287 | 0.946667 | 0.315419 | 0.166372 | 0.953401 | 0.852875 | 0.131111 | 0.020905 | 0.239154 |
| 307 | 0.289289 | 0.182222 | 0.859606 | 0.802557 | 0.996599 | 0.805750 | 0.413333 | 0.030276 | 0.317639 |
| 308 | 0.293293 | 0.917778 | 0.939317 | 0.520248 | 0.029981 | 0.842875 | 0.326667 | 0.016218 | 0.199990 |
| 309 | 0.296296 | 0.573333 | 0.409904 | 0.090473 | 0.174390 | 0.594625 | 0.653333 | 0.016592 | 0.245160 |
| 310 | 0.297297 | 0.264444 | 0.663287 | 0.400452 | 0.088164 | 0.218250 | 0.215556 | 0.034461 | 0.409548 |
| 311 | 0.301301 | 0.862222 | 0.576994 | 0.885188 | 0.660394 | 0.792750 | 0.246667 | 0.019410 | 0.240641 |
| 312 | 0.302302 | 0.384444 | 0.091751 | 0.657105 | 0.628713 | 0.273250 | 0.966667 | 0.014139 | 0.205650 |
| 313 | 0.312312 | 0.144444 | 0.773619 | 0.306877 | 0.298780 | 0.793750 | 0.748889 | 0.024503 | 0.289602 |
| 314 | 0.263253 | 0.733333 | 0.732542 | 0.525844 | 0.424383 | 0.558875 | 0.402222 | 0.021610 | 0.247680 |
| 315 | 0.263453 | 0.580000 | 0.462715 | 0.761338 | 0.579268 | 0.392875 | 0.108889 | 0.035327 | 0.320594 |


| 316 | 0.263653 | 0.242222 | 0.459600 | 0.448197 | 0.411364 | 0.080250 | 0.904444 | 0.023255 | 0.268621 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 317 | 0.263853 | 0.973333 | 0.657046 | 0.670260 | 0.304603 | 0.329125 | 0.566667 | 0.016836 | 0.235905 |
| 318 | 0.264053 | 0.751111 | 0.867025 | 0.255817 | 0.622329 | 0.694000 | 0.040000 | 0.040420 | 0.371943 |
| 319 | 0.264253 | 0.655556 | 0.786100 | 0.688284 | 0.521277 | 0.411875 | 0.966667 | 0.016615 | 0.232082 |
| 320 | 0.264453 | 0.306667 | 0.014244 | 0.714981 | 0.428419 | 0.279000 | 0.100000 | 0.031133 | 0.267119 |
| 321 | 0.264653 | 0.953333 | 0.802596 | 0.918047 | 0.329840 | 0.222875 | 0.551111 | 0.018167 | 0.249551 |
| 322 | 0.264853 | 0.240000 | 0.130189 | 0.378779 | 0.681442 | 0.109375 | 0.506667 | 0.025324 | 0.245959 |
| 323 | 0.265053 | 0.802222 | 0.831045 | 0.580138 | 0.248153 | 0.580125 | 0.142222 | 0.029392 | 0.299452 |
| 324 | 0.265253 | 0.848889 | 0.191945 | 0.772626 | 0.090752 | 0.374875 | 0.233333 | 0.016264 | 0.187021 |
| 325 | 0.265653 | 0.488889 | 0.983026 | 0.072430 | 0.655340 | 0.897125 | 0.335556 | 0.027707 | 0.288381 |
| 326 | 0.265853 | 0.168889 | 0.724662 | 0.963378 | 0.400407 | 0.979625 | 0.666667 | 0.024645 | 0.200429 |
| 327 | 0.266053 | 0.448889 | 0.298451 | 0.598433 | 0.429654 | 0.599750 | 0.666667 | 0.016659 | 0.183744 |
| 328 | 0.266253 | 0.506667 | 0.594458 | 0.592060 | 0.088745 | 0.346500 | 1.000000 | 0.016348 | 0.238950 |
| 329 | 0.266453 | 0.411111 | 0.389601 | 0.044551 | 0.241870 | 0.993000 | 0.262222 | 0.018342 | 0.117297 |
| 330 | 0.267053 | 0.615556 | 0.763244 | 0.993006 | 0.489651 | 0.499750 | 0.575556 | 0.020488 | 0.241874 |
| 331 | 0.267253 | 0.440000 | 0.064555 | 0.831387 | 0.664667 | 0.594500 | 0.924444 | 0.011551 | 0.142465 |
| 332 | 0.267453 | 0.182222 | 0.371191 | 0.706159 | 0.397287 | 0.040625 | 0.786667 | 0.025488 | 0.276537 |
| 333 | 0.267854 | 0.222222 | 0.651608 | 0.453799 | 0.311869 | 0.495500 | 0.397778 | 0.033942 | 0.327115 |
| 334 | 0.268454 | 0.084444 | 0.292998 | 0.023506 | 0.632576 | 0.299875 | 0.524444 | 0.038228 | 0.320447 |
| 335 | 0.268654 | 0.820000 | 0.589880 | 0.849089 | 0.386022 | 0.609750 | 0.044444 | 0.034483 | 0.306234 |
| 336 | 0.268854 | 0.857778 | 0.658690 | 0.748982 | 0.157566 | 0.381500 | 0.551111 | 0.017359 | 0.240641 |
| 337 | 0.269054 | 0.073333 | 0.390491 | 0.408379 | 0.913522 | 0.618875 | 0.151111 | 0.065289 | 0.458933 |
| 338 | 0.269254 | 0.040000 | 0.788107 | 0.424380 | 0.701977 | 0.576750 | 0.148889 | 0.076513 | 0.541008 |
| 339 | 0.269454 | 0.935556 | 0.222712 | 0.085012 | 0.643880 | 0.249500 | 0.113333 | 0.025964 | 0.248529 |
| 340 | 0.269654 | 0.648889 | 0.196762 | 0.526993 | 0.215262 | 0.060000 | 0.006667 | 0.039733 | 0.360664 |
| 341 | 0.269854 | 0.735556 | 0.018875 | 0.247626 | 0.271545 | 0.197500 | 0.173333 | 0.018157 | 0.188687 |
| 342 | 0.270054 | 0.033333 | 0.323503 | 0.388059 | 0.816667 | 0.233500 | 0.315556 | 0.059645 | 0.450719 |
| 343 | 0.270254 | 0.851111 | 0.364937 | 0.233240 | 0.408590 | 0.913125 | 0.133333 | 0.020230 | 0.157928 |
| 344 | 0.270454 | 0.617778 | 0.424422 | 0.299777 | 0.834590 | 0.709375 | 0.297778 | 0.023535 | 0.228142 |


| 345 | 0.270654 | 0.280000 | 0.016942 | 0.990959 | 0.261905 | 0.841750 | 0.086667 | 0.013773 | 0.066298 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 346 | 0.270854 | 0.795556 | 0.655197 | 0.304616 | 0.116102 | 0.678500 | 0.733333 | 0.013983 | 0.181095 |
| 347 | 0.271054 | 0.586667 | 0.233275 | 0.101552 | 0.230507 | 0.663500 | 0.864444 | 0.011291 | 0.132328 |
| 348 | 0.271654 | 0.148889 | 0.898686 | 0.476023 | 0.274138 | 0.375875 | 0.791111 | 0.031678 | 0.346724 |
| 349 | 0.271854 | 0.475556 | 0.456341 | 0.179714 | 0.524250 | 0.559500 | 0.533333 | 0.021204 | 0.231681 |
| 350 | 0.272054 | 0.415556 | 0.517208 | 0.605645 | 0.555457 | 0.296000 | 0.200000 | 0.035365 | 0.329690 |
| 351 | 0.400400 | 0.093333 | 0.029844 | 0.508093 | 0.147436 | 0.747750 | 0.068889 | 0.018937 | 0.076238 |
| 352 | 0.410410 | 0.055556 | 0.480910 | 0.790144 | 0.817460 | 0.952000 | 0.088889 | 0.060097 | 0.348175 |
| 353 | 0.371274 | 0.551111 | 0.385164 | 0.674658 | 0.588710 | 0.642375 | 0.528889 | 0.017649 | 0.236475 |
| 354 | 0.372474 | 0.155556 | 0.315480 | 0.798860 | 0.865741 | 0.626375 | 0.637778 | 0.022581 | 0.243879 |
| 355 | 0.376075 | 0.613333 | 0.817141 | 0.561812 | 0.104815 | 0.952000 | 0.175556 | 0.019747 | 0.173070 |
| 356 | 0.376875 | 0.528889 | 0.383841 | 1.000751 | 0.315177 | 0.647875 | 0.144444 | 0.022499 | 0.227495 |
| 357 | 0.377475 | 0.637778 | 0.082010 | 0.151675 | 0.071478 | 0.374250 | 0.562222 | 0.012868 | 0.178802 |
| 358 | 0.379076 | 0.946667 | 0.837675 | 0.598309 | 0.308853 | 0.897000 | 0.266667 | 0.018602 | 0.240728 |
| 359 | 0.379676 | 0.268889 | 0.764334 | 0.366142 | 0.324219 | 0.711125 | 0.882222 | 0.020984 | 0.305254 |
| 360 | 0.381276 | 0.040000 | 0.115038 | 0.227418 | 0.088095 | 0.623375 | 0.575556 | 0.017024 | 0.122660 |
| 361 | 0.381876 | 0.346667 | 0.985941 | 0.049957 | 0.089662 | 0.891750 | 0.702222 | 0.018679 | 0.238976 |
| 362 | 0.382476 | 0.057778 | 0.786534 | 0.098835 | 0.370219 | 0.644500 | 0.942222 | 0.029373 | 0.379485 |
| 363 | 0.384077 | 0.428889 | 0.599511 | 0.398072 | 0.402412 | 0.855750 | 0.175556 | 0.026049 | 0.252829 |
| 364 | 0.384477 | 0.440000 | 0.979075 | 0.982873 | 0.466088 | 0.380625 | 0.284444 | 0.028227 | 0.363238 |
| 365 | 0.385877 | 0.417778 | 0.775767 | 0.609626 | 0.190341 | 0.740500 | 0.264444 | 0.024691 | 0.276214 |
| 366 | 0.386277 | 0.024444 | 0.990804 | 0.047983 | 0.677536 | 0.989750 | 0.755556 | 0.034924 | 0.371419 |
| 367 | 0.386677 | 0.782222 | 0.467971 | 0.937545 | 0.326296 | 0.929125 | 0.160000 | 0.017252 | 0.164753 |
| 368 | 0.387277 | 0.026667 | 0.990010 | 0.893476 | 0.354938 | 0.677125 | 0.800000 | 0.033379 | 0.332456 |
| 369 | 0.388078 | 0.342222 | 0.921084 | 0.442962 | 0.663448 | 0.426250 | 0.188889 | 0.035463 | 0.406238 |
| 370 | 0.388878 | 0.753333 | 0.665438 | 0.630670 | 0.700711 | 0.534250 | 0.584444 | 0.018872 | 0.306643 |
| 371 | 0.389078 | 0.700000 | 0.668950 | 0.699316 | 0.022599 | 0.880125 | 0.735556 | 0.010830 | 0.124915 |
| 372 | 0.389478 | 0.415556 | 0.365922 | 0.645152 | 0.798906 | 0.286500 | 0.237778 | 0.025760 | 0.293548 |
| 373 | 0.389878 | 0.104444 | 0.398428 | 0.796294 | 0.017974 | 0.966375 | 0.106667 | 0.018614 | 0.058954 |


| 374 | 0.390478 | 0.033333 | 0.052134 | 0.051329 | 0.714286 | 0.753000 | 0.235556 | 0.034126 | 0.241840 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 375 | 0.390878 | 0.191111 | 0.730434 | 0.066742 | 0.465686 | 0.680750 | 0.615556 | 0.027020 | 0.369431 |
| 376 | 0.391278 | 0.080000 | 0.240446 | 0.692750 | 0.087398 | 0.910750 | 0.495556 | 0.011731 | 0.056057 |
| 377 | 0.392278 | 0.642222 | 0.883542 | 0.006610 | 0.675595 | 0.673125 | 0.346667 | 0.025288 | 0.442896 |
| 378 | 0.392679 | 0.673333 | 0.584931 | 0.285041 | 0.254902 | 0.569750 | 0.040000 | 0.032027 | 0.331000 |
| 379 | 0.392879 | 0.326667 | 0.933754 | 0.710022 | 0.333333 | 0.682500 | 0.926667 | 0.020362 | 0.305105 |
| 380 | 0.393279 | 0.320000 | 0.128082 | 0.450191 | 0.896032 | 0.579875 | 0.311111 | 0.021076 | 0.215467 |
| 381 | 0.394279 | 0.775556 | 0.655022 | 0.268396 | 0.854575 | 0.894375 | 0.413333 | 0.019687 | 0.286878 |
| 382 | 0.395079 | 0.413333 | 0.716740 | 0.272586 | 0.919326 | 0.612500 | 0.728889 | 0.024333 | 0.483782 |
| 383 | 0.396679 | 0.315556 | 0.391632 | 0.368579 | 0.905556 | 0.643375 | 0.533333 | 0.022493 | 0.278656 |
| 384 | 0.397479 | 0.208889 | 0.495209 | 0.109426 | 0.952381 | 0.861375 | 0.597778 | 0.024554 | 0.293682 |
| 385 | 0.398680 | 0.124444 | 0.857939 | 0.118774 | 0.653409 | 0.652500 | 0.922222 | 0.027461 | 0.396248 |
| 386 | 0.398880 | 0.402222 | 0.847139 | 0.815354 | 0.689024 | 0.895625 | 0.711111 | 0.020000 | 0.271280 |
| 387 | 0.399080 | 0.488889 | 0.321719 | 0.758430 | 0.774476 | 0.286250 | 0.182222 | 0.025252 | 0.278725 |
| 388 | 0.399480 | 0.015556 | 0.316359 | 0.314549 | 0.278736 | 0.917750 | 0.673333 | 0.022494 | 0.151066 |
| 389 | 0.399680 | 0.951111 | 0.673882 | 0.853562 | 0.086048 | 0.909625 | 0.073333 | 0.018443 | 0.156295 |
| 390 | 0.400880 | 0.368889 | 0.013854 | 0.022837 | 0.450893 | 0.529750 | 0.451111 | 0.014809 | 0.163516 |
| 391 | 0.401480 | 0.171111 | 0.391013 | 0.687840 | 0.688272 | 0.304250 | 0.635556 | 0.024422 | 0.290195 |
| 392 | 0.403881 | 0.611111 | 0.414870 | 0.506992 | 0.613465 | 0.905000 | 0.728889 | 0.014489 | 0.190904 |
| 393 | 0.338268 | 0.582222 | 0.824613 | 0.885233 | 0.799645 | 0.756750 | 0.137778 | 0.034667 | 0.364872 |
| 394 | 0.338468 | 0.273333 | 0.098104 | 0.134029 | 0.008258 | 0.438250 | 0.197778 | 0.017454 | 0.188047 |
| 395 | 0.339068 | 0.042222 | 0.125991 | 0.580033 | 0.216667 | 0.896750 | 0.300000 | 0.017346 | 0.089872 |
| 396 | 0.340868 | 0.324444 | 0.547760 | 0.182405 | 0.220370 | 0.653125 | 0.553333 | 0.021043 | 0.332375 |
| 397 | 0.341268 | 0.491111 | 0.545505 | 0.776762 | 0.253205 | 0.401125 | 0.051111 | 0.035453 | 0.402348 |
| 398 | 0.341868 | 0.373333 | 0.273226 | 0.993980 | 0.989255 | 0.154625 | 0.920000 | 0.017490 | 0.309899 |
| 399 | 0.342068 | 0.024444 | 0.474365 | 0.690699 | 0.826023 | 0.394250 | 1.000000 | 0.028877 | 0.361578 |
| 400 | 0.342268 | 0.691111 | 0.692081 | 0.384102 | 0.784335 | 0.926625 | 0.020000 | 0.034633 | 0.356999 |
| 401 | 0.342669 | 0.011111 | 0.026376 | 0.897046 | 0.810284 | 0.273000 | 0.437778 | 0.033217 | 0.321287 |
| 402 | 0.343269 | 0.693333 | 0.367180 | 0.866703 | 0.542535 | 0.883000 | 0.477778 | 0.017039 | 0.240207 |


| 403 | 0.343869 | 0.595556 | 0.366054 | 0.429792 | 0.656818 | 0.535875 | 0.120000 | 0.026174 | 0.330801 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 404 | 0.346069 | 0.260000 | 0.374159 | 0.791021 | 0.908879 | 0.604750 | 0.057778 | 0.042043 | 0.365207 |
| 405 | 0.346269 | 0.855556 | 0.044810 | 0.322612 | 0.217372 | 0.744500 | 0.273333 | 0.008422 | 0.080714 |
| 406 | 0.346469 | 0.480000 | 0.350432 | 0.210429 | 0.562652 | 0.939625 | 0.337778 | 0.018181 | 0.221663 |
| 407 | 0.346869 | 0.022222 | 0.590683 | 0.421153 | 0.186782 | 0.411500 | 0.840000 | 0.031965 | 0.413464 |
| 408 | 0.348070 | 0.386667 | 0.282084 | 0.503533 | 0.505848 | 0.823125 | 0.842222 | 0.014254 | 0.196744 |
| 409 | 0.348870 | 0.444444 | 0.518809 | 0.624190 | 0.589080 | 0.622375 | 0.253333 | 0.024719 | 0.323115 |
| 410 | 0.349270 | 0.835556 | 0.808729 | 0.102933 | 0.598291 | 0.785125 | 0.428889 | 0.020368 | 0.392470 |
| 411 | 0.349670 | 0.782222 | 0.008797 | 0.044961 | 0.185897 | 0.898125 | 0.520000 | 0.002824 | 0.012790 |
| 412 | 0.351270 | 0.526667 | 0.405366 | 0.749456 | 0.595029 | 0.714375 | 0.033333 | 0.032133 | 0.299542 |
| 413 | 0.351470 | 0.591111 | 0.523954 | 0.919950 | 0.234167 | 0.744000 | 0.255556 | 0.018974 | 0.243812 |
| 414 | 0.356271 | 0.691111 | 0.022984 | 0.189981 | 0.131609 | 0.846125 | 0.111111 | 0.004704 | 0.016801 |
| 415 | 0.357271 | 0.544444 | 0.741050 | 0.759661 | 0.511281 | 0.652125 | 0.002222 | 0.042830 | 0.394317 |
| 416 | 0.358072 | 0.831111 | 0.278459 | 0.942490 | 0.916667 | 0.864750 | 0.040000 | 0.023512 | 0.236062 |
| 417 | 0.358872 | 0.664444 | 0.354975 | 0.589509 | 0.824266 | 0.774125 | 0.297778 | 0.019054 | 0.252430 |
| 418 | 0.360072 | 0.215556 | 0.456390 | 0.190628 | 0.028589 | 0.385625 | 0.982222 | 0.019719 | 0.357341 |
| 419 | 0.360272 | 0.362222 | 0.481376 | 0.867107 | 0.550000 | 0.619750 | 0.175556 | 0.027862 | 0.307262 |
| 420 | 0.362472 | 0.142222 | 0.784601 | 0.731389 | 0.162602 | 0.794750 | 0.977778 | 0.020511 | 0.270726 |
| 421 | 0.367073 | 0.600000 | 0.396892 | 0.184235 | 0.737624 | 0.639750 | 0.044444 | 0.032528 | 0.354710 |
| 422 | 0.317263 | 0.428889 | 0.227673 | 0.849902 | 0.974099 | 0.550750 | 0.415556 | 0.019484 | 0.230063 |
| 423 | 0.317463 | 0.586667 | 0.922422 | 0.642706 | 0.677165 | 0.726125 | 0.211111 | 0.026706 | 0.328567 |
| 424 | 0.318064 | 0.757778 | 0.801749 | 0.368577 | 0.279902 | 0.575875 | 0.457778 | 0.020795 | 0.334892 |
| 425 | 0.318264 | 0.713333 | 0.447803 | 0.560172 | 0.532407 | 0.798750 | 0.166667 | 0.021254 | 0.231188 |
| 426 | 0.319064 | 0.086667 | 0.335507 | 0.859153 | 0.641667 | 0.533500 | 0.484444 | 0.028777 | 0.271360 |
| 427 | 0.319864 | 0.537778 | 0.545111 | 0.834670 | 0.083333 | 0.442875 | 0.144444 | 0.025571 | 0.295699 |
| 428 | 0.320064 | 0.377778 | 0.718167 | 0.854542 | 0.254444 | 0.133250 | 0.031111 | 0.046062 | 0.414449 |
| 429 | 0.320264 | 0.462222 | 0.338496 | 0.932515 | 0.333333 | 0.482875 | 0.215556 | 0.022197 | 0.247809 |
| 430 | 0.320664 | 0.248889 | 0.138403 | 0.384814 | 0.144097 | 0.201250 | 0.326667 | 0.021490 | 0.243847 |
| 431 | 0.429429 | 0.311111 | 0.228811 | 0.196763 | 0.386612 | 0.513500 | 0.782222 | 0.017870 | 0.191815 |


| 432 | 0.430430 | 0.146667 | 0.613613 | 0.282667 | 0.678839 | 0.590625 | 0.953333 | 0.027053 | 0.278639 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 433 | 0.431431 | 0.446667 | 0.245754 | 0.914159 | 0.495169 | 0.660625 | 0.951111 | 0.013291 | 0.163108 |
| 434 | 0.432432 | 0.031111 | 0.139463 | 0.710706 | 0.642857 | 0.983000 | 0.404444 | 0.030275 | 0.181101 |
| 435 | 0.433433 | 0.602222 | 0.522198 | 0.384180 | 0.779557 | 0.742750 | 0.062222 | 0.039465 | 0.337318 |
| 436 | 0.434434 | 0.564444 | 0.787104 | 0.511780 | 0.334354 | 0.640625 | 0.680000 | 0.019184 | 0.232302 |
| 437 | 0.435435 | 0.848889 | 0.246492 | 0.289578 | 0.056187 | 0.561500 | 0.524444 | 0.011240 | 0.137050 |
| 438 | 0.436436 | 0.266667 | 0.164930 | 0.335920 | 0.711765 | 0.261250 | 0.060000 | 0.052920 | 0.400013 |
| 439 | 0.437437 | 0.551111 | 0.771949 | 0.411348 | 0.303962 | 0.615625 | 0.046667 | 0.044448 | 0.411588 |
| 440 | 0.439439 | 0.337778 | 0.167022 | 0.948037 | 0.530142 | 0.132125 | 0.488889 | 0.021301 | 0.236426 |
| 441 | 0.441441 | 0.044444 | 0.220075 | 0.306938 | 0.206250 | 0.232250 | 0.131111 | 0.065396 | 0.510650 |
| 442 | 0.443443 | 0.217778 | 0.655832 | 0.864684 | 0.364780 | 0.692750 | 0.586667 | 0.026863 | 0.256408 |
| 443 | 0.445445 | 0.755556 | 0.718131 | 0.931147 | 0.832185 | 0.485500 | 0.326667 | 0.023976 | 0.264141 |
| 444 | 0.446446 | 0.902222 | 0.527045 | 0.860273 | 0.985192 | 0.754750 | 0.100000 | 0.028732 | 0.260775 |
| 445 | 0.447447 | 0.344444 | 0.647897 | 0.698046 | 0.338923 | 0.221250 | 0.184444 | 0.040431 | 0.391383 |
| 446 | 0.448448 | 0.504444 | 0.182298 | 0.903089 | 0.125661 | 0.444500 | 0.926667 | 0.011379 | 0.153957 |
| 447 | 0.449449 | 0.982222 | 0.594702 | 0.557640 | 0.249563 | 0.568625 | 0.853333 | 0.013145 | 0.192371 |
| 448 | 0.450450 | 0.522222 | 0.329960 | 0.275880 | 0.696219 | 0.275250 | 0.633333 | 0.019334 | 0.234596 |
| 449 | 0.453453 | 0.688889 | 0.732463 | 0.988976 | 0.151593 | 0.600625 | 0.706667 | 0.016194 | 0.210536 |
| 450 | 0.454454 | 0.928889 | 0.668335 | 0.181362 | 0.330502 | 0.503500 | 0.924444 | 0.014370 | 0.225676 |
| 451 | 0.455455 | 0.542222 | 0.045956 | 0.914920 | 0.356061 | 0.833875 | 0.226667 | 0.011327 | 0.082010 |
| 452 | 0.456456 | 0.391111 | 0.762745 | 0.641556 | 0.998391 | 0.147125 | 0.635556 | 0.025740 | 0.291911 |
| 453 | 0.457457 | 0.784444 | 0.087500 | 0.926658 | 0.676941 | 0.544500 | 0.673333 | 0.011273 | 0.152672 |
| 454 | 0.458458 | 0.075556 | 0.802811 | 0.425385 | 0.150000 | 0.819875 | 0.464444 | 0.037048 | 0.305325 |
| 455 | 0.459459 | 0.175556 | 0.376549 | 0.732433 | 0.839041 | 0.758750 | 0.455556 | 0.030628 | 0.257677 |
| 456 | 0.460460 | 0.240000 | 0.926242 | 0.992353 | 0.061198 | 0.724750 | 0.506667 | 0.027334 | 0.265704 |
| 457 | 0.461461 | 0.828889 | 0.095859 | 0.436815 | 0.981707 | 0.663625 | 0.235556 | 0.017657 | 0.177887 |
| 458 | 0.463463 | 0.942222 | 0.620189 | 0.060779 | 0.266849 | 0.586625 | 0.311111 | 0.020316 | 0.250904 |
| 459 | 0.465465 | 0.095556 | 0.272434 | 0.724877 | 0.492647 | 0.450500 | 0.093333 | 0.062480 | 0.464004 |
| 460 | 0.466466 | 0.511111 | 0.059741 | 0.184946 | 0.449468 | 0.391375 | 0.417778 | 0.016795 | 0.179442 |


| 461 | 0.467467 | 0.044444 | 0.436958 | 0.256082 | 0.591463 | 0.767750 | 0.313333 | 0.050239 | 0.367702 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 462 | 0.468468 | 0.571111 | 0.813377 | 0.832381 | 0.894543 | 0.412375 | 0.848889 | 0.018951 | 0.243595 |
| 463 | 0.469469 | 0.153333 | 0.959625 | 0.630520 | 0.524401 | 0.299250 | 0.640000 | 0.035700 | 0.367572 |
| 464 | 0.470470 | 0.131111 | 0.594118 | 0.862297 | 0.174242 | 0.950000 | 0.097778 | 0.042946 | 0.256965 |
| 465 | 0.471471 | 0.097778 | 0.685263 | 0.758448 | 0.482323 | 0.400375 | 0.344444 | 0.048188 | 0.423293 |
| 466 | 0.421284 | 0.162222 | 0.538655 | 0.163873 | 0.787778 | 0.829375 | 0.960000 | 0.024290 | 0.239164 |
| 467 | 0.421484 | 0.808889 | 0.707877 | 0.236545 | 0.413435 | 0.528875 | 0.071111 | 0.035192 | 0.348457 |
| 468 | 0.421684 | 0.015556 | 1.005934 | 0.172830 | 0.476908 | 0.327875 | 0.035556 | 0.134847 | 0.941836 |
| 469 | 0.422084 | 0.711111 | 0.875246 | 0.580176 | 0.154130 | 0.219625 | 0.382222 | 0.024564 | 0.314966 |
| 470 | 0.422284 | 0.697778 | 0.382919 | 0.705366 | 0.008185 | 0.655875 | 0.491111 | 0.012376 | 0.140434 |
| 471 | 0.422484 | 0.371111 | 0.187454 | 0.062890 | 0.668465 | 0.488500 | 0.428889 | 0.022976 | 0.220062 |
| 472 | 0.422885 | 0.331111 | 0.974911 | 0.571419 | 0.519892 | 0.836375 | 0.097778 | 0.046196 | 0.383418 |
| 473 | 0.423085 | 0.277778 | 0.660899 | 0.694061 | 0.334314 | 0.436500 | 0.873333 | 0.022837 | 0.266785 |
| 474 | 0.423485 | 0.500000 | 0.260494 | 0.062228 | 0.322917 | 0.531125 | 0.804444 | 0.014898 | 0.185341 |
| 475 | 0.423685 | 0.571111 | 0.825311 | 0.340644 | 0.387705 | 0.482250 | 0.040000 | 0.046596 | 0.435081 |
| 476 | 0.423885 | 0.708889 | 0.183913 | 0.188914 | 0.343567 | 0.851750 | 0.900000 | 0.008802 | 0.093120 |
| 477 | 0.424085 | 0.255556 | 0.842052 | 0.654463 | 0.380288 | 0.379125 | 0.164444 | 0.048150 | 0.445506 |
| 478 | 0.424285 | 0.542222 | 0.244915 | 0.875288 | 0.753425 | 0.809750 | 0.342222 | 0.018899 | 0.180327 |
| 479 | 0.425085 | 0.304444 | 0.714636 | 0.778882 | 0.283991 | 0.593125 | 0.526667 | 0.026267 | 0.269887 |
| 480 | 0.425285 | 0.304444 | 0.725529 | 0.169928 | 0.603509 | 0.432250 | 0.720000 | 0.025812 | 0.303853 |
| 481 | 0.425685 | 0.455556 | 0.892231 | 0.051574 | 0.232008 | 0.324500 | 0.826667 | 0.022033 | 0.326230 |
| 482 | 0.475295 | 0.253333 | 0.848282 | 0.764270 | 0.723829 | 0.815625 | 0.324444 | 0.029480 | 0.308745 |
| 483 | 0.475695 | 0.093333 | 0.022624 | 0.766495 | 0.555556 | 0.911375 | 0.362222 | 0.016262 | 0.098043 |
| 484 | 0.475895 | 0.028889 | 0.430207 | 0.953788 | 0.297619 | 0.863625 | 0.557778 | 0.026888 | 0.199672 |
| 485 | 0.480896 | 0.328889 | 0.234982 | 0.134567 | 0.216967 | 0.632500 | 0.088889 | 0.027098 | 0.231976 |
| 486 | 0.481096 | 0.113333 | 0.249038 | 0.131400 | 0.159314 | 0.486125 | 0.595556 | 0.022575 | 0.250603 |
| 487 | 0.481296 | 0.288889 | 0.309174 | 0.479534 | 0.956522 | 0.850625 | 0.282222 | 0.024519 | 0.237246 |
| 488 | 0.482897 | 0.091111 | 0.545256 | 0.491484 | 0.241935 | 0.668500 | 0.235556 | 0.039075 | 0.311332 |
| 489 | 0.483097 | 0.522222 | 0.457860 | 0.972317 | 0.831579 | 0.626500 | 0.348889 | 0.021004 | 0.260103 |


| 490 | 0.485897 | 0.108889 | 0.025955 | 0.901253 | 0.305556 | 0.881750 | 0.406667 | 0.010909 | 0.055985 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 491 | 0.488098 | 0.060000 | 0.506677 | 0.445604 | 0.214103 | 0.467250 | 0.946667 | 0.025978 | 0.310174 |
| 492 | 0.488698 | 0.000000 | 0.788543 | 0.492165 | 0.231481 | 0.811000 | 0.742222 | 0.034748 | 0.289664 |
| 493 | 0.491898 | 0.157778 | 0.295223 | 0.677521 | 0.992424 | 0.342500 | 0.788889 | 0.022492 | 0.266432 |
| 494 | 0.492298 | 0.831111 | 0.448159 | 0.404567 | 0.057270 | 0.831125 | 0.813333 | 0.009189 | 0.113104 |
| 495 | 0.501900 | 0.071111 | 0.955558 | 0.885925 | 0.174731 | 0.889000 | 0.773333 | 0.026001 | 0.241708 |
| 496 | 0.504501 | 0.208889 | 0.049399 | 0.637862 | 0.947031 | 0.103375 | 0.344444 | 0.023552 | 0.239803 |
| 497 | 0.506501 | 0.620000 | 0.976649 | 0.397952 | 0.297799 | 0.774000 | 0.993333 | 0.017163 | 0.290311 |
| 498 | 0.506901 | 0.315556 | 0.068471 | 0.360580 | 0.317511 | 0.886750 | 0.362222 | 0.010440 | 0.069065 |
| 499 | 0.508702 | 0.257778 | 0.593945 | 0.188487 | 0.766150 | 0.220500 | 0.397778 | 0.030358 | 0.396035 |
| 500 | 0.509302 | 0.540000 | 0.184106 | 0.412249 | 0.595960 | 0.942750 | 0.233333 | 0.015862 | 0.141285 |
| 501 | 0.509502 | 0.062222 | 0.269465 | 0.678671 | 0.192308 | 0.292000 | 0.373333 | 0.031967 | 0.300217 |
| 502 | 0.510302 | 0.388889 | 0.594030 | 0.475019 | 0.645833 | 0.802375 | 0.746667 | 0.019277 | 0.268201 |
| 503 | 0.510502 | 0.080000 | 0.882179 | 0.802022 | 0.052469 | 0.546875 | 0.411111 | 0.037103 | 0.361143 |
| 504 | 0.510902 | 0.835556 | 0.112295 | 0.006962 | 0.609776 | 0.670750 | 0.748889 | 0.011429 | 0.188290 |
| 505 | 0.512302 | 0.077778 | 0.344359 | 0.498384 | 0.952899 | 0.418250 | 0.557778 | 0.030862 | 0.308702 |
| 506 | 0.512703 | 0.753333 | 0.892863 | 0.749365 | 0.741784 | 0.891625 | 0.573333 | 0.018318 | 0.274624 |
| 507 | 0.514303 | 0.842222 | 0.520222 | 0.881694 | 0.579032 | 0.589125 | 0.811111 | 0.017208 | 0.394747 |
| 508 | 0.514703 | 0.086667 | 0.493214 | 0.638060 | 0.567610 | 0.811750 | 0.408889 | 0.031312 | 0.266611 |
| 509 | 0.514903 | 0.106667 | 0.670511 | 0.530171 | 0.268750 | 0.567375 | 0.617778 | 0.028815 | 0.324864 |
| 510 | 0.516103 | 0.200000 | 0.214313 | 0.730606 | 0.914141 | 0.888125 | 0.191111 | 0.027319 | 0.209394 |
| 511 | 0.517103 | 0.231111 | 0.432977 | 0.769383 | 0.797814 | 0.472500 | 0.431111 | 0.025425 | 0.284241 |
| 512 | 0.517303 | 0.428889 | 0.354567 | 0.125067 | 0.983568 | 0.694750 | 0.324444 | 0.023919 | 0.290629 |
| 513 | 0.519304 | 0.126667 | 0.468305 | 0.902245 | 0.610215 | 0.217000 | 0.662222 | 0.026168 | 0.303237 |
| 514 | 0.521304 | 0.393333 | 0.743451 | 0.135154 | 0.672555 | 0.169000 | 0.804444 | 0.027706 | 0.639616 |
| 515 | 0.535536 | 0.151111 | 0.145320 | 0.058220 | 0.080846 | 0.557500 | 0.426667 | 0.020720 | 0.150673 |
| 516 | 0.536537 | 0.048889 | 0.508547 | 0.888920 | 0.602713 | 0.818875 | 0.948889 | 0.029881 | 0.258363 |
| 517 | 0.537538 | 0.080000 | 0.298596 | 0.763674 | 0.688306 | 0.105125 | 0.182222 | 0.059741 | 0.485291 |
| 518 | 0.538539 | 0.613333 | 0.135006 | 0.173006 | 0.278287 | 0.430375 | 0.744444 | 0.012542 | 0.166300 |


| 519 | 0.539540 | 0.960000 | 0.245361 | 0.947692 | 0.471774 | 0.749750 | 0.813333 | 0.010116 | 0.130207 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 520 | 0.541542 | 0.268889 | 0.191382 | 0.413826 | 0.383333 | 0.662625 | 0.300000 | 0.024852 | 0.198782 |
| 521 | 0.542543 | 0.322222 | 0.807384 | 0.152069 | 0.920509 | 0.217250 | 0.228889 | 0.041781 | 0.421213 |
| 522 | 0.544545 | 0.544444 | 0.051814 | 0.558168 | 0.595139 | 0.300250 | 0.597778 | 0.014214 | 0.184414 |
| 523 | 0.545546 | 0.626667 | 0.702610 | 0.446368 | 0.896636 | 0.296250 | 0.862222 | 0.018305 | 0.252864 |
| 524 | 0.546547 | 0.875556 | 0.405862 | 0.031366 | 0.205840 | 0.459500 | 0.686667 | 0.014425 | 0.214090 |
| 525 | 0.547548 | 0.211111 | 0.440888 | 0.977591 | 0.619565 | 0.475500 | 0.555556 | 0.028184 | 0.276045 |
| 526 | 0.548549 | 0.524444 | 0.977332 | 0.694569 | 0.039593 | 0.855875 | 0.660000 | 0.016828 | 0.171477 |
| 527 | 0.550551 | 0.595556 | 0.697191 | 0.139815 | 0.514977 | 0.801750 | 0.908889 | 0.016071 | 0.207800 |
| 528 | 0.552553 | 0.677778 | 0.003146 | 0.525263 | 0.095411 | 0.888875 | 0.395556 | 0.000097 | 0.000012 |
| 529 | 0.553554 | 0.904444 | 0.150672 | 0.502182 | 0.981481 | 0.523500 | 0.902222 | 0.011604 | 0.165350 |
| 530 | 0.554555 | 0.428889 | 0.490980 | 0.961922 | 0.456758 | 0.285250 | 0.337778 | 0.027685 | 0.289710 |
| 531 | 0.555556 | 0.166667 | 0.590448 | 0.509221 | 0.278509 | 0.928875 | 0.946667 | 0.019612 | 0.169627 |
| 532 | 0.556557 | 0.166667 | 0.377893 | 0.018182 | 0.666667 | 0.637625 | 0.666667 | 0.027546 | 0.258835 |
| 533 | 0.557558 | 0.273333 | 0.850364 | 0.248293 | 0.028736 | 0.125125 | 0.726667 | 0.027665 | 0.373188 |
| 534 | 0.558559 | 0.700000 | 0.146722 | 0.214765 | 0.694779 | 0.830875 | 0.840000 | 0.010788 | 0.125399 |
| 535 | 0.560561 | 0.726667 | 0.533643 | 0.436539 | 0.046099 | 0.565625 | 0.357778 | 0.018732 | 0.219326 |
| 536 | 0.561562 | 0.602222 | 0.980034 | 0.417172 | 0.411847 | 0.375375 | 0.488889 | 0.024504 | 0.308812 |
| 537 | 0.562563 | 0.986667 | 0.706203 | 0.093412 | 0.777006 | 0.524500 | 0.037778 | 0.037363 | 0.374646 |
| 538 | 0.565566 | 0.955556 | 0.936847 | 0.333583 | 0.231085 | 0.541500 | 0.617778 | 0.017226 | 0.252295 |
| 539 | 0.567568 | 0.080000 | 0.435035 | 0.796548 | 0.154255 | 0.967000 | 0.677778 | 0.019193 | 0.119980 |
| 540 | 0.570571 | 0.788889 | 0.432273 | 0.074956 | 0.551515 | 0.962000 | 0.806667 | 0.012444 | 0.144017 |
| 541 | 0.572573 | 0.495556 | 0.465320 | 0.393255 | 0.826667 | 0.399375 | 0.077778 | 0.043263 | 0.370968 |
| 542 | 0.573574 | 0.860000 | 0.428222 | 0.922645 | 0.256790 | 0.686625 | 0.217778 | 0.019218 | 0.198759 |
| 543 | 0.577578 | 0.171111 | 0.084460 | 0.029589 | 0.239247 | 0.180125 | 0.384444 | 0.028049 | 0.244652 |
| 544 | 0.526705 | 0.322222 | 0.945282 | 0.887545 | 0.048246 | 0.542125 | 0.280000 | 0.034879 | 0.357031 |
| 545 | 0.527105 | 0.680000 | 0.273996 | 0.047254 | 0.931895 | 0.684375 | 0.211111 | 0.024448 | 0.227634 |
| 546 | 0.527906 | 0.380000 | 0.043320 | 0.956300 | 0.044985 | 0.564750 | 0.013333 | 0.019538 | 0.122887 |
| 547 | 0.528706 | 0.288889 | 0.340703 | 0.290247 | 0.773603 | 0.127000 | 0.057778 | 0.057782 | 0.450219 |


| 548 | 0.528906 | 0.184444 | 0.390843 | 0.891730 | 0.100000 | 0.569375 | 0.520000 | 0.024138 | 0.232381 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 549 | 0.529106 | 0.726667 | 0.398943 | 0.275566 | 0.508357 | 0.143250 | 0.882222 | 0.015419 | 0.232388 |
| 550 | 0.529706 | 0.962222 | 0.921615 | 0.358727 | 0.330940 | 0.755500 | 0.951111 | 0.013686 | 0.196953 |
| 551 | 0.530106 | 0.155556 | 0.974657 | 0.193143 | 0.817061 | 0.335125 | 0.635556 | 0.036413 | 0.395481 |
| 552 | 0.530306 | 0.942222 | 0.293480 | 0.733703 | 0.459834 | 0.099250 | 0.133333 | 0.024682 | 0.269021 |
| 553 | 0.530506 | 0.257778 | 0.863755 | 0.384103 | 0.943182 | 0.711375 | 0.140000 | 0.049401 | 0.424006 |
| 554 | 0.530706 | 0.995556 | 0.371536 | 0.199020 | 0.979021 | 0.731750 | 0.280000 | 0.019551 | 0.211661 |
| 555 | 0.530906 | 0.448889 | 0.846864 | 0.007198 | 0.891638 | 0.394625 | 0.166667 | 0.040328 | 0.435699 |
| 556 | 0.531106 | 0.848889 | 0.451899 | 0.087176 | 0.183099 | 0.636125 | 0.104444 | 0.025419 | 0.241525 |
| 557 | 0.531506 | 0.780000 | 0.159333 | 0.799854 | 0.631966 | 0.337875 | 0.613333 | 0.013735 | 0.189215 |
| 558 | 0.531706 | 1.000000 | 0.728946 | 0.438088 | 0.877589 | 0.765125 | 0.662222 | 0.015662 | 0.210526 |
| 559 | 0.531906 | 0.842222 | 0.016385 | 0.023391 | 0.229003 | 0.255250 | 0.188889 | 0.016216 | 0.172753 |
| 560 | 0.532106 | 0.406667 | 0.164115 | 0.876153 | 0.750000 | 0.953625 | 0.342222 | 0.017663 | 0.145129 |
| 561 | 0.532907 | 0.397778 | 0.771394 | 0.194884 | 0.869919 | 0.792000 | 0.620000 | 0.023897 | 0.265598 |
| 562 | 0.533107 | 0.262222 | 0.144769 | 0.396578 | 0.479630 | 0.605500 | 0.724444 | 0.017495 | 0.174510 |
| 563 | 0.591592 | 1.000000 | 0.933192 | 0.091108 | 0.465818 | 0.890875 | 0.837778 | 0.015429 | 0.304341 |
| 564 | 0.593594 | 0.108889 | 0.461922 | 0.209418 | 0.092424 | 0.972000 | 0.462222 | 0.017617 | 0.102084 |
| 565 | 0.597598 | 0.184444 | 0.920718 | 0.883588 | 0.290498 | 0.729750 | 0.835556 | 0.023236 | 0.296692 |
| 566 | 0.600601 | 0.117778 | 0.359376 | 0.403485 | 0.862745 | 0.569625 | 0.560000 | 0.027804 | 0.296105 |
| 567 | 0.603604 | 0.744444 | 0.827468 | 0.529065 | 0.553591 | 0.722750 | 0.404444 | 0.020762 | 0.315982 |
| 568 | 0.607608 | 0.642222 | 0.408193 | 0.900960 | 0.198517 | 0.810750 | 0.517778 | 0.012867 | 0.153603 |
| 569 | 0.611612 | 0.208889 | 0.257580 | 0.694086 | 0.717687 | 0.483500 | 0.937778 | 0.018771 | 0.243410 |
| 570 | 0.620621 | 0.195556 | 0.770458 | 0.053247 | 0.627451 | 0.736750 | 0.482222 | 0.029783 | 0.388980 |
| 571 | 0.622623 | 0.404444 | 0.303669 | 0.278694 | 0.358696 | 0.939000 | 0.442222 | 0.015047 | 0.144239 |
| 572 | 0.579116 | 0.475556 | 0.683488 | 0.264164 | 0.002427 | 0.613500 | 0.600000 | 0.018468 | 0.282333 |
| 573 | 0.579316 | 0.157778 | 0.653061 | 0.546218 | 0.115462 | 0.743125 | 0.826667 | 0.020940 | 0.243274 |
| 574 | 0.579516 | 0.068889 | 0.475815 | 0.196840 | 0.596405 | 0.702375 | 0.997778 | 0.025492 | 0.298370 |
| 575 | 0.579716 | 0.717778 | 0.991710 | 0.555368 | 0.644309 | 0.889625 | 0.473333 | 0.020334 | 0.305817 |
| 576 | 0.582917 | 0.117778 | 0.902163 | 0.686643 | 0.831633 | 0.544250 | 0.715556 | 0.030151 | 0.368711 |


| 577 | 0.591718 | 0.588889 | 0.341988 | 0.589365 | 0.332064 | 0.604875 | 0.264444 | 0.019446 | 0.231472 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 578 | 0.592118 | 0.291111 | 0.182214 | 0.720987 | 0.568519 | 0.721125 | 0.160000 | 0.024529 | 0.207027 |
| 579 | 0.593519 | 0.360000 | 0.115320 | 0.826269 | 0.555085 | 0.547750 | 0.235556 | 0.020124 | 0.199185 |
| 580 | 0.593919 | 0.566667 | 0.371948 | 0.118979 | 0.435417 | 0.917000 | 0.684444 | 0.014084 | 0.182763 |
| 581 | 0.594719 | 0.035556 | 0.029326 | 0.184561 | 0.355263 | 0.270875 | 0.342222 | 0.030445 | 0.267514 |
| 582 | 0.594919 | 0.462222 | 0.861449 | 0.223221 | 0.383880 | 0.940625 | 0.562222 | 0.021055 | 0.319878 |
| 583 | 0.596319 | 0.397778 | 0.798645 | 0.875828 | 0.231151 | 0.769375 | 0.228889 | 0.025918 | 0.270032 |
| 584 | 0.596919 | 0.377778 | 0.298510 | 0.914311 | 0.656728 | 0.938625 | 0.093333 | 0.025821 | 0.192147 |
| 585 | 0.597119 | 0.808889 | 0.272848 | 0.609649 | 0.177171 | 0.640875 | 0.026667 | 0.021954 | 0.198059 |
| 586 | 0.597319 | 0.353333 | 0.849928 | 0.926368 | 0.447065 | 0.764000 | 0.257778 | 0.027615 | 0.294195 |
| 587 | 0.597720 | 0.686667 | 0.788268 | 0.260248 | 0.350949 | 0.910625 | 0.366667 | 0.019653 | 0.261487 |
| 588 | 0.597920 | 0.586667 | 0.729947 | 0.639192 | 0.088095 | 0.883625 | 0.193333 | 0.019892 | 0.187883 |
| 589 | 0.599720 | 0.217778 | 0.147442 | 0.888536 | 0.380208 | 0.971375 | 0.060000 | 0.023040 | 0.115159 |
| 590 | 0.600920 | 0.024444 | 0.619210 | 0.876077 | 0.461934 | 0.277000 | 0.046667 | 0.094438 | 0.592048 |
| 591 | 0.602120 | 0.148889 | 0.354082 | 0.041170 | 0.834699 | 0.902125 | 0.668889 | 0.023406 | 0.256953 |
| 592 | 0.602320 | 0.077778 | 0.617047 | 0.021409 | 0.184426 | 0.662125 | 0.973333 | 0.025030 | 0.324644 |
| 593 | 0.602521 | 0.648889 | 0.868413 | 0.908770 | 0.820281 | 0.862375 | 0.188889 | 0.025247 | 0.291464 |
| 594 | 0.603521 | 0.320000 | 0.181960 | 0.324475 | 0.120031 | 0.591875 | 0.104444 | 0.022939 | 0.186721 |
| 595 | 0.604521 | 0.577778 | 0.224555 | 0.792676 | 0.781532 | 0.864000 | 0.240000 | 0.017768 | 0.181017 |
| 596 | 0.605521 | 0.524444 | 0.999247 | 0.501387 | 0.896288 | 0.809125 | 0.015556 | 0.045222 | 0.395005 |
| 597 | 0.606321 | 0.373333 | 0.529390 | 0.369442 | 0.841146 | 0.901750 | 0.988889 | 0.017654 | 0.257000 |
| 598 | 0.606721 | 0.180000 | 0.014728 | 0.354438 | 0.606250 | 0.417125 | 0.471111 | 0.019797 | 0.199920 |
| 599 | 0.610122 | 0.713333 | 0.569653 | 0.701008 | 0.373624 | 0.450125 | 0.346667 | 0.021189 | 0.312188 |
| 600 | 0.612122 | 0.122222 | 0.249532 | 0.820340 | 0.378415 | 0.605125 | 0.326667 | 0.026825 | 0.238091 |
| 601 | 0.612322 | 0.242222 | 0.394393 | 0.125871 | 0.041971 | 0.403250 | 0.684444 | 0.020578 | 0.311225 |
| 602 | 0.612723 | 0.244444 | 0.462733 | 0.250569 | 0.725962 | 0.655500 | 0.437778 | 0.026269 | 0.307114 |
| 603 | 0.615123 | 0.111111 | 0.066561 | 0.332107 | 0.945652 | 0.686500 | 0.015556 | 0.058461 | 0.351527 |
| 604 | 0.643644 | 0.313333 | 0.989896 | 0.338275 | 0.585632 | 0.922875 | 0.622222 | 0.025350 | 0.261738 |
| 605 | 0.644645 | 0.611111 | 0.315225 | 0.899213 | 0.532986 | 0.644625 | 0.662222 | 0.014981 | 0.182144 |


| 606 | 0.645646 | 0.106667 | 0.614452 | 0.088283 | 0.391975 | 0.529500 | 0.171111 | 0.056709 | 0.484882 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 607 | 0.646647 | 0.064444 | 0.217502 | 1.005061 | 0.127104 | 0.203250 | 0.262222 | 0.042852 | 0.406570 |
| 608 | 0.647648 | 0.295556 | 0.502541 | 0.225441 | 0.861337 | 0.089125 | 0.860000 | 0.023803 | 0.283695 |
| 609 | 0.648649 | 0.355556 | 0.709977 | 0.020583 | 0.959163 | 0.343375 | 0.951111 | 0.022810 | 0.307828 |
| 610 | 0.650651 | 0.935556 | 0.281522 | 0.729704 | 0.279540 | 0.308250 | 0.102222 | 0.024851 | 0.261163 |
| 611 | 0.651652 | 0.284444 | 0.026201 | 0.238864 | 0.498792 | 0.948000 | 0.144444 | 0.018383 | 0.090704 |
| 612 | 0.652653 | 0.991111 | 0.462303 | 0.260123 | 0.463941 | 0.366375 | 0.797778 | 0.014324 | 0.220182 |
| 613 | 0.653654 | 0.093333 | 0.658709 | 0.184792 | 0.768315 | 0.425375 | 0.464444 | 0.044351 | 0.412129 |
| 614 | 0.654655 | 0.593333 | 0.923633 | 0.862550 | 0.575581 | 0.266250 | 0.642222 | 0.021969 | 0.277182 |
| 615 | 0.657658 | 0.317778 | 0.921337 | 0.455285 | 0.936620 | 0.908875 | 0.900000 | 0.022356 | 0.247173 |
| 616 | 0.658659 | 0.724444 | 0.203166 | 0.465202 | 0.610990 | 0.533500 | 0.786667 | 0.012984 | 0.175896 |
| 617 | 0.660661 | 0.957778 | 0.308304 | 0.078250 | 0.827206 | 0.690750 | 0.122222 | 0.024781 | 0.232027 |
| 618 | 0.662663 | 0.408889 | 0.908816 | 0.547093 | 0.057832 | 0.765750 | 0.035556 | 0.045488 | 0.370064 |
| 619 | 0.663664 | 0.282222 | 0.777519 | 0.412029 | 0.003501 | 0.960000 | 0.813333 | 0.013814 | 0.101574 |
| 620 | 0.668669 | 0.317778 | 0.190912 | 0.075921 | 0.250000 | 0.774750 | 0.228889 | 0.021142 | 0.142592 |
| 621 | 0.670671 | 0.197778 | 0.347812 | 0.947284 | 0.212025 | 0.718750 | 0.917778 | 0.016977 | 0.171210 |
| 622 | 0.671672 | 0.933333 | 0.137702 | 0.181729 | 0.841975 | 0.210250 | 0.604444 | 0.014017 | 0.199980 |
| 623 | 0.672673 | 0.344444 | 0.673306 | 0.199936 | 0.048826 | 0.068125 | 0.204444 | 0.038267 | 0.437484 |
| 624 | 0.675676 | 0.217778 | 0.716009 | 0.753095 | 0.354938 | 0.499500 | 0.088889 | 0.056628 | 0.477557 |
| 625 | 0.676677 | 0.951111 | 0.750487 | 0.963668 | 0.433117 | 0.620625 | 0.382222 | 0.019352 | 0.231937 |
| 626 | 0.677678 | 0.448889 | 0.303278 | 0.842394 | 0.243659 | 0.462500 | 0.215556 | 0.026053 | 0.253151 |
| 627 | 0.678679 | 0.115556 | 0.855428 | 0.961797 | 0.665000 | 0.197250 | 0.533333 | 0.041528 | 0.401395 |
| 628 | 0.679680 | 0.480000 | 0.521756 | 0.019847 | 0.498555 | 0.263250 | 0.762222 | 0.020467 | 0.280842 |
| 629 | 0.680681 | 0.806667 | 0.977279 | 0.515747 | 0.986472 | 0.982000 | 0.337778 | 0.022925 | 0.244526 |
| 630 | 0.681682 | 0.015556 | 0.430307 | 0.317676 | 0.481061 | 0.481500 | 0.515556 | 0.049510 | 0.412835 |
| 631 | 0.682683 | 0.302222 | 0.254524 | 0.240473 | 0.032780 | 0.243250 | 0.251111 | 0.028355 | 0.287994 |
| 632 | 0.683684 | 0.397778 | 0.166295 | 0.361344 | 0.054207 | 0.889875 | 0.113333 | 0.009142 | 0.028737 |
| 633 | 0.631726 | 0.753333 | 0.088595 | 0.211307 | 0.259306 | 0.606500 | 0.655556 | 0.009729 | 0.116048 |
| 634 | 0.631926 | 0.571111 | 0.684767 | 0.772282 | 0.013248 | 0.986250 | 0.180000 | 0.015800 | 0.093748 |


| 635 | 0.632126 | 0.368889 | 0.440039 | 0.756619 | 0.717681 | 0.266000 | 0.451111 | 0.026706 | 0.281298 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 636 | 0.632326 | 0.757778 | 0.219809 | 0.737693 | 0.697415 | 0.844750 | 0.126667 | 0.021810 | 0.185560 |
| 637 | 0.632527 | 0.473333 | 0.322637 | 0.285769 | 0.948198 | 0.496500 | 0.248889 | 0.029089 | 0.268933 |
| 638 | 0.632727 | 0.160000 | 0.603142 | 0.916179 | 0.702941 | 0.685875 | 0.617778 | 0.030635 | 0.283206 |
| 639 | 0.632927 | 0.235556 | 0.841131 | 0.938894 | 0.529614 | 0.718375 | 0.640000 | 0.027913 | 0.279400 |
| 640 | 0.633127 | 0.168889 | 0.822870 | 0.400383 | 0.278907 | 0.106250 | 0.097778 | 0.063740 | 0.568438 |
| 641 | 0.633327 | 0.386667 | 0.432092 | 0.070867 | 0.871212 | 0.651125 | 0.864444 | 0.019518 | 0.231599 |
| 642 | 0.633527 | 0.226667 | 0.073311 | 0.470930 | 0.129781 | 0.992750 | 0.017778 | 0.005538 | 0.005923 |
| 643 | 0.633727 | 0.724444 | 0.900954 | 0.371511 | 0.510763 | 0.199000 | 0.735556 | 0.019325 | 0.281344 |
| 644 | 0.633927 | 0.953333 | 0.417759 | 0.986423 | 0.236010 | 0.790000 | 0.357778 | 0.013983 | 0.149389 |
| 645 | 0.634327 | 0.588889 | 0.175612 | 0.796806 | 0.221655 | 0.824125 | 0.388889 | 0.011041 | 0.091111 |
| 646 | 0.634527 | 0.075556 | 0.961316 | 0.209121 | 0.234375 | 0.894000 | 0.302222 | 0.045191 | 0.366809 |
| 647 | 0.634727 | 0.331111 | 0.606831 | 0.428494 | 0.724462 | 0.873125 | 0.997778 | 0.018755 | 0.206089 |
| 648 | 0.634927 | 0.340000 | 0.963347 | 0.239535 | 0.873620 | 0.953750 | 0.526667 | 0.027074 | 0.280626 |
| 649 | 0.635327 | 0.742222 | 0.899996 | 0.774063 | 0.995455 | 0.995000 | 0.588889 | 0.018460 | 0.207486 |
| 650 | 0.635527 | 0.742222 | 0.925019 | 0.985825 | 0.671725 | 0.717000 | 0.466667 | 0.021107 | 0.241858 |
| 651 | 0.635727 | 0.466667 | 0.731627 | 0.082892 | 0.618896 | 0.344125 | 0.037778 | 0.052556 | 0.502117 |
| 652 | 0.635927 | 0.324444 | 0.260645 | 0.992633 | 0.838685 | 0.661375 | 0.293333 | 0.027158 | 0.247133 |
| 653 | 0.636127 | 0.455556 | 0.449701 | 0.479083 | 0.141870 | 0.470750 | 0.488889 | 0.020898 | 0.238222 |
| 654 | 0.636327 | 0.251111 | 0.476378 | 0.756367 | 0.176011 | 0.086375 | 0.582222 | 0.026527 | 0.311113 |
| 655 | 0.636527 | 0.951111 | 0.026108 | 0.852912 | 0.317493 | 0.556500 | 0.537778 | 0.008682 | 0.108280 |
| 656 | 0.637327 | 0.208889 | 0.719649 | 0.865512 | 0.314565 | 0.648500 | 0.375556 | 0.034327 | 0.316642 |
| 657 | 0.637728 | 0.884444 | 0.131149 | 0.361980 | 0.931973 | 0.880750 | 0.386667 | 0.013437 | 0.142301 |
| 658 | 0.638128 | 0.740000 | 0.528734 | 0.876936 | 0.090256 | 0.194250 | 0.417778 | 0.019910 | 0.272937 |
| 659 | 0.638328 | 0.135556 | 0.424901 | 0.502623 | 0.552564 | 0.764750 | 0.075556 | 0.059859 | 0.410856 |
| 660 | 0.638528 | 0.573333 | 0.121521 | 0.015072 | 0.210276 | 0.464500 | 0.675556 | 0.012514 | 0.156585 |
| 661 | 0.638928 | 0.884444 | 0.166124 | 0.555166 | 0.544393 | 0.407625 | 0.017778 | 0.032290 | 0.283072 |
| 662 | 0.714743 | 0.468889 | 0.572033 | 0.760397 | 0.419919 | 0.547125 | 0.337778 | 0.026329 | 0.269863 |
| 663 | 0.716343 | 0.055556 | 0.732534 | 0.498772 | 0.079932 | 0.994375 | 0.653333 | 0.022311 | 0.145571 |


| 664 | 0.716943 | 0.222222 | 0.976095 | 0.507008 | 0.584596 | 0.657125 | 0.208889 | 0.038719 | 0.385519 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 665 | 0.718344 | 0.473333 | 0.470276 | 0.779188 | 0.264840 | 0.983250 | 0.853333 | 0.011627 | 0.127414 |
| 666 | 0.718944 | 0.195556 | 0.603712 | 0.875863 | 0.759091 | 0.550875 | 0.282222 | 0.032573 | 0.318959 |
| 667 | 0.720144 | 0.044444 | 0.834304 | 0.574177 | 0.923469 | 0.965000 | 0.155556 | 0.064192 | 0.408086 |
| 668 | 0.720744 | 0.711111 | 0.109924 | 0.853139 | 0.409184 | 0.425250 | 0.508889 | 0.013410 | 0.199944 |
| 669 | 0.720944 | 0.260000 | 0.606680 | 0.751651 | 0.007310 | 0.706375 | 0.626667 | 0.018137 | 0.215324 |
| 670 | 0.721744 | 0.153333 | 0.628810 | 0.834191 | 0.754016 | 0.707000 | 0.680000 | 0.024922 | 0.284061 |
| 671 | 0.721944 | 0.051111 | 0.095783 | 0.355352 | 0.443548 | 0.890375 | 0.431111 | 0.019509 | 0.122805 |
| 672 | 0.722344 | 0.786667 | 0.863315 | 0.687580 | 0.070847 | 0.775000 | 0.131111 | 0.023102 | 0.254588 |
| 673 | 0.722545 | 0.317778 | 0.198640 | 0.786178 | 0.852041 | 0.711500 | 0.144444 | 0.026993 | 0.234790 |
| 674 | 0.723145 | 0.337778 | 0.630126 | 0.348539 | 0.107323 | 0.799375 | 0.253333 | 0.023638 | 0.230081 |
| 675 | 0.724945 | 0.380000 | 0.256268 | 0.007978 | 0.333634 | 0.213875 | 0.177778 | 0.028371 | 0.350949 |
| 676 | 0.725145 | 0.348889 | 0.959413 | 0.398036 | 0.390504 | 0.712125 | 0.731111 | 0.022575 | 0.351294 |
| 677 | 0.725545 | 0.568889 | 0.370106 | 0.922823 | 0.138686 | 0.375125 | 0.564444 | 0.016460 | 0.256015 |
| 678 | 0.726745 | 0.008889 | 0.941931 | 0.013771 | 0.344203 | 0.678750 | 0.237778 | 0.062822 | 0.558555 |
| 679 | 0.727145 | 0.273333 | 0.210882 | 0.370931 | 0.824297 | 0.831375 | 0.595556 | 0.018181 | 0.201366 |
| 680 | 0.727546 | 0.402222 | 0.203181 | 0.208589 | 0.350295 | 0.780125 | 0.640000 | 0.013720 | 0.157587 |
| 681 | 0.729546 | 0.084444 | 0.466373 | 0.493257 | 0.278205 | 0.544750 | 0.560000 | 0.028812 | 0.295516 |
| 682 | 0.729946 | 0.875556 | 0.938026 | 0.016824 | 0.844810 | 0.877375 | 0.375556 | 0.022036 | 0.410242 |
| 683 | 0.730146 | 0.344444 | 0.300070 | 0.277115 | 0.471905 | 0.373500 | 0.062222 | 0.036894 | 0.341023 |
| 684 | 0.730546 | 0.340000 | 0.419545 | 0.636539 | 0.063725 | 0.756125 | 0.057778 | 0.028054 | 0.194136 |
| 685 | 0.731346 | 0.964444 | 0.368389 | 0.730375 | 0.069600 | 0.792500 | 0.368889 | 0.010612 | 0.118986 |
| 686 | 0.706707 | 0.191111 | 0.712484 | 0.465990 | 0.498428 | 0.633625 | 0.275556 | 0.034460 | 0.350951 |
| 687 | 0.723724 | 0.333333 | 0.529530 | 0.296296 | 0.131356 | 0.887875 | 0.015556 | 0.035630 | 0.216273 |
| 688 | 0.725726 | 0.426667 | 0.572792 | 0.317176 | 0.941419 | 0.500500 | 0.722222 | 0.021244 | 0.343209 |
| 689 | 0.731732 | 0.988889 | 0.114227 | 0.687373 | 0.165978 | 0.672625 | 0.351111 | 0.009209 | 0.109205 |
| 690 | 0.732733 | 0.502222 | 0.727516 | 0.381011 | 0.151802 | 0.872875 | 0.411111 | 0.018588 | 0.215578 |
| 691 | 0.684737 | 0.417778 | 0.834660 | 0.978645 | 0.569444 | 0.898375 | 0.282222 | 0.024598 | 0.267115 |
| 692 | 0.685937 | 0.602222 | 0.146187 | 0.155590 | 0.781346 | 0.427500 | 0.384444 | 0.018452 | 0.258099 |


| 693 | 0.688938 | 0.746667 | 0.956867 | 0.995861 | 0.149729 | 0.564375 | 0.226667 | 0.023979 | 0.330649 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 694 | 0.689338 | 0.328889 | 0.098155 | 0.717370 | 0.356410 | 0.423500 | 0.440000 | 0.017040 | 0.194390 |
| 695 | 0.710711 | 0.742222 | 0.697732 | 0.656666 | 0.198029 | 0.703750 | 0.060000 | 0.037722 | 0.517060 |
| 696 | 0.713714 | 0.191111 | 0.001266 | 0.560974 | 0.680303 | 0.800750 | 0.426667 | 0.015127 | 0.120330 |
| 697 | 0.684337 | 0.126667 | 0.638895 | 0.076703 | 0.792793 | 0.712750 | 0.608889 | 0.030319 | 0.379444 |
| 698 | 0.686337 | 0.382222 | 0.937336 | 0.868003 | 0.238506 | 0.784750 | 0.931111 | 0.018229 | 0.265510 |
| 699 | 0.003003 | 0.224444 | 0.045360 | 0.300447 | 0.101852 | 0.803750 | 0.402222 | 0.005692 | 0.023482 |
| 700 | 0.046409 | 0.135556 | 0.895051 | 0.553781 | 0.134259 | 0.351875 | 0.493333 | 0.030692 | 0.377364 |
| 701 | 0.059212 | 0.688889 | 0.147926 | 0.370472 | 0.576176 | 0.837625 | 0.840000 | 0.009600 | 0.102815 |
| 702 | 0.125125 | 0.113333 | 0.487028 | 0.735033 | 0.843220 | 0.854875 | 0.362222 | 0.030556 | 0.251821 |
| 703 | 0.195239 | 0.273333 | 0.945191 | 0.231794 | 0.924048 | 0.151375 | 0.771111 | 0.028003 | 0.390159 |
| 704 | 0.188638 | 0.840000 | 0.884317 | 0.568731 | 0.414843 | 0.096875 | 0.862222 | 0.016656 | 0.245928 |
| 705 | 0.217043 | 0.180000 | 0.712199 | 0.007740 | 0.240637 | 0.816625 | 0.877778 | 0.023048 | 0.230201 |
| 706 | 0.528106 | 0.128889 | 0.621197 | 0.416268 | 0.220149 | 0.921625 | 0.553333 | 0.022981 | 0.219937 |
| 707 | 0.531532 | 0.691111 | 0.855196 | 0.528633 | 0.162242 | 0.359375 | 0.280000 | 0.027038 | 0.325083 |
| 708 | 0.534535 | 0.584444 | 0.563628 | 0.388670 | 0.805556 | 0.892875 | 0.424444 | 0.021166 | 0.213357 |
| 709 | 0.495495 | 0.435556 | 0.006624 | 0.460809 | 0.742938 | 0.585625 | 0.817778 | 0.012462 | 0.143031 |
| 710 | 0.323065 | 0.286667 | 0.485518 | 0.688043 | 0.406627 | 0.399500 | 0.302222 | 0.026944 | 0.301478 |
| 711 | 0.425425 | 0.333333 | 0.411229 | 0.192503 | 0.133333 | 0.336375 | 0.000000 | 0.059491 | 0.491138 |
| 712 | 0.426426 | 0.566667 | 0.113304 | 0.066333 | 0.915692 | 0.539500 | 0.880000 | 0.013485 | 0.173962 |
| 713 | 0.427427 | 0.875556 | 0.891587 | 0.307406 | 0.619478 | 0.554500 | 0.931111 | 0.015664 | 0.234522 |
| 714 | 0.428428 | 0.348889 | 0.842677 | 0.311770 | 0.327909 | 0.659625 | 0.484444 | 0.027108 | 0.289951 |
| 715 | 0.521504 | 0.022222 | 0.224946 | 0.674521 | 0.190476 | 0.917375 | 0.731111 | 0.015501 | 0.089886 |
| 716 | 0.522505 | 0.797778 | 0.039473 | 0.140558 | 0.287341 | 0.695750 | 0.360000 | 0.009660 | 0.097492 |
| 717 | 0.522705 | 0.664444 | 0.301550 | 0.457559 | 0.933333 | 0.475500 | 0.295556 | 0.020904 | 0.270351 |
| 718 | 0.524505 | 0.675556 | 0.336129 | 0.134438 | 0.432249 | 0.514500 | 0.848889 | 0.016650 | 0.346890 |
| 719 | 0.530531 | 0.773333 | 0.881690 | 0.545566 | 0.414444 | 0.804750 | 0.980000 | 0.014355 | 0.191468 |
| 720 | 0.531532 | 0.984444 | 0.575914 | 0.792039 | 0.104182 | 0.094125 | 0.231111 | 0.022688 | 0.296183 |
| 721 | 0.532533 | 0.628889 | 0.399908 | 0.013657 | 0.884409 | 0.497500 | 0.055556 | 0.040292 | 0.347414 |


| 722 | 0.713343 | 0.671111 | 0.113971 | 0.303766 | 0.522331 | 0.862125 | 0.195556 | 0.014352 | 0.132537 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 723 | 0.686687 | 0.444444 | 0.105105 | 0.143143 | 0.883648 | 0.904875 | 0.400000 | 0.015927 | 0.166398 |
| 724 | 0.693694 | 0.024444 | 0.519796 | 0.084713 | 0.042398 | 0.405375 | 0.728889 | 0.032377 | 0.392808 |
| 725 | 0.616923 | 0.100000 | 0.979455 | 0.489863 | 0.170091 | 0.874125 | 0.962222 | 0.023609 | 0.261459 |
| 726 | 0.617323 | 0.344444 | 0.116611 | 0.435471 | 0.063953 | 0.993375 | 0.573333 | 0.000455 | 0.000180 |
| 727 | 0.642643 | 0.204444 | 0.108265 | 0.893452 | 0.768519 | 0.619625 | 0.006667 | 0.056376 | 0.418440 |
| 728 | 0.687337 | 0.717778 | 0.744013 | 0.263685 | 0.019588 | 0.892000 | 0.513333 | 0.012995 | 0.149334 |
| 729 | 0.692739 | 0.006667 | 0.912827 | 0.158668 | 0.030488 | 0.781375 | 0.140000 | 0.070739 | 0.446739 |
| 730 | 0.194639 | 0.764444 | 0.415951 | 0.767385 | 0.365385 | 0.562875 | 0.604444 | 0.014889 | 0.224634 |
| 731 | 0.194839 | 0.813333 | 0.405404 | 0.891190 | 0.021368 | 0.807000 | 0.740000 | 0.007862 | 0.096757 |
| 732 | 0.215215 | 0.326667 | 0.939860 | 0.535848 | 0.015915 | 0.093125 | 0.395556 | 0.033950 | 0.403061 |
| 733 | -0.049734 | 0.157778 | 0.004422 | 0.427442 | 0.384409 | 0.527500 | 0.108889 | 0.028299 | 0.225920 |
| 734 | -0.048733 | 0.811111 | 0.817226 | 0.352913 | 0.818525 | 0.936875 | 0.637778 | 0.014825 | 0.166482 |
| 735 | 0.144144 | 0.368889 | 0.589866 | 0.183900 | 0.594398 | 0.340375 | 0.317778 | 0.027780 | 0.365626 |
| 736 | 0.194039 | 0.926667 | 0.151627 | 0.406855 | 0.683579 | 0.538875 | 0.793333 | 0.011122 | 0.181952 |
| 737 | 0.583584 | 0.073333 | 0.723611 | 0.621099 | 0.899718 | 0.764750 | 0.222222 | 0.049617 | 0.372045 |
| 738 | 0.272855 | 0.026667 | 0.809609 | 0.938537 | 0.138889 | 0.817375 | 0.688889 | 0.034965 | 0.281050 |
| 739 | 0.273455 | 0.471111 | 0.977296 | 0.016472 | 0.180969 | 0.098000 | 0.568889 | 0.026626 | 0.375948 |
| 740 | 0.387387 | 0.640000 | 0.365528 | 0.945602 | 0.558712 | 0.938000 | 0.320000 | 0.016111 | 0.169527 |
| 741 | 0.388388 | 0.022222 | 0.674159 | 0.742125 | 0.843333 | 0.314375 | 0.186667 | 0.067875 | 0.470019 |
| 742 | 0.639640 | 0.595556 | 0.535623 | 0.893000 | 0.997053 | 0.388375 | 0.520000 | 0.021501 | 0.250399 |
| 743 | 0.128826 | 0.226667 | 0.123419 | 0.399820 | 0.311983 | 0.336625 | 0.717778 | 0.017803 | 0.209217 |
| 744 | 0.129226 | 0.435556 | 0.113511 | 0.251923 | 0.170168 | 0.685500 | 0.173333 | 0.014445 | 0.115209 |
| 745 | 0.114114 | 0.131111 | 0.859012 | 0.561691 | 0.977273 | 0.943000 | 0.820000 | 0.025127 | 0.286389 |
| 746 | 0.063013 | 0.222222 | 0.017449 | 0.614891 | 0.111502 | 0.602875 | 0.168889 | 0.015643 | 0.094042 |
| 747 | 0.128128 | 0.066667 | 0.076493 | 0.933273 | 0.231013 | 0.240250 | 0.542222 | 0.022267 | 0.218198 |
| 748 | 0.043043 | 0.733333 | 0.307927 | 0.002477 | 0.727954 | 0.306250 | 0.066667 | 0.027338 | 0.322551 |
| 749 | -0.048126 | 0.146667 | 0.876959 | 0.412573 | 0.093602 | 0.221875 | 0.642222 | 0.031639 | 0.350189 |
| 750 | -0.047726 | 0.122222 | 0.356019 | 0.273286 | 0.555556 | 0.539125 | 0.268889 | 0.036109 | 0.325793 |


| 751 | 0.378378 | 0.111111 | 0.966079 | 0.466836 | 0.865530 | 0.654625 | 0.988889 | 0.027520 | 0.363816 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 752 | 0.384384 | 0.457778 | 0.922841 | 0.971862 | 0.328483 | 0.914875 | 0.677778 | 0.017826 | 0.226604 |
| 753 | 0.427886 | 0.228889 | 0.534214 | 0.965900 | 0.709064 | 0.567500 | 0.131111 | 0.049043 | 0.407043 |
| 754 | 0.486486 | 0.364444 | 0.625352 | 0.223017 | 0.348485 | 0.954000 | 0.162222 | 0.026900 | 0.232619 |
| 755 | 0.694539 | 0.208889 | 0.487062 | 0.468627 | 0.039683 | 0.861000 | 0.315556 | 0.018888 | 0.136016 |
| 756 | 0.017403 | 0.248889 | 0.097334 | 0.378984 | 0.771127 | 0.801500 | 0.177778 | 0.028156 | 0.193278 |
| 757 | 0.321264 | 0.417778 | 0.737722 | 0.288253 | 0.680723 | 0.805750 | 0.484444 | 0.023017 | 0.309666 |
| 758 | -0.050735 | 0.168889 | 0.553105 | 0.204408 | 0.377637 | 0.283250 | 0.960000 | 0.023027 | 0.270504 |
| 759 | 0.634635 | 0.408889 | 0.875143 | 0.044079 | 0.818352 | 0.779750 | 0.975556 | 0.020378 | 0.272107 |
| 760 | -0.046125 | 0.337778 | 0.284595 | 0.007255 | 0.878662 | 0.104375 | 0.677778 | 0.019351 | 0.227739 |
| 761 | 0.214214 | 0.151111 | 0.707453 | 0.588866 | 0.521667 | 0.552500 | 0.108889 | 0.060382 | 0.468958 |
| 762 | 0.426885 | 0.548889 | 0.361528 | 0.674695 | 0.637550 | 0.763000 | 0.042222 | 0.037977 | 0.294404 |
| 763 | 0.507508 | 0.453333 | 0.548878 | 0.176426 | 0.522464 | 0.436375 | 0.268889 | 0.027046 | 0.361995 |
| 764 | 0.515516 | 0.500000 | 0.465664 | 0.347449 | 0.517521 | 0.573625 | 0.877778 | 0.019009 | 0.384412 |
| 765 | 0.516517 | 0.222222 | 0.474039 | 0.289410 | 0.363126 | 0.255250 | 0.368889 | 0.029319 | 0.361360 |
| 766 | 0.686937 | 0.531111 | 0.928569 | 0.930773 | 0.714674 | 0.369875 | 0.917778 | 0.019451 | 0.340968 |
| 767 | 0.195039 | 0.271111 | 0.446431 | 0.056607 | 0.627252 | 0.167250 | 0.455556 | 0.030585 | 0.368509 |
| 768 | 0.404081 | 0.822222 | 0.562540 | 0.514784 | 0.480820 | 0.909375 | 0.351111 | 0.017034 | 0.213169 |
| 769 | 0.427085 | 0.197778 | 0.743097 | 0.907563 | 0.951031 | 0.810750 | 0.075556 | 0.060573 | 0.443579 |
| 770 | 0.425885 | 0.051111 | 0.543810 | 0.625273 | 0.936047 | 0.975375 | 0.491111 | 0.041401 | 0.305299 |
| 771 | 0.494494 | 0.393333 | 0.686888 | 0.040164 | 0.904321 | 0.555500 | 0.273333 | 0.030344 | 0.415869 |
| 772 | 0.194239 | 0.084444 | 0.430405 | 0.514181 | 0.209459 | 0.435250 | 0.837778 | 0.028635 | 0.329603 |
| 773 | 0.534307 | 0.448889 | 0.734447 | 0.352453 | 0.306061 | 0.540500 | 0.002222 | 0.056650 | 0.479570 |
| 774 | 0.533907 | 0.308889 | 0.519206 | 0.447596 | 0.218680 | 0.516750 | 0.224444 | 0.034580 | 0.319847 |
| 775 | 0.426485 | 0.206667 | 0.520744 | 0.455701 | 0.884085 | 0.085000 | 0.313333 | 0.041466 | 0.374881 |
| 776 | 0.018004 | 0.104444 | 0.219905 | 0.678466 | 0.169540 | 0.558250 | 0.906667 | 0.018656 | 0.170121 |
| 777 | 0.615723 | 0.371111 | 0.032583 | 0.885596 | 0.361684 | 0.692125 | 0.340000 | 0.012556 | 0.112539 |
| 778 | 0.640641 | 0.664444 | 0.112428 | 0.800311 | 0.987374 | 0.233250 | 0.562222 | 0.015310 | 0.200988 |
| 779 | 0.641642 | 0.215556 | 0.259947 | 0.918920 | 0.111111 | 0.721750 | 0.322222 | 0.020108 | 0.159760 |


| 780 | 0.687738 | 0.673333 | 0.925279 | 0.662759 | 0.766026 | 0.925625 | 0.544444 | 0.019379 | 0.286633 |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 781 | 0.686737 | 0.780000 | 0.687654 | 0.609442 | 0.158382 | 0.537250 | 0.926667 | 0.015046 | 0.279644 |
| 782 | 0.357471 | 0.848889 | 0.837322 | 0.908743 | 0.184821 | 0.579125 | 0.162222 | 0.023558 | 0.348296 |
| 783 | 0.616123 | 0.962222 | 0.997854 | 0.838145 | 0.516816 | 0.616375 | 0.406667 | 0.019993 | 0.326147 |

Table E. 2 Waterside meta-model data

| Case\# | D1 | Pl/D1 | LR | $\theta$ | Velocity | Nu_CFD | ADP/Segment_CFD |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | [-] | [-] | [-] | [-] | [-] | [-] | [Pa] |
| 1 | 0.53106 | 0.29259 | 0.69138 | 0.36874 | 0.56713 | 27.13 | 346.79 |
| 2 | 0.58517 | 0.45892 | 0.05411 | 0.29259 | 0.90381 | 44.05 | 438.37 |
| 3 | 0.98196 | 0.75752 | 0.41683 | 0.05411 | 0.81162 | 40.87 | 1257.90 |
| 4 | 0.06012 | 0.63327 | 0.41884 | 0.59118 | 0.40882 | 13.20 | 414.32 |
| 5 | 0.60321 | 0.07816 | 0.96393 | 0.84168 | 0.11824 | 15.27 | 28.91 |
| 6 | 0.72946 | 0.21242 | 0.21643 | 0.50701 | 0.22846 | 24.28 | 44.99 |
| 7 | 0.22846 | 0.56914 | 0.94389 | 0.51303 | 0.77355 | 22.45 | 913.34 |
| 8 | 0.18437 | 0.92585 | 0.85772 | 0.57515 | 0.13828 | 7.73 | 110.35 |
| 9 | 0.53908 | 0.72144 | 0.61924 | 0.71343 | 0.44088 | 28.34 | 239.58 |
| 10 | 0.40882 | 0.60120 | 0.01002 | 0.15030 | 0.69739 | 25.01 | 461.39 |
| 11 | 0.65731 | 0.68136 | 0.20240 | 0.71743 | 0.47695 | 39.67 | 188.03 |
| 12 | 0.60922 | 0.08617 | 0.75752 | 0.60521 | 0.57315 | 37.21 | 299.43 |
| 13 | 0.59519 | 0.81964 | 0.55511 | 0.23046 | 0.32064 | 16.43 | 214.50 |
| 14 | 0.54509 | 0.84770 | 0.89780 | 0.14429 | 0.22044 | 10.50 | 235.74 |
| 15 | 0.65331 | 0.76353 | 0.28858 | 0.82565 | 0.15230 | 20.32 | 32.23 |
| 16 | 0.72745 | 0.43888 | 0.03808 | 0.52305 | 0.41483 | 37.09 | 106.11 |
| 17 | 0.39880 | 0.11824 | 0.27455 | 0.71142 | 0.36473 | 29.12 | 135.53 |
| 18 | 0.55912 | 0.22244 | 0.48697 | 0.96192 | 0.70140 | 55.83 | 519.28 |
| 19 | 0.60721 | 0.88577 | 0.27054 | 0.73547 | 0.80361 | 51.24 | 501.79 |
| 20 | 0.59920 | 0.42485 | 0.49699 | 0.77956 | 0.03607 | 8.02 | 1.92 |
| 21 | 0.69739 | 0.80160 | 0.35872 | 0.19439 | 0.54910 | 26.98 | 395.78 |


| 22 | 0.07214 | 0.28056 | 0.19238 | 0.51102 | 0.24649 | 11.26 | 138.89 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 23 | 0.06413 | 0.09018 | 0.89980 | 0.79559 | 0.62325 | 18.10 | 746.80 |
| 24 | 0.32265 | 0.36072 | 0.65531 | 0.57315 | 0.67735 | 28.03 | 502.25 |
| 25 | 0.41884 | 0.12625 | 0.23447 | 0.27255 | 0.09218 | 8.63 | 18.84 |
| 26 | 0.54108 | 0.98998 | 0.38277 | 0.22645 | 0.18838 | 11.20 | 102.70 |
| 27 | 0.43287 | 0.62725 | 0.07615 | 0.83768 | 0.67936 | 47.31 | 360.84 |
| 28 | 0.95792 | 0.87375 | 0.92786 | 0.63727 | 0.05210 | 8.22 | 6.78 |
| 29 | 0.76553 | 0.86373 | 0.34469 | 0.55711 | 0.02204 | 5.42 | 3.34 |
| 30 | 0.39479 | 0.04810 | 0.95391 | 0.85371 | 0.89178 | 45.98 | 984.45 |
| 31 | 0.89379 | 0.62525 | 0.76152 | 0.94790 | 0.82164 | 62.53 | 599.05 |
| 32 | 0.44890 | 0.53307 | 0.02806 | 0.54108 | 0.38878 | 28.46 | 121.95 |
| 33 | 0.66533 | 0.95792 | 0.31062 | 0.59920 | 0.48898 | 34.52 | 229.12 |
| 34 | 0.08417 | 0.26453 | 0.71944 | 0.50301 | 0.82365 | 18.38 | 1037.48 |
| 35 | 0.81563 | 0.60321 | 0.26052 | 0.37475 | 0.10822 | 13.09 | 18.52 |
| 36 | 0.88778 | 0.83166 | 0.80561 | 0.44088 | 0.84369 | 48.30 | 695.54 |
| 37 | 0.01403 | 0.59519 | 0.63928 | 0.41483 | 0.18236 | 6.19 | 267.23 |
| 38 | 0.49699 | 0.67936 | 0.44489 | 0.98798 | 0.83768 | 53.15 | 683.25 |
| 39 | 0.57715 | 0.35471 | 0.67535 | 0.91583 | 0.33667 | 30.16 | 151.42 |
| 40 | 0.89178 | 0.50100 | 0.97796 | 0.89780 | 0.60321 | 47.75 | 367.34 |
| 41 | 0.80962 | 0.82365 | 0.07816 | 0.38076 | 0.05611 | 9.78 | 4.69 |
| 42 | 0.92184 | 0.89980 | 0.32465 | 0.32665 | 0.90180 | 51.69 | 610.81 |
| 43 | 0.57315 | 0.76754 | 0.31864 | 0.63928 | 0.30261 | 24.71 | 105.98 |
| 44 | 0.58116 | 0.86974 | 0.52705 | 0.49900 | 0.49900 | 28.22 | 295.26 |
| 45 | 0.74749 | 0.07615 | 0.00601 | 0.57114 | 0.05812 | 14.56 | 0.99 |
| 46 | 0.06814 | 0.40281 | 0.88978 | 0.34469 | 0.84569 | 14.47 | 1607.03 |
| 47 | 0.17234 | 0.38477 | 0.47695 | 0.48096 | 0.17435 | 9.55 | 88.31 |
| 48 | 0.84770 | 0.38677 | 0.26253 | 0.15631 | 0.76553 | 40.92 | 494.91 |
| 49 | 0.19038 | 0.68737 | 0.75150 | 0.91182 | 0.08417 | 7.34 | 36.72 |
| 50 | 0.07615 | 0.93387 | 0.12826 | 0.47495 | 0.33467 | 12.05 | 266.42 |


| 51 | 0.88176 | 0.58517 | 0.43888 | 0.84569 | 0.44689 | 44.11 | 181.05 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 52 | 0.80561 | 0.54509 | 0.39078 | 0.72144 | 0.14830 | 20.51 | 28.70 |
| 53 | 0.36473 | 0.16232 | 0.52906 | 0.51503 | 0.76954 | 33.96 | 502.04 |
| 54 | 0.75150 | 0.02405 | 0.64930 | 0.39078 | 0.12425 | 12.18 | 21.32 |
| 55 | 0.11222 | 0.51102 | 0.04810 | 0.45892 | 0.73948 | 22.94 | 544.25 |
| 56 | 0.05611 | 0.42685 | 0.88377 | 0.94589 | 0.64529 | 18.35 | 918.43 |
| 57 | 0.09218 | 0.48697 | 0.70541 | 0.56313 | 0.69339 | 17.33 | 868.57 |
| 58 | 0.89980 | 0.30661 | 0.43487 | 0.03607 | 0.23647 | 14.70 | 188.45 |
| 59 | 0.54309 | 0.02204 | 0.13627 | 0.64729 | 0.96192 | 70.14 | 615.96 |
| 60 | 0.65932 | 0.16032 | 0.36473 | 0.69940 | 0.76353 | 57.86 | 443.82 |
| 61 | 0.22044 | 0.74749 | 0.67335 | 0.68537 | 0.76152 | 25.67 | 801.52 |
| 62 | 0.74148 | 0.27455 | 0.54509 | 0.13828 | 0.71543 | 34.24 | 619.32 |
| 63 | 0.85571 | 0.95992 | 0.45090 | 0.80160 | 0.72545 | 55.40 | 442.79 |
| 64 | 0.96192 | 0.23647 | 0.81964 | 0.44489 | 0.86172 | 53.82 | 535.72 |
| 65 | 0.02605 | 0.47896 | 0.30261 | 0.33467 | 0.64128 | 12.33 | 902.89 |
| 66 | 0.97595 | 0.20842 | 0.68337 | 0.47695 | 0.06613 | 10.45 | 7.03 |
| 67 | 0.64529 | 0.40481 | 0.05010 | 0.13226 | 0.25251 | 15.51 | 85.50 |
| 68 | 0.17836 | 0.08417 | 0.14429 | 0.56513 | 0.20641 | 14.47 | 66.45 |
| 69 | 0.31463 | 0.81162 | 0.07415 | 0.63126 | 0.78758 | 38.34 | 491.57 |
| 70 | 0.95591 | 0.52705 | 0.72946 | 0.70541 | 0.88377 | 60.95 | 604.29 |
| 71 | 0.56313 | 0.47495 | 0.03006 | 0.34870 | 0.10621 | 12.52 | 16.43 |
| 72 | 0.50701 | 0.06012 | 0.50301 | 0.70341 | 0.55912 | 40.26 | 306.53 |
| 73 | 0.25251 | 0.17836 | 0.66533 | 0.32265 | 0.83166 | 23.57 | 783.48 |
| 74 | 0.57916 | 0.40882 | 0.07014 | 0.91784 | 0.94389 | 75.16 | 607.88 |
| 75 | 0.95391 | 0.82966 | 0.59920 | 0.62926 | 0.11623 | 15.91 | 24.67 |
| 76 | 0.35271 | 0.58317 | 0.19038 | 0.72345 | 0.86974 | 43.77 | 588.88 |
| 77 | 0.42285 | 0.40681 | 0.28657 | 0.10020 | 0.09619 | 6.45 | 55.02 |
| 78 | 0.47094 | 0.99599 | 0.08818 | 0.66733 | 0.66333 | 40.22 | 344.01 |
| 79 | 0.48096 | 0.79559 | 0.81363 | 0.81764 | 0.61122 | 32.16 | 468.02 |


| 80 | 0.75952 | 0.99198 | 0.55711 | 0.87174 | 0.34269 | 30.75 | 147.92 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 81 | 0.50301 | 0.98397 | 0.22244 | 0.92585 | 0.98397 | 58.79 | 747.15 |
| 82 | 0.61323 | 0.25651 | 0.82966 | 0.77355 | 0.40281 | 30.75 | 197.59 |
| 83 | 0.37876 | 0.39078 | 0.20040 | 0.73747 | 0.50301 | 33.25 | 231.79 |
| 84 | 0.92585 | 0.00401 | 0.59719 | 0.91984 | 0.20040 | 34.05 | 68.54 |
| 85 | 0.94790 | 0.31263 | 0.55912 | 0.27856 | 0.86373 | 49.39 | 547.86 |
| 86 | 0.93387 | 0.79158 | 0.88577 | 0.73948 | 0.74749 | 52.52 | 542.90 |
| 87 | 0.69138 | 0.73747 | 0.72144 | 0.22846 | 0.16633 | 11.13 | 89.10 |
| 88 | 0.29459 | 0.34669 | 0.78958 | 0.69138 | 0.50701 | 23.54 | 361.76 |
| 89 | 0.39078 | 0.44689 | 0.70741 | 0.07615 | 0.54309 | 16.71 | 938.23 |
| 90 | 0.84970 | 0.33267 | 0.85371 | 0.33267 | 0.15832 | 13.40 | 49.33 |
| 91 | 0.73747 | 0.34469 | 0.18236 | 0.89579 | 0.52305 | 55.29 | 223.29 |
| 92 | 0.82766 | 0.60721 | 0.31463 | 0.80561 | 0.75752 | 61.79 | 406.50 |
| 93 | 0.21844 | 0.63126 | 0.71343 | 0.93186 | 0.99800 | 35.11 | 1222.45 |
| 94 | 0.79960 | 0.33667 | 0.50902 | 0.27455 | 0.40681 | 25.78 | 178.27 |
| 95 | 0.67335 | 0.29659 | 0.33467 | 0.53307 | 0.03006 | 5.75 | 4.35 |
| 96 | 0.96994 | 0.96994 | 0.24449 | 0.30661 | 0.03407 | 5.70 | 0.66 |
| 97 | 0.97996 | 0.39279 | 0.86974 | 0.56914 | 0.13427 | 15.61 | 26.96 |
| 98 | 0.66132 | 0.64529 | 0.83367 | 0.15832 | 0.87174 | 35.77 | 1274.50 |
| 99 | 0.70541 | 0.43086 | 0.74749 | 0.25651 | 0.47896 | 25.04 | 315.76 |
| 100 | 0.87776 | 0.87575 | 0.65932 | 0.69539 | 0.55110 | 41.81 | 306.33 |
| 101 | 0.19439 | 0.73146 | 0.25251 | 0.81964 | 0.39479 | 21.70 | 236.96 |
| 102 | 0.16834 | 0.13627 | 0.58918 | 0.98397 | 0.77956 | 35.07 | 980.77 |
| 103 | 0.56112 | 0.37675 | 0.76754 | 0.88978 | 0.04208 | 7.42 | 3.38 |
| 104 | 0.49900 | 0.31062 | 0.22044 | 0.90982 | 0.02605 | 8.22 | 4.11 |
| 105 | 0.84569 | 0.88176 | 0.45491 | 0.29860 | 0.20240 | 15.81 | 73.44 |
| 106 | 0.66333 | 0.55511 | 0.30461 | 0.98998 | 0.35471 | 39.19 | 136.14 |
| 107 | 0.87575 | 0.85571 | 0.41082 | 0.67936 | 0.32866 | 31.60 | 113.83 |
| 108 | 0.13226 | 0.89579 | 0.87174 | 0.92986 | 0.37675 | 15.34 | 405.44 |


| 109 | 0.81764 | 0.30862 | 0.42685 | 0.87575 | 0.86774 | 70.82 | 574.63 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 110 | 0.87174 | 0.73347 | 0.49900 | 0.58317 | 0.70942 | 50.27 | 403.43 |
| 111 | 0.96593 | 0.83367 | 0.13828 | 0.74549 | 0.57114 | 56.25 | 218.15 |
| 112 | 0.87976 | 0.72745 | 0.99800 | 0.90782 | 0.98798 | 61.74 | 895.34 |
| 113 | 0.94990 | 0.80962 | 0.29259 | 0.97194 | 0.46693 | 52.73 | 191.63 |
| 114 | 0.18036 | 0.51503 | 0.53707 | 0.51703 | 0.93186 | 26.67 | 1017.45 |
| 115 | 0.67735 | 0.52305 | 0.57114 | 0.60321 | 0.70741 | 44.34 | 429.33 |
| 116 | 0.47896 | 0.00000 | 0.95190 | 0.58717 | 0.06012 | 7.39 | 9.51 |
| 117 | 0.31263 | 0.45291 | 0.37475 | 0.81563 | 0.80160 | 39.61 | 610.37 |
| 118 | 0.30862 | 0.56513 | 0.98397 | 0.87776 | 0.75952 | 31.06 | 793.19 |
| 119 | 0.62926 | 0.71743 | 0.42084 | 0.84369 | 0.66533 | 46.90 | 396.73 |
| 120 | 0.65130 | 0.72345 | 0.89379 | 0.60922 | 0.92385 | 47.22 | 844.53 |
| 121 | 0.79760 | 0.51703 | 0.47896 | 0.09018 | 0.68537 | 33.05 | 777.19 |
| 122 | 0.78557 | 0.19439 | 0.66132 | 0.79960 | 0.26653 | 30.03 | 91.52 |
| 123 | 0.21242 | 0.69940 | 0.24248 | 0.40481 | 0.12224 | 8.41 | 51.19 |
| 124 | 0.63327 | 0.24048 | 0.51102 | 0.44890 | 0.80561 | 44.86 | 466.82 |
| 125 | 0.24048 | 0.25451 | 0.20441 | 0.57715 | 0.47495 | 24.13 | 222.92 |
| 126 | 0.93788 | 0.03407 | 0.87375 | 0.16433 | 0.46493 | 28.03 | 275.22 |
| 127 | 0.45691 | 0.49900 | 0.46092 | 0.54709 | 0.60120 | 31.53 | 347.07 |
| 128 | 0.34469 | 0.64128 | 0.62525 | 0.16633 | 0.73547 | 21.23 | 1078.27 |
| 129 | 0.36273 | 0.41283 | 0.07214 | 0.25251 | 0.24850 | 14.32 | 88.56 |
| 130 | 0.46293 | 0.47094 | 0.13026 | 0.19038 | 0.46894 | 20.85 | 244.97 |
| 131 | 0.61122 | 0.71944 | 0.80962 | 0.75351 | 0.01202 | 4.13 | 2.28 |
| 132 | 0.80762 | 0.48497 | 0.29659 | 0.35271 | 0.30461 | 23.74 | 88.99 |
| 133 | 0.64128 | 0.64729 | 0.01403 | 0.14830 | 0.43687 | 22.93 | 197.97 |
| 134 | 0.41283 | 0.35070 | 0.27655 | 0.63327 | 0.24048 | 20.43 | 71.91 |
| 135 | 0.14429 | 0.43487 | 0.45291 | 0.62325 | 0.59920 | 20.96 | 504.51 |
| 136 | 0.86974 | 0.03808 | 0.63527 | 0.39279 | 0.52705 | 37.72 | 204.79 |
| 137 | 0.71543 | 0.30060 | 0.42886 | 0.65130 | 0.36874 | 33.01 | 126.47 |


| 138 | 0.85772 | 0.44489 | 0.64529 | 0.84970 | 0.29058 | 31.46 | 98.68 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 139 | 0.73948 | 0.79359 | 0.04008 | 0.51904 | 0.96994 | 62.60 | 479.82 |
| 140 | 0.76152 | 0.30461 | 0.78557 | 0.61924 | 0.24449 | 22.07 | 74.77 |
| 141 | 0.84369 | 0.18236 | 0.41283 | 0.47295 | 0.12024 | 15.38 | 16.66 |
| 142 | 0.31663 | 0.42084 | 0.30661 | 0.53507 | 0.71343 | 31.06 | 436.89 |
| 143 | 0.10621 | 0.93788 | 0.36273 | 0.68938 | 0.62725 | 20.24 | 669.27 |
| 144 | 0.37675 | 0.49299 | 0.74950 | 0.11022 | 0.21844 | 8.86 | 250.43 |
| 145 | 0.36072 | 0.90581 | 0.08617 | 0.78758 | 0.18637 | 18.67 | 53.95 |
| 146 | 0.28457 | 0.09619 | 0.91984 | 0.57916 | 0.78958 | 29.18 | 689.69 |
| 147 | 0.67936 | 0.05411 | 0.09419 | 0.66934 | 0.57715 | 59.07 | 247.03 |
| 148 | 0.68537 | 0.10020 | 0.82766 | 0.31463 | 0.41884 | 23.83 | 199.12 |
| 149 | 0.74349 | 0.38277 | 0.81563 | 0.97996 | 0.97996 | 66.32 | 887.32 |
| 150 | 0.63727 | 0.32265 | 0.94589 | 0.94389 | 0.57515 | 41.54 | 388.94 |
| 151 | 0.23447 | 0.95591 | 0.21443 | 0.98597 | 0.39279 | 25.61 | 242.72 |
| 152 | 0.42485 | 0.46293 | 0.58116 | 0.96994 | 0.78557 | 46.82 | 657.65 |
| 153 | 0.45892 | 0.67335 | 0.67735 | 0.82766 | 0.94990 | 46.72 | 880.26 |
| 154 | 0.52906 | 0.09820 | 0.98196 | 0.74950 | 0.40481 | 28.44 | 216.76 |
| 155 | 0.24248 | 0.89178 | 0.17635 | 0.10421 | 0.55311 | 14.31 | 755.80 |
| 156 | 0.31062 | 0.42886 | 0.73347 | 0.43888 | 0.84770 | 28.15 | 827.72 |
| 157 | 0.33267 | 0.25852 | 0.84168 | 0.25451 | 0.60521 | 19.52 | 602.00 |
| 158 | 0.29058 | 0.44289 | 0.12024 | 0.53106 | 0.17034 | 14.30 | 46.15 |
| 159 | 0.70942 | 0.42285 | 0.54108 | 0.65932 | 0.91984 | 57.12 | 625.33 |
| 160 | 0.49299 | 0.18838 | 0.10220 | 0.49299 | 0.26052 | 23.30 | 59.04 |
| 161 | 0.75551 | 0.31864 | 0.70942 | 0.63527 | 0.78357 | 51.26 | 487.97 |
| 162 | 0.28056 | 0.96794 | 0.88778 | 0.29459 | 0.82565 | 21.68 | 1421.19 |
| 163 | 0.85170 | 0.21443 | 0.25451 | 0.47896 | 0.51303 | 43.20 | 162.57 |
| 164 | 0.51303 | 0.39880 | 0.84770 | 0.67335 | 0.25050 | 18.72 | 102.22 |
| 165 | 0.33868 | 0.78557 | 0.53106 | 0.90180 | 0.49499 | 28.58 | 334.46 |
| 166 | 0.28257 | 0.71543 | 0.40080 | 0.46894 | 0.87976 | 30.03 | 807.46 |


| 167 | 0.83367 | 0.48096 | 0.03206 | 0.71543 | 0.17836 | 28.37 | 27.48 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 168 | 0.61924 | 0.59920 | 0.87976 | 0.87375 | 0.88778 | 50.37 | 768.73 |
| 169 | 0.10822 | 0.24449 | 0.60922 | 0.73146 | 0.65531 | 21.38 | 625.33 |
| 170 | 0.68337 | 0.70341 | 0.26453 | 0.37074 | 0.41283 | 27.08 | 165.89 |
| 171 | 0.11623 | 0.57916 | 0.34870 | 0.22445 | 0.88978 | 16.98 | 1334.61 |
| 172 | 0.31864 | 0.52505 | 0.86573 | 0.50501 | 0.12826 | 8.70 | 62.68 |
| 173 | 0.72144 | 0.94188 | 0.00000 | 0.80962 | 0.28858 | 35.93 | 76.54 |
| 174 | 0.30060 | 0.99399 | 0.50501 | 0.73347 | 0.29459 | 17.93 | 172.99 |
| 175 | 0.74950 | 0.03006 | 0.00401 | 0.32866 | 0.74549 | 58.08 | 235.27 |
| 176 | 0.13828 | 0.00200 | 0.79760 | 0.34068 | 0.77756 | 18.03 | 840.67 |
| 177 | 0.44489 | 0.02605 | 0.23046 | 0.75752 | 0.58717 | 54.13 | 407.75 |
| 178 | 0.70140 | 0.43687 | 0.15230 | 0.86974 | 0.07214 | 15.44 | 4.90 |
| 179 | 0.29659 | 0.08818 | 0.01804 | 0.38677 | 0.35872 | 22.07 | 108.96 |
| 180 | 0.24449 | 0.94990 | 0.64329 | 0.27655 | 0.92184 | 22.43 | 1514.86 |
| 181 | 0.67535 | 0.66533 | 0.46293 | 0.79359 | 0.94589 | 59.62 | 695.05 |
| 182 | 0.90581 | 0.36874 | 0.03407 | 0.08216 | 0.08818 | 8.68 | 18.71 |
| 183 | 0.67134 | 0.66934 | 0.83768 | 0.40681 | 0.64729 | 33.35 | 499.38 |
| 184 | 0.53507 | 0.69539 | 0.59118 | 0.35872 | 0.69539 | 30.90 | 551.89 |
| 185 | 0.82365 | 0.73948 | 0.91383 | 0.37275 | 0.34469 | 21.98 | 190.13 |
| 186 | 0.38277 | 0.55110 | 0.46493 | 0.82966 | 0.30661 | 23.11 | 135.37 |
| 187 | 0.60521 | 0.59719 | 0.37876 | 0.14028 | 0.24248 | 12.96 | 141.77 |
| 188 | 0.82565 | 0.92385 | 0.97996 | 0.68737 | 0.51904 | 35.51 | 341.75 |
| 189 | 0.52705 | 0.74950 | 0.72345 | 0.54509 | 0.14429 | 10.02 | 50.37 |
| 190 | 0.25651 | 0.70140 | 0.48096 | 0.64529 | 0.57916 | 24.32 | 436.52 |
| 191 | 0.07014 | 0.13026 | 0.35471 | 0.16834 | 0.36273 | 8.57 | 434.33 |
| 192 | 0.03808 | 0.99800 | 0.75351 | 0.45090 | 0.65331 | 11.81 | 1387.81 |
| 193 | 0.40281 | 0.35671 | 0.73747 | 0.35671 | 0.29659 | 14.71 | 168.47 |
| 194 | 0.81363 | 0.26052 | 0.70341 | 0.66333 | 0.47295 | 38.94 | 208.83 |
| 195 | 0.21042 | 0.91383 | 0.38878 | 0.52505 | 0.41082 | 17.20 | 309.26 |


| 196 | 0.33066 | 0.53507 | 0.24048 | 0.99198 | 0.51703 | 39.12 | 330.40 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 197 | 0.24850 | 0.67134 | 0.29058 | 0.69739 | 0.16834 | 13.42 | 63.80 |
| 198 | 0.16032 | 0.61323 | 0.96794 | 0.68337 | 0.37475 | 14.19 | 359.85 |
| 199 | 0.62124 | 0.66132 | 0.46894 | 0.18036 | 0.91383 | 36.92 | 969.57 |
| 200 | 0.77555 | 0.09218 | 0.80160 | 0.28056 | 0.68337 | 37.29 | 412.81 |
| 201 | 0.90381 | 0.70942 | 0.11824 | 0.30461 | 0.32265 | 26.19 | 89.29 |
| 202 | 0.71743 | 0.89379 | 0.09619 | 0.67535 | 0.60721 | 48.45 | 258.22 |
| 203 | 0.46894 | 0.02806 | 0.40481 | 0.08016 | 0.75551 | 26.38 | 787.68 |
| 204 | 0.32665 | 0.22846 | 0.31663 | 0.67735 | 0.79960 | 40.64 | 542.98 |
| 205 | 0.40681 | 0.90982 | 0.43086 | 0.56112 | 0.73747 | 32.98 | 576.20 |
| 206 | 0.22645 | 0.79960 | 0.74549 | 0.26854 | 0.68136 | 17.21 | 1027.60 |
| 207 | 0.35671 | 0.98196 | 0.19439 | 0.22244 | 0.77154 | 24.97 | 743.46 |
| 208 | 0.16433 | 0.68537 | 0.43687 | 0.92385 | 0.95190 | 33.87 | 1070.39 |
| 209 | 0.69940 | 0.74549 | 0.14228 | 0.46693 | 0.71142 | 45.58 | 327.26 |
| 210 | 0.30261 | 0.89780 | 0.64729 | 0.24048 | 0.31062 | 11.41 | 321.96 |
| 211 | 0.92385 | 0.67735 | 0.68737 | 0.07415 | 0.65932 | 34.26 | 993.89 |
| 212 | 0.77154 | 0.56313 | 0.13226 | 0.20842 | 0.62525 | 34.92 | 304.21 |
| 213 | 0.57515 | 0.65731 | 0.94188 | 0.26453 | 0.09419 | 6.66 | 43.84 |
| 214 | 0.93186 | 0.85371 | 0.65731 | 0.33667 | 0.49098 | 31.70 | 293.37 |
| 215 | 0.66733 | 0.28457 | 0.02605 | 0.76353 | 0.67134 | 65.73 | 313.81 |
| 216 | 0.99399 | 0.26854 | 0.36874 | 0.58517 | 0.28657 | 33.01 | 68.71 |
| 217 | 0.98597 | 0.34068 | 0.44689 | 0.93587 | 0.52104 | 58.13 | 232.00 |
| 218 | 0.78156 | 0.10822 | 0.21042 | 0.80361 | 0.29259 | 41.05 | 86.44 |
| 219 | 0.50100 | 0.79760 | 0.25050 | 0.22044 | 0.37074 | 17.82 | 216.28 |
| 220 | 0.09018 | 0.49098 | 0.14028 | 0.35471 | 0.17635 | 8.65 | 108.93 |
| 221 | 0.09820 | 0.10421 | 0.90982 | 0.53908 | 0.09820 | 6.33 | 62.17 |
| 222 | 0.15431 | 0.24649 | 0.39479 | 0.45691 | 0.39078 | 15.63 | 244.86 |
| 223 | 0.63126 | 0.76152 | 0.01202 | 0.49699 | 0.32665 | 28.17 | 85.12 |
| 224 | 0.71343 | 0.18437 | 0.06413 | 0.45491 | 0.59319 | 46.02 | 180.68 |


| 225 | 0.91984 | 0.21042 | 0.41483 | 0.58116 | 0.71743 | 58.77 | 330.58 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 226 | 0.85972 | 0.05812 | 0.99599 | 0.36072 | 0.85571 | 48.74 | 574.31 |
| 227 | 0.27856 | 0.61122 | 0.63327 | 0.46293 | 0.67335 | 22.93 | 604.27 |
| 228 | 0.74549 | 0.66733 | 0.35271 | 0.50100 | 0.93387 | 54.60 | 578.42 |
| 229 | 0.48297 | 0.57114 | 0.93186 | 0.20641 | 0.42084 | 16.80 | 432.43 |
| 230 | 0.73547 | 0.65531 | 0.80361 | 0.86172 | 0.28257 | 25.89 | 112.93 |
| 231 | 0.95190 | 0.53908 | 0.29860 | 0.93988 | 0.04409 | 11.74 | 4.90 |
| 232 | 0.60120 | 0.50902 | 0.38477 | 0.60120 | 0.25451 | 22.20 | 74.97 |
| 233 | 0.97796 | 0.37275 | 0.44088 | 0.81363 | 0.21643 | 30.40 | 51.42 |
| 234 | 0.94188 | 0.17034 | 0.94990 | 0.97796 | 0.27655 | 34.59 | 115.99 |
| 235 | 0.89579 | 0.11623 | 0.51303 | 0.71944 | 0.06814 | 13.93 | 5.24 |
| 236 | 0.88577 | 0.93186 | 0.28457 | 0.13427 | 0.27856 | 17.61 | 156.73 |
| 237 | 0.21443 | 0.83768 | 0.38677 | 0.72946 | 0.02405 | 5.03 | 2.09 |
| 238 | 0.94389 | 0.84369 | 0.44890 | 0.90381 | 0.10421 | 18.54 | 15.40 |
| 239 | 0.41683 | 0.97996 | 0.53307 | 0.06814 | 0.85371 | 24.28 | 2136.04 |
| 240 | 0.04609 | 0.85170 | 0.61122 | 0.70140 | 0.13226 | 6.59 | 124.86 |
| 241 | 0.15230 | 0.50701 | 0.24850 | 0.77555 | 0.61523 | 26.61 | 494.28 |
| 242 | 0.77956 | 0.58116 | 0.73547 | 0.60721 | 0.16433 | 16.62 | 43.14 |
| 243 | 0.81162 | 0.55311 | 0.72745 | 0.55511 | 0.48297 | 34.08 | 234.84 |
| 244 | 0.99599 | 0.19639 | 0.09820 | 0.18437 | 0.43888 | 32.36 | 126.40 |
| 245 | 0.58717 | 0.46493 | 0.30060 | 0.88577 | 0.75150 | 55.52 | 447.82 |
| 246 | 0.78758 | 0.81764 | 0.99198 | 0.20441 | 0.82766 | 38.83 | 1163.25 |
| 247 | 0.43687 | 0.97395 | 0.09018 | 0.65731 | 0.99198 | 50.37 | 670.69 |
| 248 | 0.49098 | 0.15631 | 0.48297 | 0.38277 | 0.25651 | 16.83 | 87.02 |
| 249 | 0.77355 | 0.58717 | 0.98798 | 0.58918 | 0.20842 | 17.95 | 77.06 |
| 250 | 0.16633 | 0.40080 | 0.34669 | 0.74148 | 0.14629 | 11.37 | 59.24 |
| 251 | 0.18637 | 0.32064 | 0.42285 | 0.04208 | 0.36673 | 9.16 | 729.33 |
| 252 | 0.46693 | 0.82565 | 0.94790 | 0.75551 | 0.20441 | 15.09 | 97.96 |
| 253 | 0.23046 | 0.38076 | 0.33868 | 0.83166 | 0.42485 | 24.60 | 232.49 |


| 254 | 0.37374 | 0.57576 | 0.68687 | 0.84848 | 0.17172 | 15.29 | 65.72 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 255 | 0.06212 | 0.34870 | 0.83567 | 0.82365 | 0.18437 | 8.38 | 149.79 |
| 256 | 0.52305 | 0.96593 | 0.21242 | 0.40080 | 0.36072 | 22.15 | 158.69 |
| 257 | 0.51515 | 0.73737 | 0.30303 | 0.69697 | 0.34343 | 26.55 | 130.85 |
| 258 | 0.66667 | 0.25253 | 0.48485 | 0.29293 | 0.52525 | 28.95 | 258.29 |
| 259 | 0.54910 | 0.54709 | 0.37675 | 0.05010 | 0.73146 | 25.89 | 1168.65 |
| 260 | 0.96794 | 0.44088 | 0.27856 | 0.61523 | 0.79559 | 65.31 | 376.08 |
| 261 | 0.33667 | 0.77355 | 0.95591 | 0.30862 | 0.08016 | 5.69 | 55.46 |
| 262 | 0.83567 | 0.04008 | 0.90782 | 0.61122 | 0.38277 | 32.94 | 149.60 |
| 263 | 0.75351 | 0.06613 | 0.79559 | 0.89379 | 0.43287 | 43.88 | 280.75 |

Table F. 1 Design problem I results

| N_r | OD | PI/OD | Pt/OD | LR | $\theta$ | V_a | v_w | N_tpr | Volume | material volume | Power | Q_total' | Afr_air | A_air | A_water | Height | Length | AHTC | ADP | WHTC | WDP |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| - | - | - | - | - | - | m/s | $\mathrm{m} / \mathrm{s}$ | - | $\mathrm{cm}^{3}$ | cm3 | w | w | $\mathrm{m}^{2}$ | $\mathrm{m}^{2}$ | $\mathrm{m}^{2}$ | m | m | $\mathrm{w} / \mathrm{m}^{2} \mathrm{~K}$ | Pa | W/ m ${ }^{\mathbf{K}} \mathrm{K}$ | kPa |
| 4 | 0.8 | 1.6 | 1.5 | 6.4 | 46 | 2.8 | 0.4 | 108 | 85.41 | 10.12 | 2.41 | 849.18 | 0.02 | 0.14 | 0.11 | 0.13 | 0.14 | 430.02 | 52.47 | 9787 | 4.44 |
| 4 | 0.8 | 1.6 | 1.5 | 6.5 | 46 | 2.8 | 0.4 | 108 | 85.70 | 10.16 | 2.40 | 850.77 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 429.71 | 52.15 | 9775 | 4.45 |
| 4 | 0.8 | 1.6 | 1.5 | 6.5 | 47 | 2.8 | 0.4 | 109 | 85.99 | 10.20 | 2.39 | 850.96 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 429.23 | 51.88 | 9752 | 4.38 |
| 4 | 0.8 | 1.6 | 1.5 | 6.4 | 46 | 2.8 | 0.4 | 109 | 86.15 | 10.20 | 2.39 | 851.33 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 429.64 | 51.93 | 9741 | 4.38 |
| 4 | 0.8 | 1.6 | 1.5 | 6.4 | 46 | 2.8 | 0.4 | 109 | 86.21 | 10.21 | 2.38 | 850.26 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 428.67 | 51.61 | 9734 | 4.38 |
| 4 | 0.8 | 1.6 | 1.5 | 6.5 | 46 | 2.8 | 0.4 | 109 | 86.29 | 10.23 | 2.38 | 851.56 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 428.99 | 51.61 | 9733 | 4.39 |
| 4 | 0.8 | 1.6 | 1.5 | 6.5 | 46 | 2.8 | 0.4 | 109 | 86.37 | 10.24 | 2.37 | 852.89 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 429.30 | 51.61 | 9730 | 4.39 |
| 4 | 0.8 | 1.6 | 1.5 | 6.5 | 46 | 2.8 | 0.4 | 109 | 86.49 | 10.25 | 2.37 | 854.39 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 429.74 | 51.64 | 9723 | 4.40 |
| 4 | 0.8 | 1.6 | 1.5 | 6.5 | 46 | 2.8 | 0.4 | 109 | 86.56 | 10.25 | 2.37 | 854.39 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 429.74 | 51.63 | 9723 | 4.40 |
| 4 | 0.8 | 1.6 | 1.5 | 6.5 | 46 | 2.8 | 0.4 | 109 | 86.97 | 10.24 | 2.37 | 852.78 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 429.15 | 51.54 | 9726 | 4.39 |
| 4 | 0.8 | 1.6 | 1.5 | 6.5 | 46 | 2.8 | 0.4 | 109 | 87.04 | 10.25 | 2.37 | 854.11 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 429.46 | 51.54 | 9723 | 4.40 |
| 4 | 0.8 | 1.6 | 1.5 | 6.6 | 47 | 2.7 | 0.4 | 109 | 87.47 | 10.35 | 2.33 | 855.61 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 427.91 | 50.34 | 9710 | 4.46 |
| 4 | 0.8 | 1.6 | 1.5 | 6.6 | 46 | 2.7 | 0.4 | 108 | 87.67 | 10.37 | 2.33 | 858.95 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 428.86 | 50.37 | 9705 | 4.54 |
| 4 | 0.8 | 1.6 | 1.5 | 6.6 | 47 | 2.7 | 0.4 | 109 | 87.75 | 10.38 | 2.32 | 858.64 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 428.93 | 50.41 | 9700 | 4.47 |
| 4 | 0.8 | 1.6 | 1.5 | 6.6 | 47 | 2.7 | 0.4 | 109 | 88.13 | 10.42 | 2.31 | 857.64 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 427.22 | 49.77 | 9680 | 4.48 |
| 4 | 0.8 | 1.6 | 1.5 | 6.6 | 46 | 2.7 | 0.4 | 109 | 88.35 | 10.45 | 2.30 | 860.59 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 428.06 | 49.82 | 9661 | 4.48 |
| 4 | 0.8 | 1.6 | 1.5 | 6.6 | 46 | 2.7 | 0.4 | 109 | 88.48 | 10.46 | 2.29 | 859.28 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 426.91 | 49.49 | 9654 | 4.48 |
| 4 | 0.8 | 1.6 | 1.5 | 6.6 | 46 | 2.7 | 0.4 | 109 | 88.76 | 10.49 | 2.28 | 859.49 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 426.34 | 49.20 | 9645 | 4.49 |
| 4 | 0.8 | 1.6 | 1.5 | 6.6 | 47 | 2.7 | 0.4 | 109 | 90.32 | 10.51 | 2.12 | 853.15 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 419.79 | 46.09 | 9643 | 4.49 |
| 4 | 0.8 | 1.6 | 1.5 | 6.6 | 46 | 2.6 | 0.4 | 109 | 90.93 | 10.58 | 2.10 | 854.46 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 418.37 | 45.48 | 9604 | 4.50 |
| 4 | 0.8 | 1.6 | 1.5 | 6.6 | 46 | 2.6 | 0.4 | 109 | 91.36 | 10.62 | 2.08 | 855.91 | 0.02 | 0.15 | 0.11 | 0.13 | 0.14 | 418.02 | 45.22 | 9588 | 4.51 |


| 4 | 0.8 | 1.6 | 1.5 | 7.2 | 46 | 2.5 | 0.4 | 109 | 95.15 | 11.18 | 2.06 | 874.13 | 0.02 | 0.16 | 0.12 | 0.14 | 0.14 | 412.91 | 43.68 | 9506 | 4.72 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4 | 0.8 | 1.6 | 1.5 | 7.2 | 46 | 2.5 | 0.4 | 109 | 95.57 | 11.23 | 2.04 | 875.67 | 0.02 | 0.16 | 0.12 | 0.14 | 0.14 | 411.90 | 43.33 | 9497 | 4.73 |
| 4 | 0.8 | 1.6 | 1.5 | 7.2 | 47 | 2.5 | 0.4 | 108 | 96.08 | 11.11 | 1.93 | 864.70 | 0.02 | 0.16 | 0.12 | 0.14 | 0.14 | 406.04 | 41.24 | 9534 | 4.78 |
| 4 | 0.8 | 1.6 | 1.5 | 7.2 | 47 | 2.5 | 0.4 | 109 | 96.97 | 11.21 | 1.90 | 865.71 | 0.02 | 0.16 | 0.12 | 0.14 | 0.14 | 403.88 | 40.60 | 9487 | 4.72 |
| 4 | 0.8 | 1.6 | 1.5 | 7.4 | 46 | 2.4 | 0.4 | 109 | 100.41 | 11.58 | 1.78 | 867.37 | 0.02 | 0.17 | 0.12 | 0.15 | 0.14 | 392.62 | 37.63 | 9422 | 4.82 |
| 4 | 0.8 | 1.6 | 1.5 | 7.4 | 46 | 2.4 | 0.4 | 109 | 100.71 | 11.61 | 1.73 | 855.15 | 0.02 | 0.17 | 0.12 | 0.15 | 0.14 | 381.02 | 36.17 | 9421 | 4.83 |
| 4 | 0.8 | 1.6 | 1.5 | 7.2 | 46 | 2.3 | 0.4 | 113 | 103.52 | 11.87 | 1.72 | 870.11 | 0.02 | 0.17 | 0.13 | 0.14 | 0.15 | 385.17 | 36.17 | 9222 | 4.55 |

Table F. 2 Design problem II results

| N_r | OD | PI/OD | Pt/OD | LR | $\theta$ | V_a | V_w | N_tpr | Volume | material volume | Power | Q_total' | Afr_air | A_air | A_water | Height | Length | AHTC | ADP | WHTC | WDP |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| - | - | - | - | - | - | m/s | m/s | - | $\mathrm{cm}^{3}$ | cm3 | w | w | $\mathrm{m}^{2}$ | $\mathrm{m}^{2}$ | $\mathrm{m}^{2}$ | m | m | W/ m ${ }^{2} \mathrm{~K}$ | Pa | $\mathrm{w} / \mathrm{m}^{2} \mathrm{~K}$ | kPa |
| 4 | 0.5 | 1.9 | 1.5 | 6.7 | 63 | 2.3 | 0.5 | 200 | 62 | 7.28 | 1.51 | 839.87 | 0.02 | 0.15 | 0.12 | 0.11 | 0.20 | 386.38 | 20.91 | 14346 | 9.20 |
| 4 | 0.5 | 1.9 | 1.5 | 6.7 | 63 | 2.3 | 0.5 | 200 | 62 | 7.28 | 1.51 | 839.87 | 0.02 | 0.15 | 0.12 | 0.11 | 0.20 | 386.38 | 20.91 | 14346 | 9.20 |
| 4 | 0.5 | 1.9 | 1.5 | 6.1 | 63 | 2.4 | 0.4 | 200 | 63 | 7.48 | 1.50 | 835.53 | 0.02 | 0.15 | 0.12 | 0.10 | 0.20 | 384.91 | 21.84 | 14255 | 8.03 |
| 4 | 0.5 | 1.9 | 1.5 | 6.1 | 63 | 2.4 | 0.4 | 200 | 63 | 7.48 | 1.50 | 835.74 | 0.02 | 0.15 | 0.12 | 0.10 | 0.20 | 384.80 | 21.81 | 14249 | 8.03 |
| 4 | 0.5 | 1.9 | 1.5 | 6.2 | 63 | 2.3 | 0.4 | 200 | 64 | 7.53 | 1.48 | 835.85 | 0.02 | 0.16 | 0.12 | 0.11 | 0.20 | 382.48 | 21.32 | 14194 | 8.06 |
| 4 | 0.5 | 1.9 | 1.5 | 6.2 | 63 | 2.3 | 0.4 | 200 | 64 | 7.57 | 1.47 | 837.62 | 0.02 | 0.16 | 0.12 | 0.11 | 0.20 | 381.97 | 21.20 | 14136 | 8.08 |
| 4 | 0.5 | 1.9 | 1.5 | 6.2 | 63 | 2.3 | 0.4 | 200 | 64 | 7.58 | 1.46 | 837.00 | 0.02 | 0.16 | 0.12 | 0.11 | 0.20 | 381.12 | 21.02 | 14134 | 8.08 |
| 4 | 0.5 | 1.9 | 1.5 | 6.3 | 63 | 2.3 | 0.4 | 200 | 65 | 7.66 | 1.44 | 839.74 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 379.43 | 20.64 | 14026 | 8.12 |
| 4 | 0.5 | 1.9 | 1.5 | 6.3 | 63 | 2.3 | 0.4 | 200 | 65 | 7.68 | 1.43 | 839.76 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 378.26 | 20.37 | 14002 | 8.13 |
| 4 | 0.5 | 1.9 | 1.5 | 6.7 | 63 | 2.3 | 0.4 | 200 | 65 | 7.62 | 1.43 | 843.06 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 377.63 | 19.71 | 14011 | 8.78 |
| 4 | 0.5 | 1.9 | 1.5 | 6.4 | 63 | 2.3 | 0.4 | 200 | 65 | 7.73 | 1.42 | 840.83 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 376.88 | 20.09 | 13944 | 8.16 |
| 4 | 0.5 | 1.9 | 1.5 | 6.1 | 63 | 2.3 | 0.4 | 200 | 66 | 7.82 | 1.42 | 837.82 | 0.02 | 0.16 | 0.13 | 0.11 | 0.21 | 376.08 | 20.53 | 13913 | 7.65 |
| 4 | 0.5 | 1.9 | 1.5 | 6.5 | 63 | 2.3 | 0.4 | 200 | 66 | 7.81 | 1.40 | 843.47 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 375.22 | 19.74 | 13840 | 8.20 |
| 4 | 0.5 | 1.9 | 1.5 | 6.5 | 63 | 2.2 | 0.4 | 200 | 66 | 7.82 | 1.40 | 843.37 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 374.78 | 19.67 | 13838 | 8.22 |
| 4 | 0.5 | 1.9 | 1.5 | 6.5 | 63 | 2.2 | 0.4 | 200 | 66 | 7.84 | 1.40 | 844.11 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 374.37 | 19.58 | 13806 | 8.22 |


| 4 | 0.5 | 1.9 | 1.5 | 6.5 | 63 | 2.2 | 0.4 | 200 | 66 | 7.84 | 1.40 | 844.32 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 374.28 | 19.55 | 13798 | 8.22 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4 | 0.5 | 1.9 | 1.5 | 6.5 | 63 | 2.2 | 0.4 | 200 | 66 | 7.84 | 1.39 | 844.11 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 374.09 | 19.53 | 13798 | 8.22 |
| 4 | 0.5 | 1.9 | 1.5 | 6.5 | 63 | 2.2 | 0.4 | 200 | 67 | 7.85 | 1.39 | 843.38 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 373.14 | 19.37 | 13796 | 8.22 |
| 4 | 0.5 | 1.9 | 1.5 | 6.5 | 63 | 2.2 | 0.4 | 200 | 67 | 7.86 | 1.39 | 843.59 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 373.04 | 19.34 | 13789 | 8.23 |
| 4 | 0.5 | 1.9 | 1.5 | 6.6 | 63 | 2.2 | 0.4 | 200 | 67 | 7.90 | 1.38 | 845.30 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 372.61 | 19.26 | 13737 | 8.25 |
| 4 | 0.5 | 1.9 | 1.5 | 6.6 | 63 | 2.2 | 0.4 | 200 | 68 | 7.96 | 1.36 | 845.85 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 370.00 | 18.79 | 13666 | 8.30 |
| 4 | 0.5 | 1.9 | 1.5 | 6.6 | 63 | 2.2 | 0.4 | 200 | 68 | 7.96 | 1.36 | 845.85 | 0.02 | 0.16 | 0.13 | 0.11 | 0.20 | 370.00 | 18.79 | 13666 | 8.30 |
| 4 | 0.6 | 1.9 | 1.5 | 6.0 | 63 | 2.2 | 0.4 | 200 | 68 | 8.14 | 1.36 | 840.90 | 0.02 | 0.16 | 0.13 | 0.11 | 0.21 | 369.77 | 19.79 | 13561 | 7.23 |
| 4 | 0.5 | 1.9 | 1.5 | 6.7 | 63 | 2.2 | 0.4 | 200 | 68 | 8.05 | 1.34 | 848.01 | 0.02 | 0.17 | 0.13 | 0.11 | 0.20 | 367.78 | 18.36 | 13567 | 8.36 |
| 4 | 0.5 | 1.9 | 1.5 | 6.7 | 63 | 2.2 | 0.4 | 200 | 69 | 8.05 | 1.34 | 847.80 | 0.02 | 0.17 | 0.13 | 0.11 | 0.20 | 367.60 | 18.34 | 13567 | 8.36 |
| 4 | 0.5 | 1.9 | 1.5 | 6.7 | 63 | 2.2 | 0.4 | 200 | 69 | 8.05 | 1.34 | 847.80 | 0.02 | 0.17 | 0.13 | 0.11 | 0.20 | 367.60 | 18.34 | 13567 | 8.36 |
| 4 | 0.6 | 1.9 | 1.5 | 6.1 | 63 | 2.2 | 0.4 | 200 | 69 | 8.22 | 1.34 | 842.73 | 0.02 | 0.16 | 0.13 | 0.11 | 0.21 | 367.54 | 19.38 | 13469 | 7.28 |
| 4 | 0.6 | 1.9 | 1.5 | 6.1 | 63 | 2.2 | 0.4 | 200 | 69 | 8.22 | 1.34 | 842.73 | 0.02 | 0.16 | 0.13 | 0.11 | 0.21 | 367.54 | 19.38 | 13469 | 7.28 |
| 4 | 0.6 | 1.9 | 1.5 | 6.1 | 63 | 2.2 | 0.4 | 200 | 69 | 8.24 | 1.33 | 842.41 | 0.02 | 0.16 | 0.13 | 0.11 | 0.21 | 366.39 | 19.15 | 13453 | 7.29 |
| 4 | 0.6 | 1.9 | 1.5 | 6.1 | 63 | 2.2 | 0.4 | 200 | 69 | 8.25 | 1.33 | 842.51 | 0.02 | 0.16 | 0.13 | 0.11 | 0.21 | 366.19 | 19.11 | 13445 | 7.29 |
| 4 | 0.6 | 1.9 | 1.5 | 6.1 | 63 | 2.2 | 0.4 | 200 | 69 | 8.25 | 1.33 | 842.51 | 0.02 | 0.16 | 0.13 | 0.11 | 0.21 | 366.19 | 19.11 | 13445 | 7.29 |
| 4 | 0.6 | 1.9 | 1.5 | 6.1 | 63 | 2.2 | 0.4 | 200 | 69 | 8.25 | 1.33 | 842.61 | 0.02 | 0.16 | 0.13 | 0.11 | 0.21 | 365.98 | 19.08 | 13439 | 7.30 |
| 4 | 0.6 | 1.9 | 1.5 | 6.2 | 63 | 2.2 | 0.4 | 200 | 70 | 8.34 | 1.31 | 845.37 | 0.02 | 0.17 | 0.13 | 0.11 | 0.21 | 364.34 | 18.73 | 13335 | 7.34 |
| 4 | 0.6 | 1.9 | 1.5 | 6.2 | 63 | 2.2 | 0.4 | 200 | 70 | 8.33 | 1.30 | 844.32 | 0.02 | 0.17 | 0.13 | 0.11 | 0.21 | 363.72 | 18.64 | 13347 | 7.33 |
| 4 | 0.6 | 1.9 | 1.5 | 6.2 | 63 | 2.2 | 0.4 | 200 | 70 | 8.35 | 1.30 | 844.85 | 0.02 | 0.17 | 0.13 | 0.11 | 0.21 | 363.31 | 18.55 | 13326 | 7.34 |
| 4 | 0.6 | 1.9 | 1.5 | 6.2 | 63 | 2.2 | 0.4 | 200 | 70 | 8.35 | 1.30 | 844.74 | 0.02 | 0.17 | 0.13 | 0.11 | 0.21 | 363.22 | 18.54 | 13326 | 7.34 |
| 4 | 0.6 | 1.9 | 1.5 | 6.2 | 63 | 2.2 | 0.4 | 200 | 71 | 8.35 | 1.30 | 844.64 | 0.02 | 0.17 | 0.13 | 0.11 | 0.21 | 363.13 | 18.53 | 13326 | 7.34 |
| 4 | 0.5 | 1.9 | 1.5 | 6.6 | 63 | 2.1 | 0.4 | 200 | 71 | 8.34 | 1.29 | 850.04 | 0.02 | 0.17 | 0.14 | 0.11 | 0.21 | 362.52 | 17.87 | 13294 | 7.88 |
| 4 | 0.5 | 1.9 | 1.5 | 6.6 | 63 | 2.1 | 0.4 | 200 | 71 | 8.36 | 1.28 | 849.71 | 0.02 | 0.17 | 0.14 | 0.11 | 0.21 | 361.39 | 17.67 | 13279 | 7.89 |
| 4 | 0.6 | 1.9 | 1.5 | 6.3 | 63 | 2.2 | 0.4 | 200 | 71 | 8.44 | 1.28 | 846.68 | 0.02 | 0.17 | 0.13 | 0.11 | 0.21 | 360.94 | 18.10 | 13229 | 7.38 |
| 4 | 0.6 | 1.9 | 1.5 | 6.5 | 63 | 2.1 | 0.4 | 200 | 72 | 8.58 | 1.26 | 850.21 | 0.02 | 0.17 | 0.14 | 0.11 | 0.21 | 358.16 | 17.60 | 13081 | 7.44 |


| 4 | 0.6 | 1.9 | 1.5 | 6.4 | 63 | 2.1 | 0.4 | 200 | 73 | 8.55 | 1.26 | 848.72 | 0.02 | 0.17 | 0.14 | 0.11 | 0.21 | 357.98 | 17.58 | 13112 | 7.42 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4 | 0.6 | 1.9 | 1.5 | 6.1 | 63 | 2.1 | 0.4 | 200 | 73 | 8.63 | 1.26 | 845.79 | 0.02 | 0.17 | 0.14 | 0.11 | 0.21 | 357.87 | 18.04 | 13061 | 6.97 |
| 4 | 0.6 | 1.9 | 1.5 | 6.5 | 63 | 2.1 | 0.4 | 200 | 73 | 8.58 | 1.26 | 850.20 | 0.02 | 0.17 | 0.14 | 0.11 | 0.21 | 357.88 | 17.55 | 13075 | 7.44 |
| 4 | 0.6 | 1.9 | 1.5 | 6.5 | 63 | 2.1 | 0.4 | 200 | 73 | 8.65 | 1.24 | 850.80 | 0.02 | 0.17 | 0.14 | 0.11 | 0.21 | 355.66 | 17.14 | 13012 | 7.47 |
| 4 | 0.6 | 1.9 | 1.5 | 6.5 | 63 | 2.1 | 0.4 | 200 | 73 | 8.65 | 1.24 | 850.80 | 0.02 | 0.17 | 0.14 | 0.11 | 0.21 | 355.66 | 17.14 | 13012 | 7.47 |
| 4 | 0.6 | 1.9 | 1.5 | 6.5 | 63 | 2.1 | 0.4 | 200 | 73 | 8.65 | 1.24 | 850.80 | 0.02 | 0.17 | 0.14 | 0.11 | 0.21 | 355.66 | 17.14 | 13012 | 7.47 |
| 4 | 0.6 | 1.9 | 1.5 | 6.6 | 63 | 2.1 | 0.4 | 200 | 74 | 8.75 | 1.22 | 852.67 | 0.02 | 0.17 | 0.14 | 0.12 | 0.21 | 352.97 | 16.66 | 12908 | 7.52 |
| 4 | 0.6 | 1.9 | 1.5 | 6.0 | 63 | 2.1 | 0.4 | 200 | 75 | 8.95 | 1.21 | 847.08 | 0.02 | 0.17 | 0.14 | 0.11 | 0.22 | 351.90 | 17.46 | 12773 | 6.62 |
| 4 | 0.6 | 1.9 | 1.5 | 6.7 | 63 | 2.0 | 0.4 | 200 | 75 | 8.86 | 1.20 | 854.57 | 0.02 | 0.18 | 0.14 | 0.12 | 0.21 | 350.37 | 16.21 | 12807 | 7.59 |
| 4 | 0.6 | 1.9 | 1.5 | 6.2 | 63 | 2.1 | 0.4 | 200 | 77 | 9.11 | 1.18 | 850.22 | 0.02 | 0.18 | 0.14 | 0.11 | 0.22 | 347.82 | 16.69 | 12612 | 6.69 |
| 4 | 0.6 | 1.9 | 1.5 | 6.2 | 63 | 2.1 | 0.4 | 200 | 77 | 9.11 | 1.18 | 850.22 | 0.02 | 0.18 | 0.14 | 0.11 | 0.22 | 347.82 | 16.69 | 12612 | 6.69 |
| 4 | 0.6 | 1.9 | 1.5 | 6.5 | 63 | 2.0 | 0.4 | 200 | 77 | 9.06 | 1.17 | 853.83 | 0.02 | 0.18 | 0.14 | 0.12 | 0.21 | 347.38 | 16.13 | 12639 | 7.14 |
| 4 | 0.6 | 1.9 | 1.5 | 6.3 | 63 | 2.0 | 0.4 | 200 | 78 | 9.25 | 1.15 | 853.63 | 0.02 | 0.18 | 0.14 | 0.11 | 0.22 | 345.22 | 16.21 | 12477 | 6.74 |

Table F. 3 Automotive radiator optimization results

| N_r | OD | PI/OD | Pt/OD | LR | $\theta$ | V_a | V_w | N_tpr | Volume | material volume | Power | Q_total' | Afr_air | A_air | A_water | Height | Length | AHTC | ADP | WHTC | WDP |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| - | - | - | - | - | - | $\mathrm{m} / \mathrm{s}$ | m/s | - | $\mathrm{cm}^{3}$ | cm3 | w | w | $\mathrm{m}^{2}$ | $\mathrm{m}^{2}$ | $\mathrm{m}^{2}$ | m | m | $\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}$ | Pa | W/ m ${ }^{\mathbf{2}}$ | kPa |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 183 | 2112 | 155.7 | 63.6 | 10345.2 | 0.19 | 2.06 | 1.54 | 0.44 | 0.44 | 389.0 | 72.7 | 8868 | 15.8 |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 183 | 2112 | 155.7 | 63.6 | 10345.2 | 0.19 | 2.06 | 1.54 | 0.44 | 0.44 | 389.0 | 72.7 | 8868 | 15.8 |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2126 | 157.6 | 62.3 | 10435.9 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 389.1 | 71.3 | 8816 | 15.4 |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2137 | 157.6 | 62.2 | 10431.6 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 388.7 | 71.2 | 8816 | 15.4 |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2137 | 157.6 | 62.2 | 10431.6 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 388.7 | 71.2 | 8816 | 15.4 |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2137 | 157.6 | 62.2 | 10431.6 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 388.7 | 71.2 | 8816 | 15.4 |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2141 | 157.6 | 62.1 | 10428.7 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 388.5 | 71.1 | 8816 | 15.4 |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2141 | 157.6 | 62.1 | 10428.7 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 388.5 | 71.1 | 8816 | 15.4 |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2141 | 157.6 | 62.1 | 10428.7 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 388.5 | 71.1 | 8816 | 15.4 |


| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2141 | 157.6 | 62.1 | 10428.7 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 388.5 | 71.1 | 8816 | 15.4 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2141 | 157.6 | 62.1 | 10428.7 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 388.5 | 71.1 | 8816 | 15.4 |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2141 | 157.6 | 62.1 | 10428.7 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 388.5 | 71.1 | 8816 | 15.4 |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2141 | 157.6 | 62.1 | 10428.7 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 388.5 | 71.1 | 8816 | 15.4 |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2141 | 157.6 | 62.1 | 10428.7 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 388.5 | 71.1 | 8816 | 15.4 |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2141 | 157.6 | 62.1 | 10428.7 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 388.5 | 71.1 | 8816 | 15.4 |
| 9 | 0.9 | 2.8 | 1.5 | 5.5 | 35 | 4.2 | 0.5 | 186 | 2141 | 157.6 | 62.1 | 10428.7 | 0.20 | 2.08 | 1.56 | 0.44 | 0.45 | 388.5 | 71.1 | 8816 | 15.4 |
| 9 | 0.9 | 3.0 | 1.5 | 5.5 | 40 | 4.1 | 0.5 | 186 | 2147 | 150.4 | 58.5 | 9985.8 | 0.20 | 1.99 | 1.49 | 0.42 | 0.48 | 371.8 | 66.4 | 9511 | 16.0 |
| 9 | 0.9 | 3.0 | 1.5 | 5.5 | 40 | 4.1 | 0.5 | 186 | 2147 | 150.4 | 58.5 | 9985.8 | 0.20 | 1.99 | 1.49 | 0.42 | 0.48 | 371.8 | 66.4 | 9511 | 16.0 |
| 9 | 0.9 | 3.0 | 1.5 | 5.5 | 40 | 4.1 | 0.5 | 186 | 2147 | 150.4 | 58.5 | 9985.8 | 0.20 | 1.99 | 1.49 | 0.42 | 0.48 | 371.8 | 66.4 | 9511 | 16.0 |
| 9 | 0.9 | 3.0 | 1.5 | 5.5 | 40 | 4.1 | 0.5 | 186 | 2147 | 150.4 | 58.5 | 9985.8 | 0.20 | 1.99 | 1.49 | 0.42 | 0.48 | 371.8 | 66.4 | 9511 | 16.0 |
| 9 | 0.9 | 3.0 | 1.5 | 5.5 | 40 | 4.0 | 0.5 | 186 | 2181 | 151.5 | 53.8 | 10089.6 | 0.20 | 2.00 | 1.50 | 0.42 | 0.48 | 377.0 | 60.6 | 9562 | 16.3 |
| 9 | 0.9 | 3.0 | 1.5 | 5.5 | 40 | 4.0 | 0.5 | 186 | 2181 | 151.5 | 53.8 | 10089.6 | 0.20 | 2.00 | 1.50 | 0.42 | 0.48 | 377.0 | 60.6 | 9562 | 16.3 |

Table F. 4 HVRF system indoor coil optimization results

| N_r | OD | PI/OD | Pt/OD | LR | $\theta$ | v_a | v_w | N_tpr | Volume | material volume | Power | Q_total' | Afr_air | A_air | A_water | Height | Length | AHTC | ADP | WHTC | WDP |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| - | - | - | - | - | - | m/s | m/s | - | $\mathrm{cm}^{3}$ | cm3 | w | w | $\mathrm{m}^{2}$ | $\mathrm{m}^{2}$ | $\mathrm{m}^{2}$ | m | m | W/ m ${ }^{\mathbf{K}}$ | Pa | $\mathrm{w} / \mathrm{m}^{2} \mathrm{~K}$ | kPa |
| 5 | 0.5 | 2.9 | 1.5 | 3.8 | 19.9 | 1.9 | 0.4 | 556 | 575.5 | 40.86 | 3.65 | 3173 | 0.15 | 0.85 | 0.68 | 0.17 | 0.88 | 327.6 | 9.58 | 8322 | 8.46 |
| 5 | 0.5 | 2.9 | 1.5 | 3.9 | 19.9 | 1.8 | 0.4 | 548 | 584.2 | 41.36 | 3.61 | 3170 | 0.16 | 0.86 | 0.69 | 0.18 | 0.86 | 324.0 | 9.22 | 8420 | 8.70 |
| 5 | 0.5 | 2.9 | 1.5 | 3.8 | 19.8 | 1.8 | 0.4 | 548 | 585.9 | 41.68 | 3.61 | 3180 | 0.15 | 0.86 | 0.69 | 0.18 | 0.87 | 324.5 | 9.37 | 8246 | 8.46 |
| 5 | 0.5 | 2.9 | 1.5 | 3.9 | 19.9 | 1.8 | 0.4 | 556 | 587.0 | 41.57 | 3.58 | 3192 | 0.16 | 0.86 | 0.69 | 0.18 | 0.88 | 324.6 | 9.25 | 8353 | 8.55 |
| 5 | 0.5 | 2.9 | 1.5 | 3.8 | 19.9 | 1.8 | 0.4 | 556 | 592.8 | 42.13 | 3.54 | 3183 | 0.16 | 0.87 | 0.69 | 0.18 | 0.89 | 322.3 | 9.17 | 8219 | 8.36 |
| 5 | 0.5 | 2.9 | 1.5 | 3.8 | 19.8 | 1.8 | 0.4 | 556 | 594.4 | 42.29 | 3.54 | 3196 | 0.16 | 0.87 | 0.70 | 0.18 | 0.89 | 322.6 | 9.19 | 8196 | 8.37 |
| 5 | 0.5 | 2.9 | 1.5 | 3.8 | 19.9 | 1.8 | 0.4 | 556 | 595.7 | 42.37 | 3.53 | 3188 | 0.16 | 0.87 | 0.70 | 0.18 | 0.89 | 321.1 | 9.07 | 8227 | 8.38 |
| 5 | 0.5 | 2.9 | 1.5 | 3.8 | 19.8 | 1.8 | 0.4 | 556 | 597.4 | 42.45 | 3.52 | 3195 | 0.16 | 0.87 | 0.70 | 0.18 | 0.89 | 321.5 | 9.09 | 8204 | 8.39 |


| 5 | 0.5 | 2.9 | 1.5 | 4.0 | 19.9 | 1.8 | 0.4 | 548 | 606.9 | 43.05 | 3.49 | 3207 | 0.16 | 0.89 | 0.71 | 0.18 | 0.87 | 317.9 | 8.78 | 8318 | 8.65 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 5 | 0.5 | 2.9 | 1.5 | 4.0 | 19.9 | 1.7 | 0.4 | 556 | 618.7 | 43.84 | 3.41 | 3220 | 0.16 | 0.90 | 0.72 | 0.18 | 0.89 | 314.9 | 8.54 | 8278 | 8.57 |
| 5 | 0.6 | 2.9 | 1.5 | 3.9 | 19.9 | 1.7 | 0.4 | 548 | 638.9 | 45.39 | 3.32 | 3212 | 0.16 | 0.91 | 0.73 | 0.18 | 0.89 | 310.4 | 8.39 | 7990 | 8.19 |
| 5 | 0.6 | 2.9 | 1.5 | 3.8 | 19.9 | 1.7 | 0.4 | 548 | 640.1 | 45.44 | 3.32 | 3191 | 0.16 | 0.91 | 0.72 | 0.18 | 0.90 | 310.5 | 8.50 | 7874 | 7.97 |
| 5 | 0.6 | 2.9 | 1.5 | 3.9 | 19.9 | 1.7 | 0.4 | 548 | 640.6 | 45.47 | 3.32 | 3217 | 0.17 | 0.91 | 0.73 | 0.18 | 0.89 | 310.4 | 8.38 | 8006 | 8.20 |
| 5 | 0.6 | 2.9 | 1.5 | 3.9 | 19.9 | 1.7 | 0.4 | 548 | 640.6 | 45.47 | 3.32 | 3217 | 0.17 | 0.91 | 0.73 | 0.18 | 0.89 | 310.4 | 8.38 | 8006 | 8.20 |
| 5 | 0.6 | 2.9 | 1.5 | 3.8 | 19.9 | 1.7 | 0.4 | 548 | 641.9 | 45.62 | 3.31 | 3204 | 0.16 | 0.91 | 0.73 | 0.18 | 0.90 | 310.7 | 8.49 | 7874 | 7.97 |
| 5 | 0.6 | 2.9 | 1.5 | 3.9 | 19.9 | 1.7 | 0.4 | 548 | 646.2 | 45.82 | 3.29 | 3213 | 0.17 | 0.92 | 0.74 | 0.19 | 0.89 | 307.9 | 8.20 | 8006 | 8.24 |
| 5 | 0.6 | 2.9 | 1.5 | 3.8 | 19.8 | 1.7 | 0.4 | 548 | 648.3 | 45.97 | 3.28 | 3201 | 0.17 | 0.92 | 0.73 | 0.18 | 0.90 | 308.5 | 8.32 | 7868 | 8.02 |
| 5 | 0.6 | 2.9 | 1.5 | 3.9 | 19.9 | 1.7 | 0.4 | 548 | 652.9 | 46.23 | 3.26 | 3207 | 0.17 | 0.92 | 0.74 | 0.19 | 0.90 | 307.5 | 8.21 | 7906 | 8.06 |
| 5 | 0.6 | 2.9 | 1.5 | 3.9 | 19.9 | 1.7 | 0.4 | 548 | 652.9 | 46.23 | 3.26 | 3207 | 0.17 | 0.92 | 0.74 | 0.19 | 0.90 | 307.5 | 8.21 | 7906 | 8.06 |
| 5 | 0.6 | 2.9 | 1.5 | 3.9 | 19.9 | 1.7 | 0.4 | 548 | 652.9 | 46.23 | 3.26 | 3207 | 0.17 | 0.92 | 0.74 | 0.19 | 0.90 | 307.5 | 8.21 | 7906 | 8.06 |
| 5 | 0.6 | 2.9 | 1.5 | 3.9 | 19.9 | 1.7 | 0.4 | 548 | 656.1 | 46.50 | 3.24 | 3211 | 0.17 | 0.93 | 0.74 | 0.19 | 0.90 | 306.3 | 8.12 | 7914 | 8.08 |
| 5 | 0.6 | 2.9 | 1.5 | 3.9 | 19.9 | 1.7 | 0.4 | 548 | 656.1 | 46.50 | 3.24 | 3211 | 0.17 | 0.93 | 0.74 | 0.19 | 0.90 | 306.3 | 8.12 | 7914 | 8.08 |
| 5 | 0.6 | 2.9 | 1.5 | 3.9 | 19.9 | 1.7 | 0.4 | 548 | 658.6 | 46.68 | 3.23 | 3209 | 0.17 | 0.93 | 0.74 | 0.19 | 0.90 | 305.0 | 8.03 | 7907 | 8.10 |
| 5 | 0.6 | 2.9 | 1.5 | 4.0 | 19.9 | 1.7 | 0.4 | 548 | 665.6 | 47.11 | 3.20 | 3227 | 0.17 | 0.94 | 0.75 | 0.19 | 0.90 | 304.4 | 7.93 | 7939 | 8.15 |
| 5 | 0.6 | 2.9 | 1.5 | 4.0 | 19.9 | 1.7 | 0.4 | 548 | 668.8 | 47.29 | 3.19 | 3225 | 0.17 | 0.94 | 0.75 | 0.19 | 0.90 | 303.2 | 7.84 | 7947 | 8.18 |
| 5 | 0.6 | 2.9 | 1.5 | 4.0 | 19.9 | 1.7 | 0.4 | 548 | 668.8 | 47.29 | 3.19 | 3225 | 0.17 | 0.94 | 0.75 | 0.19 | 0.90 | 303.2 | 7.84 | 7947 | 8.18 |
| 5 | 0.6 | 2.9 | 1.5 | 4.0 | 19.9 | 1.7 | 0.4 | 548 | 671.3 | 47.47 | 3.18 | 3222 | 0.17 | 0.95 | 0.76 | 0.19 | 0.90 | 301.8 | 7.75 | 7948 | 8.20 |
| 5 | 0.6 | 2.9 | 1.5 | 4.0 | 19.9 | 1.7 | 0.4 | 548 | 671.4 | 47.47 | 3.18 | 3222 | 0.17 | 0.95 | 0.76 | 0.19 | 0.90 | 301.9 | 7.76 | 7940 | 8.20 |
| 5 | 0.6 | 2.9 | 1.5 | 4.0 | 19.9 | 1.7 | 0.4 | 548 | 671.4 | 47.47 | 3.18 | 3222 | 0.17 | 0.95 | 0.76 | 0.19 | 0.90 | 301.9 | 7.76 | 7940 | 8.20 |
| 5 | 0.6 | 2.9 | 1.5 | 4.0 | 19.9 | 1.7 | 0.4 | 548 | 671.4 | 47.47 | 3.18 | 3222 | 0.17 | 0.95 | 0.76 | 0.19 | 0.90 | 301.9 | 7.76 | 7940 | 8.20 |
| 5 | 0.6 | 2.9 | 1.5 | 4.0 | 19.9 | 1.7 | 0.4 | 548 | 671.4 | 47.47 | 3.18 | 3222 | 0.17 | 0.95 | 0.76 | 0.19 | 0.90 | 301.9 | 7.76 | 7940 | 8.20 |
| 5 | 0.6 | 2.9 | 1.5 | 4.0 | 19.9 | 1.6 | 0.4 | 548 | 675.2 | 47.73 | 3.16 | 3226 | 0.17 | 0.95 | 0.76 | 0.19 | 0.90 | 300.8 | 7.66 | 7964 | 8.23 |
| 5 | 0.6 | 2.9 | 1.5 | 4.0 | 19.9 | 1.6 | 0.4 | 548 | 675.2 | 47.73 | 3.16 | 3226 | 0.17 | 0.95 | 0.76 | 0.19 | 0.90 | 300.8 | 7.66 | 7964 | 8.23 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 678.4 | 47.90 | 3.15 | 3240 | 0.17 | 0.96 | 0.76 | 0.19 | 0.90 | 301.4 | 7.66 | 7973 | 8.25 |


| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 678.4 | 47.90 | 3.15 | 3240 | 0.17 | 0.96 | 0.76 | 0.19 | 0.90 | 301.4 | 7.66 | 7973 | 8.25 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 678.4 | 47.90 | 3.15 | 3240 | 0.17 | 0.96 | 0.76 | 0.19 | 0.90 | 301.4 | 7.66 | 7973 | 8.25 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 678.4 | 47.90 | 3.15 | 3240 | 0.17 | 0.96 | 0.76 | 0.19 | 0.90 | 301.4 | 7.66 | 7973 | 8.25 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 681.6 | 48.08 | 3.14 | 3238 | 0.17 | 0.96 | 0.77 | 0.19 | 0.90 | 300.1 | 7.57 | 7981 | 8.28 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 681.6 | 48.08 | 3.14 | 3238 | 0.17 | 0.96 | 0.77 | 0.19 | 0.90 | 300.1 | 7.57 | 7981 | 8.28 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.8 | 1.6 | 0.4 | 548 | 683.6 | 48.28 | 3.13 | 3234 | 0.17 | 0.96 | 0.77 | 0.19 | 0.90 | 298.7 | 7.49 | 7960 | 8.29 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 684.2 | 48.27 | 3.13 | 3234 | 0.17 | 0.96 | 0.77 | 0.19 | 0.90 | 298.8 | 7.49 | 7975 | 8.29 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 684.2 | 48.27 | 3.13 | 3234 | 0.17 | 0.96 | 0.77 | 0.19 | 0.90 | 298.8 | 7.49 | 7975 | 8.29 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 684.2 | 48.27 | 3.13 | 3234 | 0.17 | 0.96 | 0.77 | 0.19 | 0.90 | 298.8 | 7.49 | 7975 | 8.29 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 684.7 | 48.34 | 3.13 | 3240 | 0.17 | 0.96 | 0.77 | 0.19 | 0.90 | 298.9 | 7.48 | 7990 | 8.30 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.7 | 1.6 | 0.4 | 548 | 686.3 | 48.46 | 3.12 | 3248 | 0.18 | 0.97 | 0.77 | 0.19 | 0.90 | 299.2 | 7.50 | 7953 | 8.30 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.8 | 1.6 | 0.4 | 548 | 686.9 | 48.45 | 3.12 | 3248 | 0.18 | 0.97 | 0.77 | 0.19 | 0.90 | 299.3 | 7.50 | 7969 | 8.31 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.8 | 1.6 | 0.4 | 548 | 686.9 | 48.45 | 3.12 | 3248 | 0.18 | 0.97 | 0.77 | 0.19 | 0.90 | 299.3 | 7.50 | 7969 | 8.31 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 687.9 | 48.52 | 3.12 | 3237 | 0.18 | 0.97 | 0.77 | 0.20 | 0.90 | 297.6 | 7.39 | 7999 | 8.32 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 687.9 | 48.52 | 3.12 | 3237 | 0.18 | 0.97 | 0.77 | 0.20 | 0.90 | 297.6 | 7.39 | 7999 | 8.32 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 687.9 | 48.52 | 3.12 | 3237 | 0.18 | 0.97 | 0.77 | 0.20 | 0.90 | 297.6 | 7.39 | 7999 | 8.32 |
| 5 | 0.6 | 2.9 | 1.5 | 4.1 | 19.9 | 1.6 | 0.4 | 548 | 687.9 | 48.52 | 3.12 | 3237 | 0.18 | 0.97 | 0.77 | 0.20 | 0.90 | 297.6 | 7.39 | 7999 | 8.32 |
| 5 | 0.6 | 2.9 | 1.5 | 4.4 | 19.9 | 1.6 | 0.4 | 548 | 705.6 | 49.70 | 3.06 | 3281 | 0.18 | 1.00 | 0.80 | 0.20 | 0.89 | 293.3 | 6.94 | 8178 | 8.70 |
| 5 | 0.5 | 2.9 | 1.5 | 5.5 | 19.9 | 1.4 | 0.4 | 548 | 773.7 | 53.64 | 2.96 | 3384 | 0.21 | 1.11 | 0.89 | 0.24 | 0.86 | 277.9 | 5.44 | 9211 | 10.28 |
| 5 | 0.5 | 2.9 | 1.5 | 5.3 | 19.9 | 1.4 | 0.4 | 556 | 777.9 | 54.15 | 2.90 | 3380 | 0.21 | 1.12 | 0.89 | 0.23 | 0.89 | 276.8 | 5.46 | 8919 | 9.85 |
| 5 | 0.5 | 2.9 | 1.5 | 5.5 | 19.9 | 1.3 | 0.4 | 556 | 801.4 | 55.69 | 2.85 | 3407 | 0.21 | 1.15 | 0.92 | 0.24 | 0.89 | 272.8 | 5.16 | 9060 | 10.00 |
| 5 | 0.5 | 2.9 | 1.5 | 5.3 | 19.9 | 1.3 | 0.4 | 556 | 816.6 | 56.67 | 2.78 | 3403 | 0.21 | 1.15 | 0.92 | 0.24 | 0.90 | 271.9 | 5.18 | 8727 | 9.58 |
| 5 | 0.5 | 2.9 | 1.5 | 5.3 | 19.9 | 1.3 | 0.4 | 556 | 819.6 | 56.84 | 2.78 | 3394 | 0.21 | 1.15 | 0.92 | 0.24 | 0.90 | 270.4 | 5.11 | 8743 | 9.59 |
| 5 | 0.6 | 2.9 | 1.5 | 5.3 | 19.9 | 1.3 | 0.4 | 548 | 844.2 | 58.63 | 2.72 | 3394 | 0.22 | 1.17 | 0.94 | 0.24 | 0.90 | 267.1 | 4.97 | 8568 | 9.42 |
| 5 | 0.6 | 2.9 | 1.5 | 5.3 | 19.9 | 1.3 | 0.4 | 548 | 845.8 | 58.86 | 2.72 | 3407 | 0.22 | 1.17 | 0.94 | 0.24 | 0.90 | 267.3 | 4.97 | 8552 | 9.43 |
| 5 | 0.6 | 2.9 | 1.5 | 5.4 | 19.9 | 1.3 | 0.4 | 548 | 856.9 | 59.51 | 2.70 | 3409 | 0.22 | 1.19 | 0.95 | 0.24 | 0.90 | 265.0 | 4.82 | 8635 | 9.49 |


| 5 | 0.6 | 2.9 | 1.5 | 5.4 | 19.9 | 1.3 | 0.4 | 548 | 858.7 | 59.69 | 2.69 | 3421 | 0.22 | 1.19 | 0.95 | 0.24 | 0.90 | 265.5 | 4.82 | 8653 | 9.51 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 5 | 0.6 | 2.9 | 1.5 | 5.4 | 19.9 | 1.3 | 0.4 | 548 | 858.7 | 59.69 | 2.69 | 3421 | 0.22 | 1.19 | 0.95 | 0.24 | 0.90 | 265.5 | 4.82 | 8653 | 9.51 |
| 5 | 0.6 | 2.9 | 1.5 | 5.4 | 19.9 | 1.3 | 0.4 | 548 | 858.7 | 59.69 | 2.69 | 3421 | 0.22 | 1.19 | 0.95 | 0.24 | 0.90 | 265.5 | 4.82 | 8653 | 9.51 |
| 5 | 0.6 | 2.9 | 1.5 | 5.4 | 19.9 | 1.3 | 0.4 | 548 | 860.1 | 59.69 | 2.69 | 3421 | 0.22 | 1.19 | 0.95 | 0.24 | 0.90 | 265.5 | 4.82 | 8653 | 9.51 |
| 5 | 0.6 | 2.9 | 1.5 | 5.4 | 19.9 | 1.3 | 0.4 | 548 | 860.1 | 59.69 | 2.69 | 3421 | 0.22 | 1.19 | 0.95 | 0.24 | 0.90 | 265.5 | 4.82 | 8653 | 9.51 |
| 5 | 0.6 | 2.9 | 1.5 | 5.4 | 19.9 | 1.3 | 0.4 | 548 | 860.1 | 59.69 | 2.69 | 3421 | 0.22 | 1.19 | 0.95 | 0.24 | 0.90 | 265.5 | 4.82 | 8653 | 9.51 |
| 5 | 0.6 | 2.9 | 1.5 | 5.4 | 19.9 | 1.3 | 0.4 | 548 | 860.1 | 59.69 | 2.69 | 3421 | 0.22 | 1.19 | 0.95 | 0.24 | 0.90 | 265.5 | 4.82 | 8653 | 9.51 |
| 5 | 0.6 | 2.9 | 1.5 | 5.4 | 19.9 | 1.3 | 0.4 | 548 | 860.1 | 59.69 | 2.69 | 3421 | 0.22 | 1.19 | 0.95 | 0.24 | 0.90 | 265.5 | 4.82 | 8653 | 9.51 |
| 5 | 0.6 | 2.9 | 1.5 | 5.4 | 19.9 | 1.3 | 0.4 | 548 | 860.1 | 59.69 | 2.69 | 3421 | 0.22 | 1.19 | 0.95 | 0.24 | 0.90 | 265.5 | 4.82 | 8653 | 9.51 |
| 5 | 0.6 | 2.9 | 1.5 | 5.4 | 19.9 | 1.3 | 0.4 | 548 | 863.4 | 59.92 | 2.69 | 3413 | 0.22 | 1.19 | 0.96 | 0.24 | 0.90 | 263.9 | 4.75 | 8650 | 9.53 |
| 5 | 0.6 | 2.9 | 1.5 | 5.5 | 19.9 | 1.3 | 0.4 | 548 | 869.7 | 60.30 | 2.68 | 3440 | 0.22 | 1.20 | 0.96 | 0.25 | 0.90 | 264.9 | 4.75 | 8707 | 9.57 |
| 5 | 0.6 | 2.9 | 1.5 | 5.5 | 19.9 | 1.3 | 0.4 | 548 | 869.7 | 60.30 | 2.68 | 3440 | 0.22 | 1.20 | 0.96 | 0.25 | 0.90 | 264.9 | 4.75 | 8707 | 9.57 |
| 5 | 0.6 | 2.9 | 1.5 | 5.5 | 19.9 | 1.3 | 0.4 | 548 | 869.7 | 60.30 | 2.68 | 3440 | 0.22 | 1.20 | 0.96 | 0.25 | 0.90 | 264.9 | 4.75 | 8707 | 9.57 |
| 5 | 0.6 | 2.9 | 1.5 | 5.5 | 19.9 | 1.3 | 0.4 | 548 | 869.7 | 60.30 | 2.68 | 3440 | 0.22 | 1.20 | 0.96 | 0.25 | 0.90 | 264.9 | 4.75 | 8707 | 9.57 |
| 5 | 0.6 | 2.9 | 1.5 | 5.5 | 19.9 | 1.3 | 0.4 | 548 | 869.7 | 60.30 | 2.68 | 3440 | 0.22 | 1.20 | 0.96 | 0.25 | 0.90 | 264.9 | 4.75 | 8707 | 9.57 |
| 5 | 0.6 | 2.9 | 1.5 | 5.5 | 19.9 | 1.3 | 0.4 | 548 | 872.9 | 60.48 | 2.67 | 3429 | 0.22 | 1.21 | 0.96 | 0.25 | 0.90 | 263.3 | 4.68 | 8726 | 9.59 |
| 5 | 0.6 | 2.9 | 1.5 | 5.5 | 19.9 | 1.3 | 0.4 | 548 | 872.9 | 60.48 | 2.67 | 3429 | 0.22 | 1.21 | 0.96 | 0.25 | 0.90 | 263.3 | 4.68 | 8726 | 9.59 |
| 5 | 0.6 | 2.9 | 1.5 | 5.5 | 19.9 | 1.3 | 0.4 | 548 | 872.9 | 60.48 | 2.67 | 3429 | 0.22 | 1.21 | 0.96 | 0.25 | 0.90 | 263.3 | 4.68 | 8726 | 9.59 |
| 5 | 0.6 | 2.9 | 1.5 | 5.5 | 19.9 | 1.3 | 0.4 | 548 | 872.9 | 60.48 | 2.67 | 3429 | 0.22 | 1.21 | 0.96 | 0.25 | 0.90 | 263.3 | 4.68 | 8726 | 9.59 |
| 5 | 0.6 | 2.9 | 1.5 | 5.5 | 19.9 | 1.3 | 0.4 | 548 | 873.0 | 60.54 | 2.67 | 3432 | 0.22 | 1.21 | 0.97 | 0.25 | 0.90 | 263.3 | 4.68 | 8705 | 9.59 |
| 5 | 0.6 | 2.9 | 1.5 | 5.5 | 19.9 | 1.3 | 0.4 | 548 | 876.2 | 60.71 | 2.67 | 3444 | 0.22 | 1.21 | 0.97 | 0.25 | 0.90 | 263.8 | 4.68 | 8724 | 9.61 |
| 5 | 0.6 | 2.9 | 1.5 | 5.5 | 19.9 | 1.3 | 0.4 | 548 | 876.2 | 60.71 | 2.67 | 3444 | 0.22 | 1.21 | 0.97 | 0.25 | 0.90 | 263.8 | 4.68 | 8724 | 9.61 |
| 5 | 0.6 | 2.9 | 1.5 | 5.5 | 19.7 | 1.3 | 0.4 | 548 | 882.2 | 61.19 | 2.66 | 3450 | 0.23 | 1.22 | 0.98 | 0.25 | 0.90 | 262.6 | 4.62 | 8715 | 9.64 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.9 | 1.3 | 0.4 | 548 | 885.6 | 61.36 | 2.66 | 3440 | 0.23 | 1.22 | 0.98 | 0.25 | 0.90 | 261.0 | 4.54 | 8804 | 9.68 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.9 | 1.3 | 0.4 | 548 | 885.8 | 61.33 | 2.66 | 3438 | 0.23 | 1.22 | 0.98 | 0.25 | 0.90 | 261.0 | 4.54 | 8782 | 9.68 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.9 | 1.2 | 0.4 | 548 | 887.5 | 61.60 | 2.65 | 3455 | 0.23 | 1.23 | 0.98 | 0.25 | 0.90 | 261.6 | 4.54 | 8803 | 9.70 |


| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.9 | 1.2 | 0.4 | 548 | 887.5 | 61.60 | 2.65 | 3455 | 0.23 | 1.23 | 0.98 | 0.25 | 0.90 | 261.6 | 4.54 | 8803 | 9.70 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.9 | 1.2 | 0.4 | 548 | 889.0 | 61.60 | 2.65 | 3455 | 0.23 | 1.23 | 0.98 | 0.25 | 0.90 | 261.5 | 4.54 | 8803 | 9.70 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.9 | 1.2 | 0.4 | 548 | 889.0 | 61.60 | 2.65 | 3455 | 0.23 | 1.23 | 0.98 | 0.25 | 0.90 | 261.5 | 4.54 | 8803 | 9.70 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.9 | 1.2 | 0.4 | 548 | 889.0 | 61.60 | 2.65 | 3455 | 0.23 | 1.23 | 0.98 | 0.25 | 0.90 | 261.5 | 4.54 | 8803 | 9.70 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.9 | 1.2 | 0.4 | 548 | 889.0 | 61.60 | 2.65 | 3455 | 0.23 | 1.23 | 0.98 | 0.25 | 0.90 | 261.5 | 4.54 | 8803 | 9.70 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.8 | 1.2 | 0.4 | 548 | 889.3 | 61.57 | 2.65 | 3453 | 0.23 | 1.23 | 0.98 | 0.25 | 0.90 | 261.5 | 4.54 | 8780 | 9.70 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.8 | 1.2 | 0.4 | 548 | 891.0 | 61.84 | 2.65 | 3447 | 0.23 | 1.23 | 0.99 | 0.25 | 0.90 | 259.9 | 4.48 | 8800 | 9.72 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.8 | 1.2 | 0.4 | 548 | 891.0 | 61.84 | 2.65 | 3447 | 0.23 | 1.23 | 0.99 | 0.25 | 0.90 | 259.9 | 4.48 | 8800 | 9.72 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.8 | 1.2 | 0.4 | 548 | 892.5 | 61.84 | 2.65 | 3446 | 0.23 | 1.23 | 0.99 | 0.25 | 0.90 | 259.9 | 4.47 | 8800 | 9.72 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.8 | 1.2 | 0.4 | 548 | 892.5 | 61.84 | 2.65 | 3446 | 0.23 | 1.23 | 0.99 | 0.25 | 0.90 | 259.9 | 4.47 | 8800 | 9.72 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.8 | 1.2 | 0.4 | 548 | 892.5 | 61.84 | 2.65 | 3446 | 0.23 | 1.23 | 0.99 | 0.25 | 0.90 | 259.9 | 4.47 | 8800 | 9.72 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.8 | 1.2 | 0.4 | 548 | 892.5 | 61.84 | 2.65 | 3446 | 0.23 | 1.23 | 0.99 | 0.25 | 0.90 | 259.9 | 4.47 | 8800 | 9.72 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.8 | 1.2 | 0.4 | 548 | 892.5 | 61.84 | 2.65 | 3446 | 0.23 | 1.23 | 0.99 | 0.25 | 0.90 | 259.9 | 4.47 | 8800 | 9.72 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.8 | 1.2 | 0.4 | 548 | 892.5 | 61.84 | 2.65 | 3446 | 0.23 | 1.23 | 0.99 | 0.25 | 0.90 | 259.9 | 4.47 | 8800 | 9.72 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.7 | 1.2 | 0.4 | 548 | 895.9 | 62.08 | 2.65 | 3461 | 0.23 | 1.24 | 0.99 | 0.25 | 0.90 | 260.3 | 4.48 | 8799 | 9.74 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.7 | 1.2 | 0.4 | 548 | 895.9 | 62.08 | 2.65 | 3461 | 0.23 | 1.24 | 0.99 | 0.25 | 0.90 | 260.3 | 4.48 | 8799 | 9.74 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.7 | 1.2 | 0.4 | 548 | 895.9 | 62.08 | 2.65 | 3461 | 0.23 | 1.24 | 0.99 | 0.25 | 0.90 | 260.3 | 4.48 | 8799 | 9.74 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.2 | 1.2 | 0.4 | 548 | 921.0 | 63.79 | 2.62 | 3469 | 0.24 | 1.27 | 1.02 | 0.26 | 0.90 | 255.2 | 4.23 | 8785 | 9.89 |
| 5 | 0.6 | 2.9 | 1.5 | 5.6 | 19.2 | 1.2 | 0.4 | 548 | 921.0 | 63.79 | 2.62 | 3469 | 0.24 | 1.27 | 1.02 | 0.26 | 0.90 | 255.2 | 4.23 | 8785 | 9.89 |

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